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NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

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MSC INTERNAL NOTE NO. 68-FM-307

December 16, 1968

THE EARTH AND THE MOON AS
VIEWED FROM THE SPACECRAFT
DURING THE APOLLO 8 MISSION

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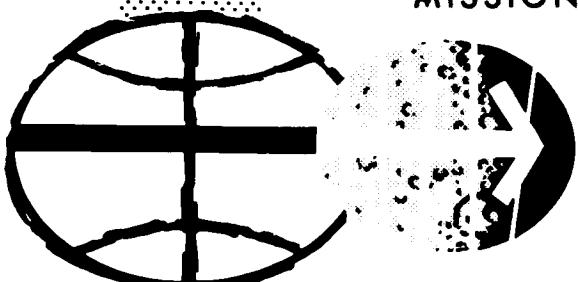
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Flight Analysis Branch

MISSION PLANNING AND ANALYSIS DIVISION

MANNED SPACECRAFT CENTER
HOUSTON, TEXAS



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MSC INTERNAL NOTE NO. 68-FM-307

PROJECT APOLLO

THE EARTH AND THE MOON AS VIEWED FROM THE
SPACECRAFT DURING THE APOLLO 8 MISSION

By Charles T. Hyle and Alfred N. Lunde
Flight Analysis Branch

December 16, 1968

MISSION PLANNING AND ANALYSIS DIVISION
NATIONAL AERONAUTICS AND SPACE ADMINISTRATION
MANNED SPACECRAFT CENTER
HOUSTON, TEXAS

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FOREWORD

Summary and Introduction

The purpose of this document is to depict the geometry of the Apollo 8 trajectory as it might appear from the spacecraft commander's window. In addition to depicting attitude orientations at TLI, LOI, TEI and entry, detailed views of the earth and moon during the trans-lunar and transearth coast phases are shown. The earth centered inertial geometry of the trajectory is also depicted. The trajectory used for this document assumes a December 21, 1968, launch date on a 72° azimuth, first-opportunity injection.

G. B. Roush and M. Jenness are acknowledged for their support with the visual computer program and the star-field chart, respectively.

The Apollo 8 Trajectory

As described in reference 1 the geometry of a lunar mission can be difficult to visualize. This document is an attempt to provide additional details to those in reference 1 to aid in visualizing the Apollo 8 mission.

As representation of the sun, earth, moon, and the outbound Apollo 8 trajectory is shown in figure 13. As shown it is winter in the northern hemisphere and the moon is below the solar plane at TLI.

An earth-centered inertial view of the sun, moon, and Apollo 8 trajectory are depicted in figure 14. Views of the trajectory from the earth's North pole and the vernal equinox are shown in figures 15(a) and 15(b). The solar and lunar orbit line of nodes are approximately coincident for this year.

Views from the Spacecraft

The change in the size of the earth and moon on the translunar and transearth coast trajectories are shown as seen from the spacecraft against the star field in figure 16.

This figure provides an overall view of the Apollo 8 mission as seen by the crew.

A more detailed representation of the trajectory is provided by the remaining figures. It is noted that the three major maneuvers are made with the crew in a heads-down attitude relative to the earth or moon, whichever is nearest. The spacecraft attitude at the beginning, middle, and end of the TLI burn are shown in figure 1. The constellation Cygnus is visible as noted.

Spacecraft window views of the earth and moon with a constant and variable field-of-view are shown during translunar coast in figures 2, 3, 4, and 5.

The apparent motion of the earth terminator relative to the spacecraft window is due to the assumed attitude of the spacecraft. This attitude assumes the spacecraft is pitched down 90° from the local horizontal in a plane defined by the radius vector to the spacecraft and the inertial vector relative to the body being observed. As the spacecraft approaches the moon the trajectory is "warped" by the moon and causes the effect depicted. The crew will not necessarily have the spacecraft oriented as stated previously.

Spacecraft windows views during LOI and TEI are shown in figures 6 and 7. During LOI the constellations Orion and Cassiopeia are visible. The sun and Venus are visible during TEI.

The earth and moon for a constant and variable field-of-view for the fast return transearth coast are shown in figures 8 through 11.

Figures 12(a) through 12(r) depict the spacecraft prior to and including entry interface. The spacecraft is holding a constant angle of 31.7° from the spacecraft X-axis to the horizon. It is of interest to note the moon coming into view and then being occulted by the earth during the 18 minutes shown in these figures.

A star catalogue (table I) identifies the stars appearing on all figures.

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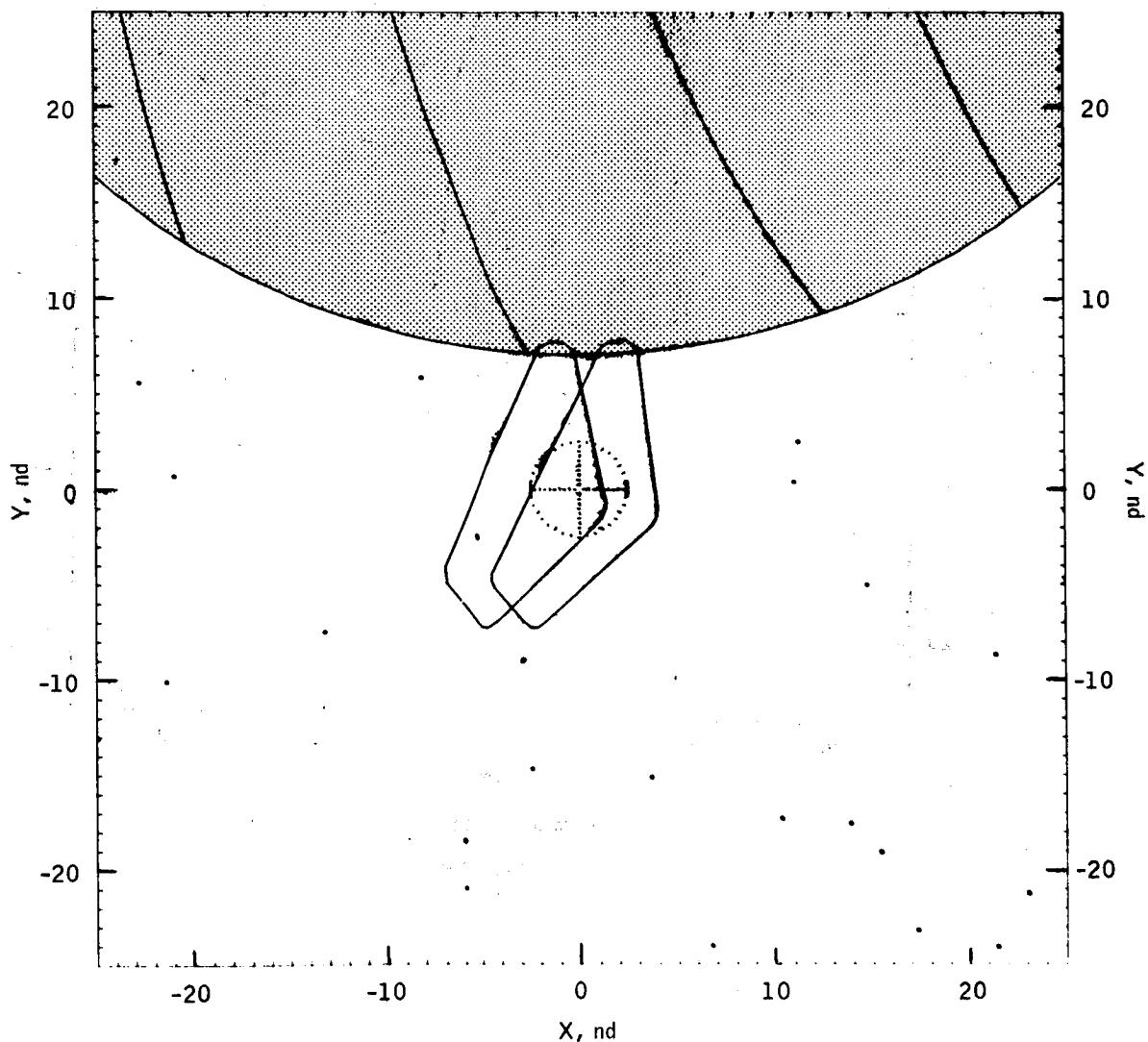
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SEQ	507	509	540	577	580	592	604	610	684	639	651	455	658
▼	-21	-23	-17	-15	-6	-13	-10	-16	-6	-3	-2	-24	21
▼	-23	-21	-23	-18	-23	-17	-17	-20	-18	-14	-14	-12	-8

673	690	700	740	745	753	765	770	798	795	813
-21	-2	-13	14	-5	-20	0	-28	11	-8	11
-9	-8	-7	-4	-2	0	-1	0	0	6	2

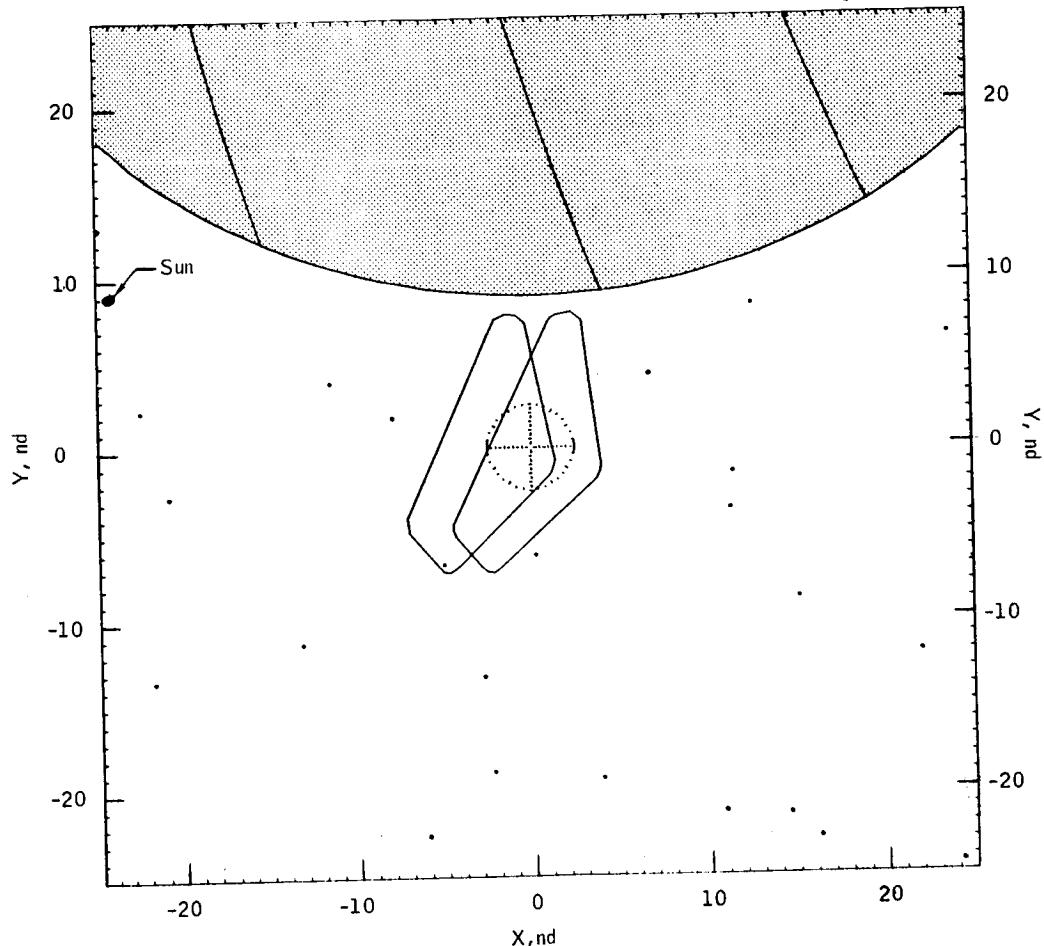


(a) Begin TLI burn.

Figure 1. - Translunar injection burn.

SEG	509	577	592	609	624	639	651	658	673	690	700	740	745
X	-24	16	14	10	-7	3	-2	22	-21	-2	-13	15	-5
Y	-24	-22	-21	-21	-22	-19	-18	-11	-13	-13	-11	-8	-6

SEG	753	755	770	792	795	803	813	849	903	971
X	-20	0	-22	11	-7	-11	11	6	12	23
Y	-2	-6	2	-3	1	4	-1	4	8	6

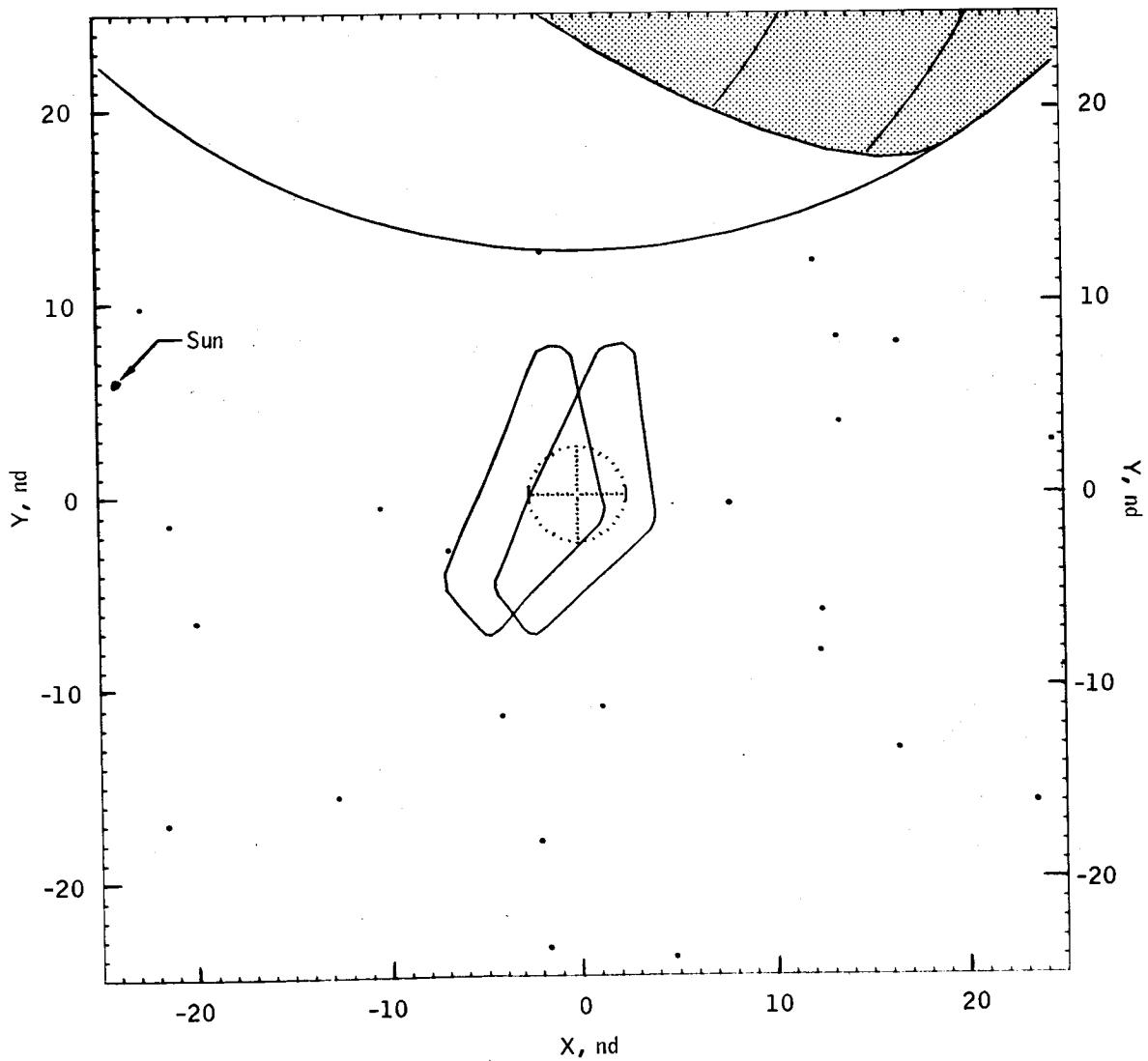


(b) Middle of TLI burn.

Figure 1. - Continued.

SEQ	639	651	659	673	690	700	740	745	753	755	770	792	795
X	4	-1	23	-21	-2	-12	16	-4	-19	1	-21	12	-6
Y	-23	-23	-15	-16	-17	-15	-13	-11	-6	-10	-1	-7	-2

SEG	803	813	844	849	903	904	931	942	950	971
X	-10	12	-22	7	13	-1	13	16	12	24
Y	0	-5	9	0	3	12	8	8	12	2



(c) End of TLI burn.
Figure 1. - Concluded.

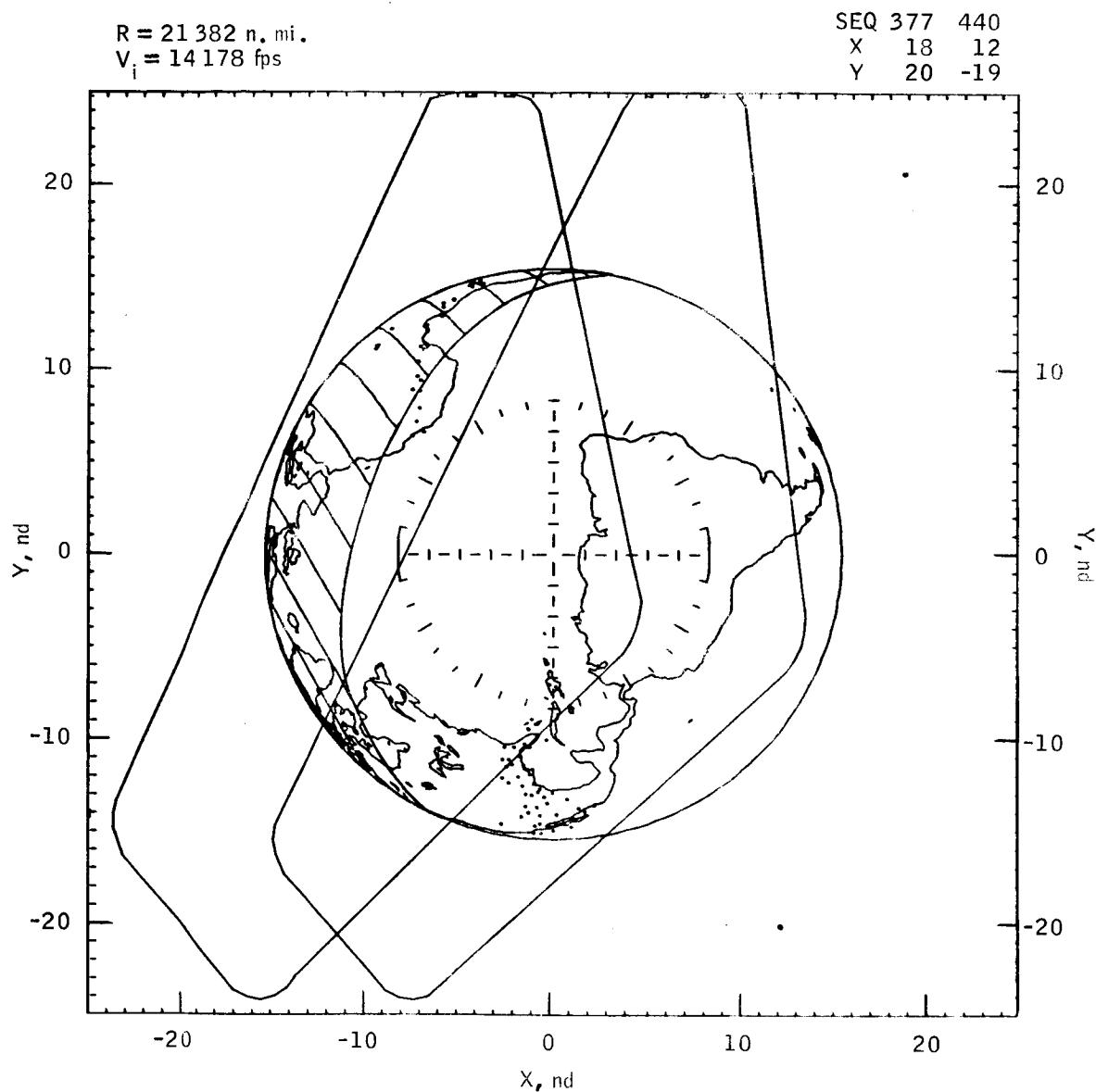
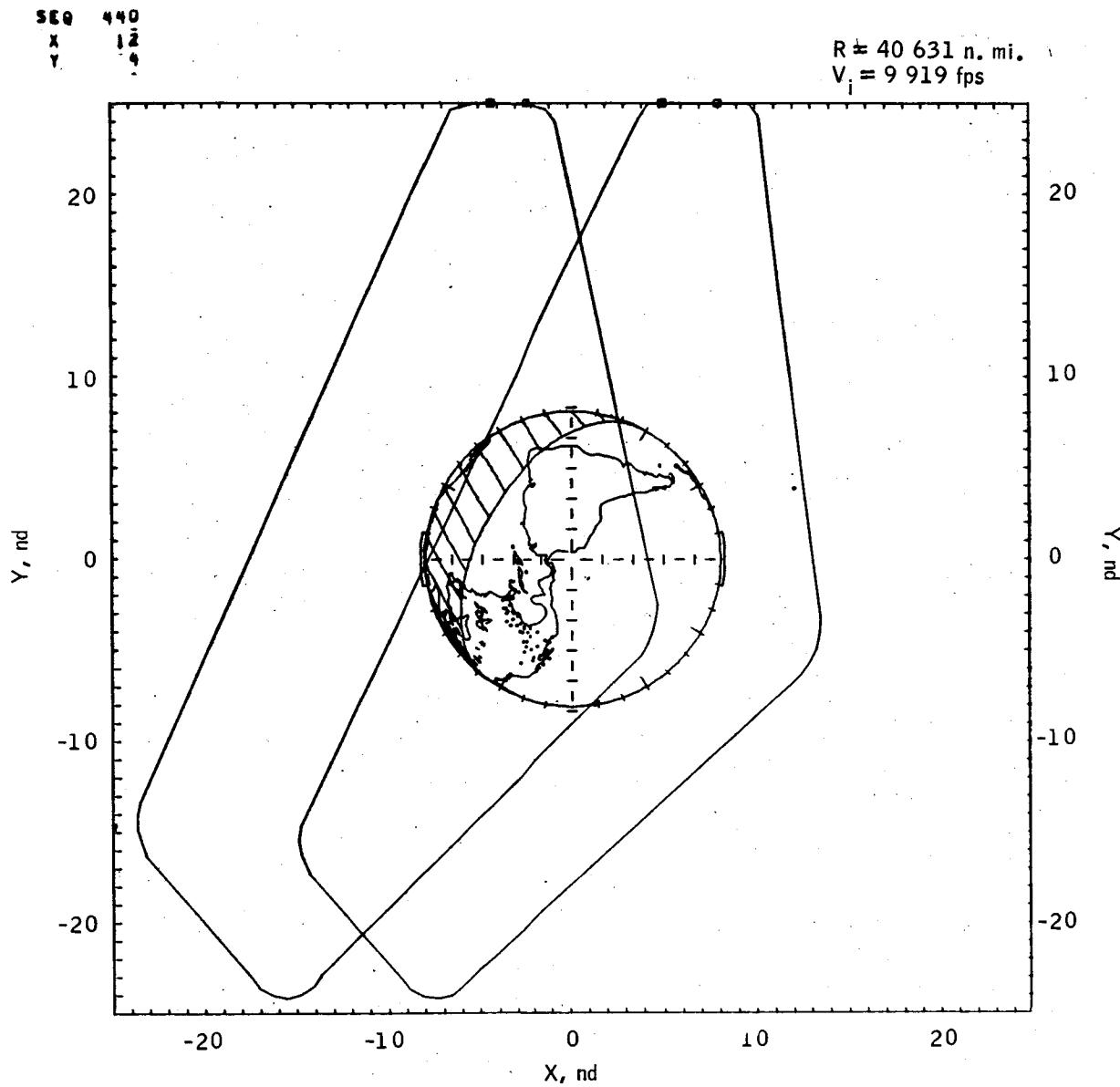
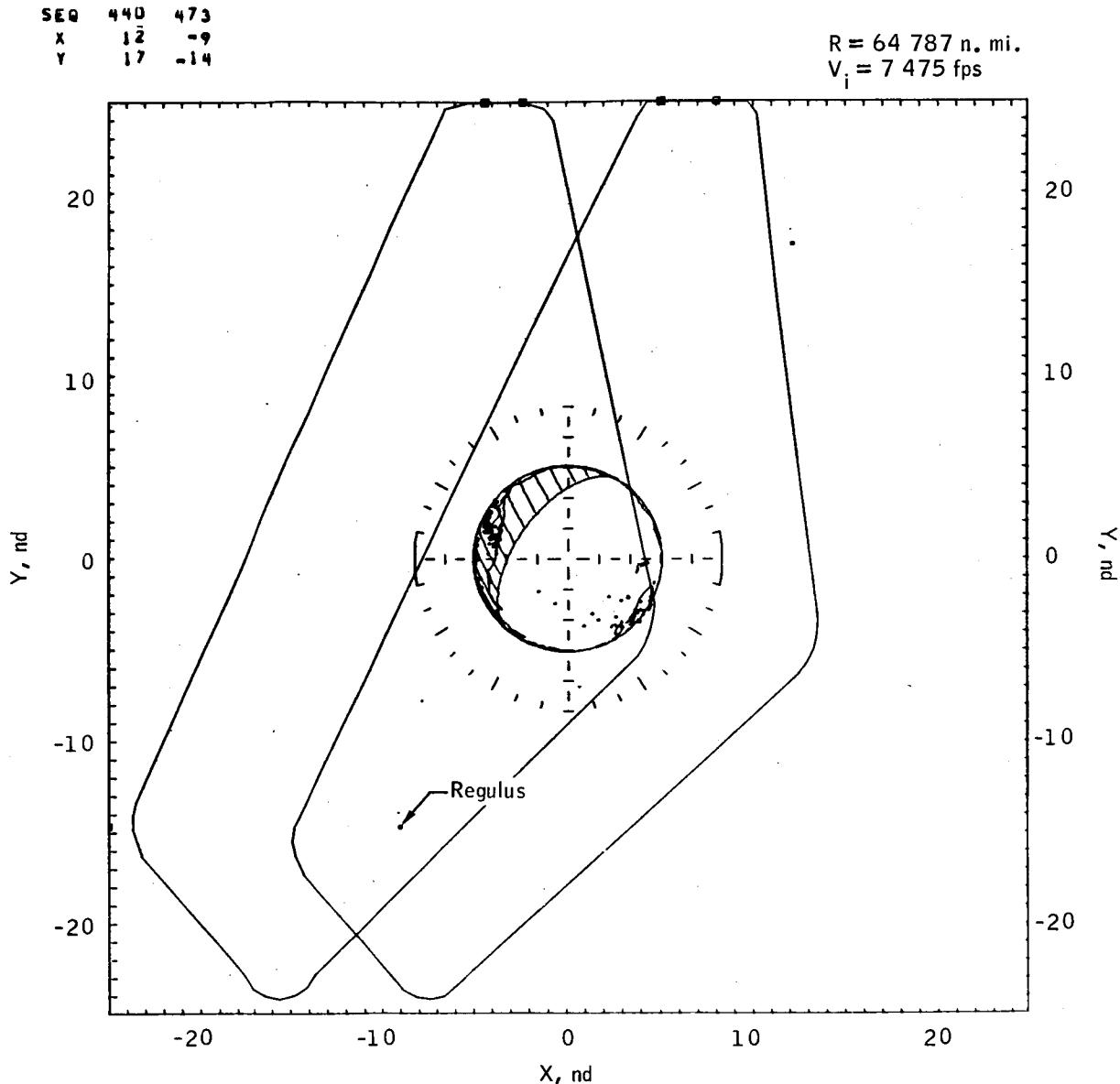


Figure 2.- Translunar coast (earth referenced) constant field of view.



(b) Time from TLI cutoff = 5 hr.

Figure 2.- Continued.

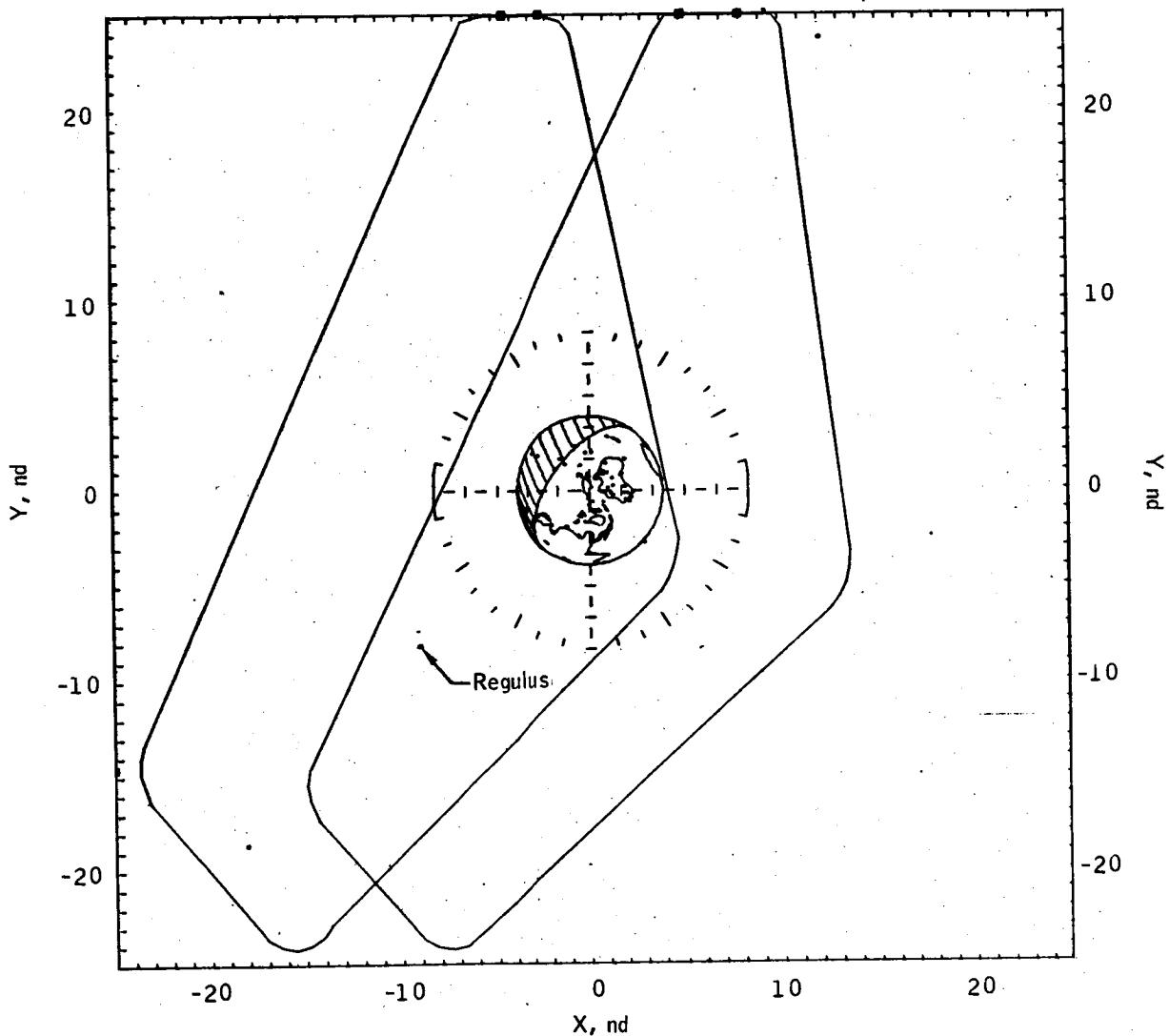


(c) Time from TLI cutoff = 10 hr.

Figure 2.- Continued.

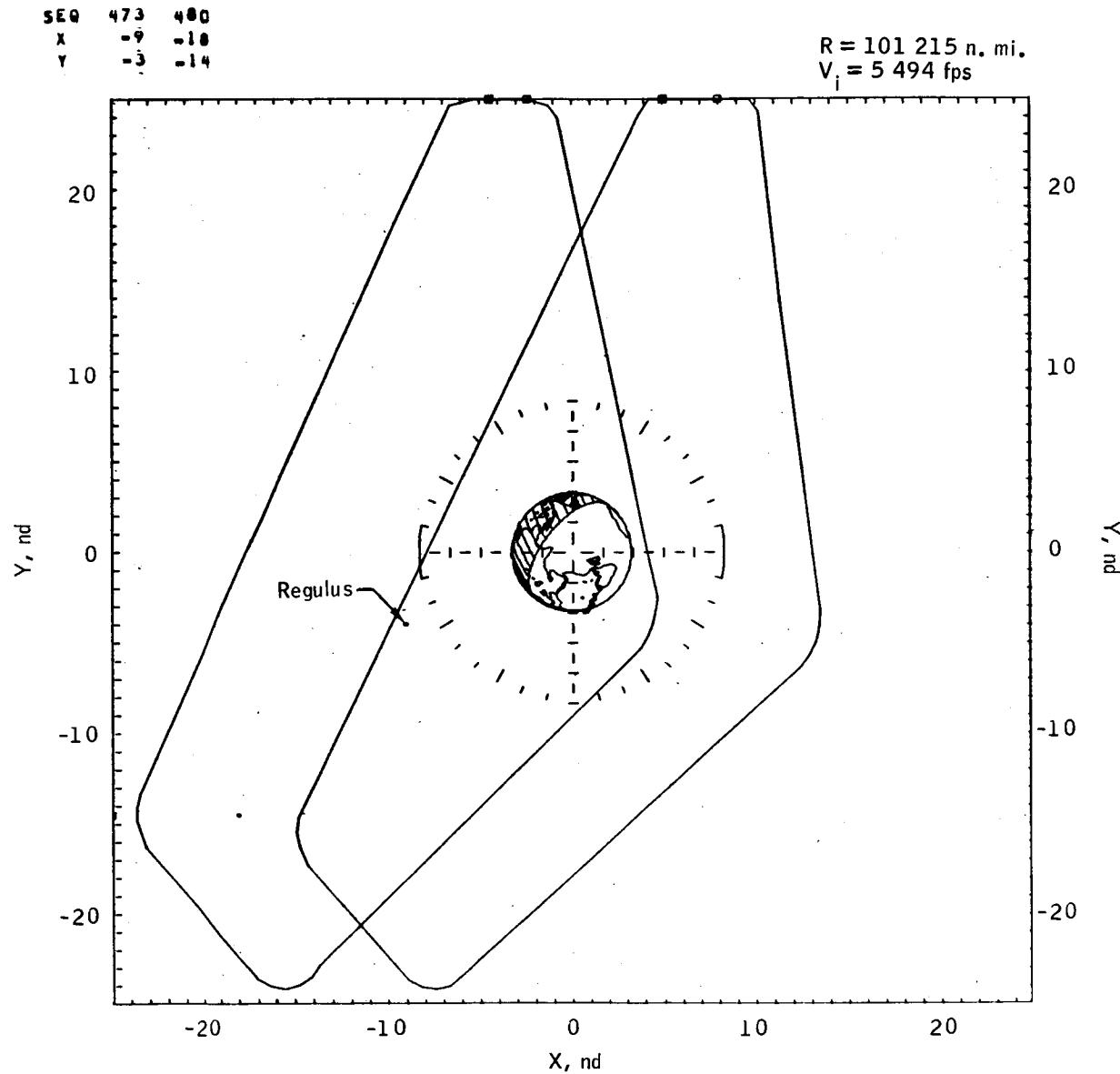
SER	440	473	480
X	-12	-9	-18
Y	-23	-7	-10

$R = 84\ 345$ n. mi.
 $V_i = 6\ 270$ fps



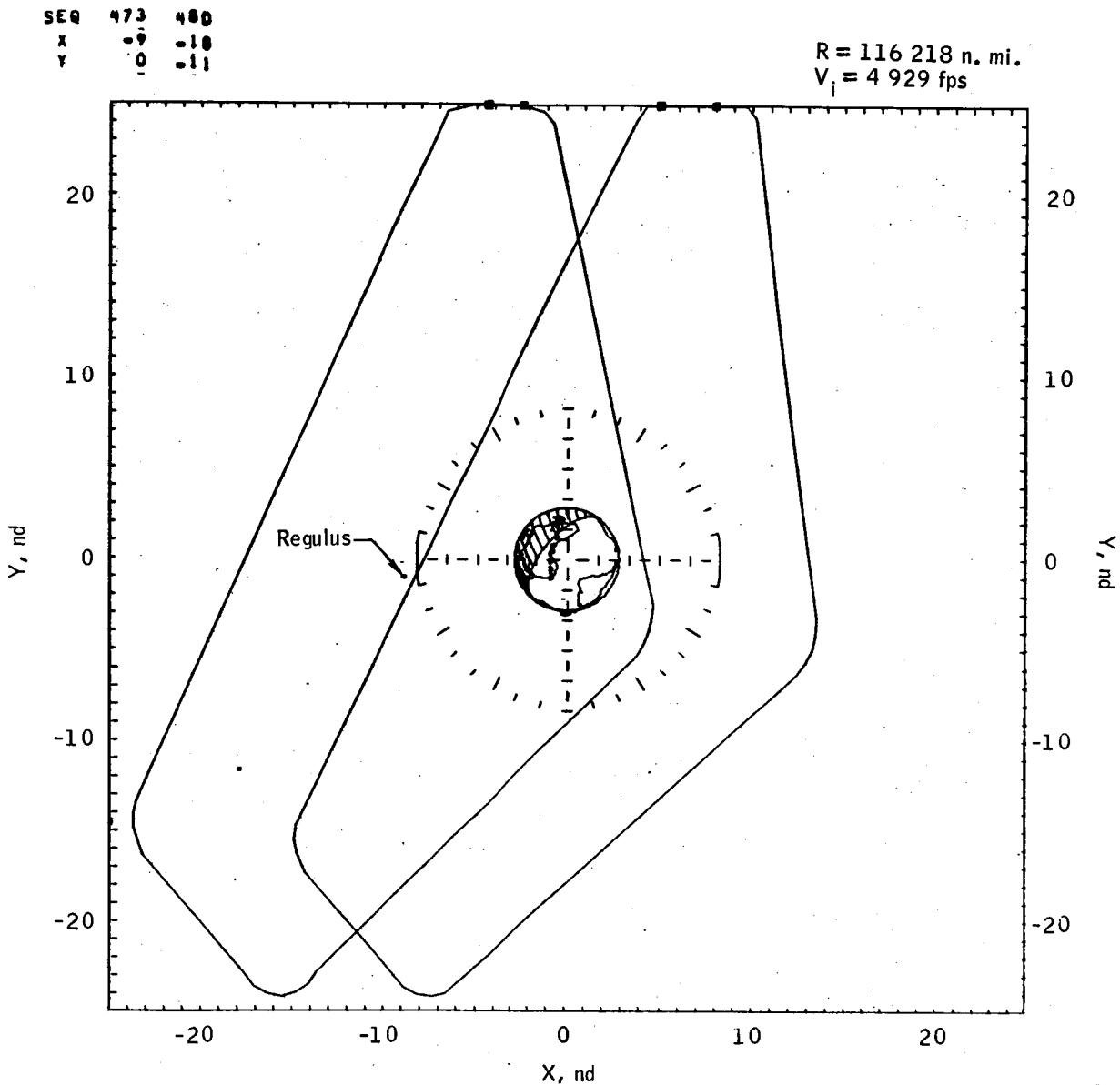
(d) Time from TLI cutoff = 15 hr.

Figure 2.- Continued.



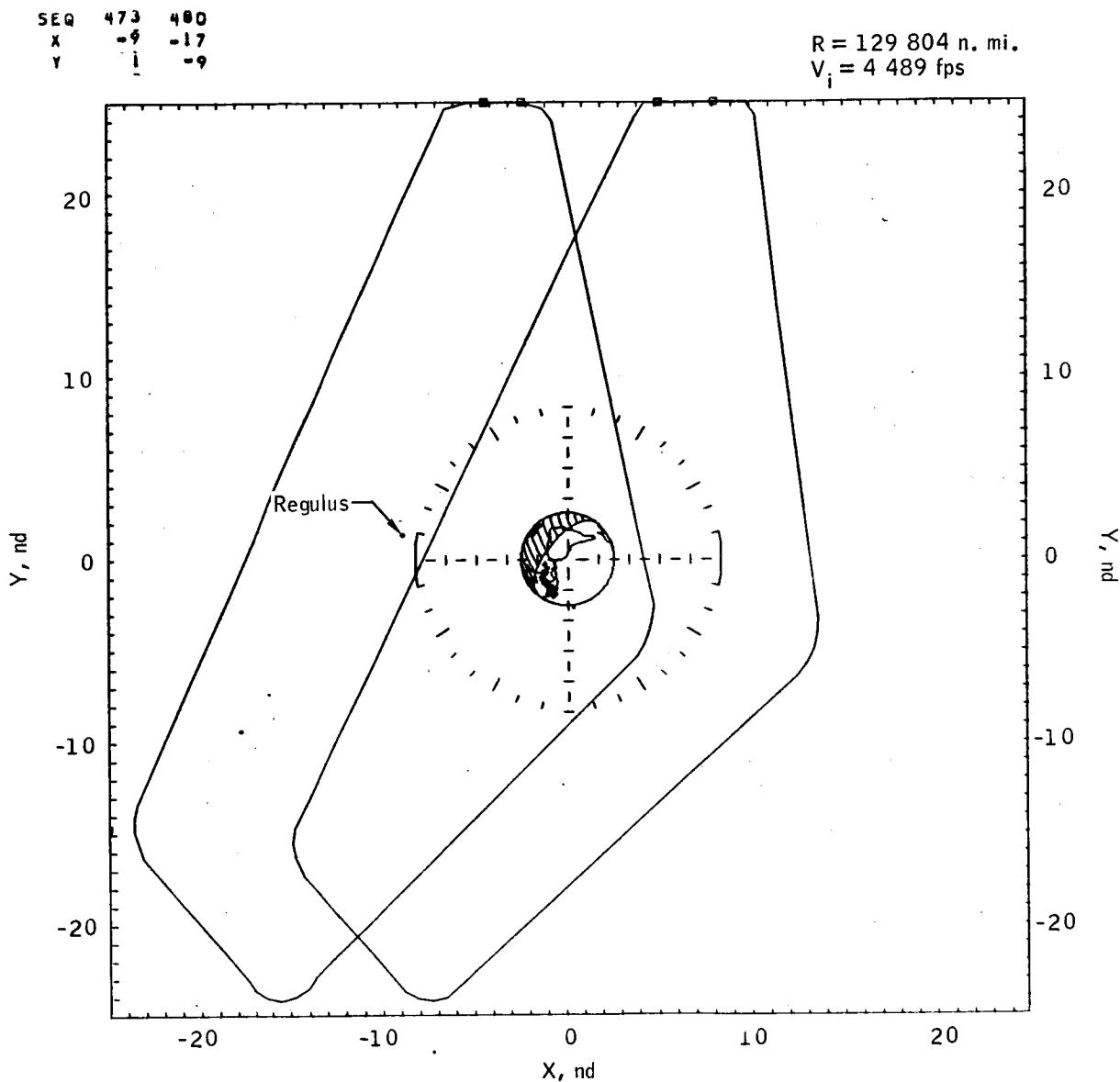
(e) Time from TLI cutoff = 20 hr.

Figure 2.- Continued.



(f) Time from TLI cutoff = 25 hr.

Figure 2.- Continued.



(g) Time from TLI cutoff = 30 hr.

Figure 2.- Continued.

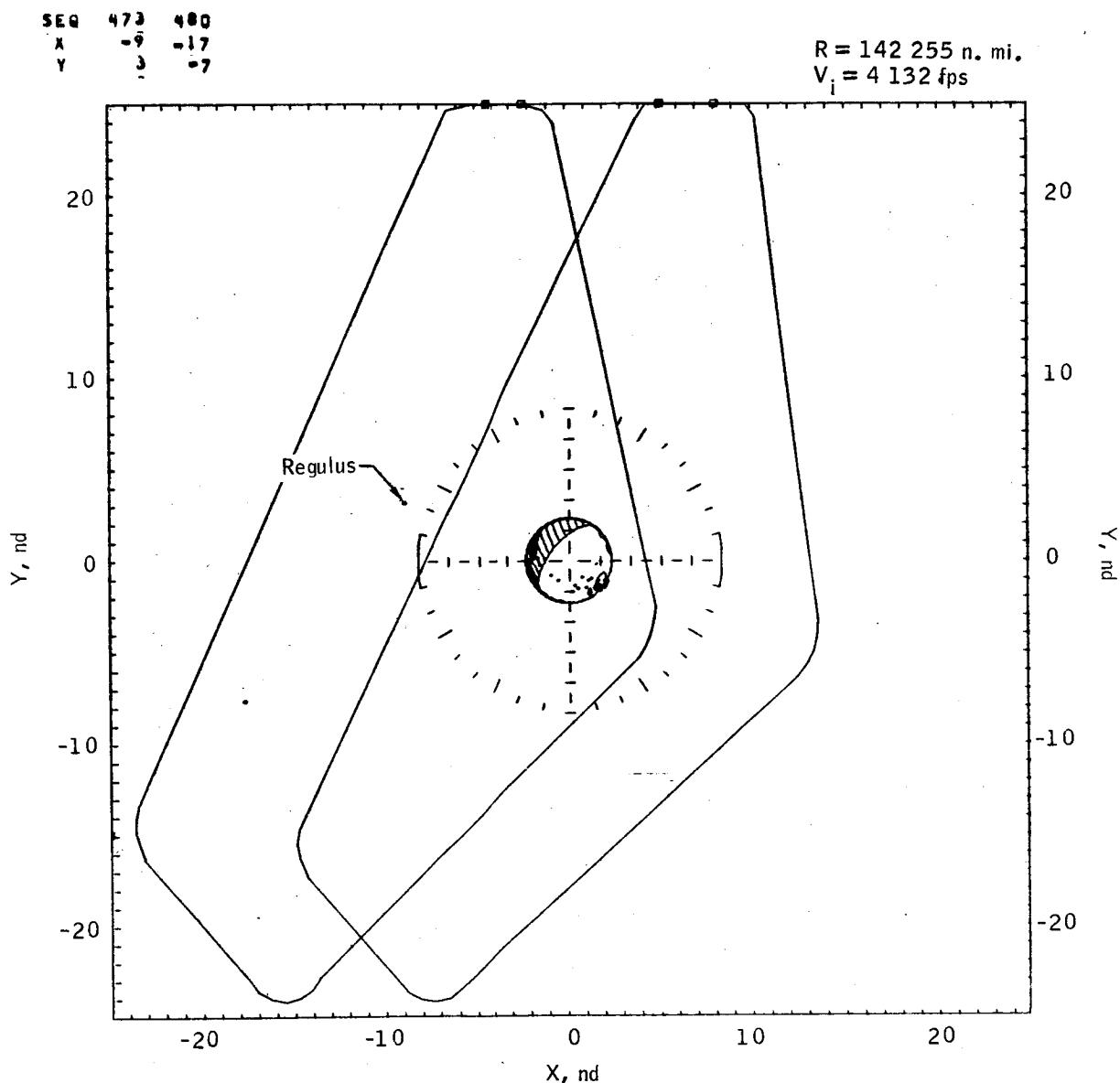
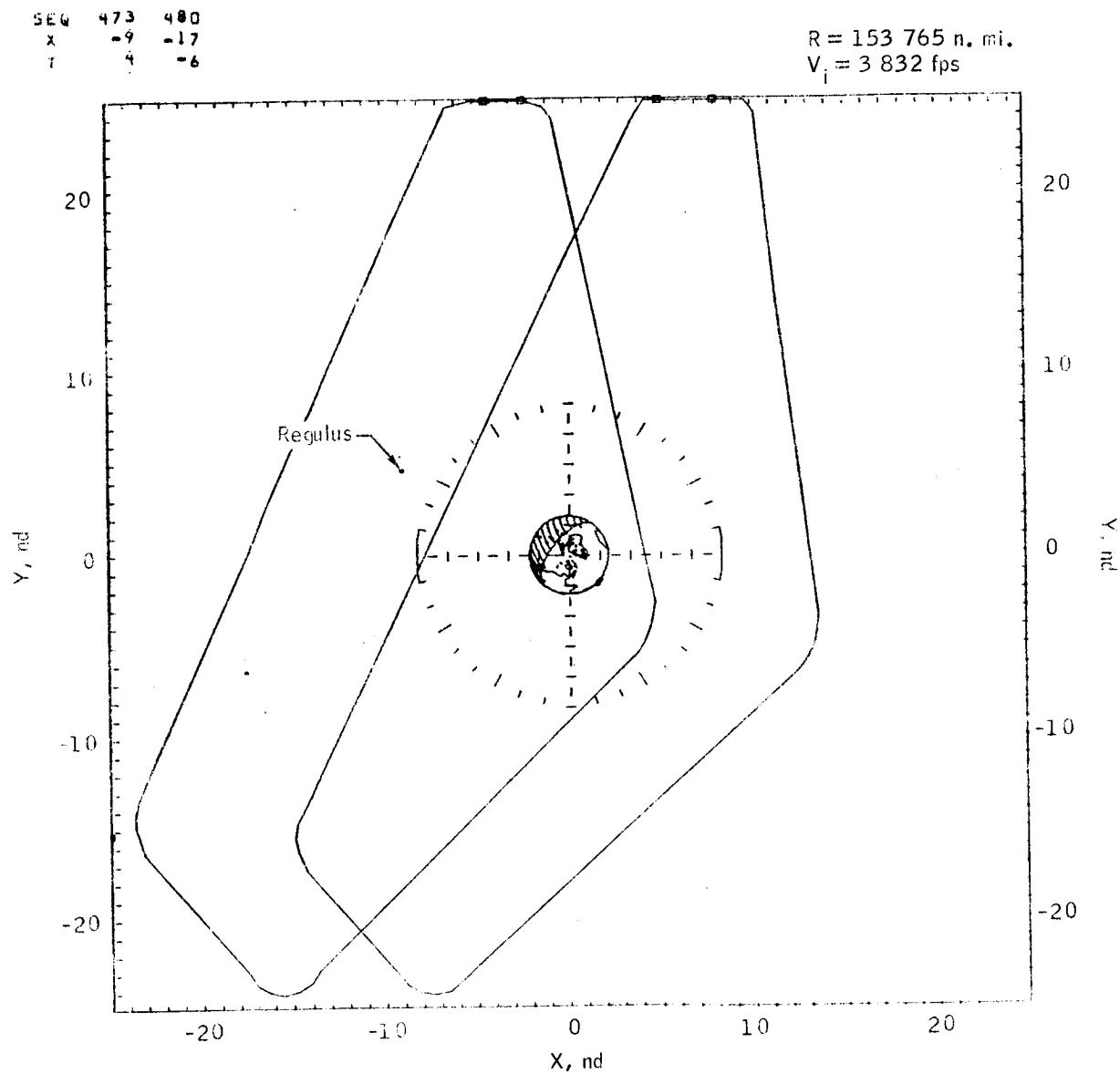


Figure 2.- Continued.



(i) Time from TLI cutoff = 40 hr.

Figure 2.- Continued.

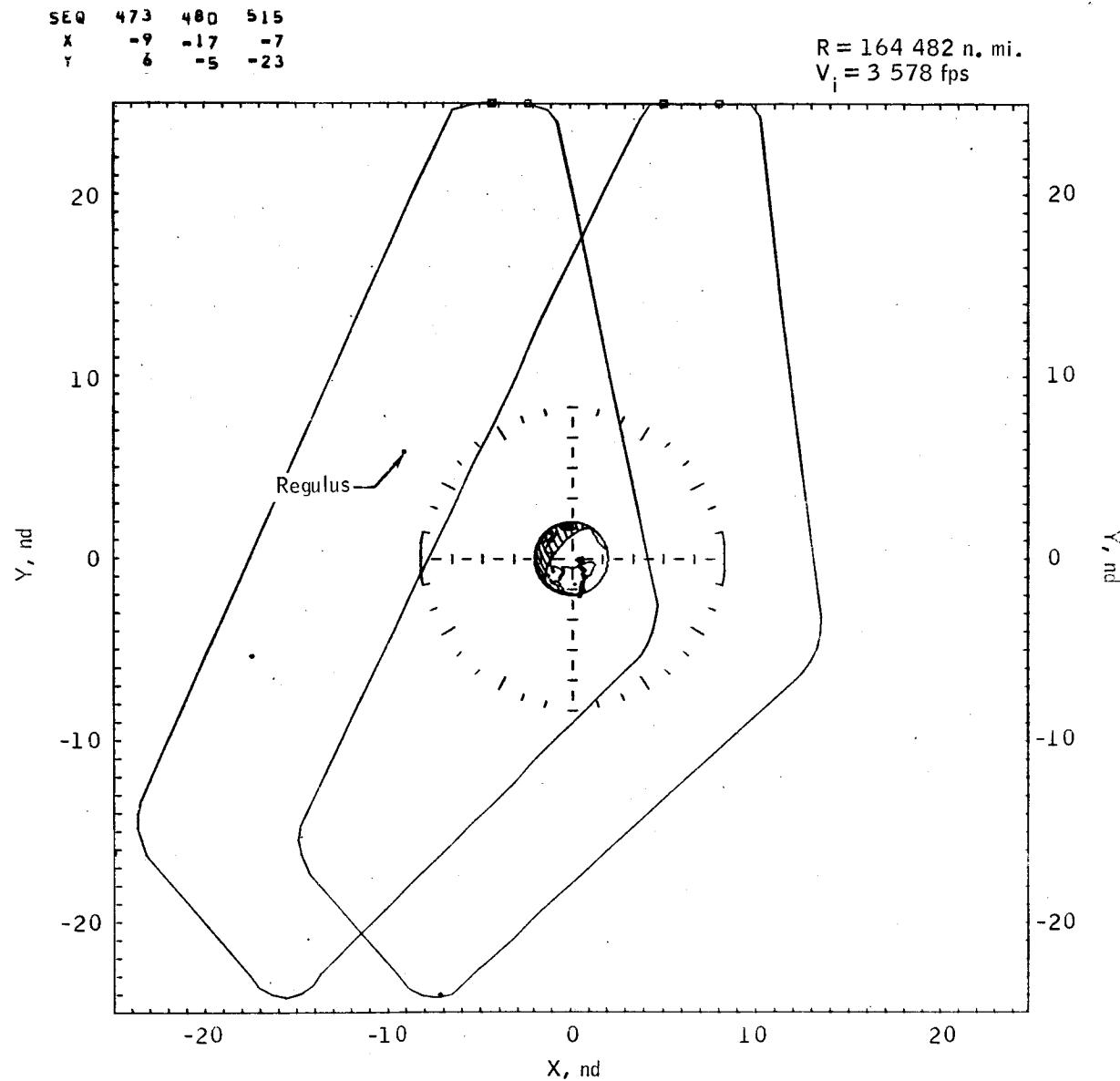


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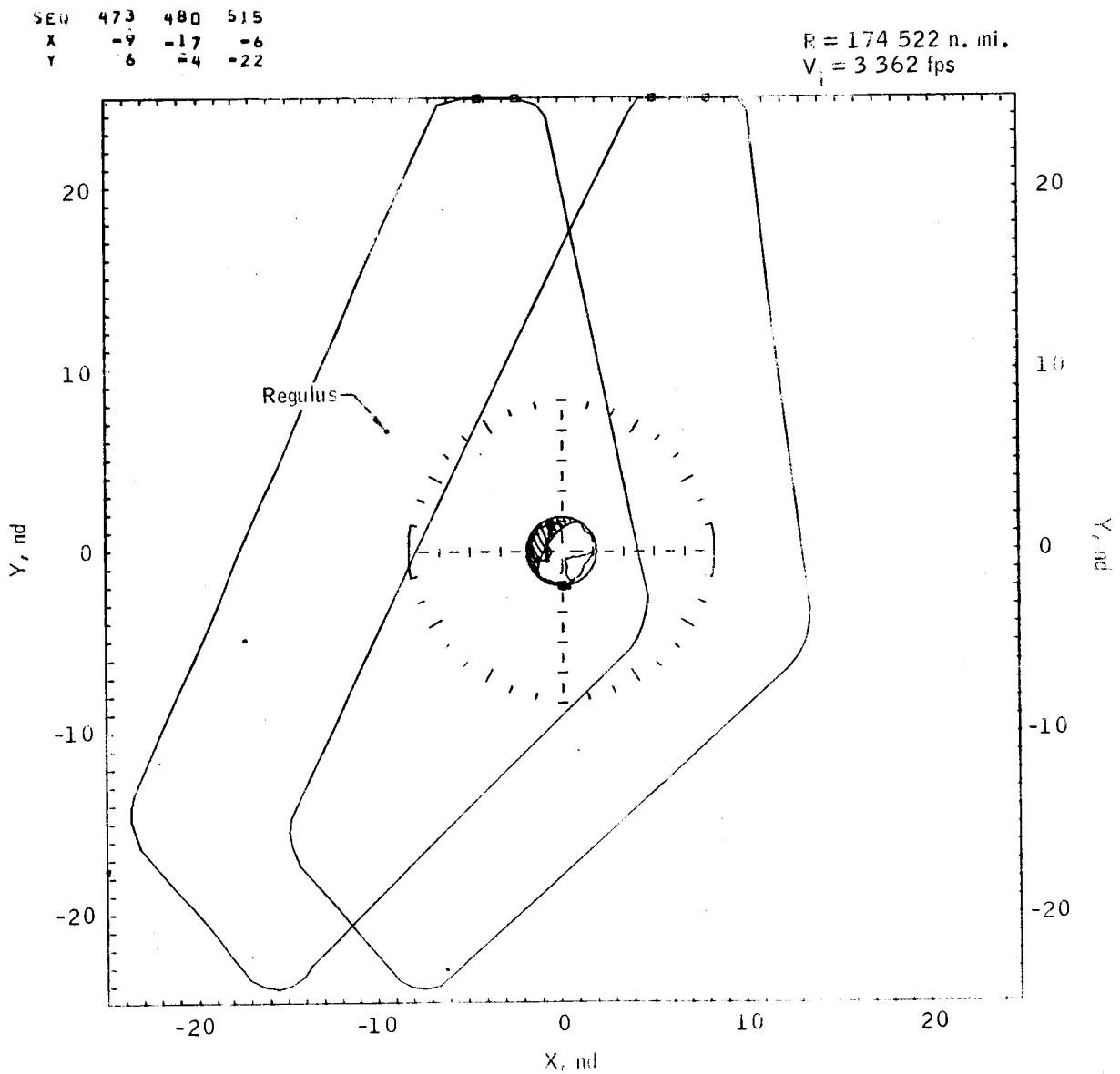
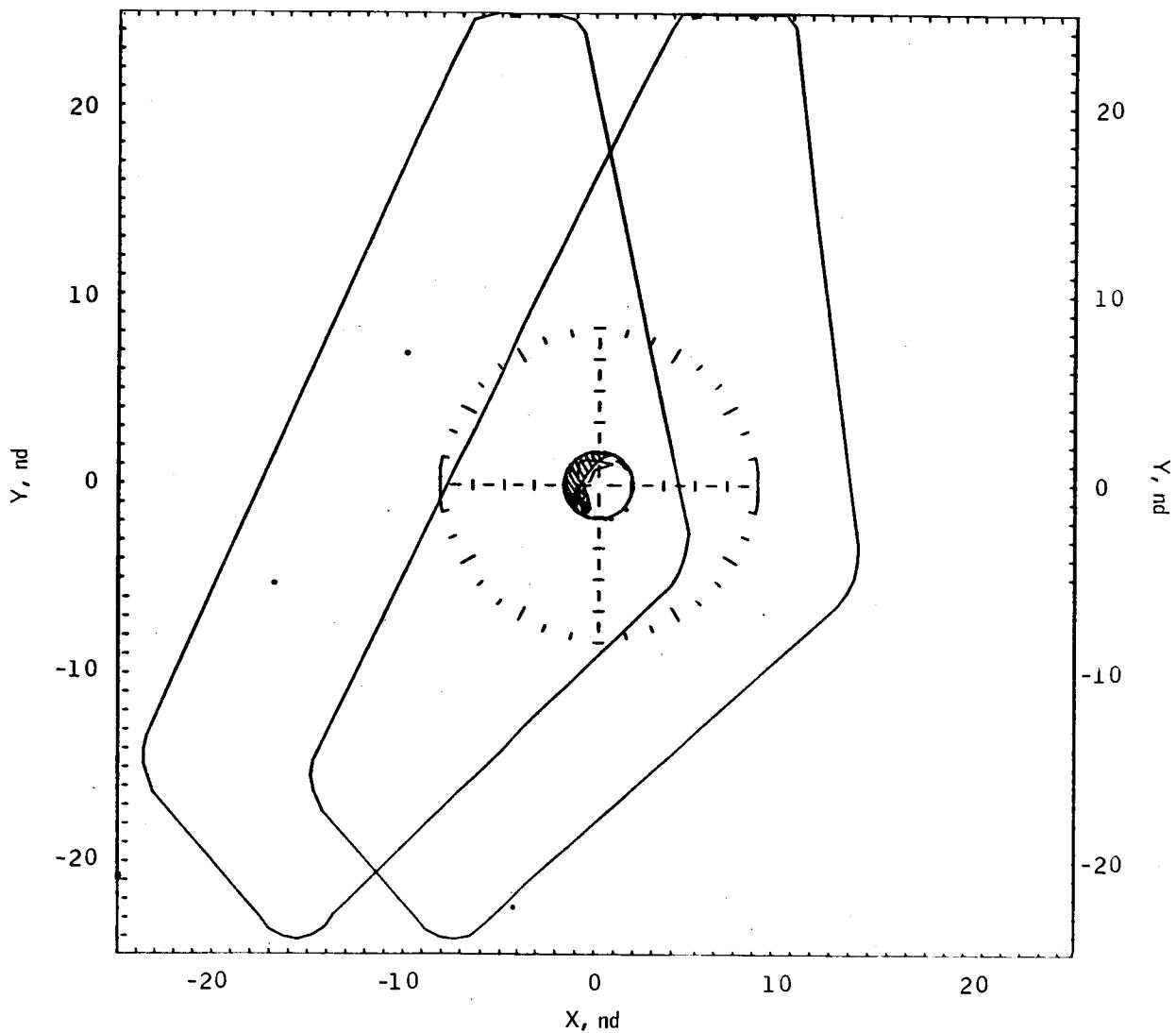
(k) Time from TLI cutoff ≈ 50 hr.

Figure 2.- Continued.

SEQ	473	480	515
X	-10	-16	-4
Y	7	-5	-22

$R_M = 29\ 319$ n. mi.
 $V_i = 4\ 025$ fps



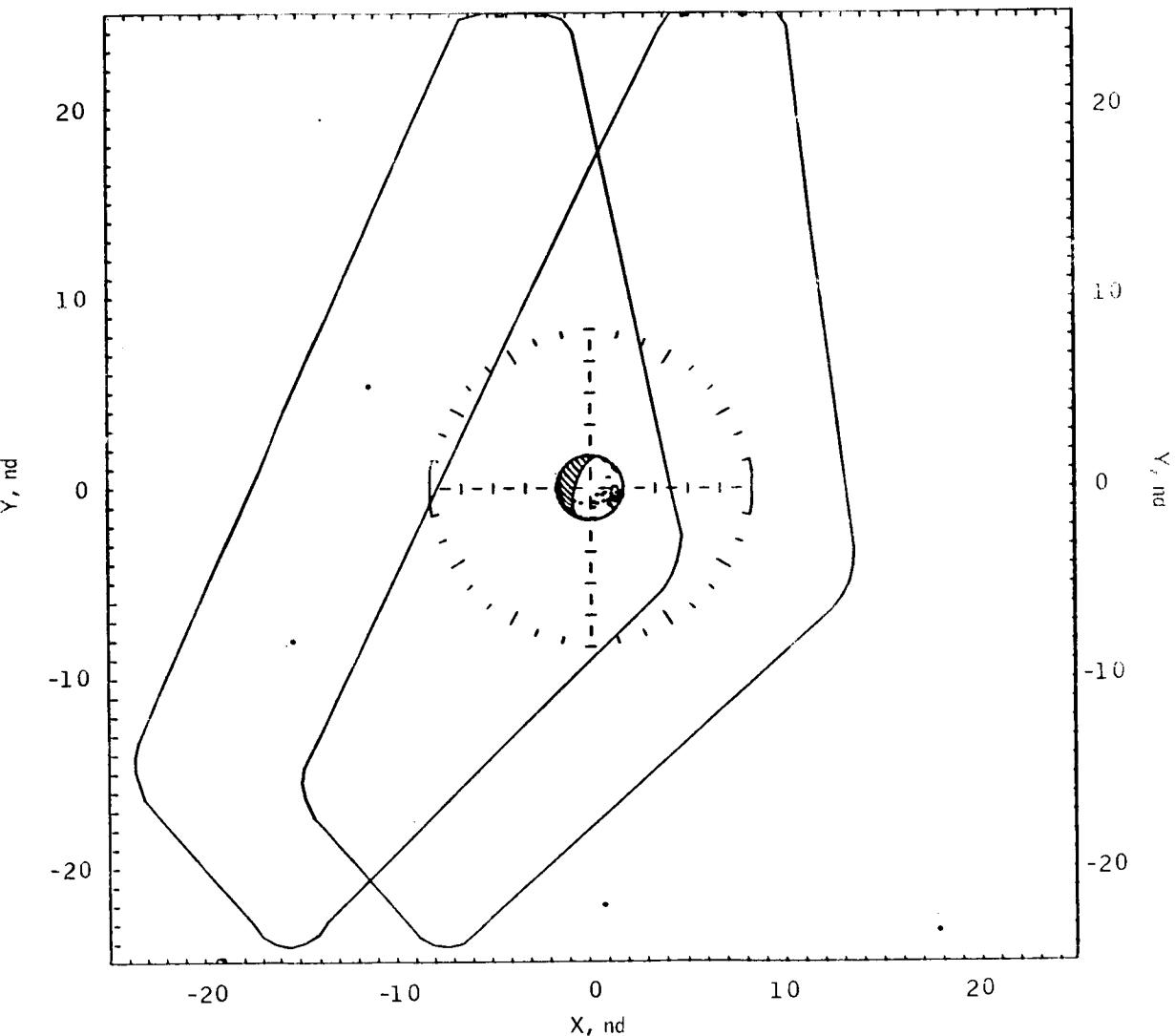
(I) Time from TLI cutoff = 55 hr.

Figure 2.- Continued.

SEQ	473	480	515	535
X	-11	-15	0	17
Y	5	-7	-21	-23

$$R_M = 17\ 254 \text{ n. mi.}$$

$$V_i = 4\ 187 \text{ fps}$$

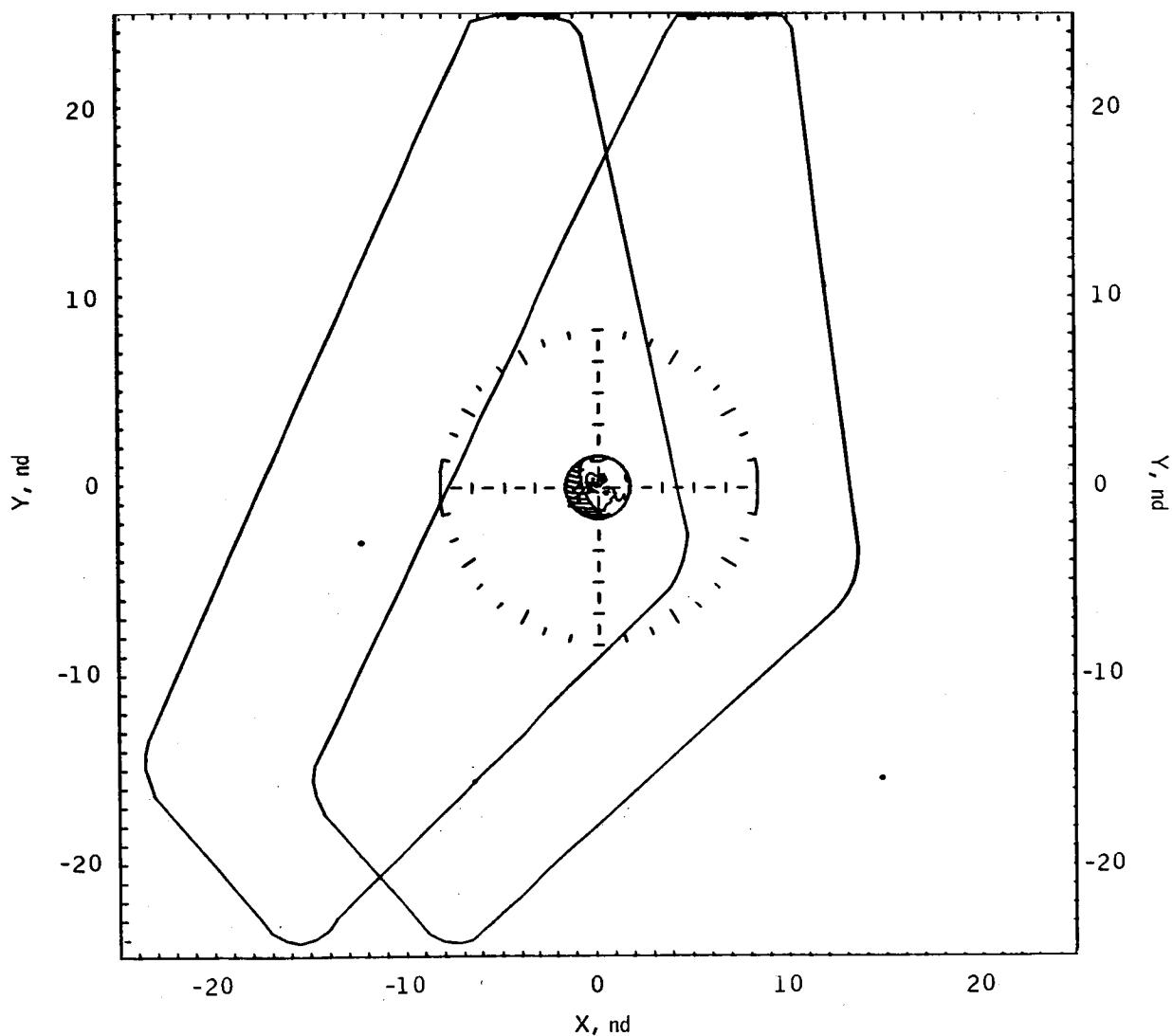


(m) Time from TLI cutoff = 60 hr.

Figure 2.- Continued.

SEQ 473 480 515
 X -12 -6 14
 Y -2 -15 -15

$R_M = 7032$ n. mi.
 $V_i = 4727$ fps

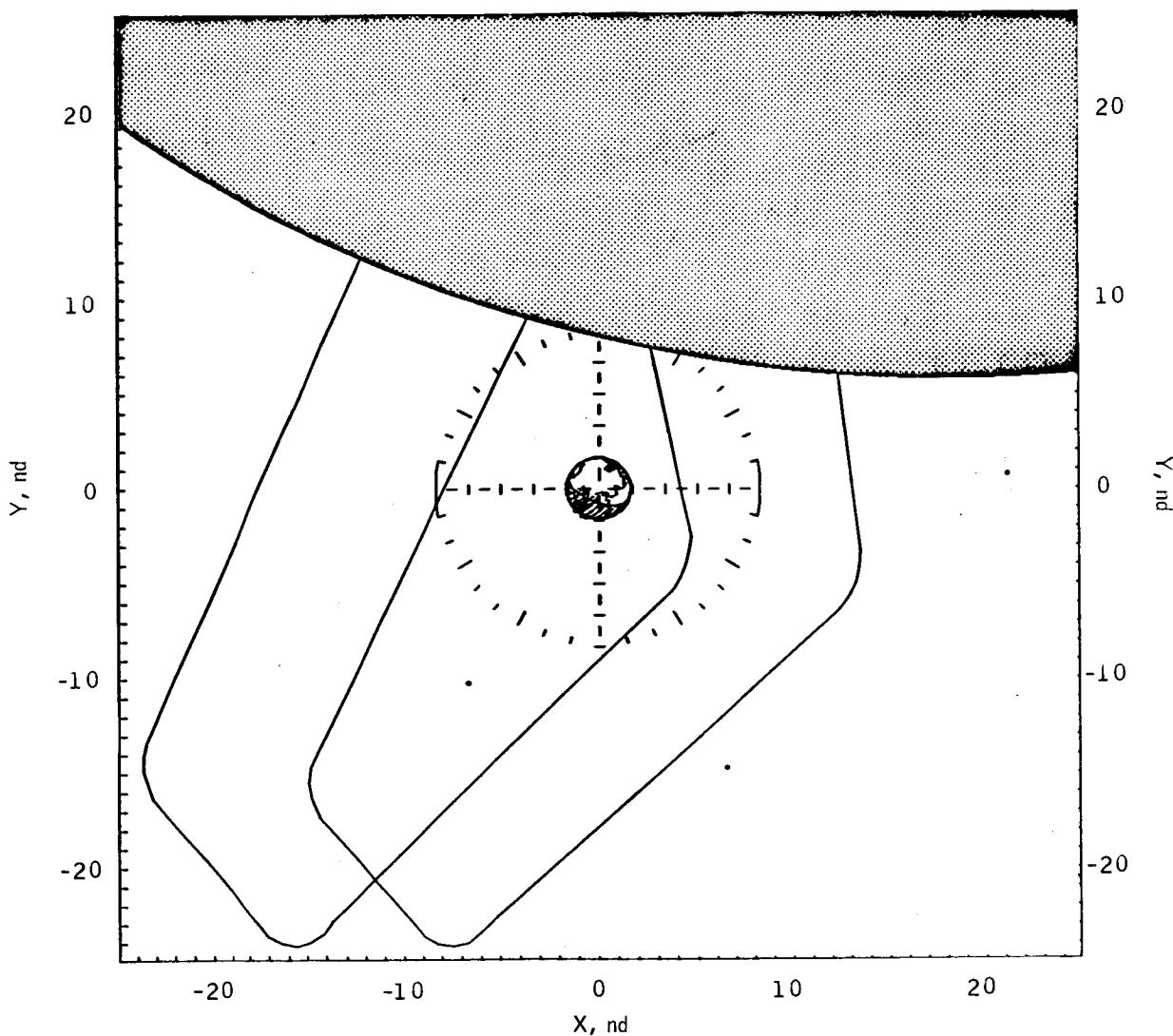


(n) Time from TLI cutoff = 64 hr.

Figure 2.- Continued.

SEQ	473	480	515
X	-6	6	21
Y	-10	-14	0

$$R_M = 1446 \text{ n. mi.}$$
$$V_i = 7324 \text{ fps}$$



(o) Time from TLI cutoff = 66 hr.

Figure 2.- Concluded.

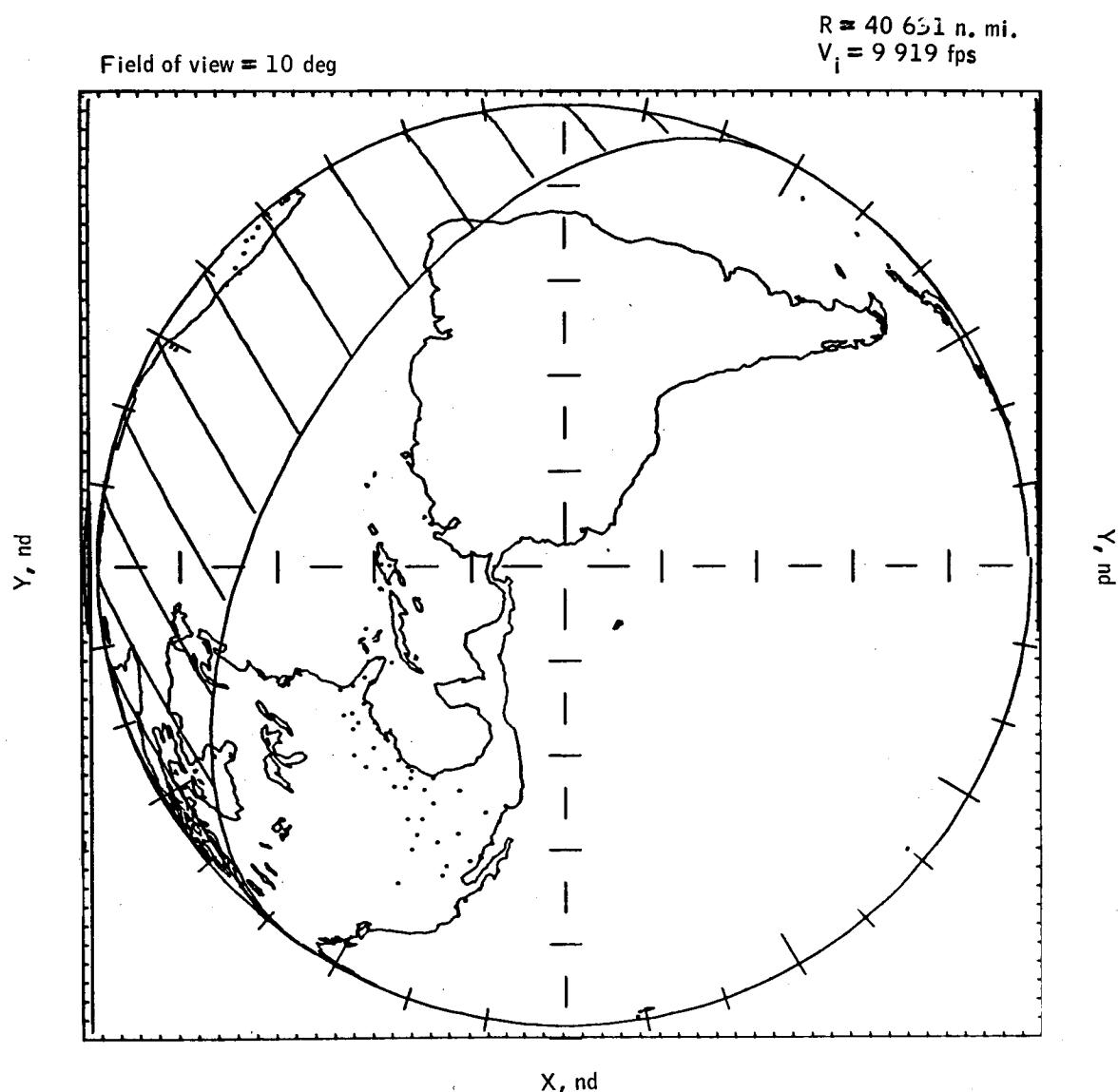
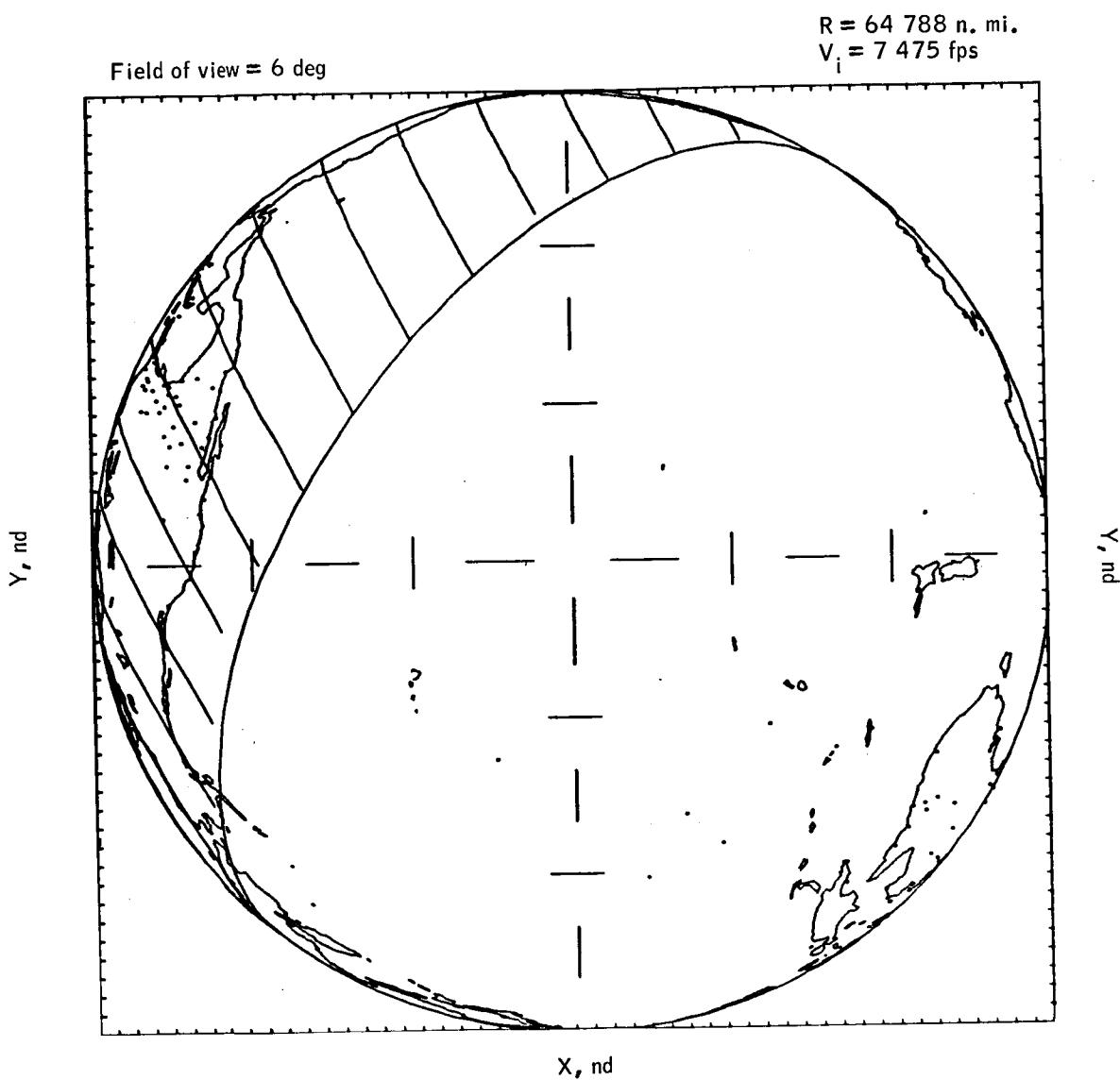
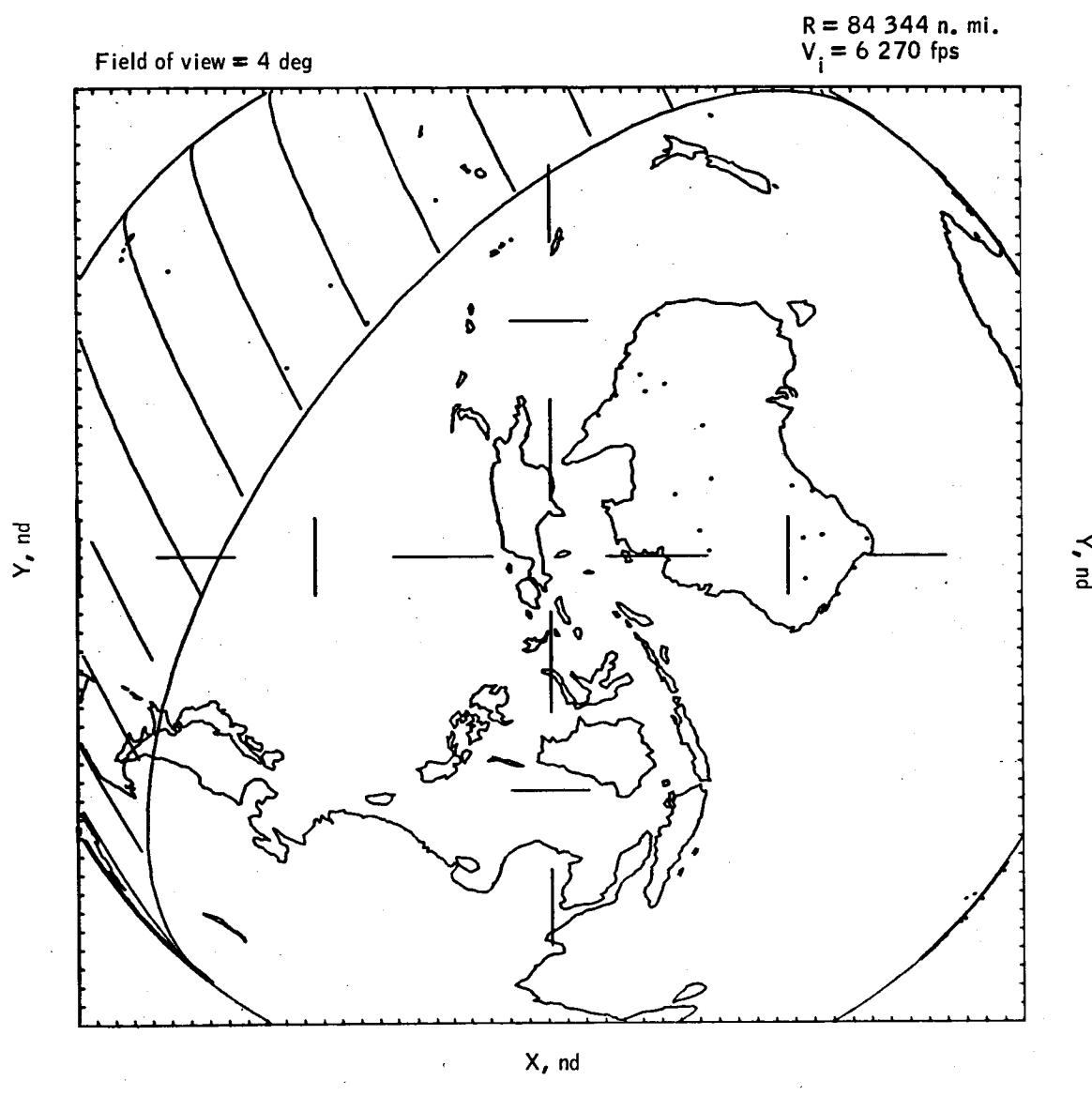
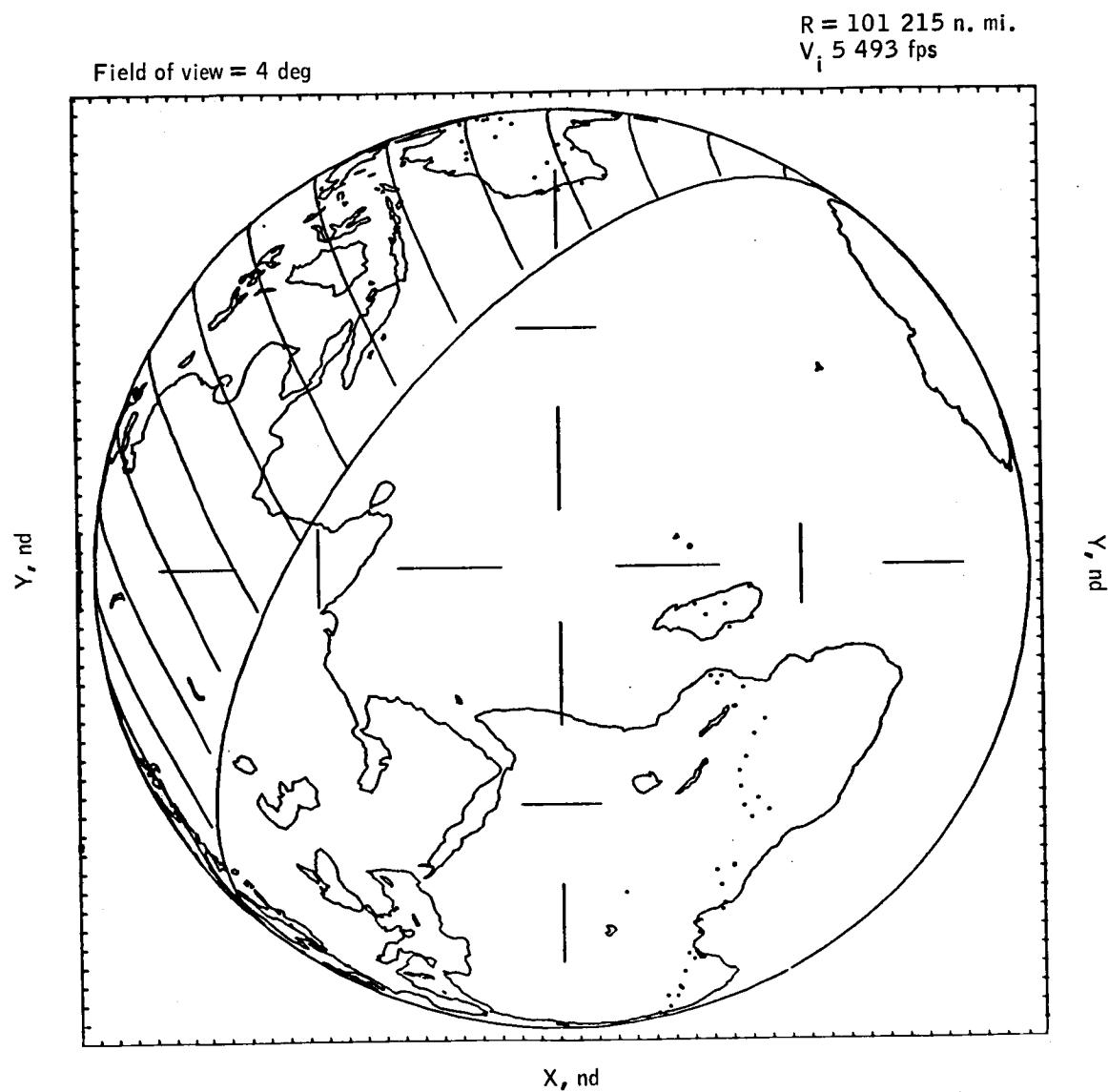


Figure 3.- Translunar coast (earth referenced) - variable field of view.





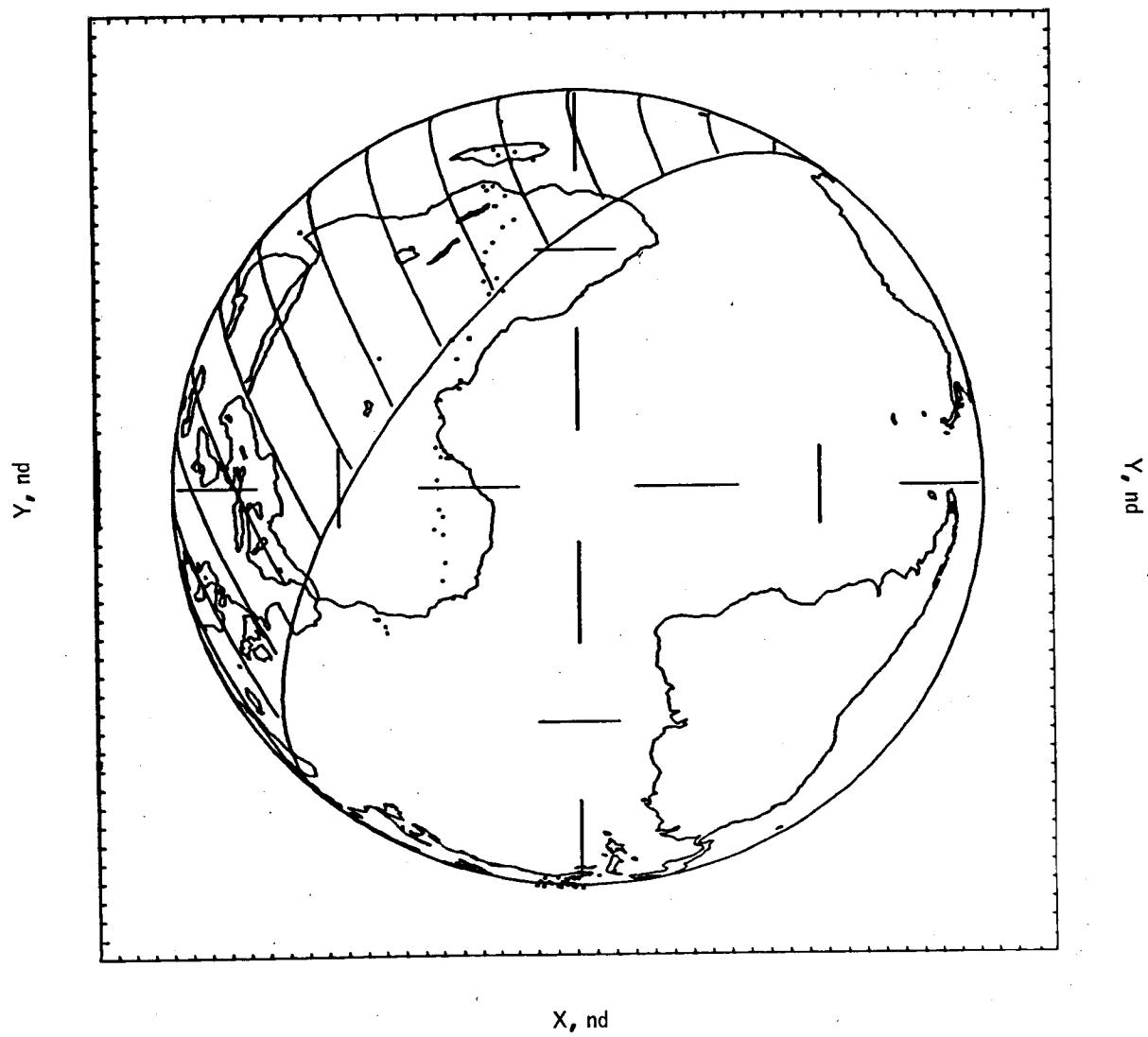


(d) Time from TLI cutoff = 20 hr.

Figure 3.- Continued.

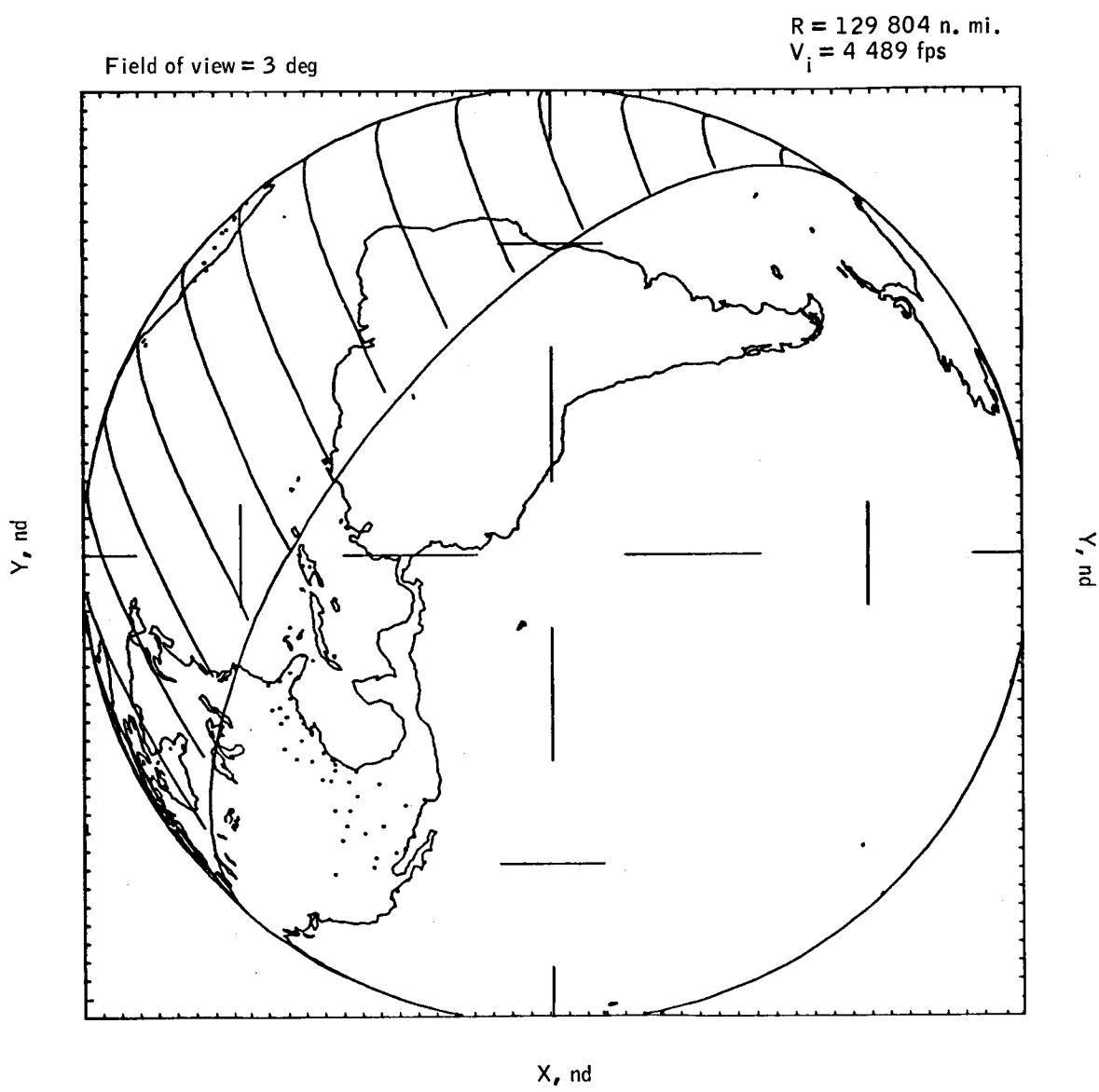
$R = 116\ 218$ n. mi.
 $V_i = 4\ 929$ fps

Field of view = 4 deg



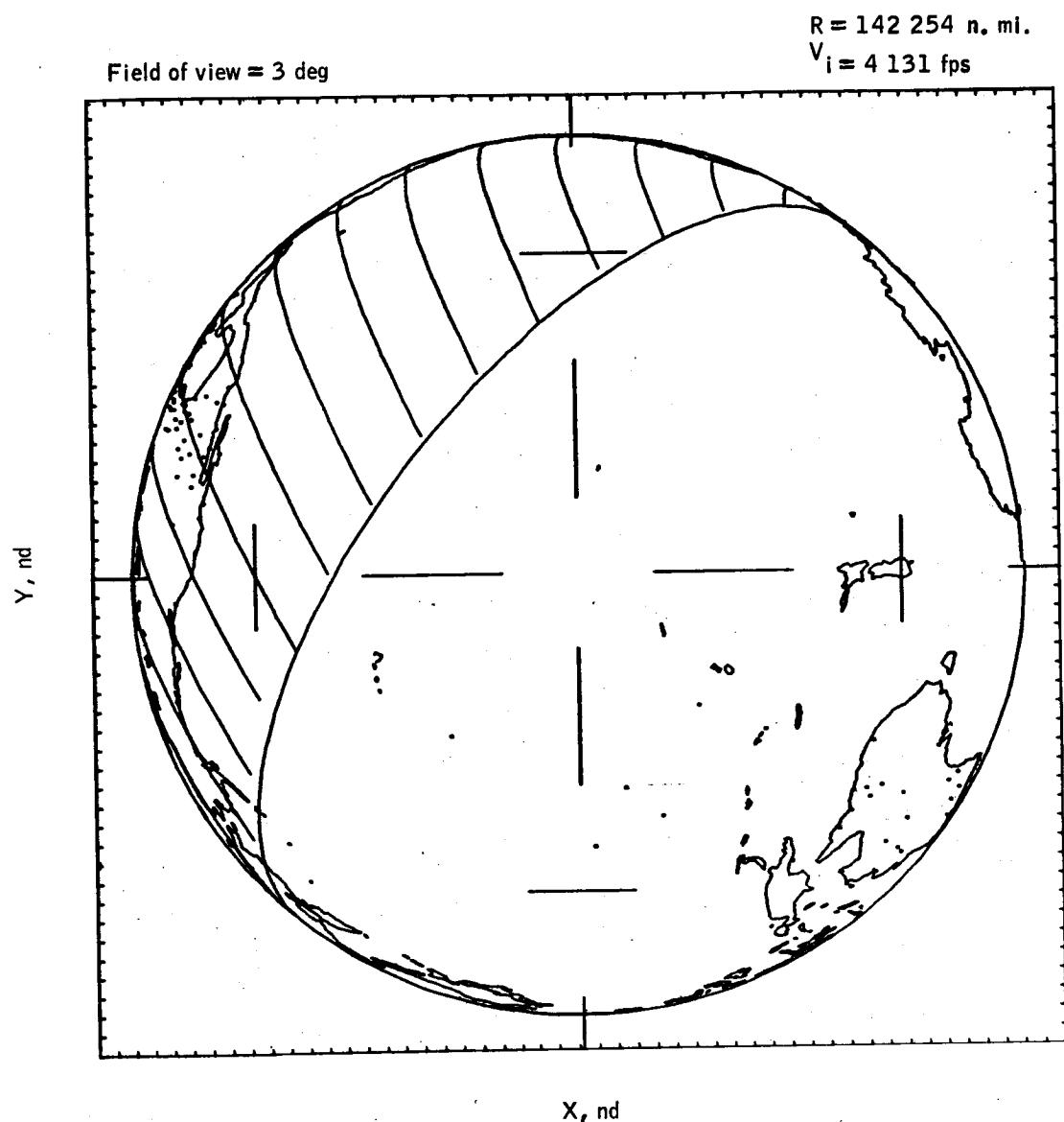
(e) Time from TLI cutoff = 25 hr.

Figure 3.- Continued.



(f) Time from TLI cutoff = 30 hr.

Figure 3.- Continued.



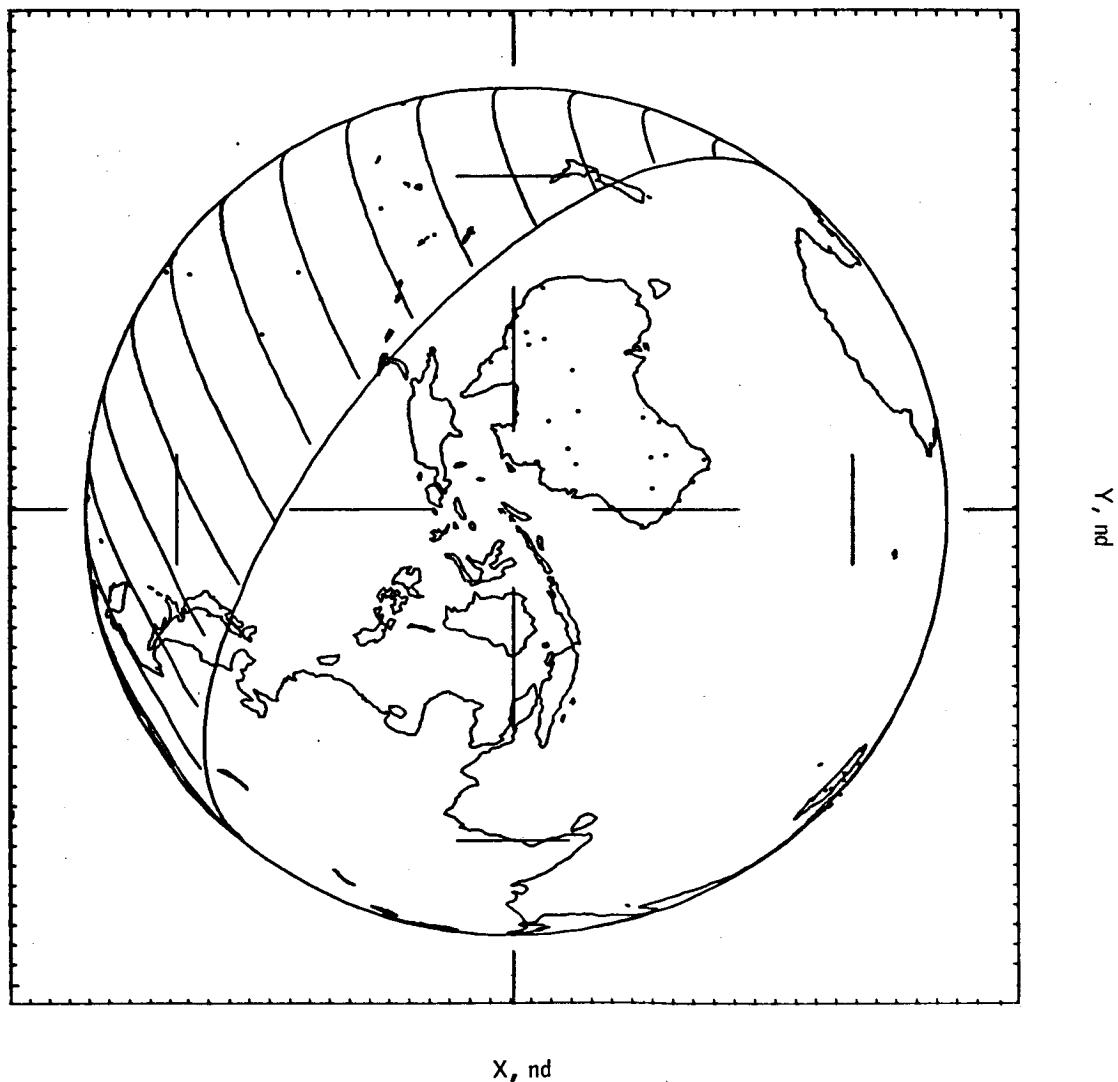
(g) Time from TLI cutoff = 35 hr.

Figure 3.- Continued.

Field of view = 3 deg

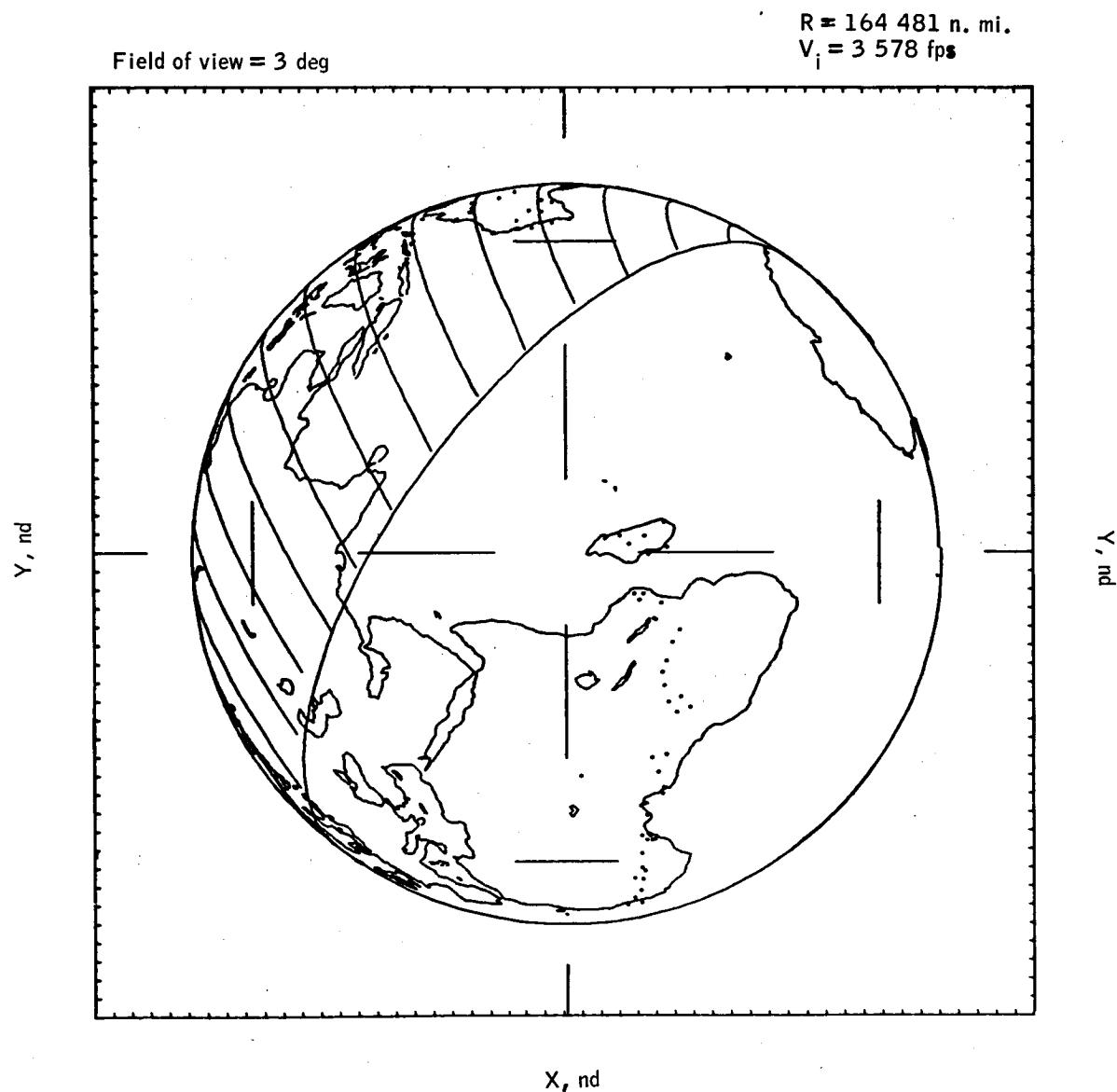
$R = 153\ 765$ n. mi.
 $V_i = 3\ 832$ fps

Y, mi



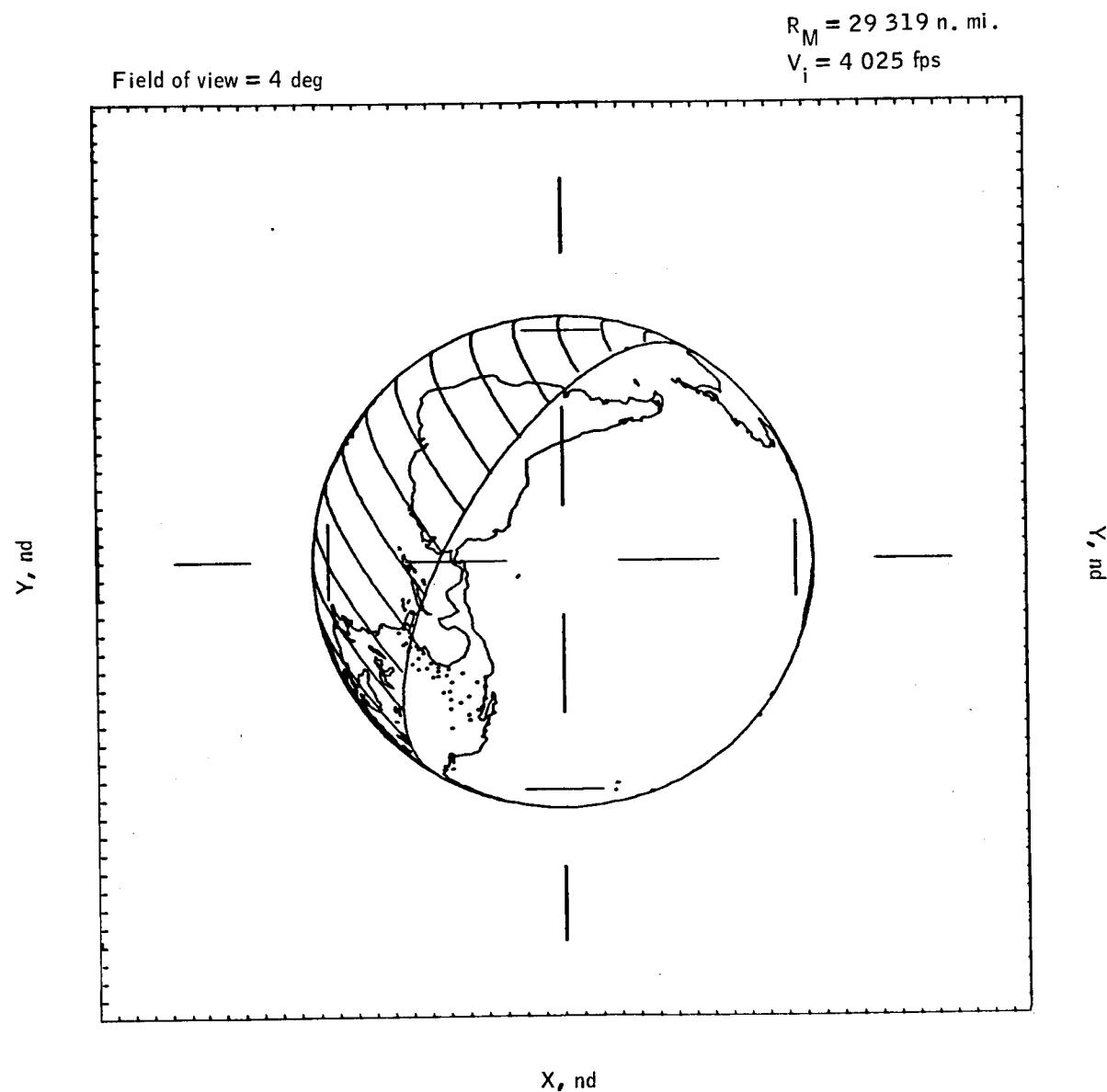
(h) Time from TLI cutoff = 40 hr.

Figure 3.- Continued.



(i) Time from TLI cutoff = 45 hr.

Figure 3.- Continued.

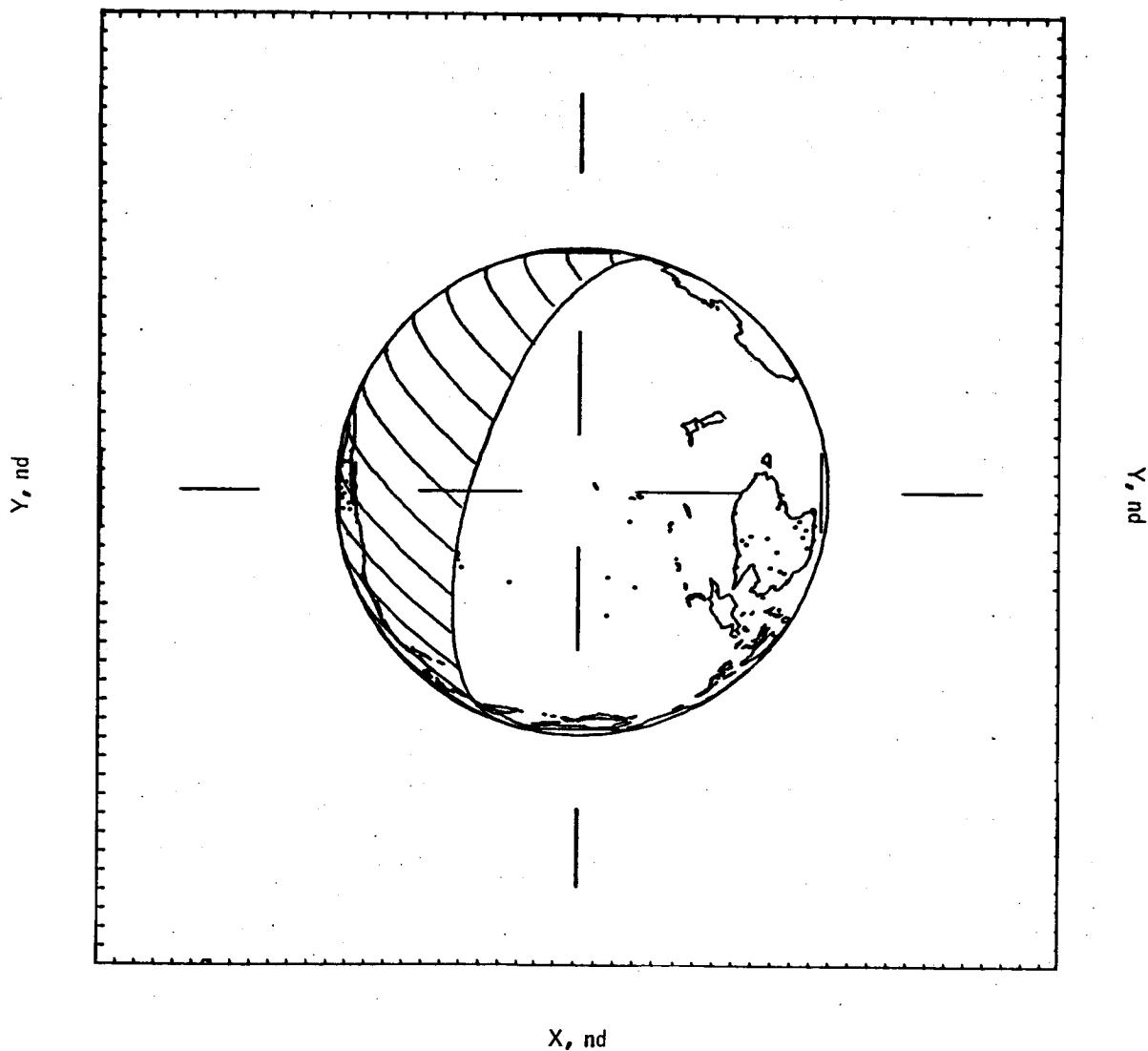


(j) Time from TLI cutoff = 55 hr.

Figure 3.- Continued.

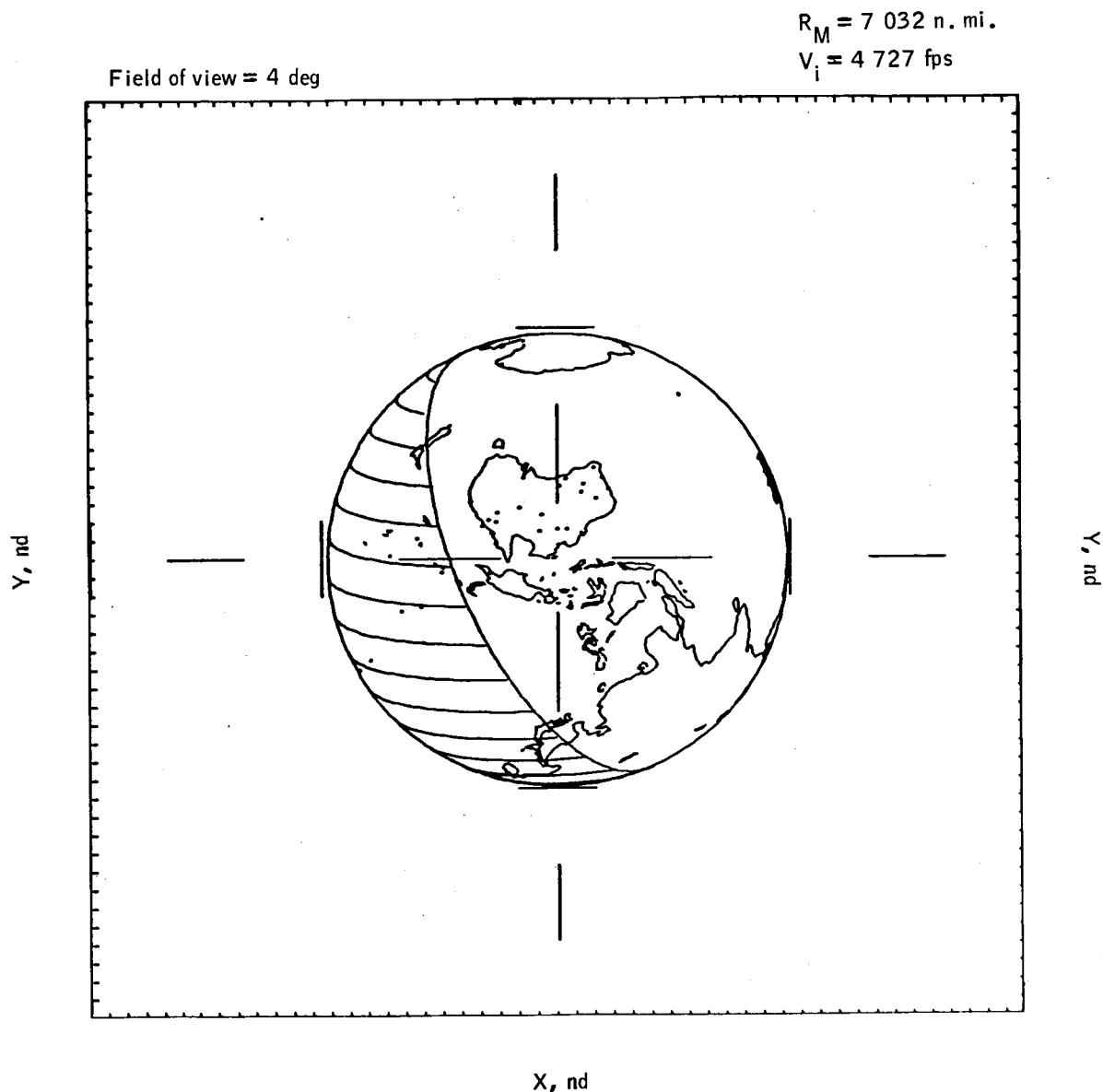
$R_M = 17\ 254$ n. mi.
 $V_i = 4\ 187$ fps

Field of view = 4 deg



(k) Time from TLI cutoff = 60 hr.

Figure 3.- Continued.

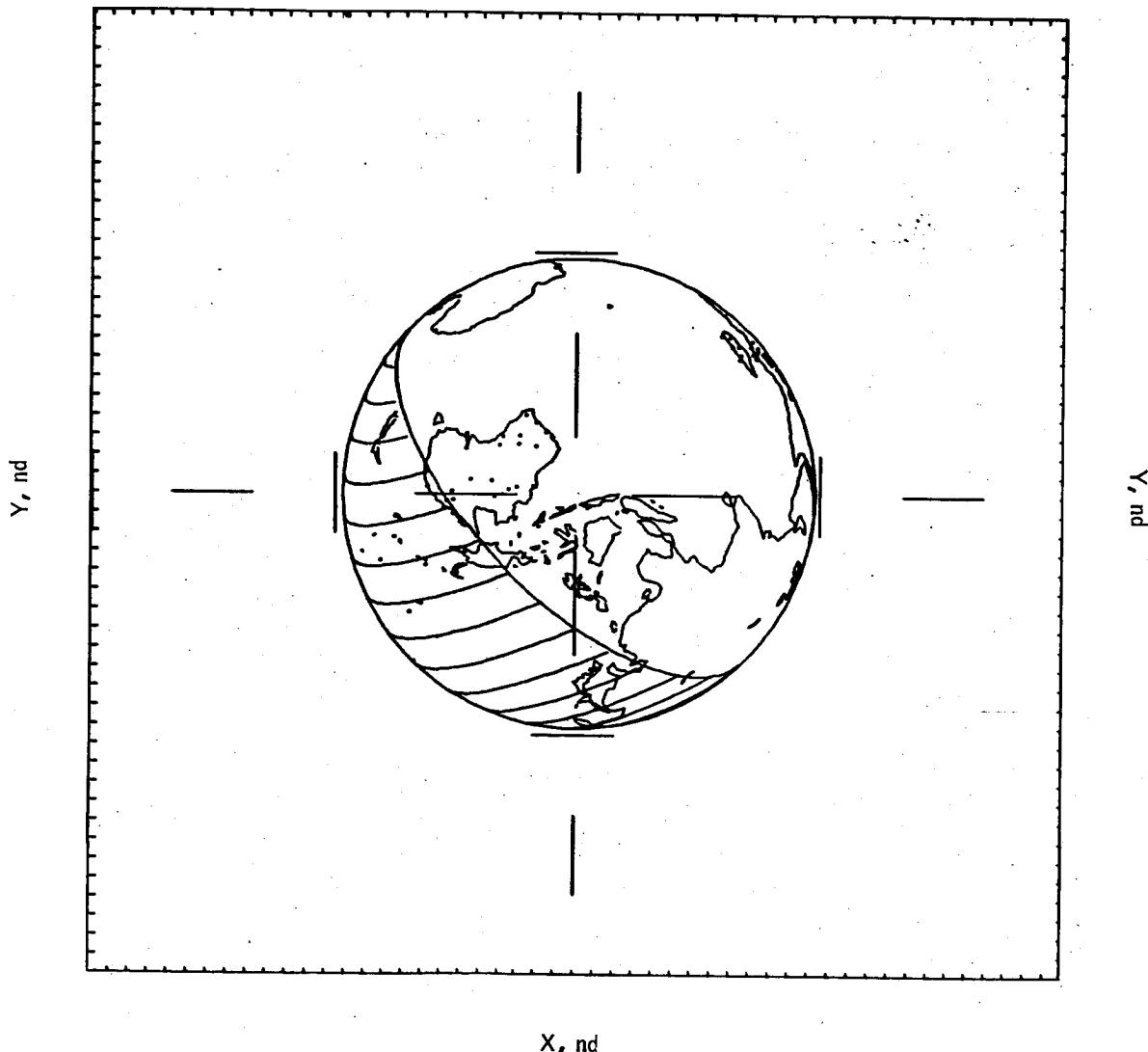


(1) Time from TLI cutoff = 64 hr.

Figure 3.- Continued.

$R_M = 1206 \text{ n. mi.}$
 $V_i = 7840 \text{ fps}$

Field of view = 4 deg



(m) Time from TLI cutoff = 65 hr.

Figure 3.- Continued.

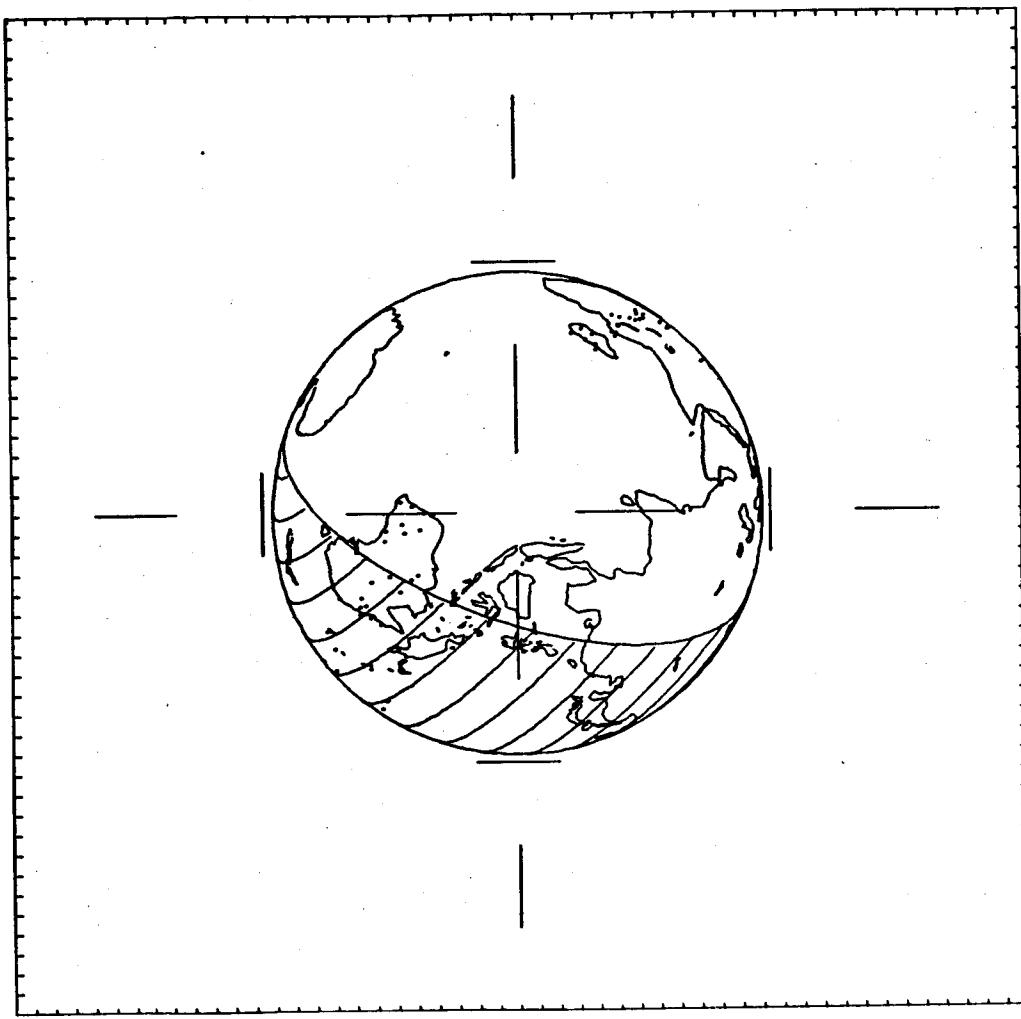
$$R_M = 4089 \text{ n.mi.}$$
$$V_i = 5308 \text{ fpm}$$

Field of view = 4 deg

Y, nd

Y, nd

X, nd



(n) Time from TLI cutoff = 66 hr.

Figure 3.- Concluded.

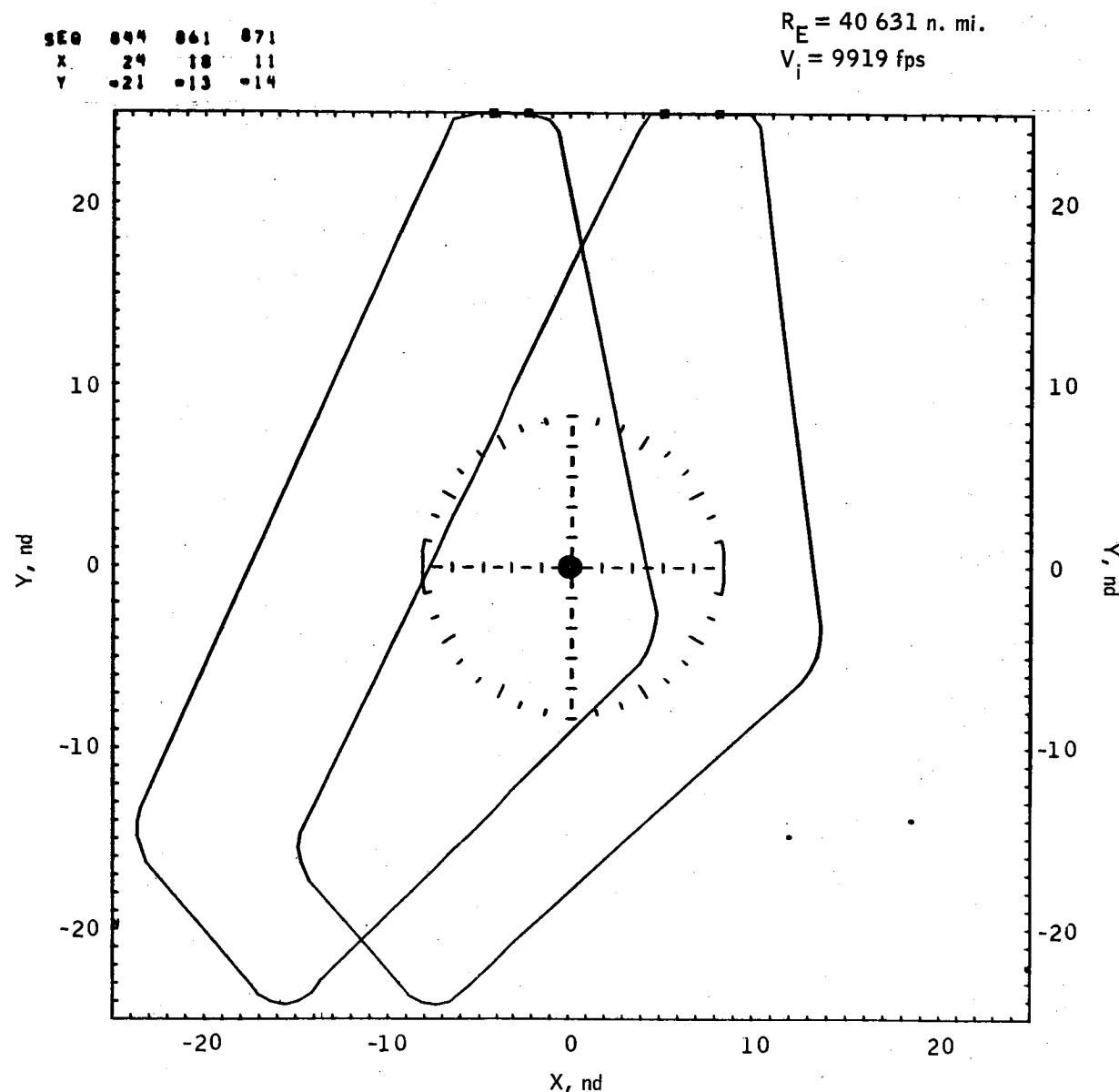
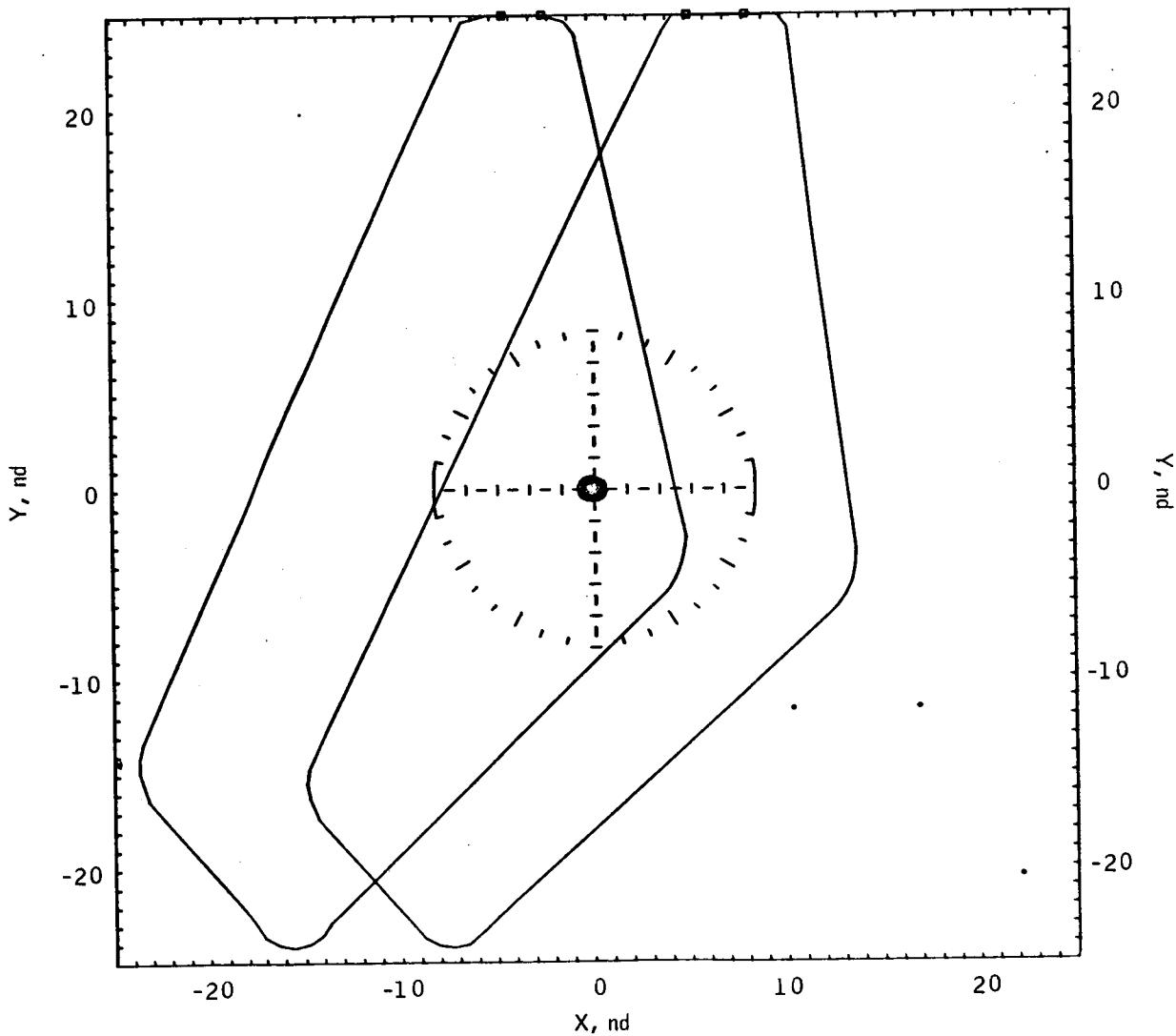


Figure 4.- Translunar coast (moon referenced)-constant field of view.

SEQ	844	861	871
X	22	16	10
Y	-20	-11	-11

$$R_E = 64788 \text{ n. mi.}$$

$$V_i = 7475 \text{ fps}$$



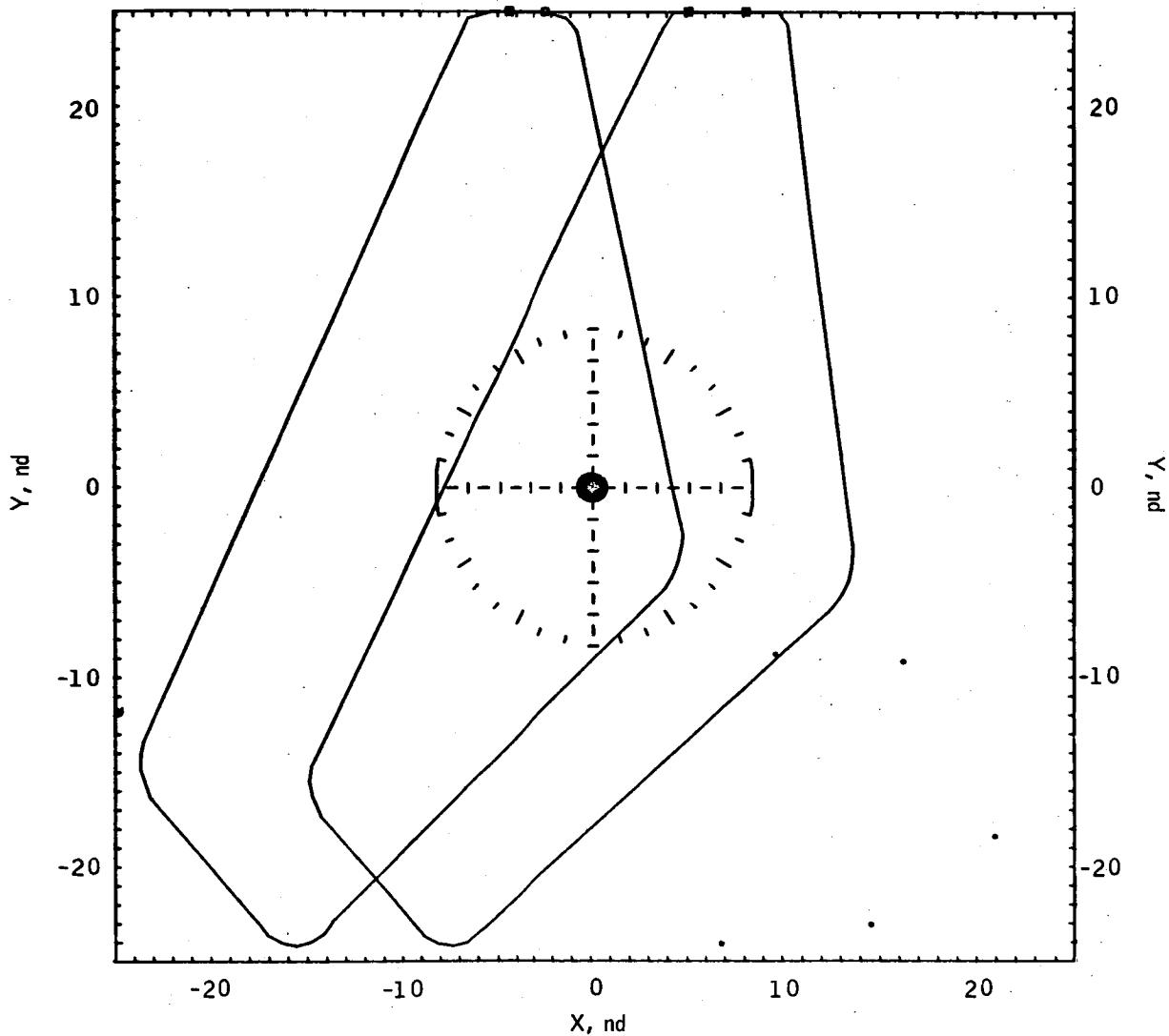
(b) Time from TLI cutoff = 10 hr.

Figure 4.- Continued.

SEG	836	841	844	861	871
X	14	6	20	16	9
Y	-22	-23	-18	-9	-8

$$R_E = 84\,345 \text{ n. mi.}$$

$$V_i = 6270 \text{ fps}$$



(c) Time from TLI cutoff = 15 hr.

Figure 4.- Continued.

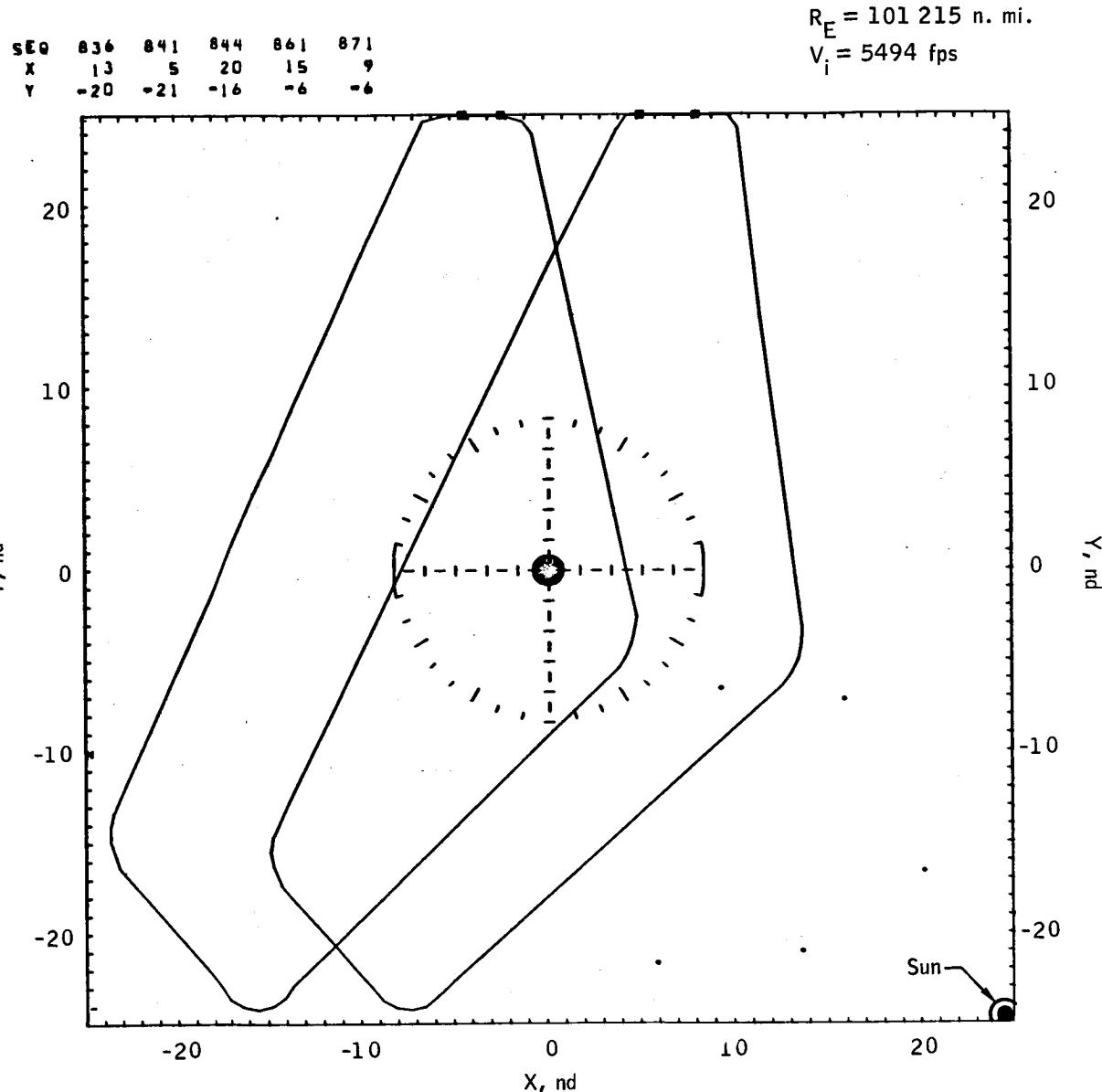


Figure 4.- Continued.

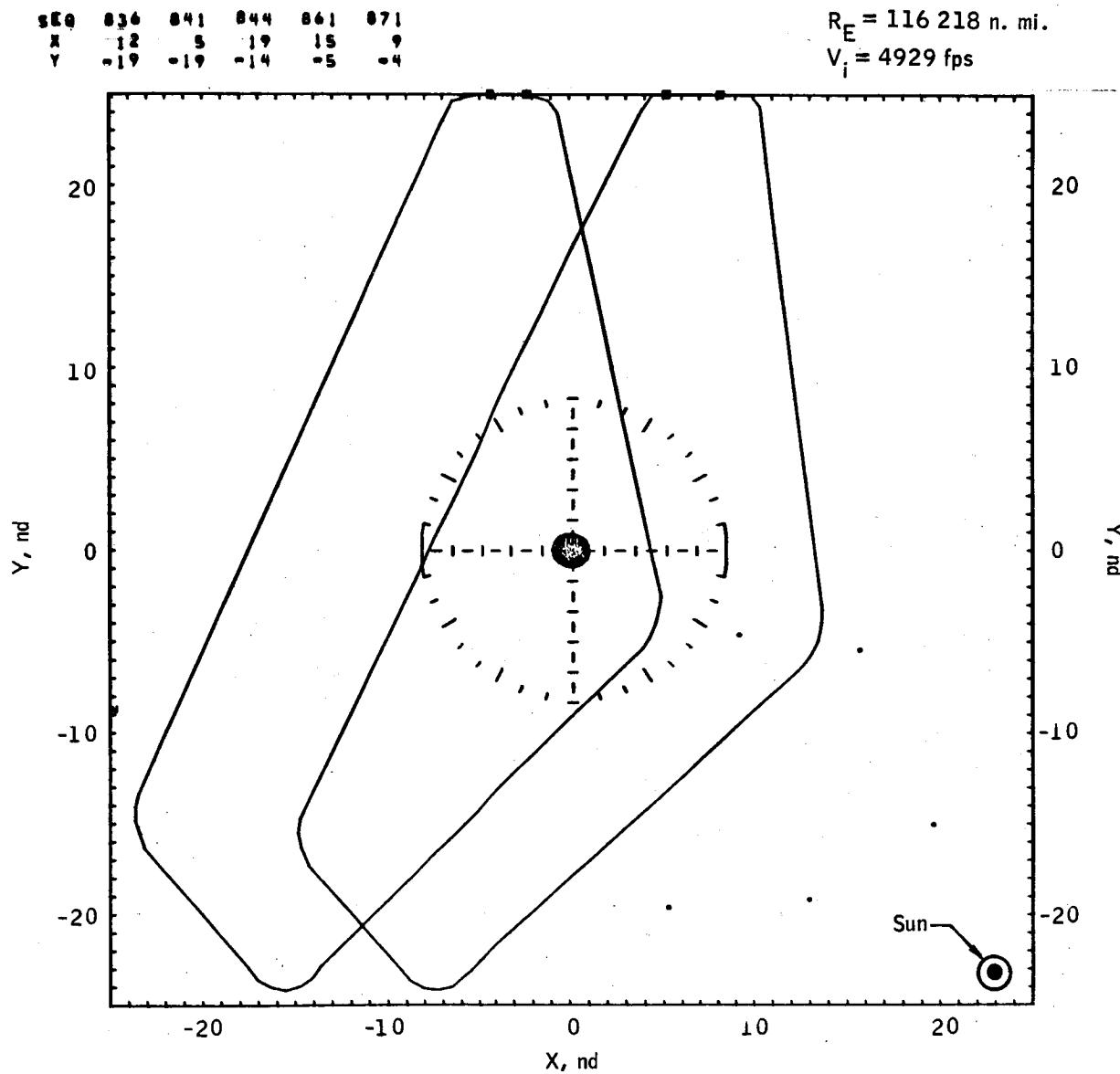
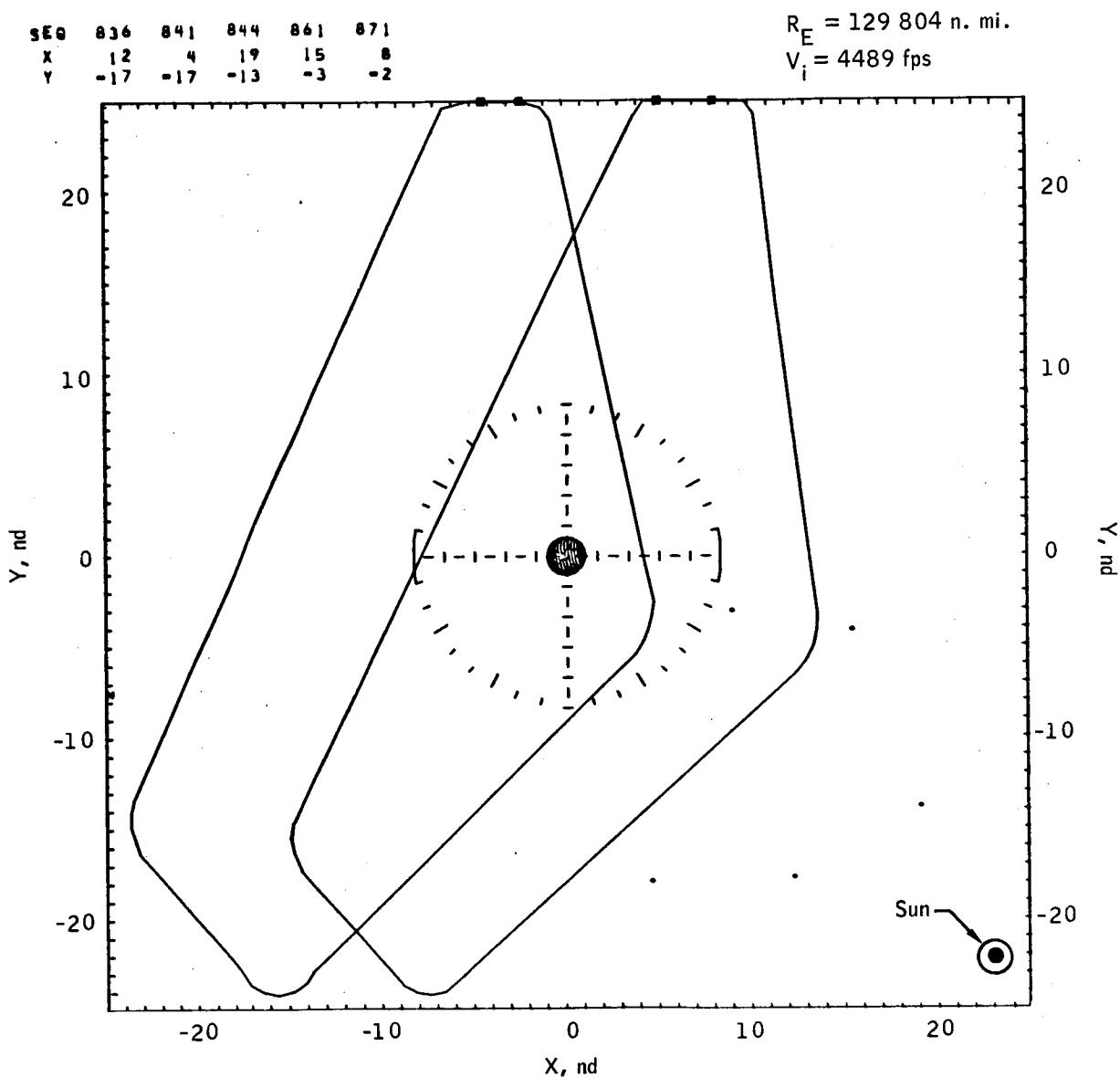


Figure 4.- Continued.



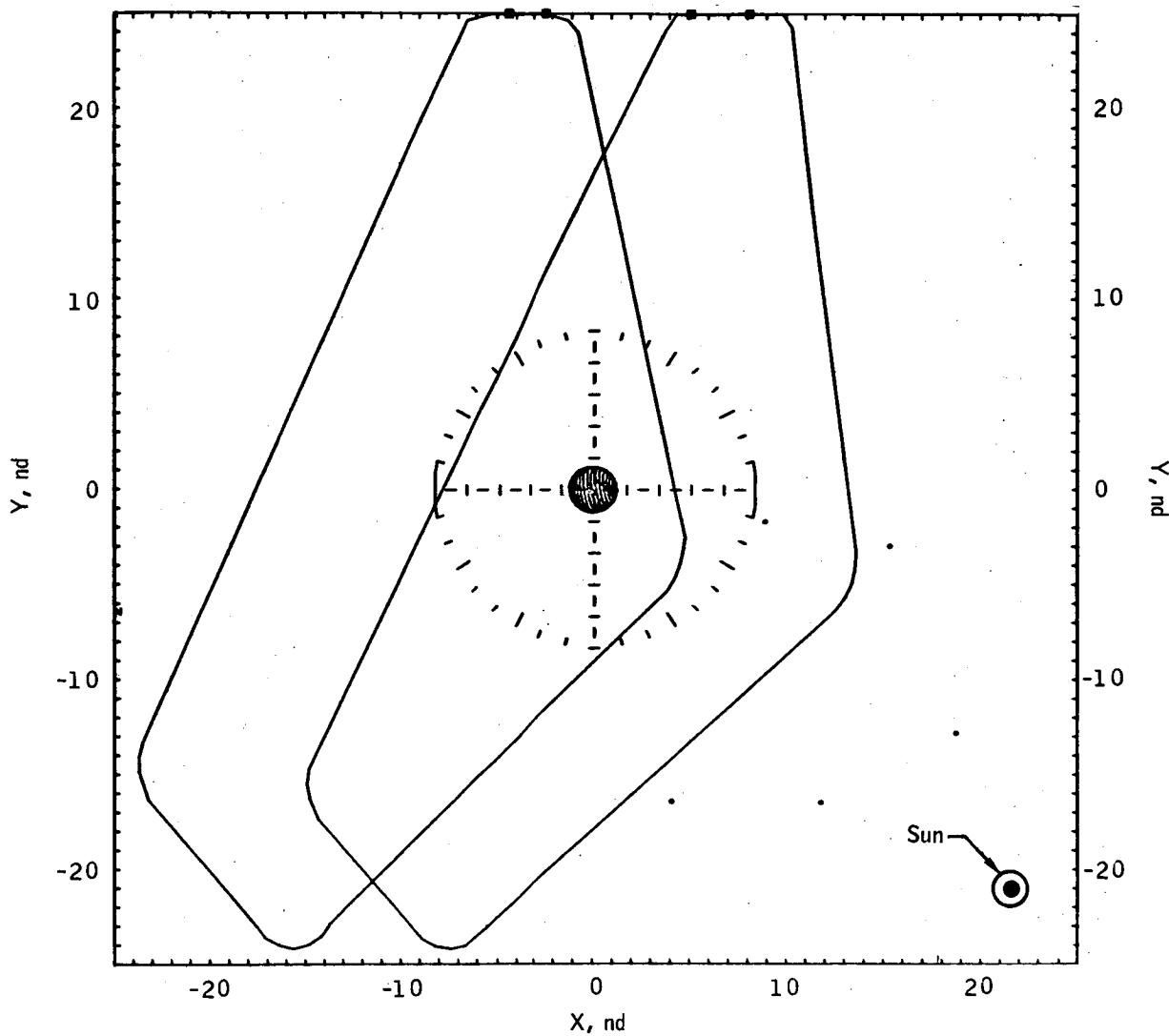
(f) Time from TLI cutoff = 30 hr.

Figure 4.- Continued.

sE0	836	841	844	861	871
X	-13	3	-18	15	0
Y	-16	-16	-12	-2	-1

$$R_E = 142\ 255 \text{ n. mi.}$$

$$V_i = 4132 \text{ fpm}$$



(g) Time from TLI cutoff = 35 hr.

Figure 4.- Continued.

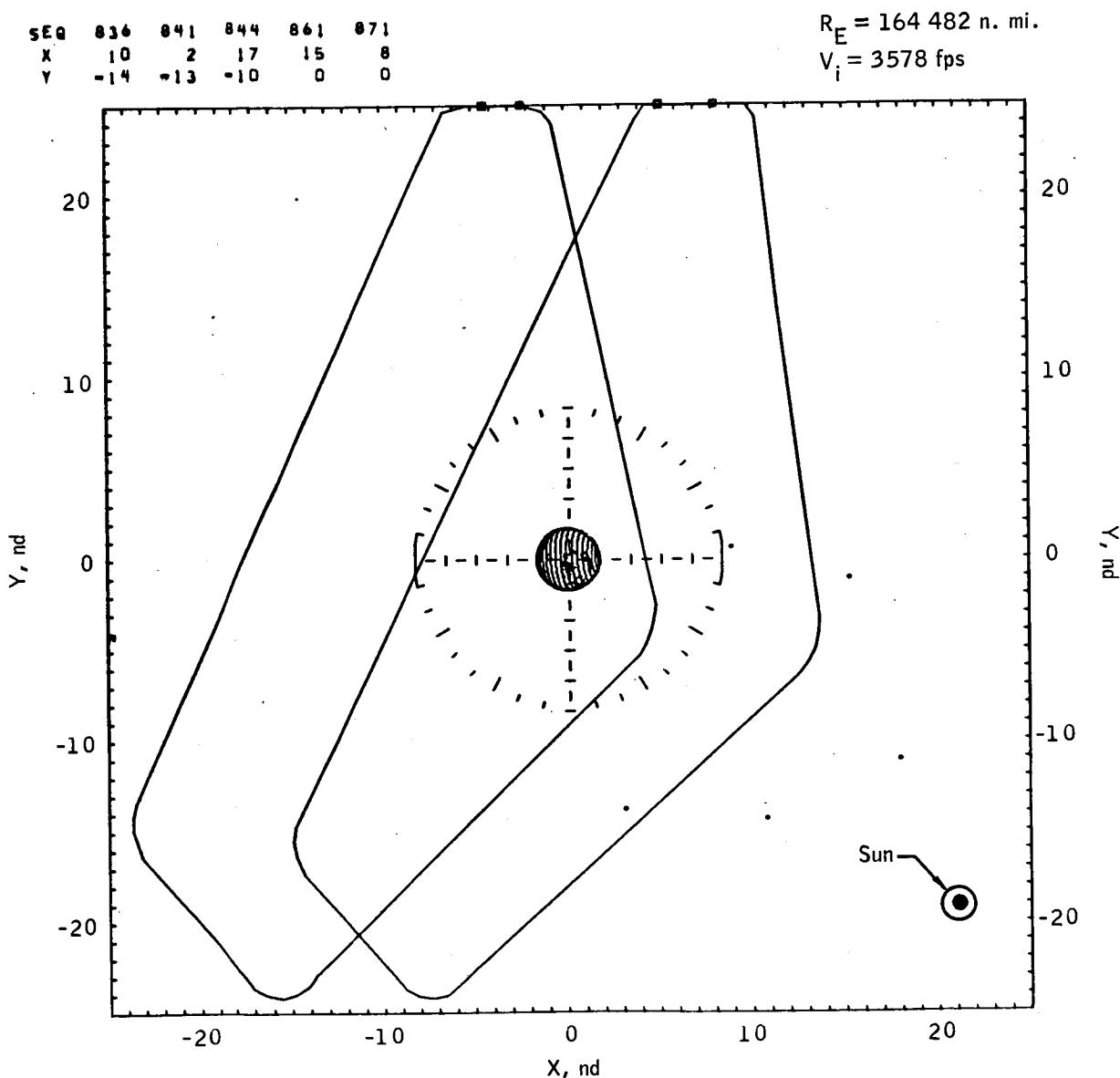


Figure 4.- Continued.

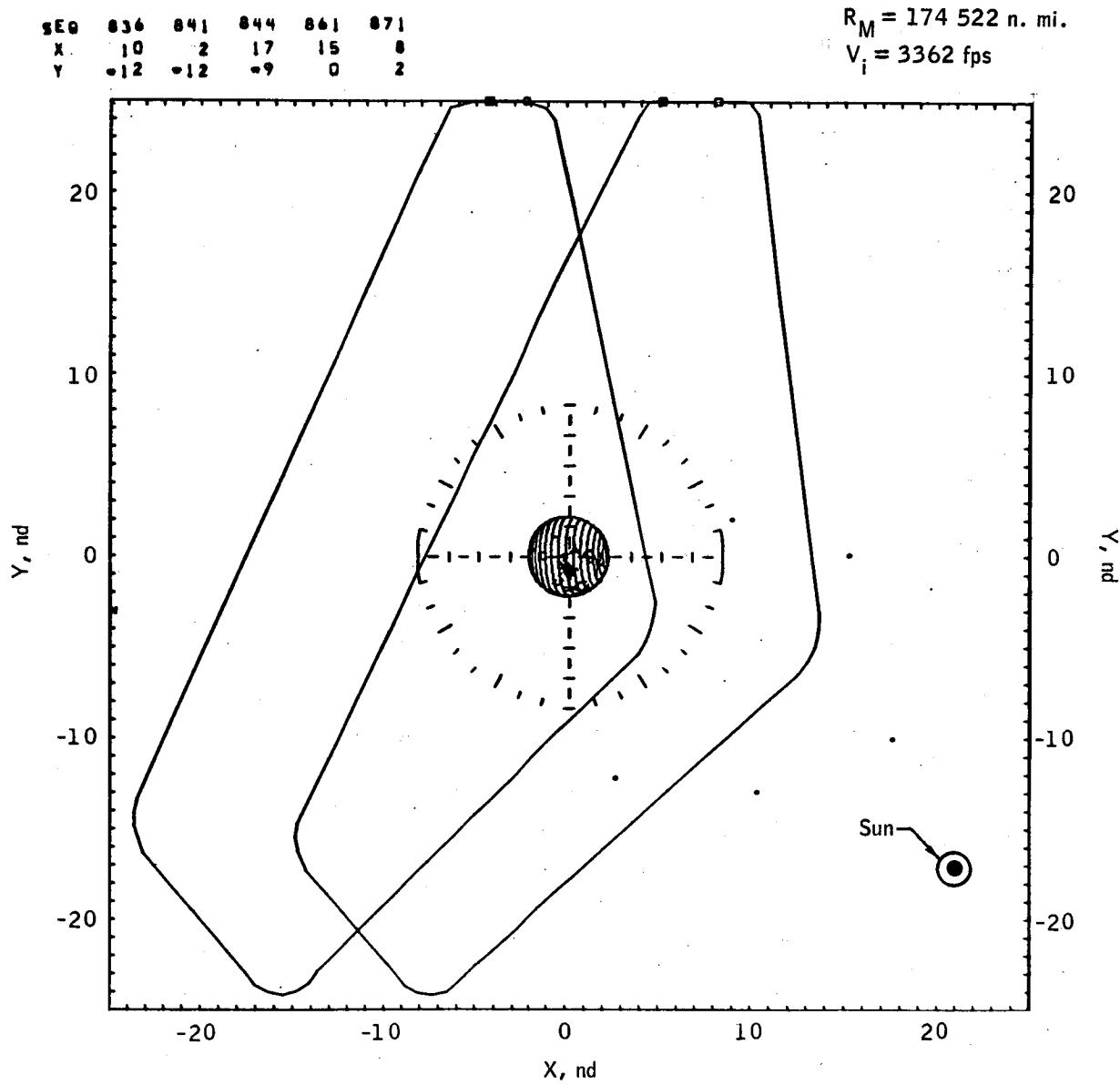


Figure 4.- Continued.

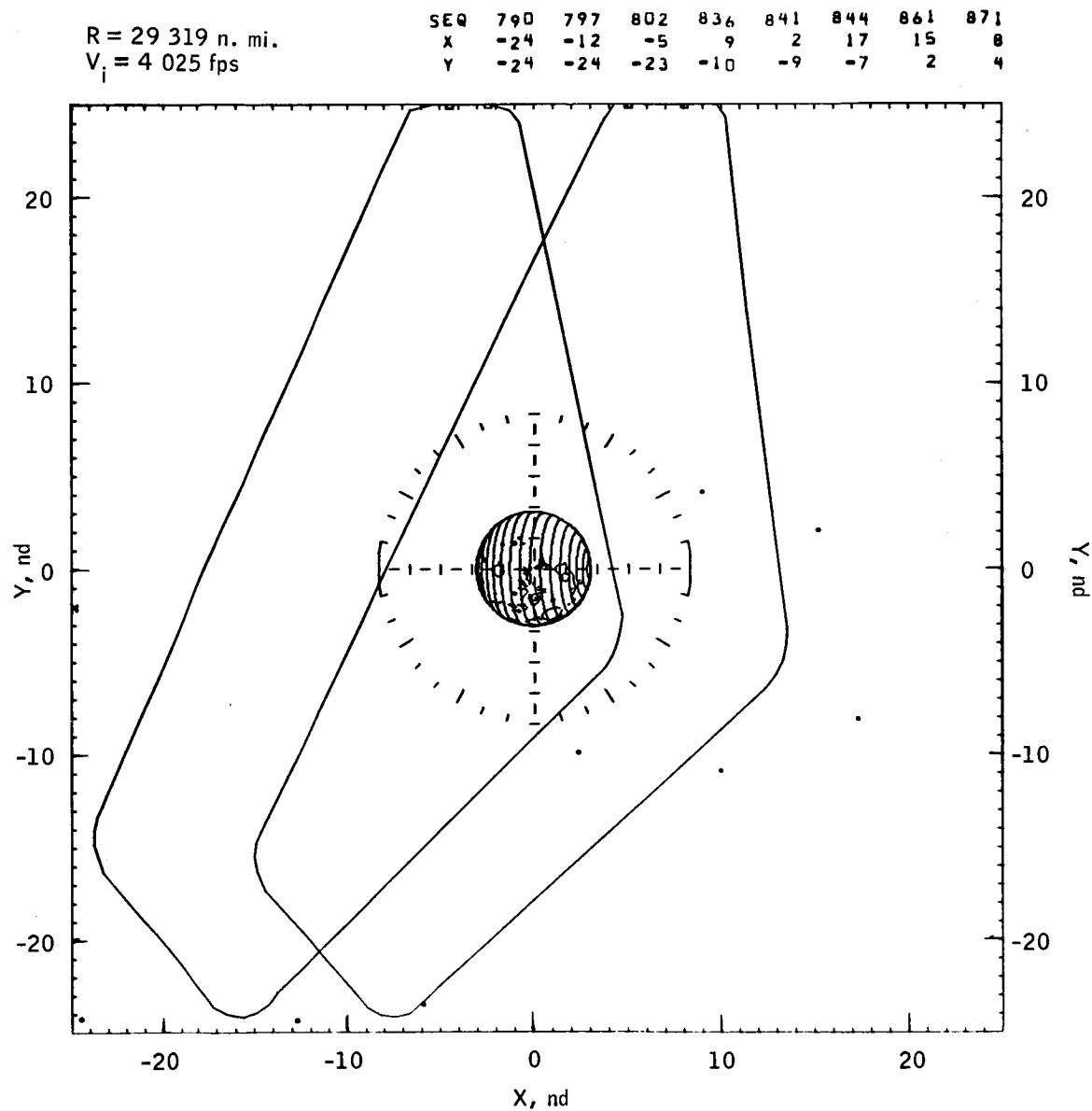


Figure 4.- Continued.

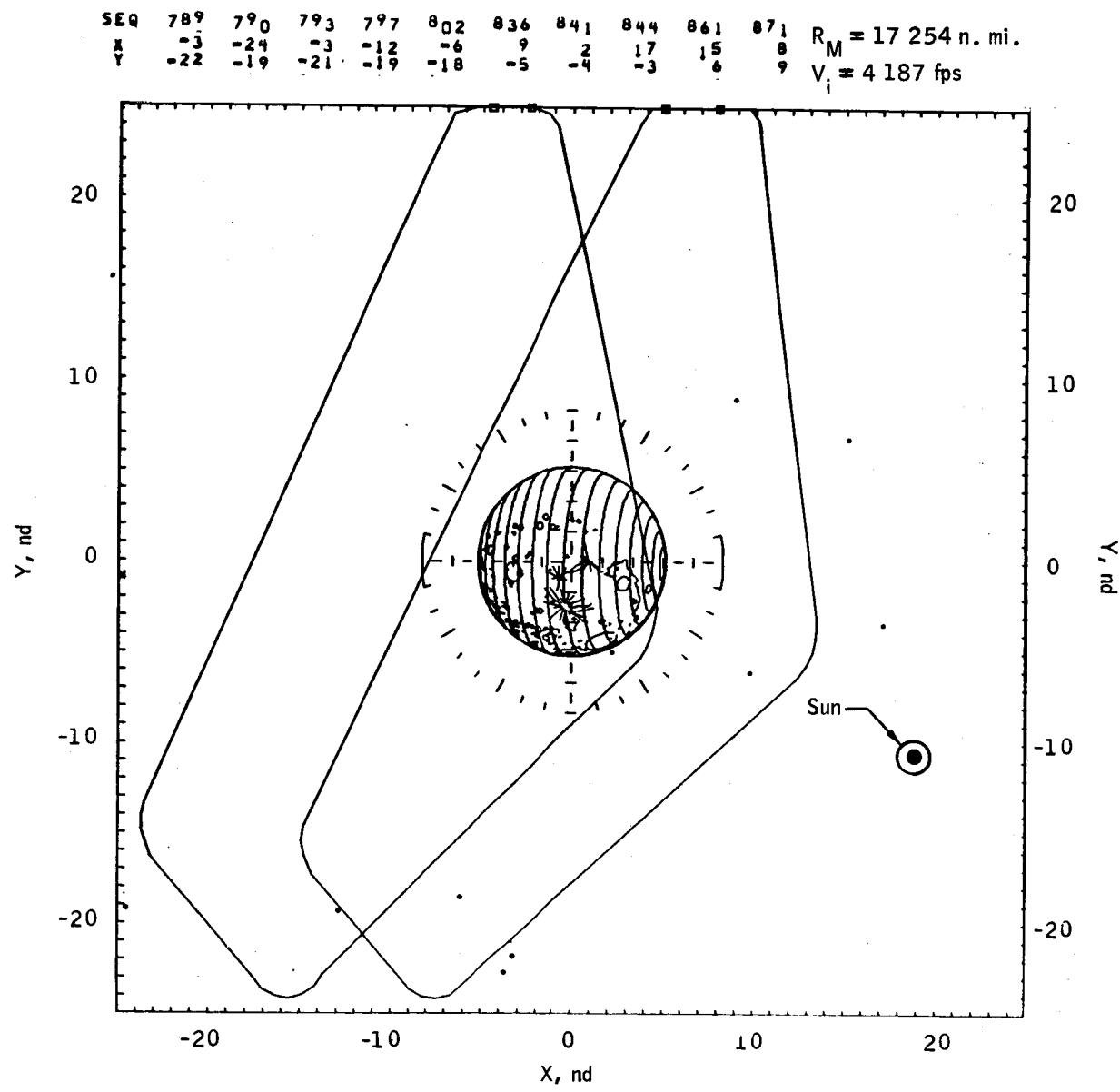
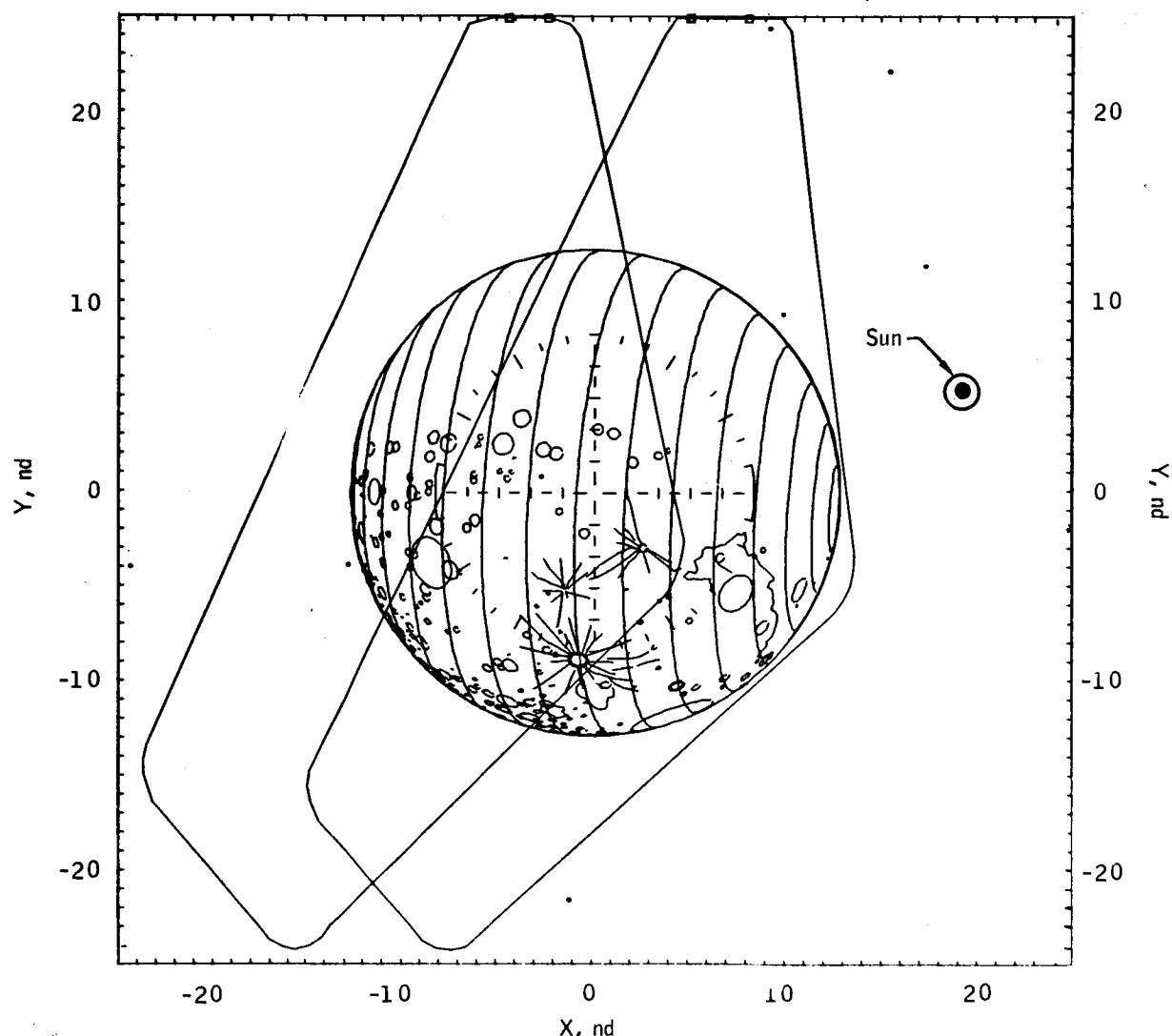


Figure 4.- Continued.

SEQ	759	790	836	844	661
X	-1	-24	9	17	15
Y	-20	-2	10	13	23

$$R_M = 7032 \text{ n.mi.}$$

$$V_i = 4726 \text{ fps}$$



(I) Time from TLI cutoff = 64 hr.

Figure 4.- Continued.

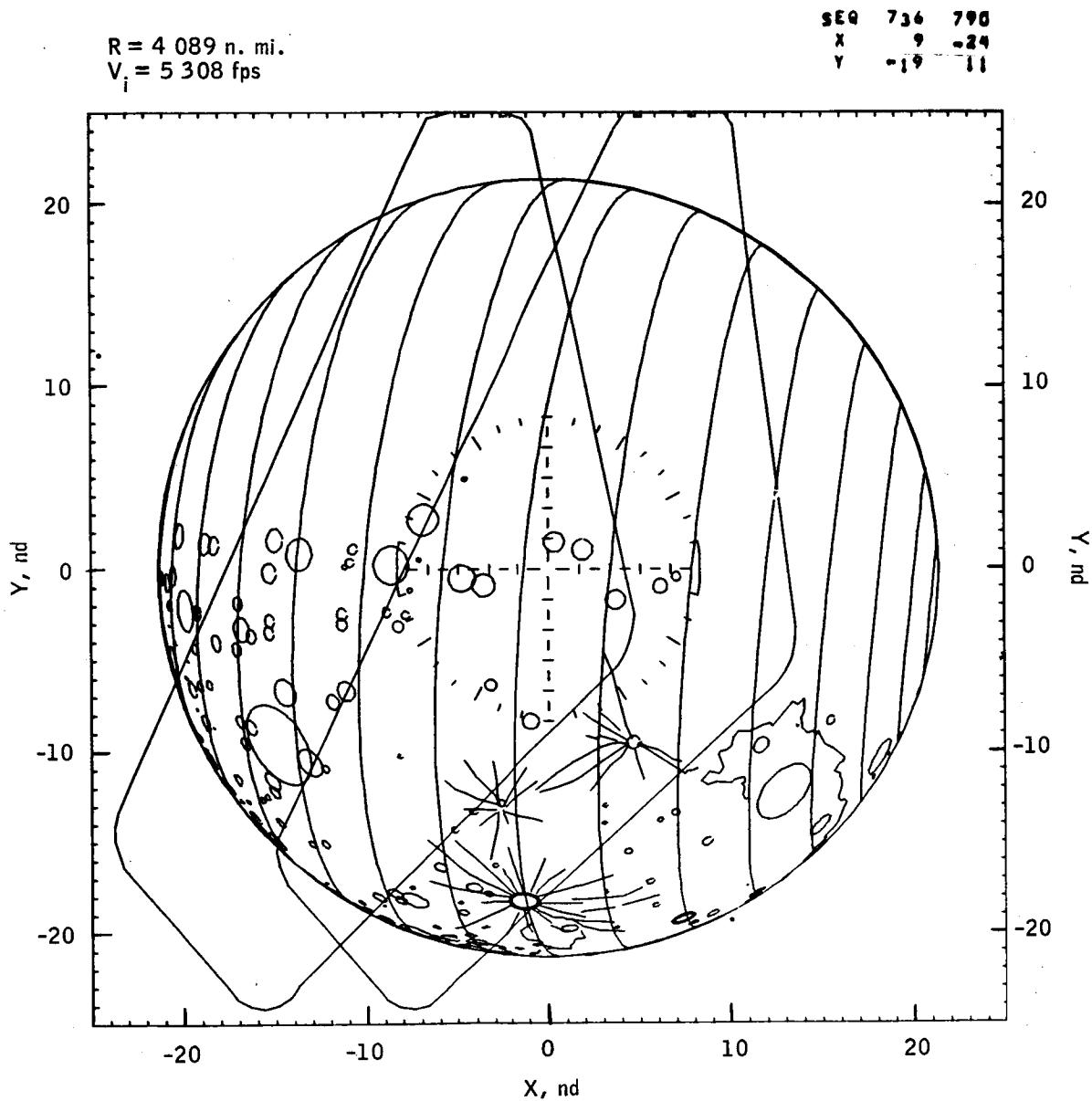
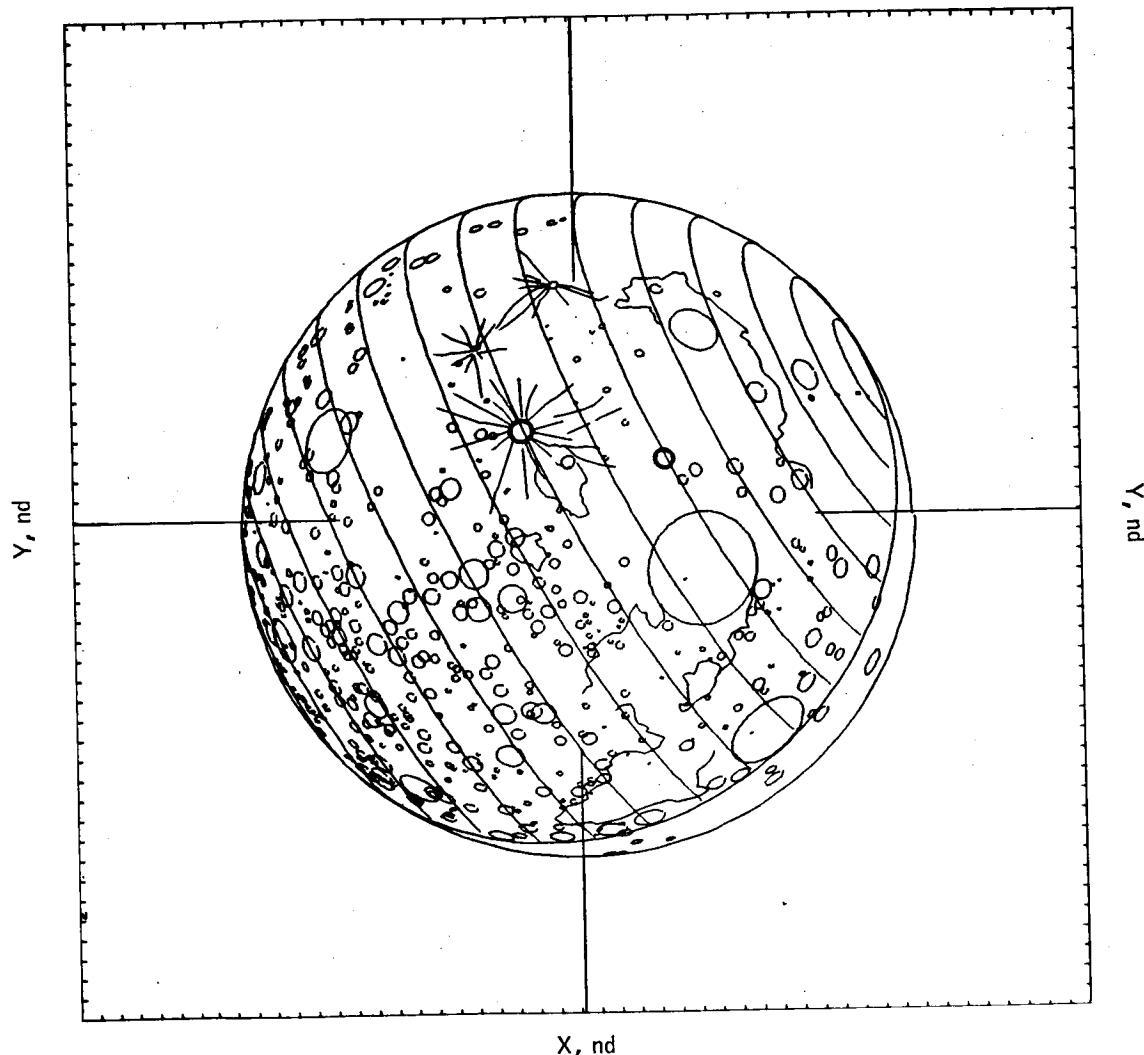


Figure 4.- Concluded.

$$R_E = 40\,631 \text{ n. mi.}$$
$$V_i = 9919 \text{ fps}$$

Field of view = 1 deg

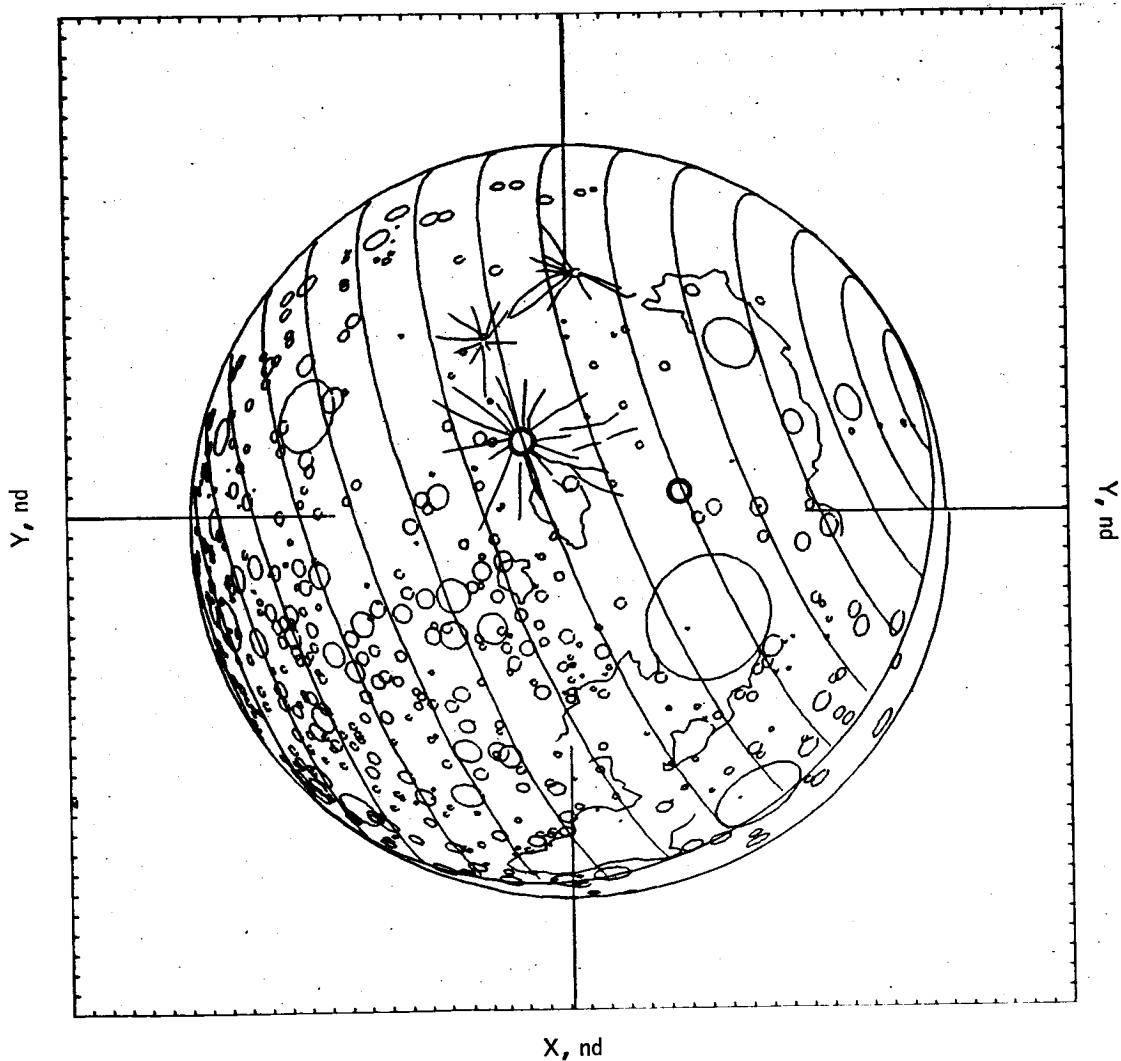


(a) Time from TLI cutoff = 5 hr.

Figure 5.- Translunar coast (moon referenced) variable field of view.

$$R_E = 64788 \text{ n. mi.}$$
$$V_i = 7475 \text{ fps}$$

Field of view = 1 deg

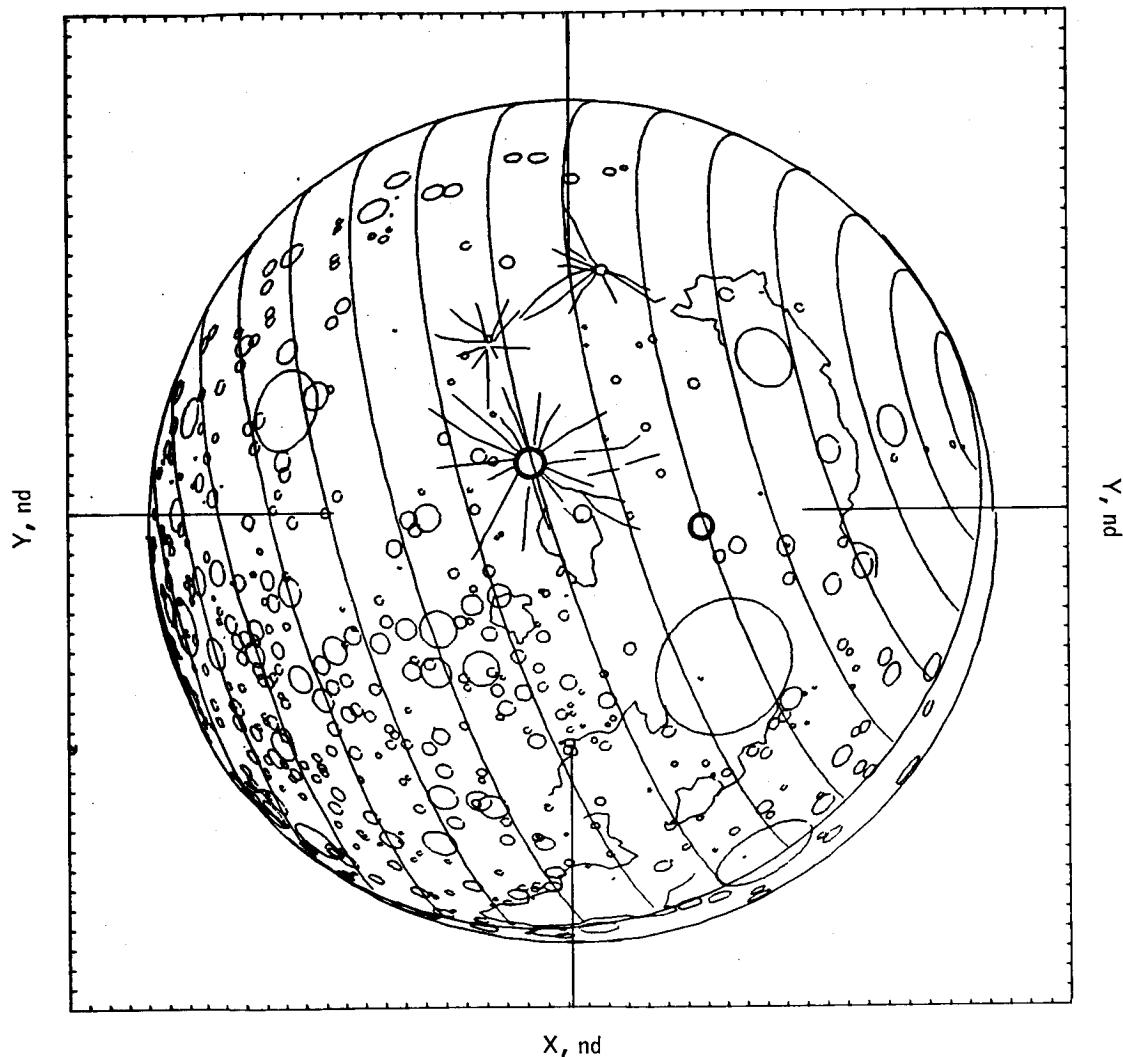


(b) Time from TLI cutoff = 10 hr.

Figure 5.- Continued.

$$R_E = 84\ 345 \text{ n. mi.}$$
$$V_i = 6270 \text{ fps}$$

Field of view = 1 deg

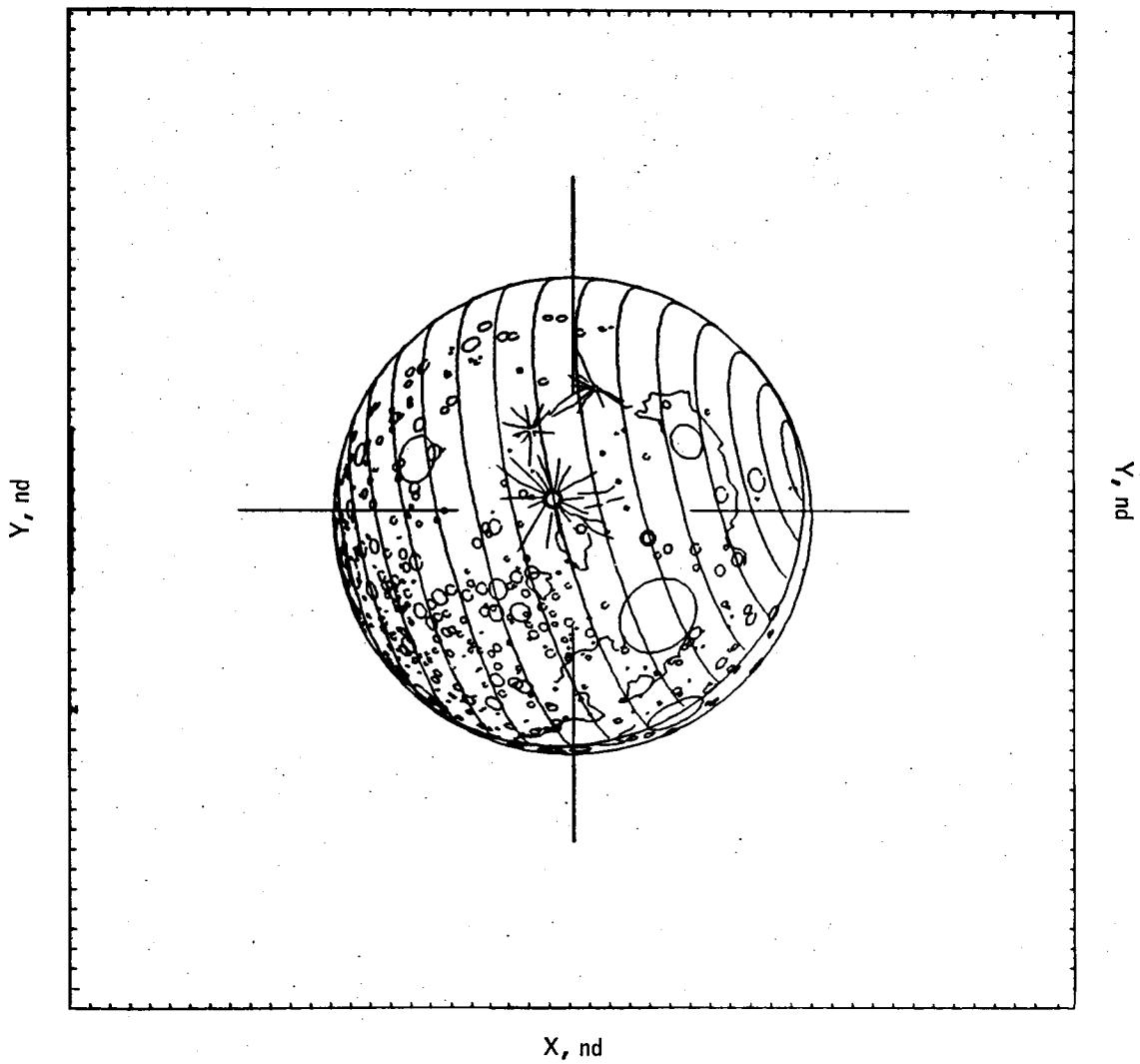


(c) Time from TLI cutoff = 15 hr.

Figure 5.- Continued.

$R_E = 101\ 215$ n. mi.
 $V_i = 5494$ fps

Field of view = 2 deg

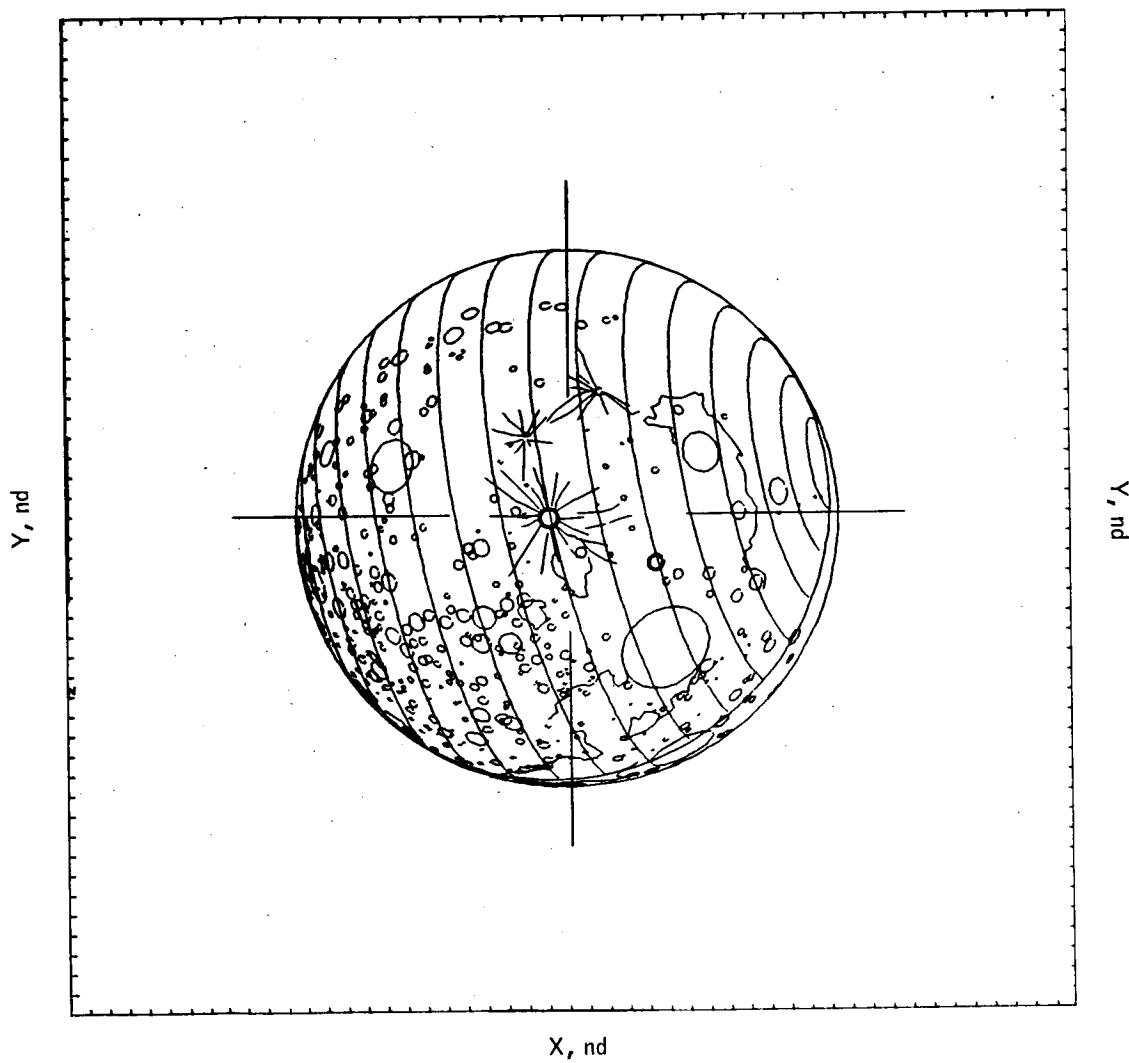


(d) Time from TLI cutoff = 20 hr.

Figure 5.- Continued.

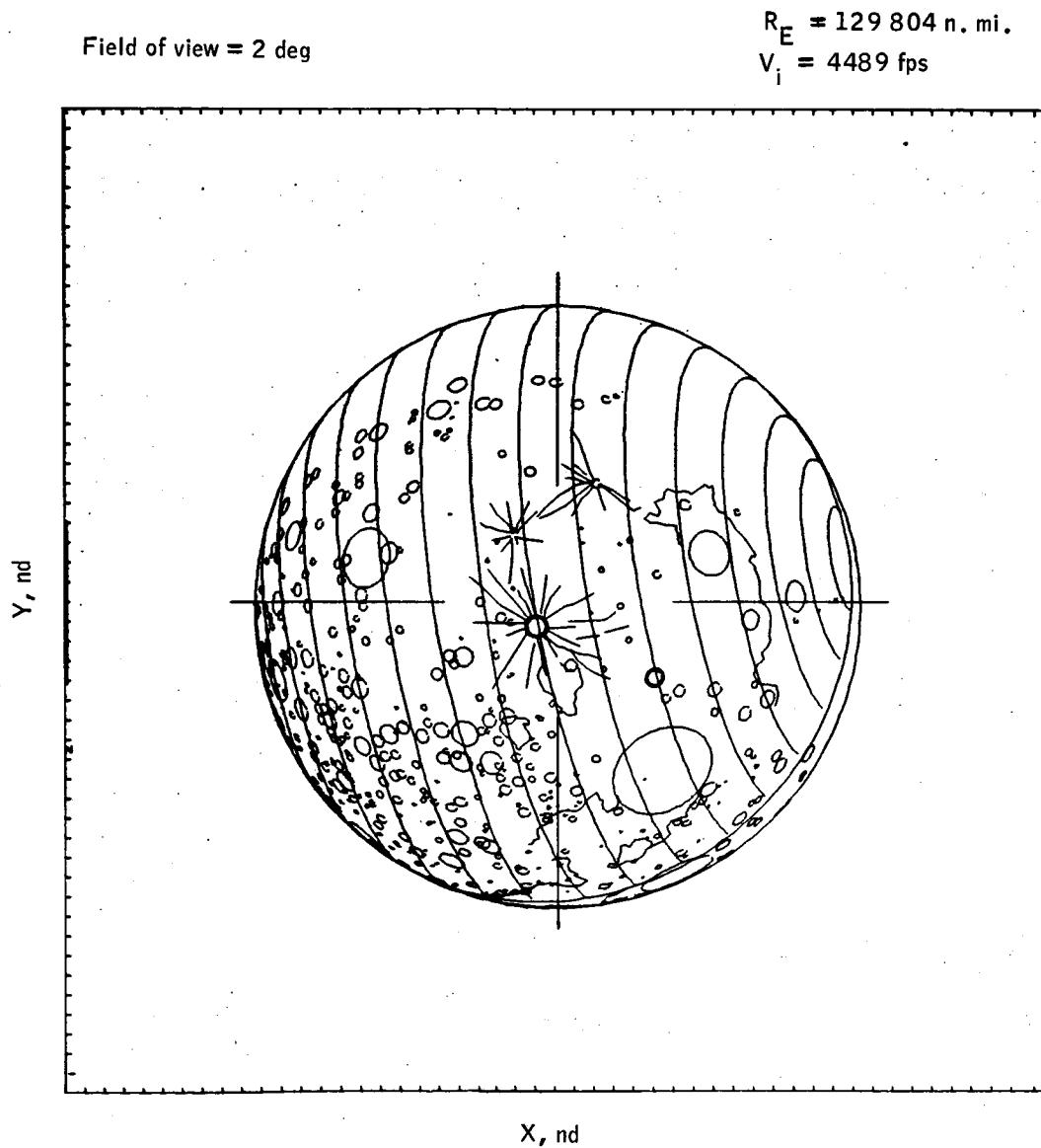
$$R_E = 116\ 218 \text{ n. mi.}$$
$$V_i = 4929 \text{ fps}$$

Field of view = 2 deg



(e) Time from TLI cutoff = 25 hr.

Figure 5.- Continued.

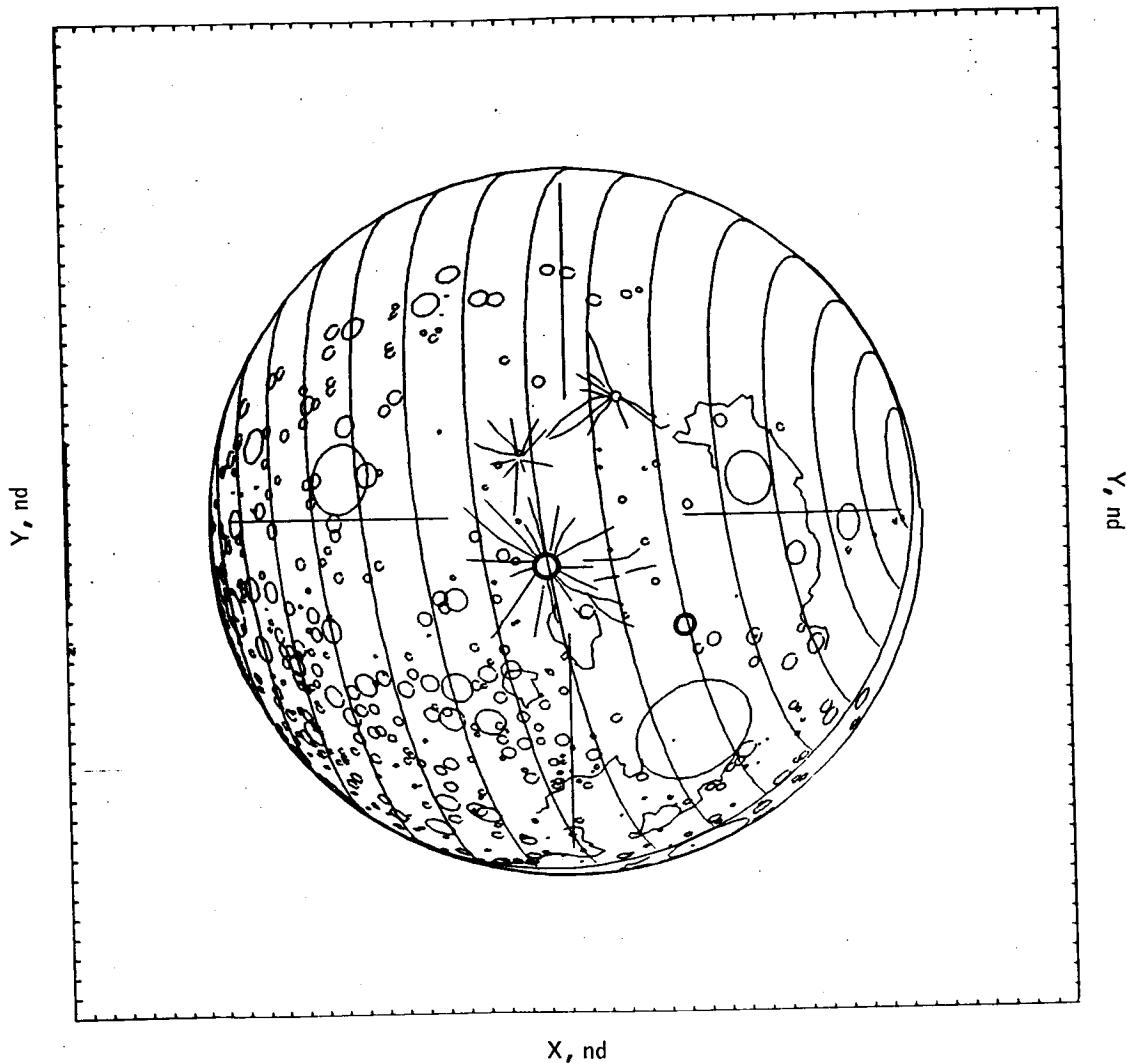


(f) Time from TLI cutoff = 30 hr.

Figure 5.- Continued.

Field of view = 2 deg

$$R_E = 142\ 255 \text{ n mi.}$$
$$V_i = 4132 \text{ fps}$$



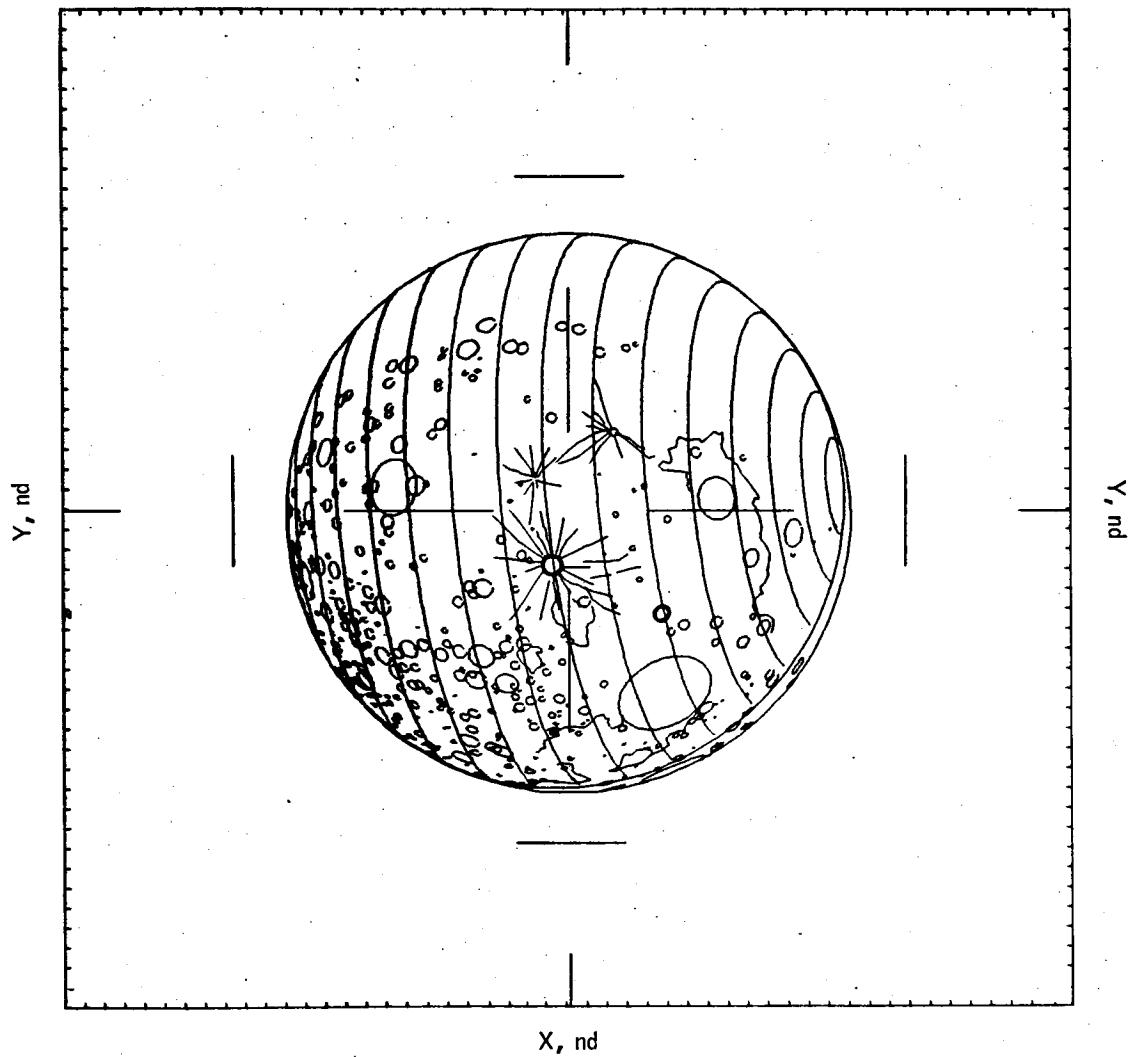
(g) Time from TLI cutoff = 35 hr.

Figure 5.- Continued.

Field of view = 3 deg

$$R_E = 153\ 765 \text{ n. mi.}$$

$$V_i = 3832 \text{ fps}$$



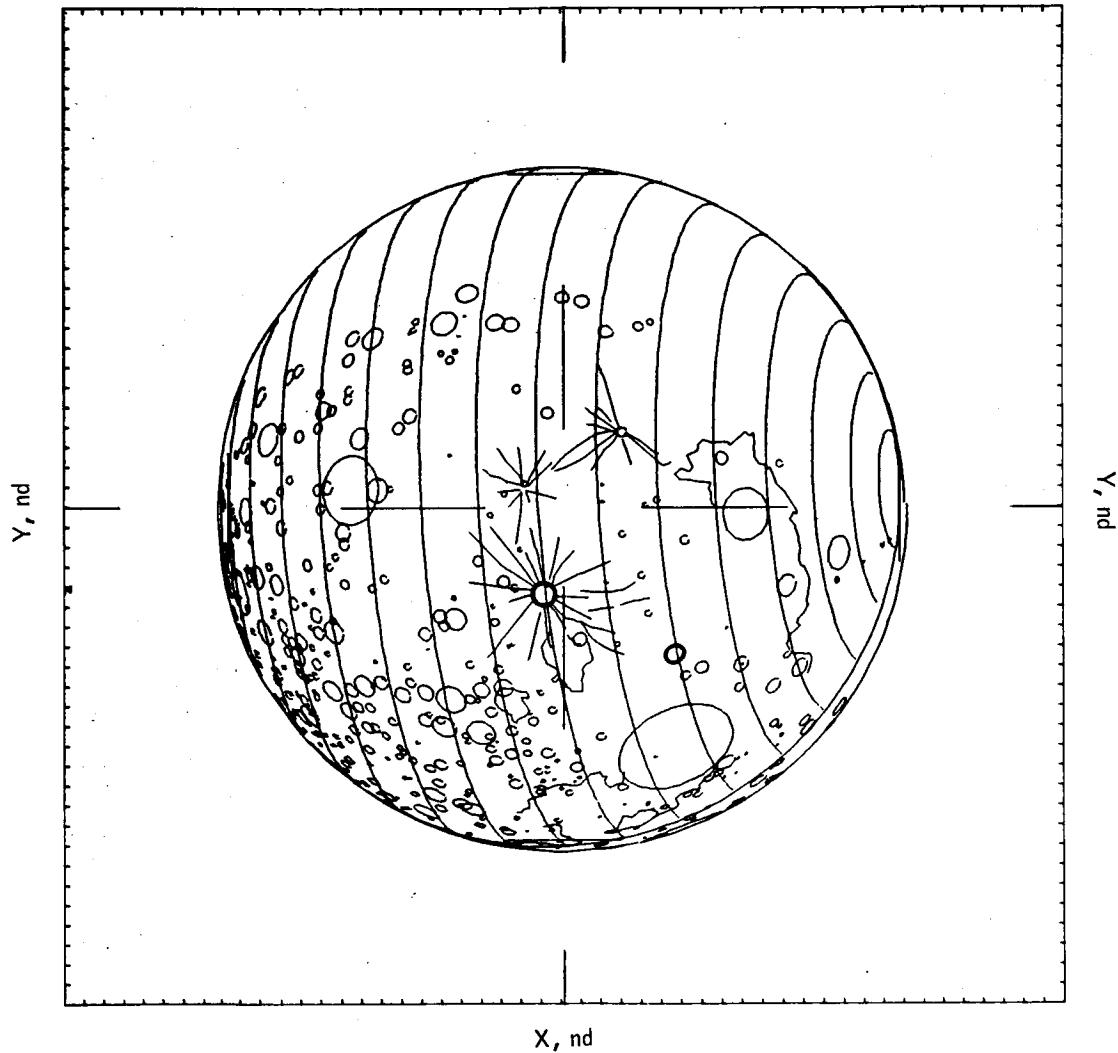
(h) Time from TLI cutoff = 40 hr.

Figure 5.- Continued.

Field of view = 3 deg

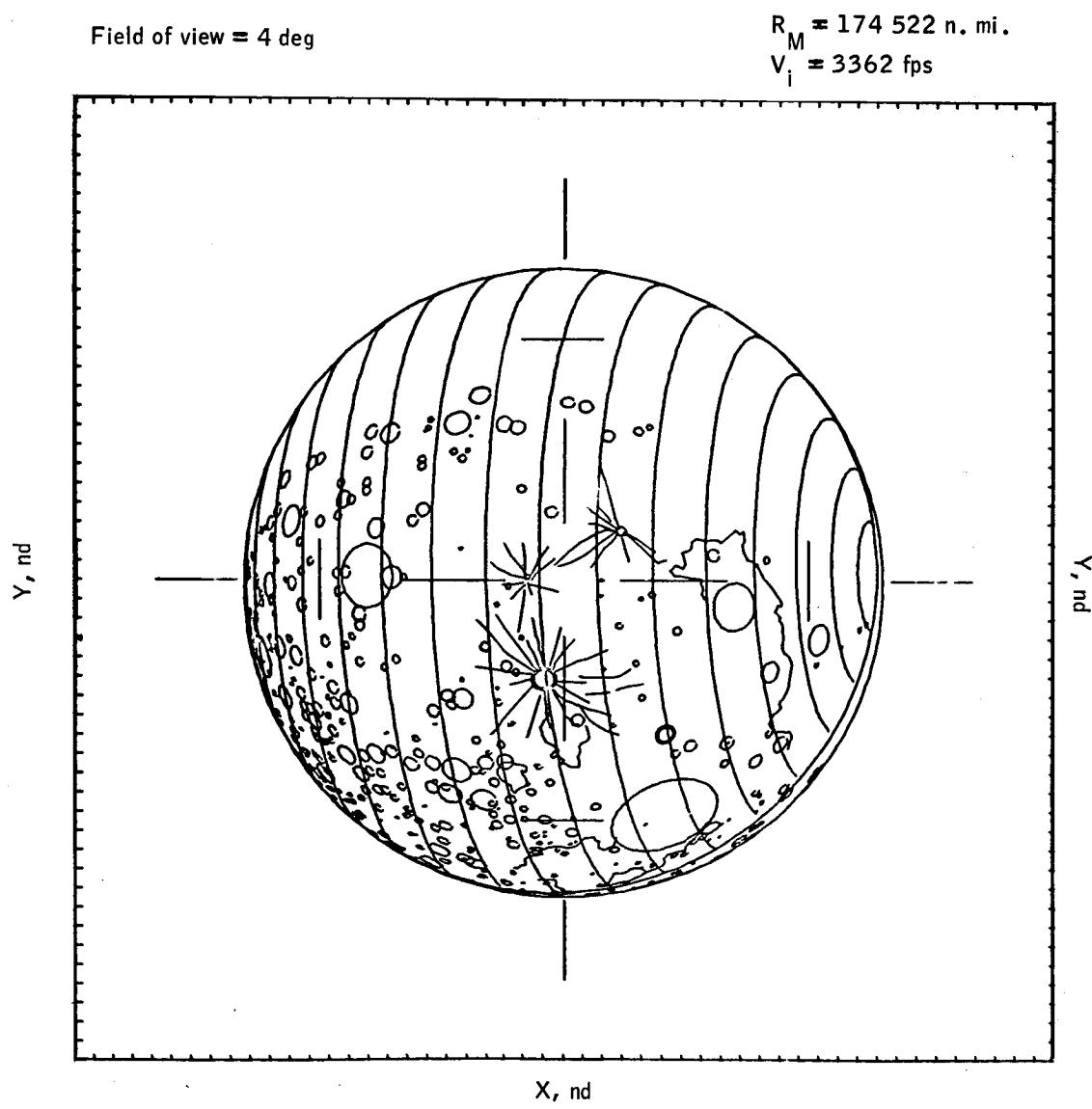
$$R_E = 164\ 482 \text{ n. mi.}$$

$$V_i = 35\ 78$$



(i) Time from TLI cutoff = 45 hr.

Figure 5.- Continued.



(j) Time from TLI cutoff = 50 hr.

Figure 5.- Continued.

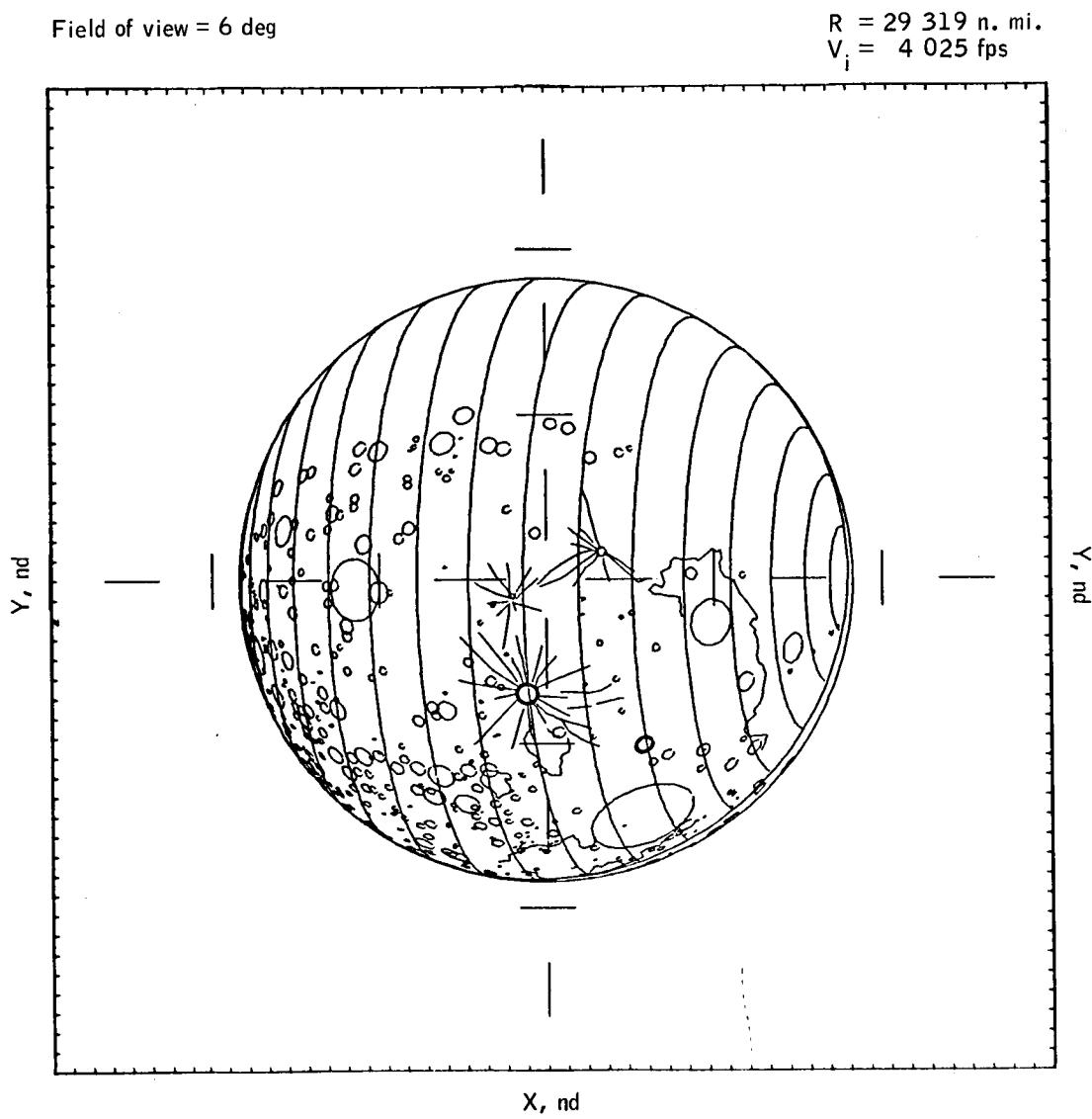
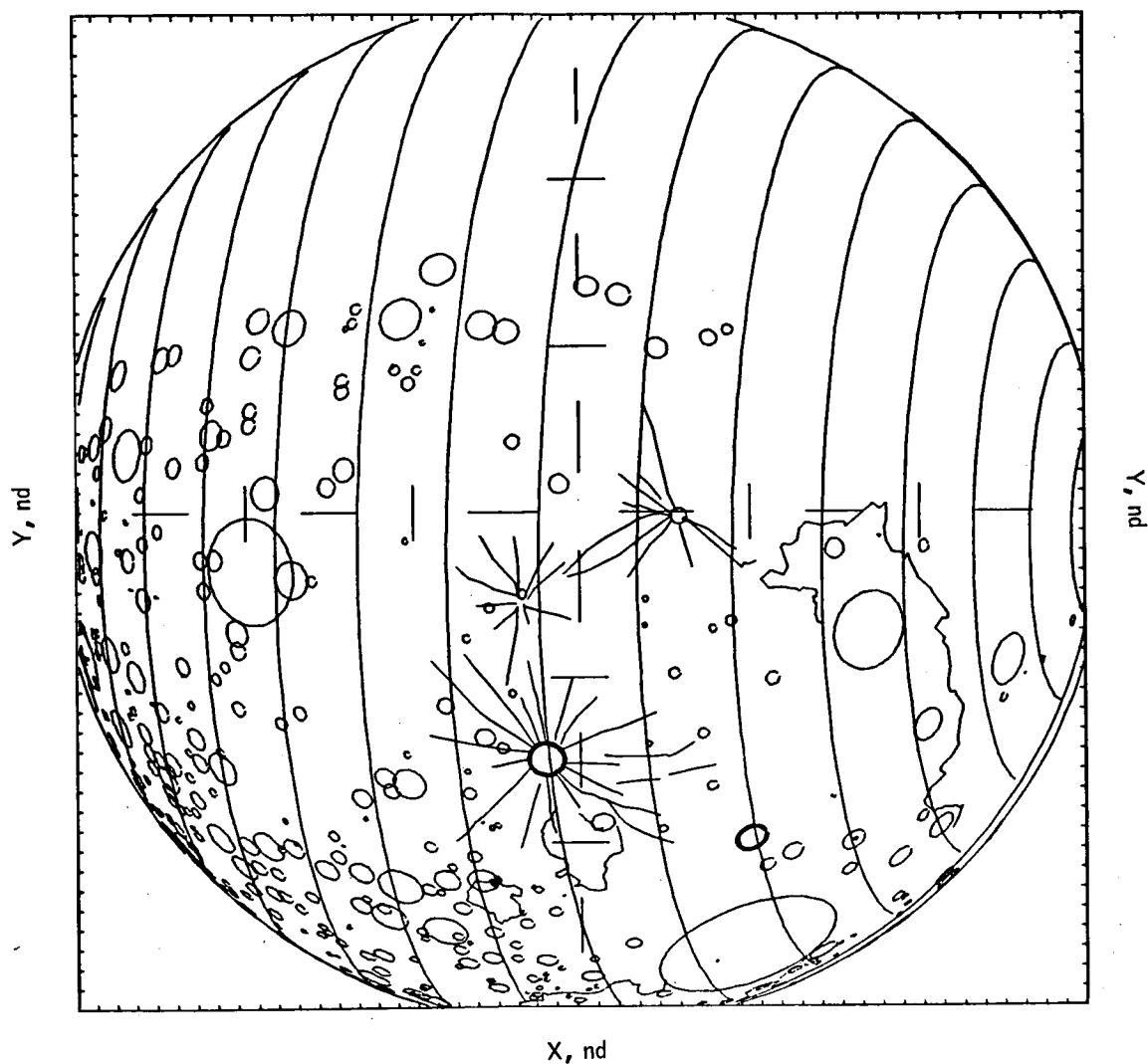


Figure 5 . - Continued.

Field of view = 6 deg

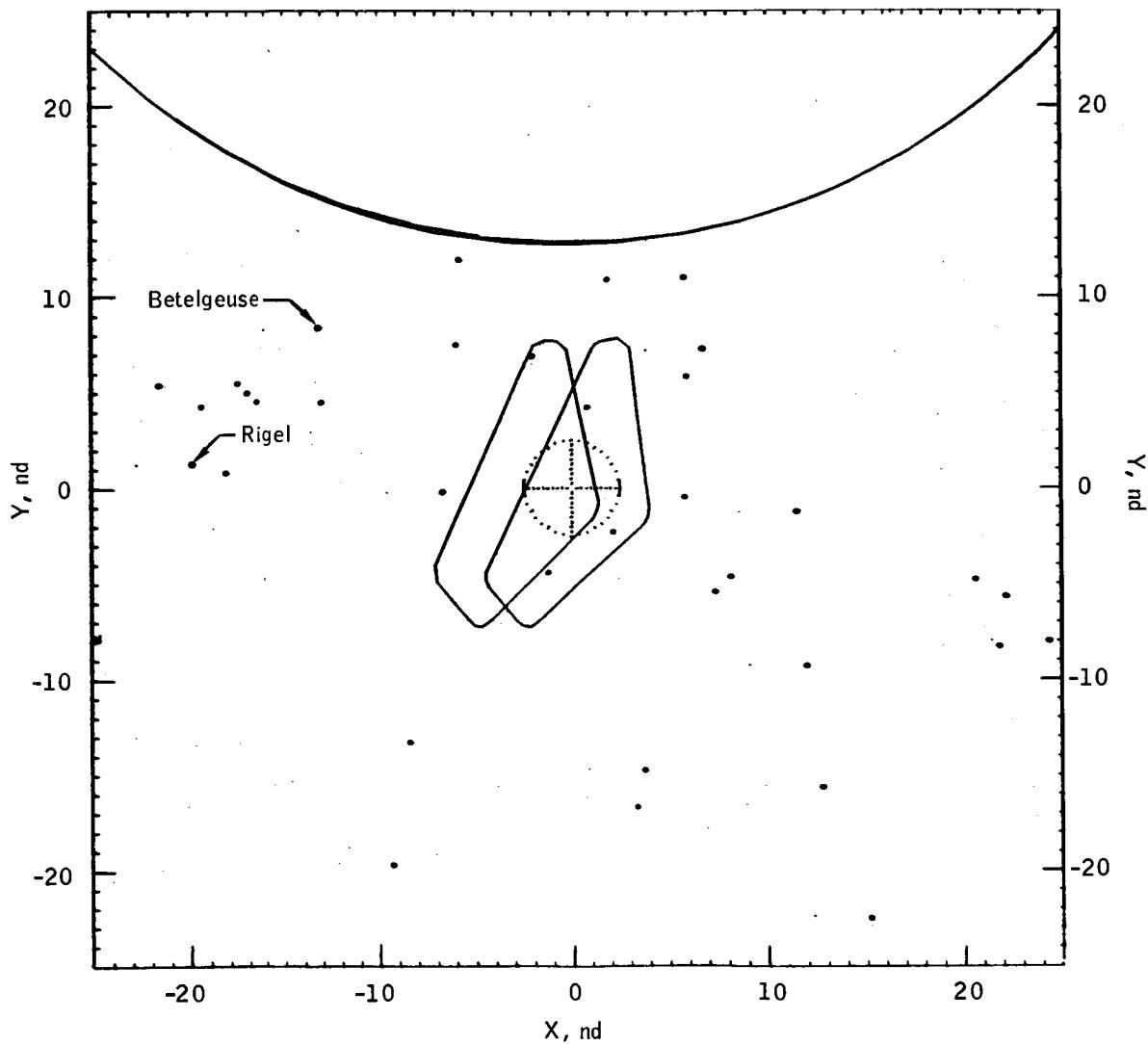
$$R_M = 17254 \text{ n. mi.}$$
$$V_i = 4187 \text{ fps}$$



(II) Time from TLI cutoff = 60 hr.

Figure 5.- Concluded.

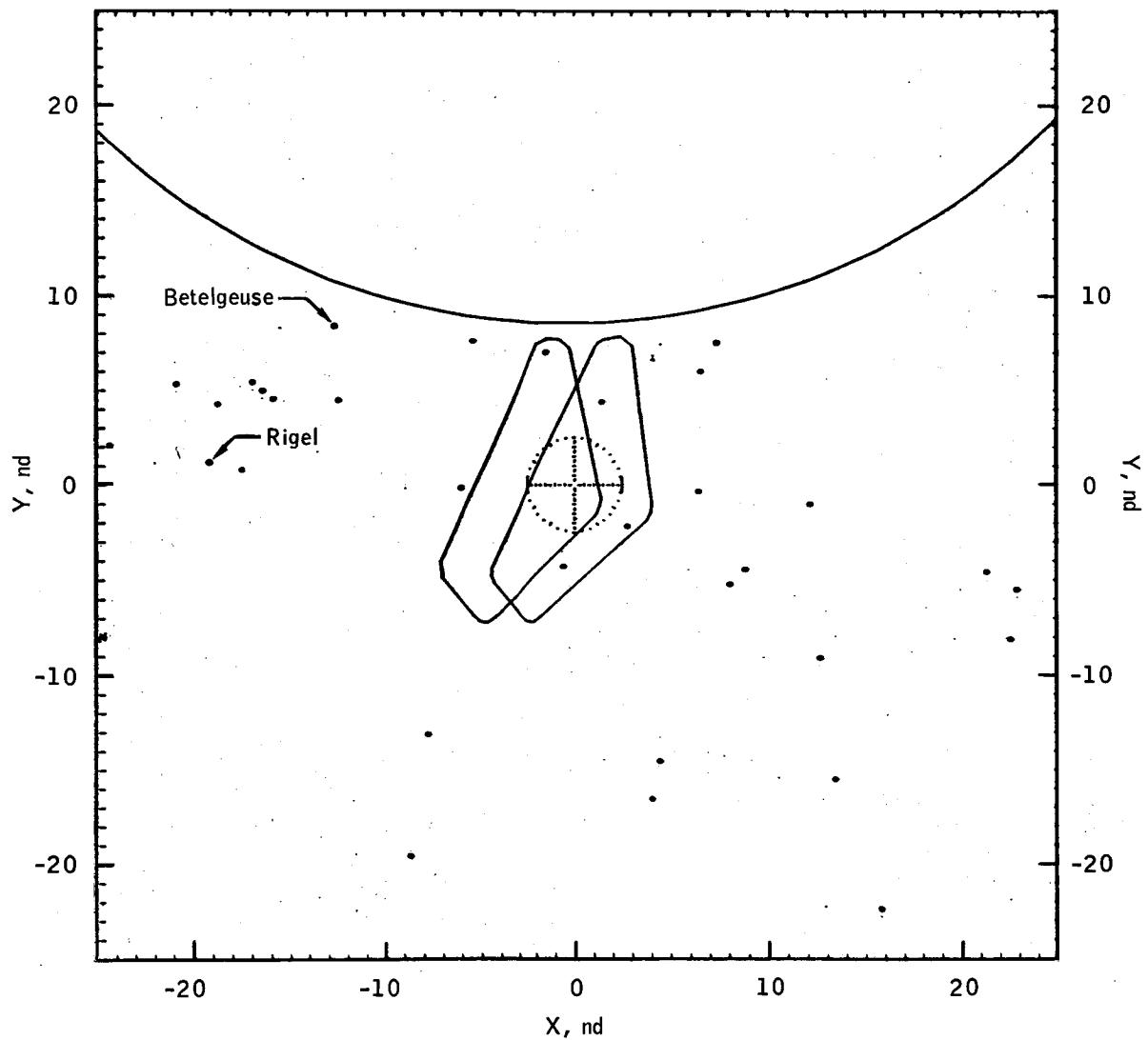
SEQ	4	5	22	31	41	47	63	73	75	80	108	111	112
X	15	24	21	22	12	20	3	11	3	-9	-8	7	8
Y	-22	-7	-8	-5	-15	-4	-16	-9	-14	-19	-13	-5	-4
SEQ	120	144	150	151	186	205	207	215	221	222	230	231	
X	11	-1	2	5	-6	0	5	-18	-19	6	-13	-2	
Y	-1	-4	-2	0	0	4	5	0	1	7	4	7	
SEQ	237	245	246	248	252	256	265	270	271	281			
X	-16	-19	-17	-5	-17	-21	-13	5	1	-5			
Y	4	4	5	7	5	5	8	11	11	12			



(a) Beginning of LOI burn.

Figure 6. - Lunar orbit insertion burn.

SEQ	4	22	31	41	47	63	73	75	80	108	111	112	120
X	15	22	22	13	21	4	12	4	-8	-7	8	8	12
Y	-22	-7	-5	-15	-4	-16	-8	-14	-19	-12	-5	-4	0
SEQ	144	150	151	186	205	207	215	221	222	230	231	237	
X	0	2	6	-6	1	6	-17	-19	7	-12	-1	-15	
Y	-4	-2	0	0	4	6	0	1	7	4	7	4	
SEQ	239	245	246	248	252	256	265						
X	-24	-18	-16	-5	-16	-20	-12						
Y	2	4	5	7	5	5	8						



(b) Middle of LOI burn.

Figure 6. - Continued.

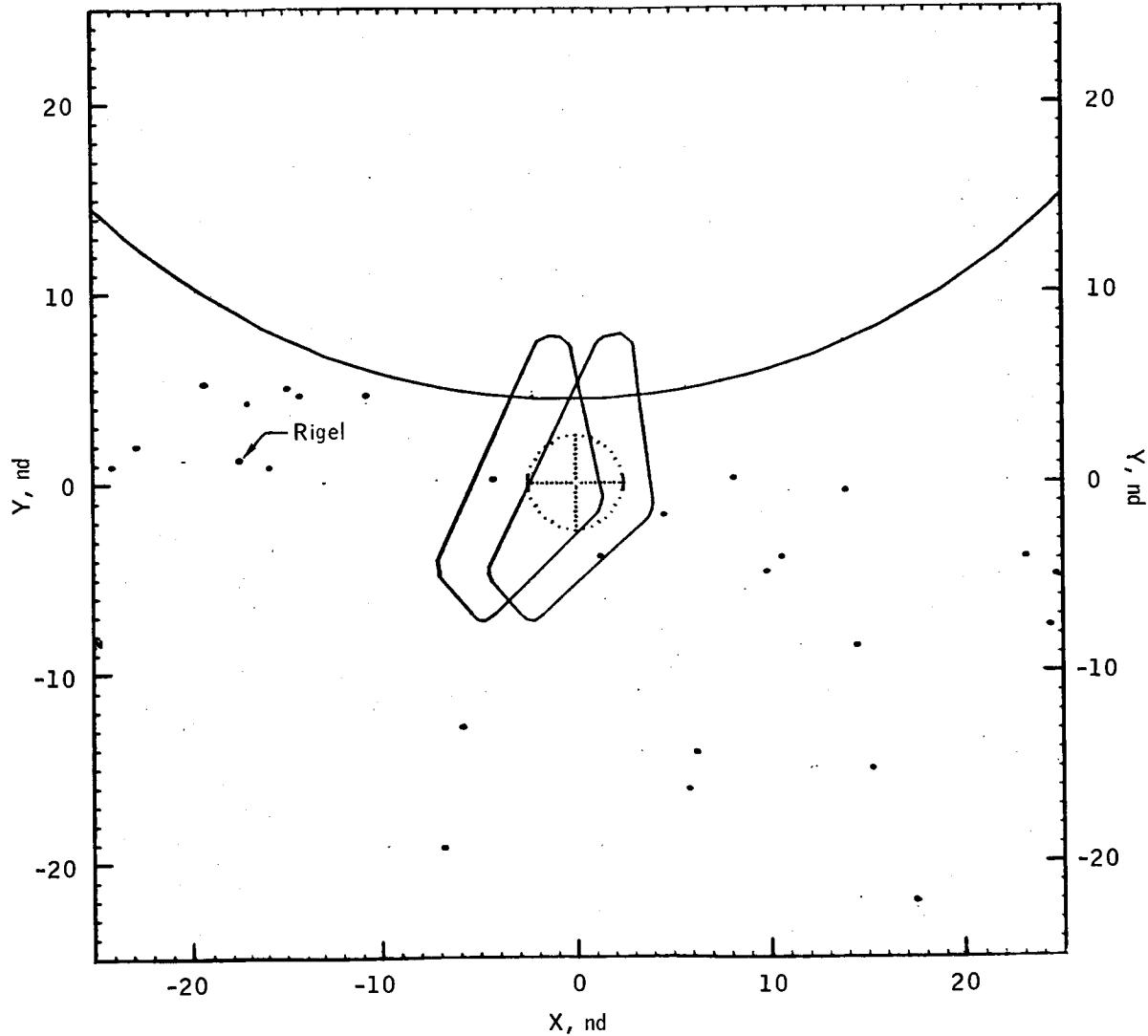
SEQ	4	22	31	41	47	63	73	75	80	108	111	112	120
X	17	24	24	15	23	5	14	6	-6	-5	9	10	13
Y	-21	-7	-4	-14	-3	-15	-8	-14	-19	-12	-4	-3	0

SEQ	144	150	151	186	215	221	230	233	237	239	245	246
X	1	4	8	-4	-15	-17	-10	-23	-14	-22	-17	-14
Y	-3	-1	0	0	1	1	4	1	4	2	4	5

SEQ 256

X -19

Y 5

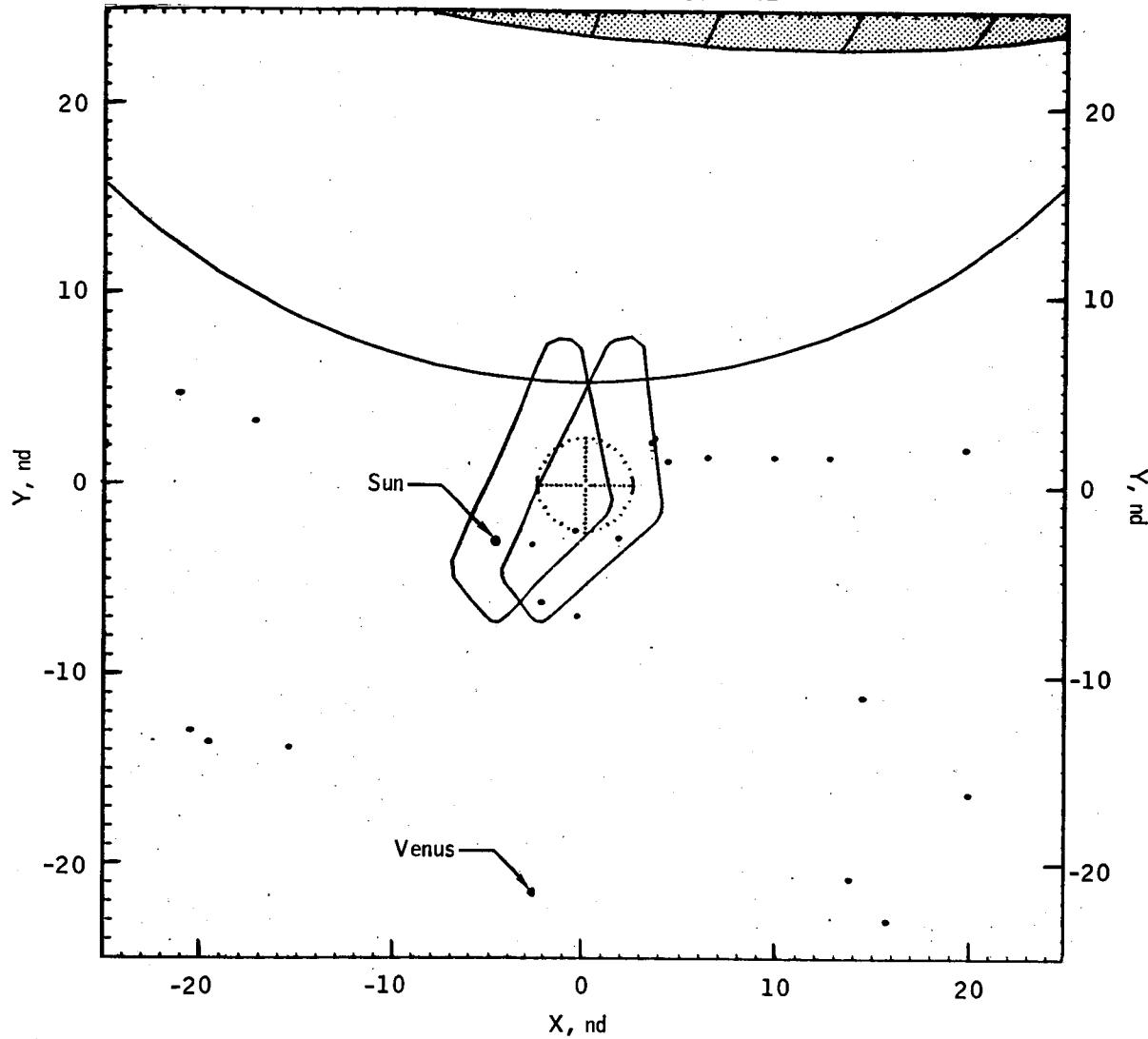


(c) End of LOI burn.

Figure 6. - Concluded.

SEQ	757	781	789	790	793	795	797	802	803	836	841	844
X	19	12	3	9	3	-21	6	4	-17	0	1	-2
Y	2	1	2	1	2	4	1	1	3	-2	-2	-2

SEQ	861	871	904	907	909	933	1001	1010	1028
X	-2	0	-20	-19	-15	14	13	19	15
Y	-6	-6	-12	-13	-13	-11	-20	-16	-22

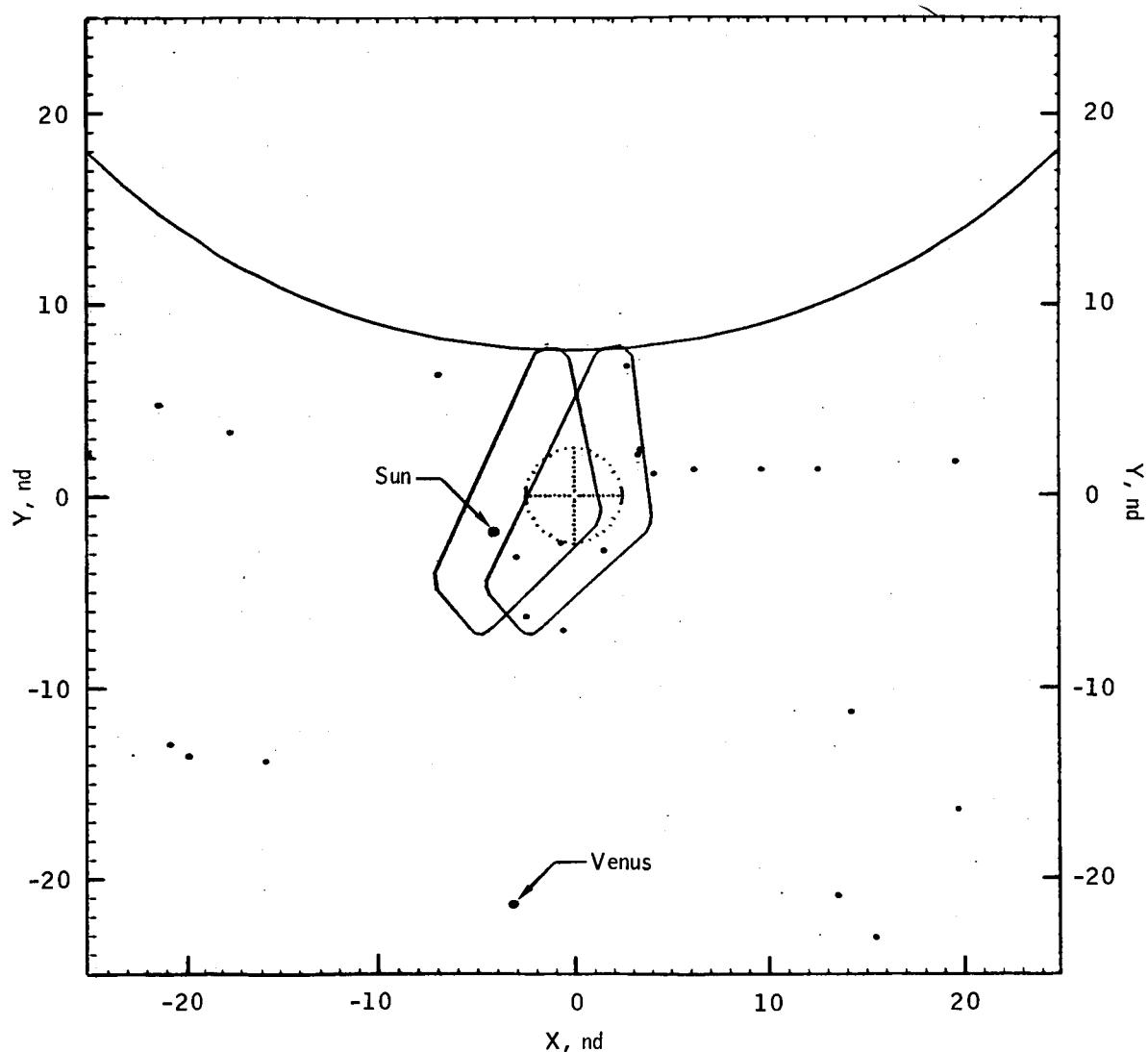


(a) Beginning of TEI burn.

Figure 7. - Transearth injection burn.

SEQ	757	759	770	781	789	790	793	795	797	802	803	836	841
X	19	2	-6	12	3	9	3	-21	6	4	-17	0	1
Y	1	6	6	1	2	1	2	4	1	1	3	-2	-2

SEQ	844	861	871	904	907	909	933	1001	1010	1028
X	-2	-2	0	-20	-19	-15	14	13	19	15
Y	-3	-6	-6	-12	-13	-13	-11	-20	-16	-22



(b) Middle of TEI burn.

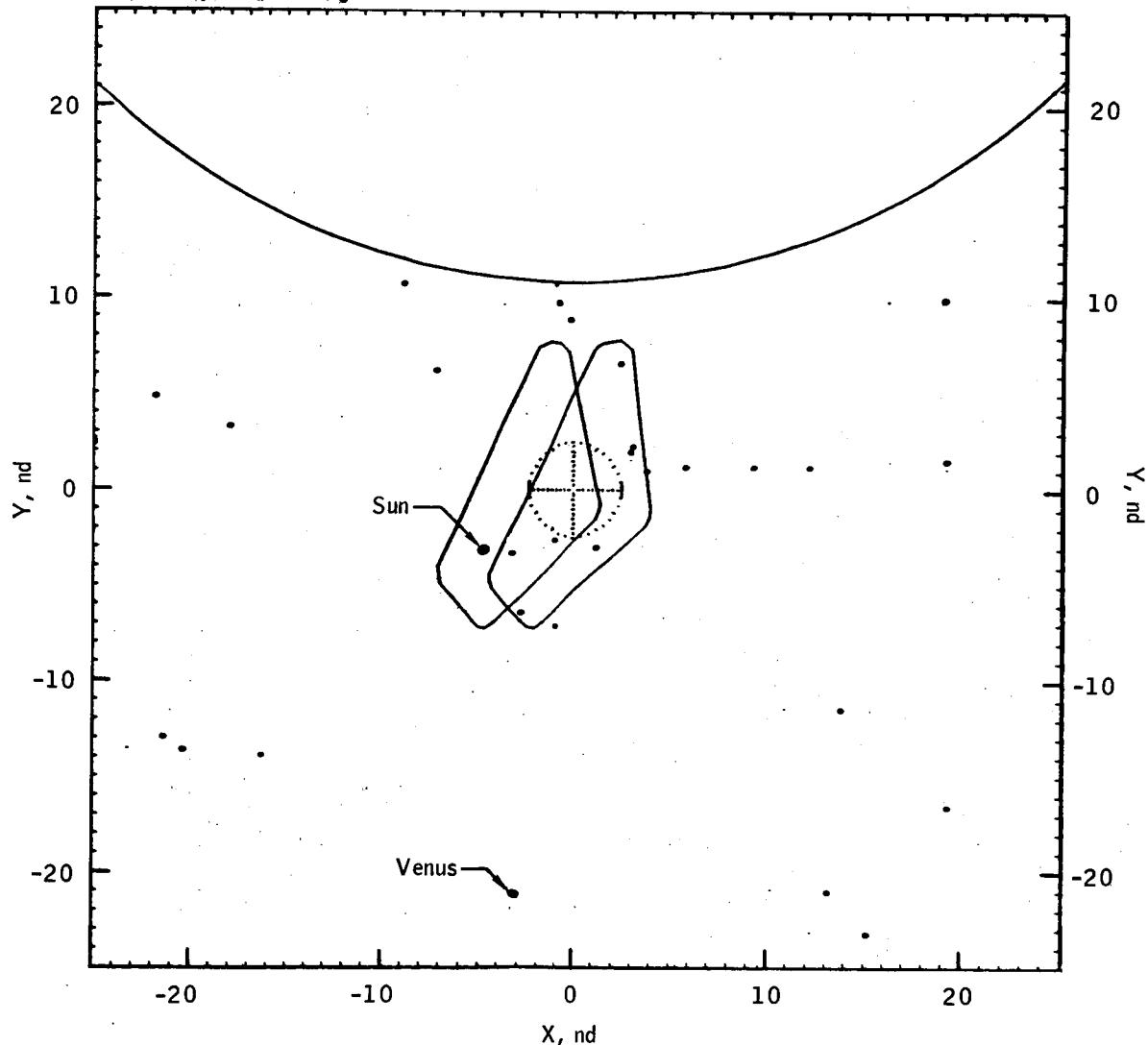
Figure 7.- Continued.

SEQ	643	736	743	751	753	757	759	770	781	789	790	793	795
X	19	-1	0	0	-8	19	2	-7	12	3	9	2	-21
Y	10	10	9	9	10	1	6	6	1	2	1	2	4

SEQ	797	802	803	836	841	844	861	871	904	907	909	933
X	5	3	-17	-1	1	-3	-2	-1	-21	-20	-16	13
Y	1	1	3	-2	-2	-3	-6	-6	-12	-13	-13	-11

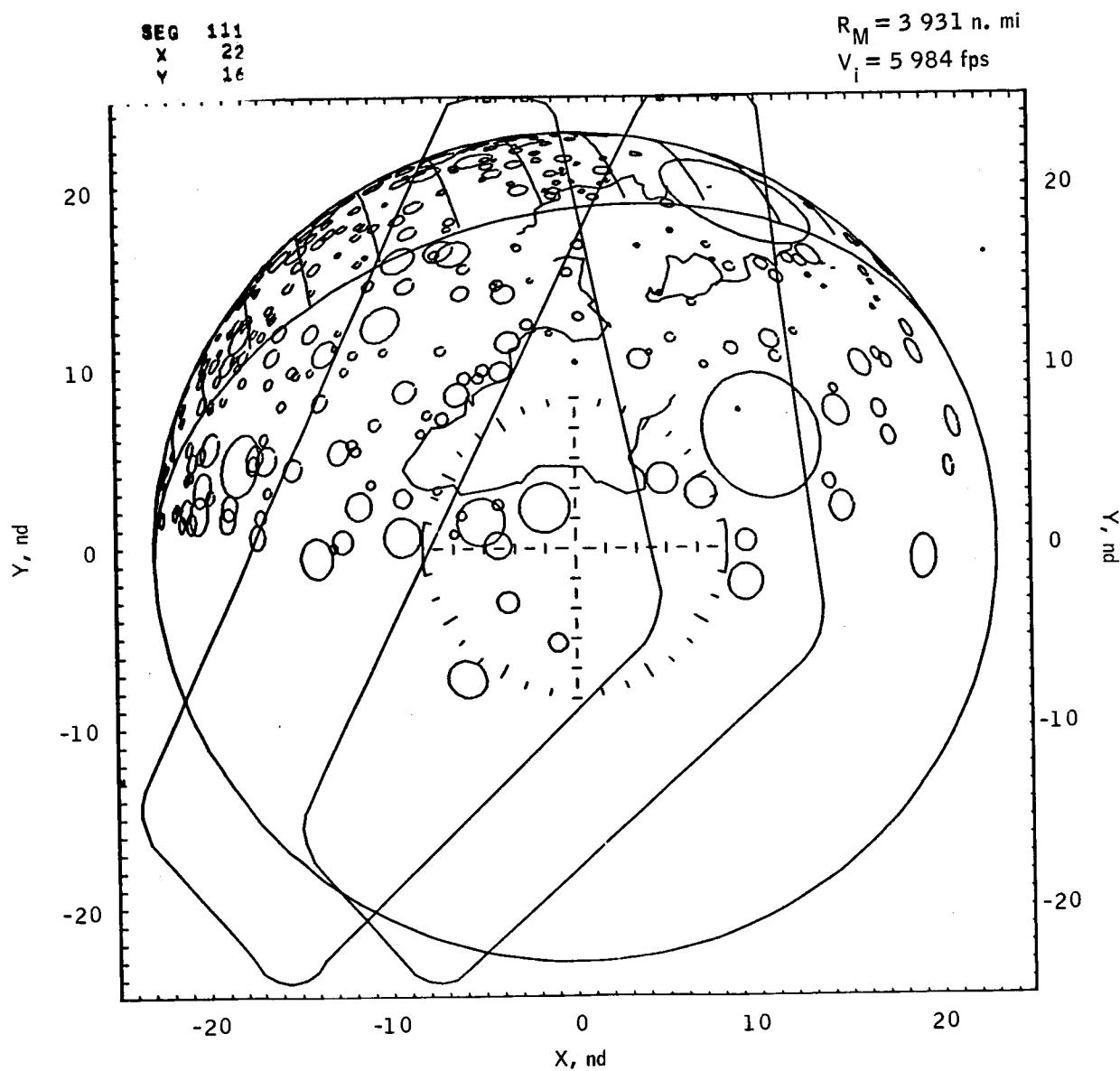
SEQ 1001 1010 1028

X	13	19	15
Y	-20	-16	-23



(c) End of TEI burn.

Figure 7.- Concluded.



(a) Time from TEI cutoff = 1 hr.

Figure 8.- Transearth coast (moon referenced) constant field of view.

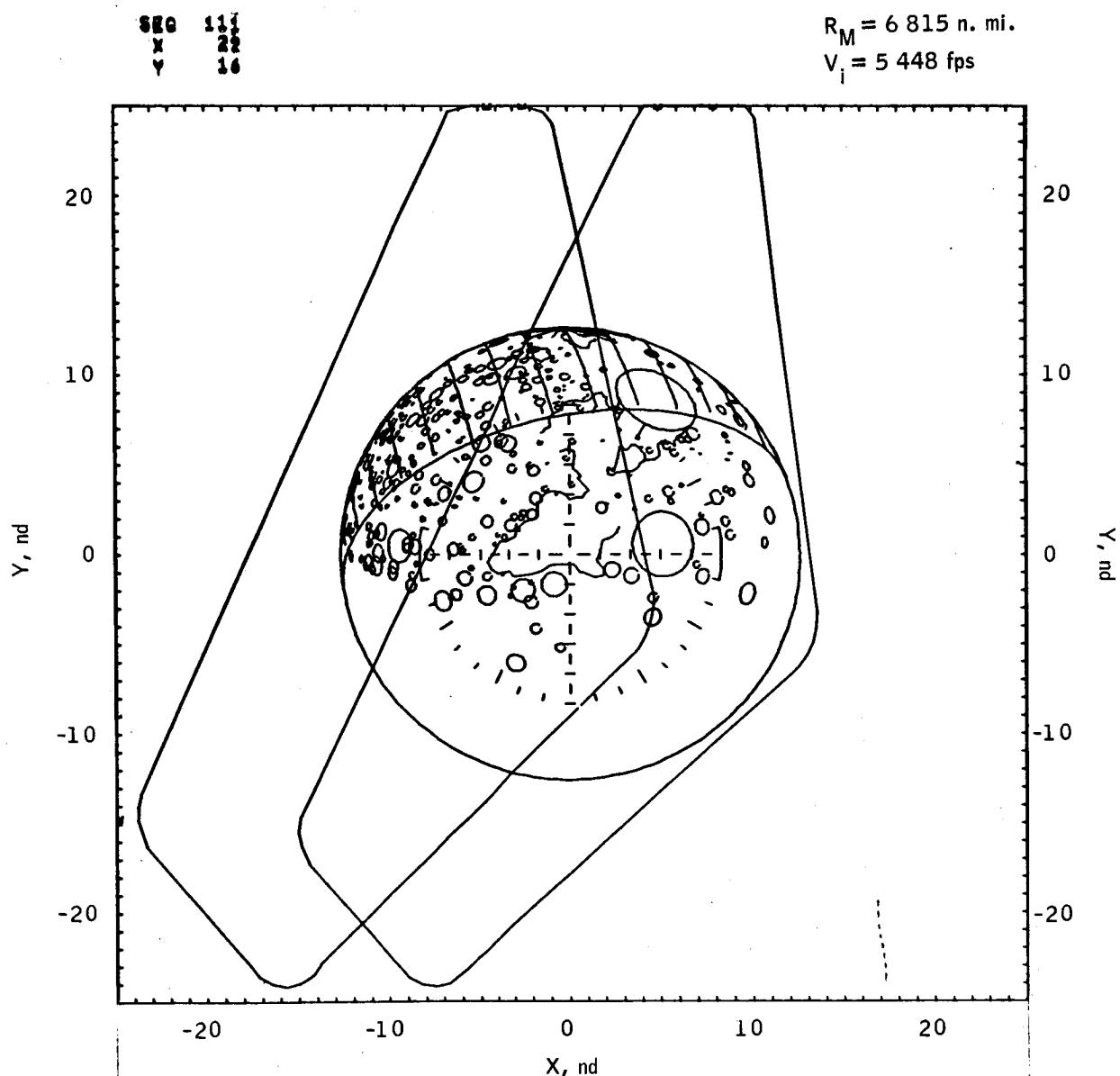
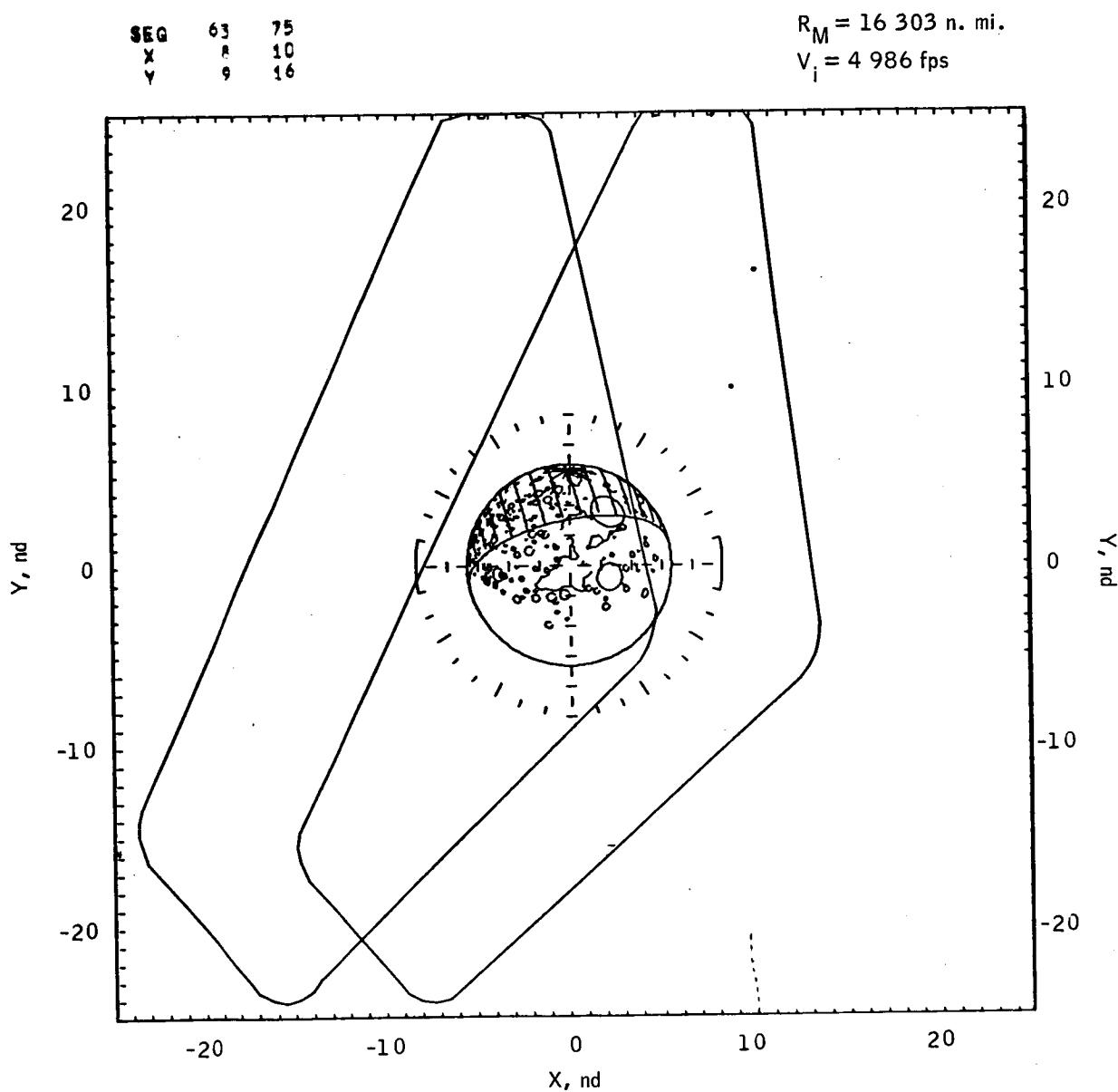


Figure 8.- Continued.



(c) Time from TEI cutoff = 5 hr.

Figure 8.- Continued.

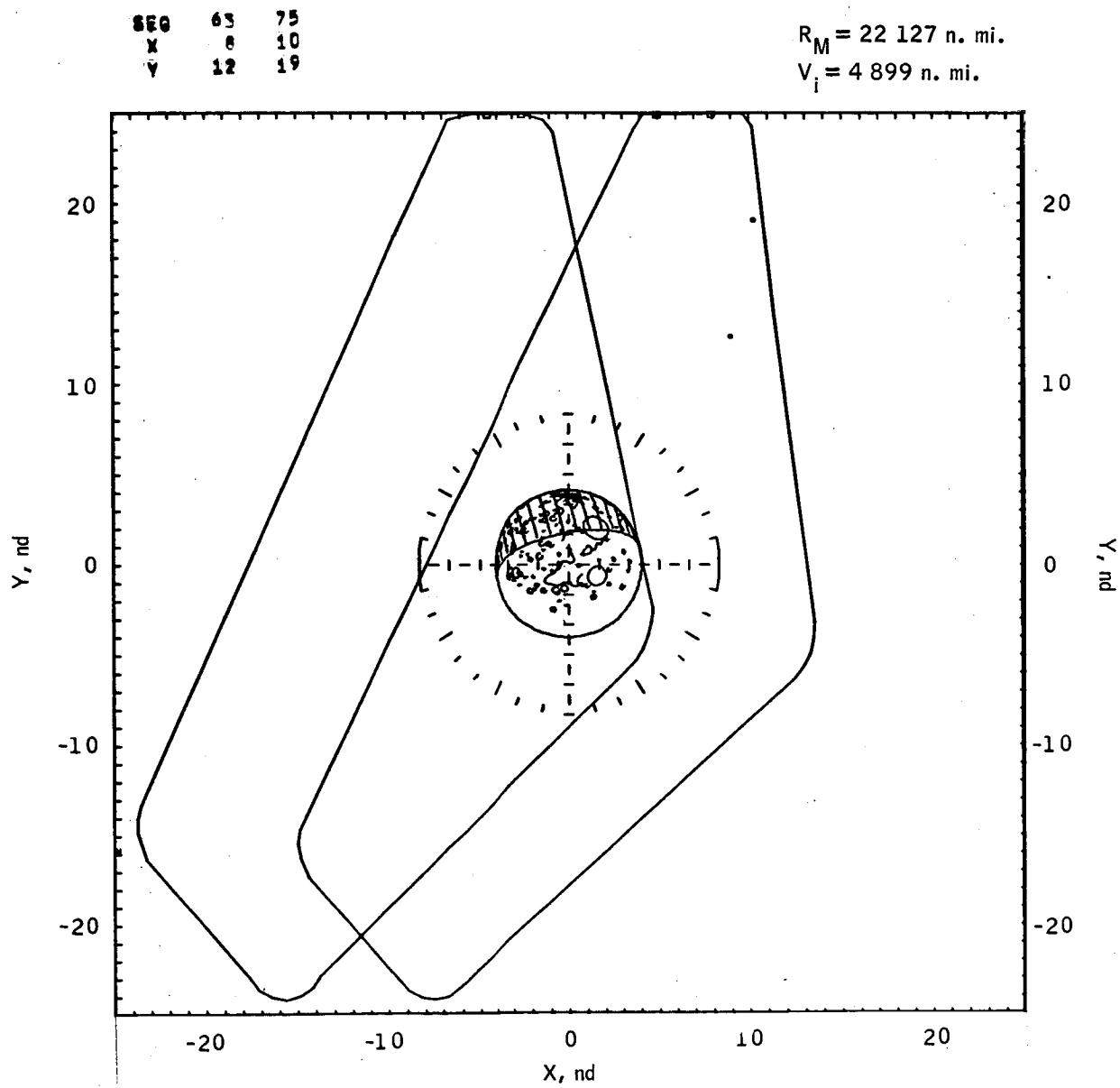
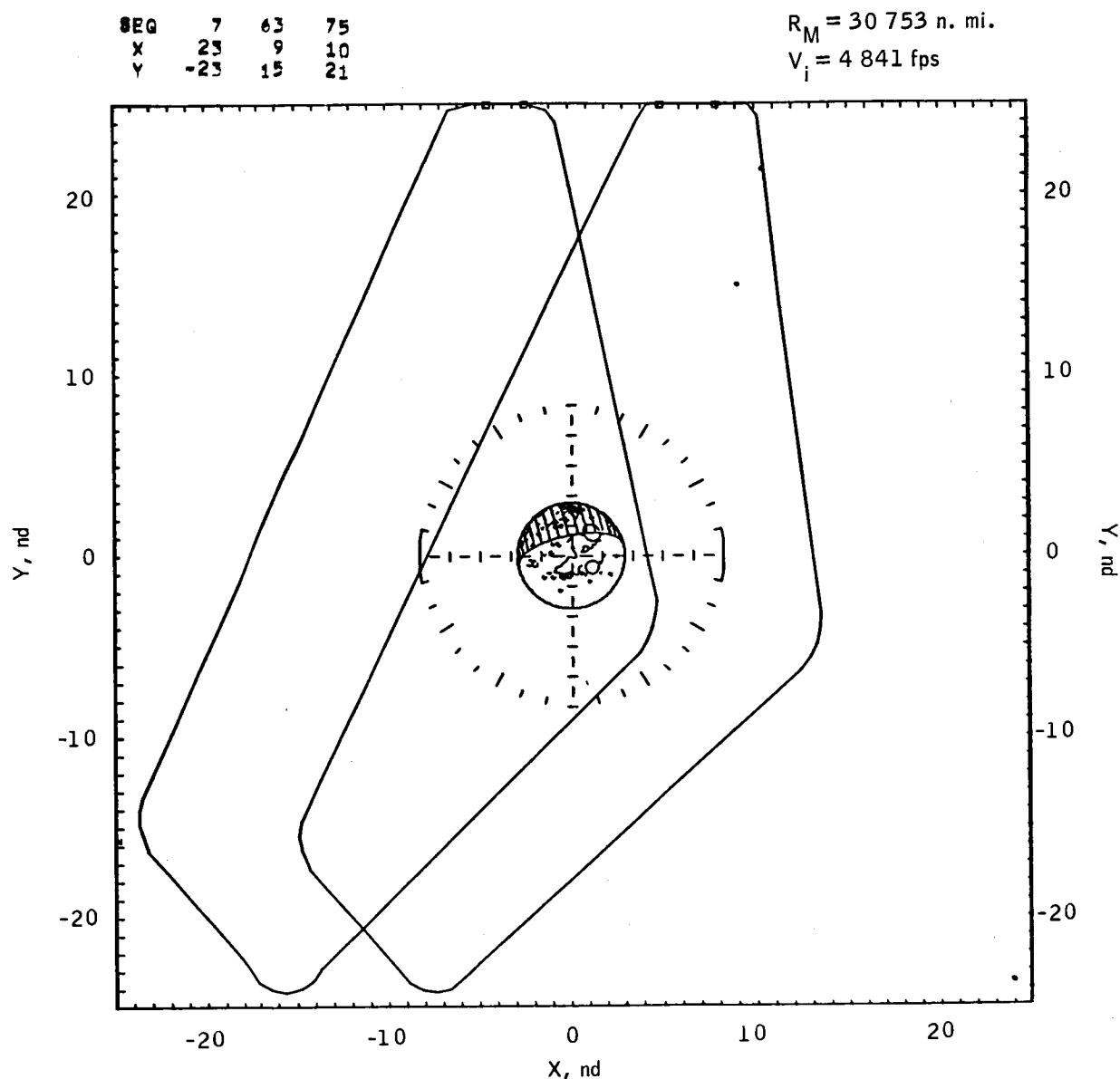


Figure 8.- Continued.



(e) Time from TEI cutoff = 10 hr.

Figure 8.- Continued.

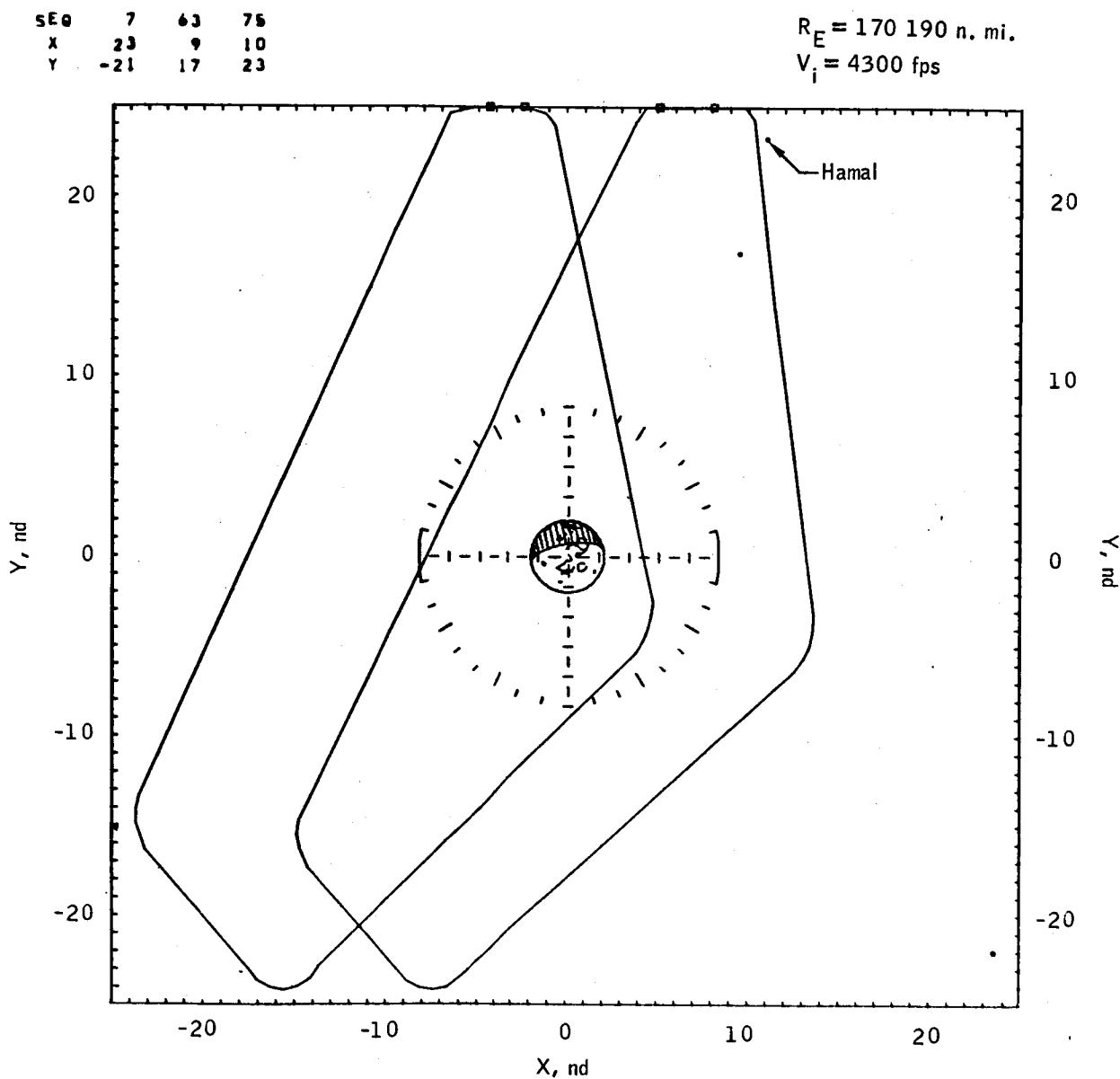
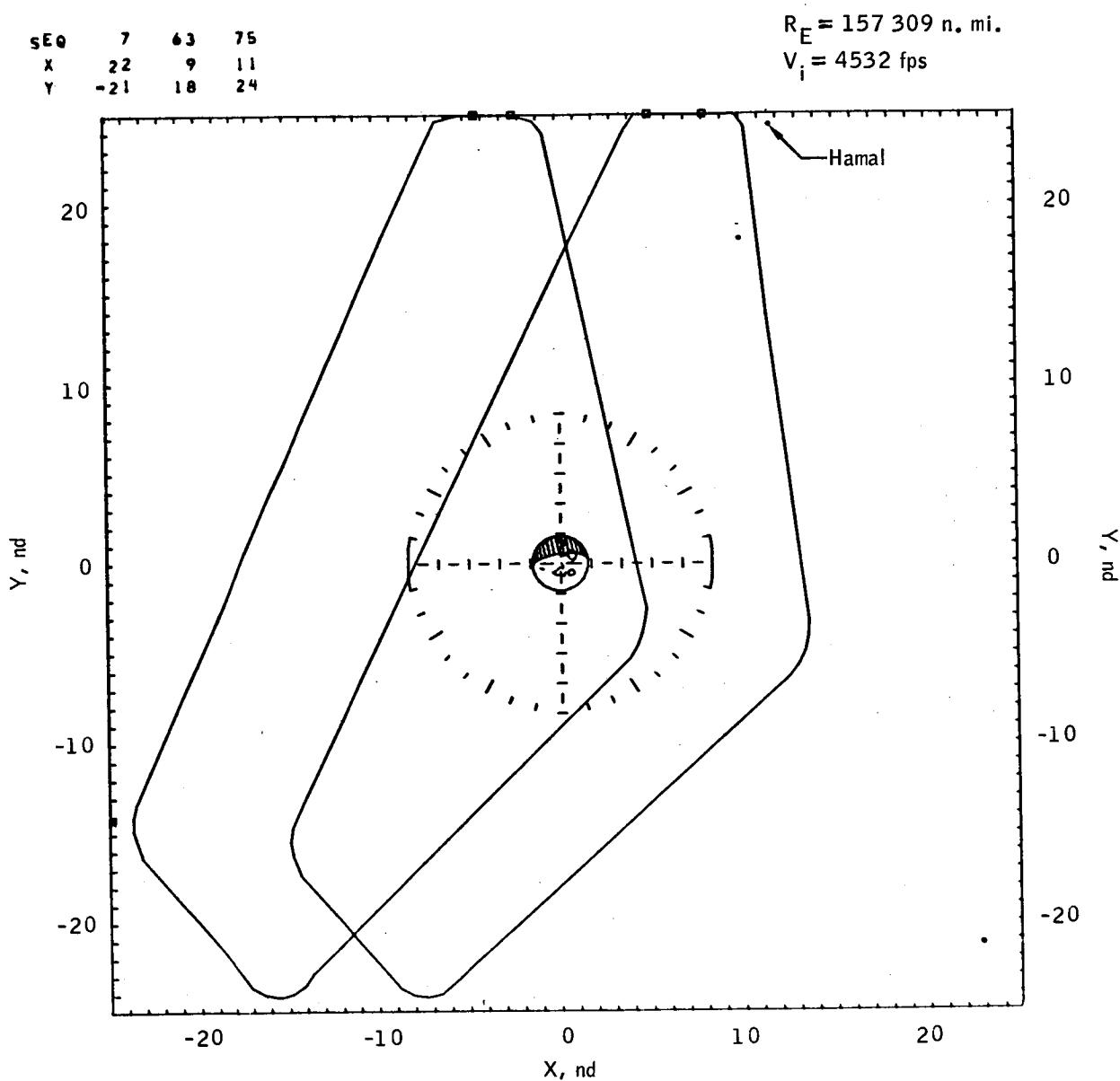


Figure 8.- Continued.



(g) Time from TEI cutoff = 20 hr.

Figure 8.- Continued.

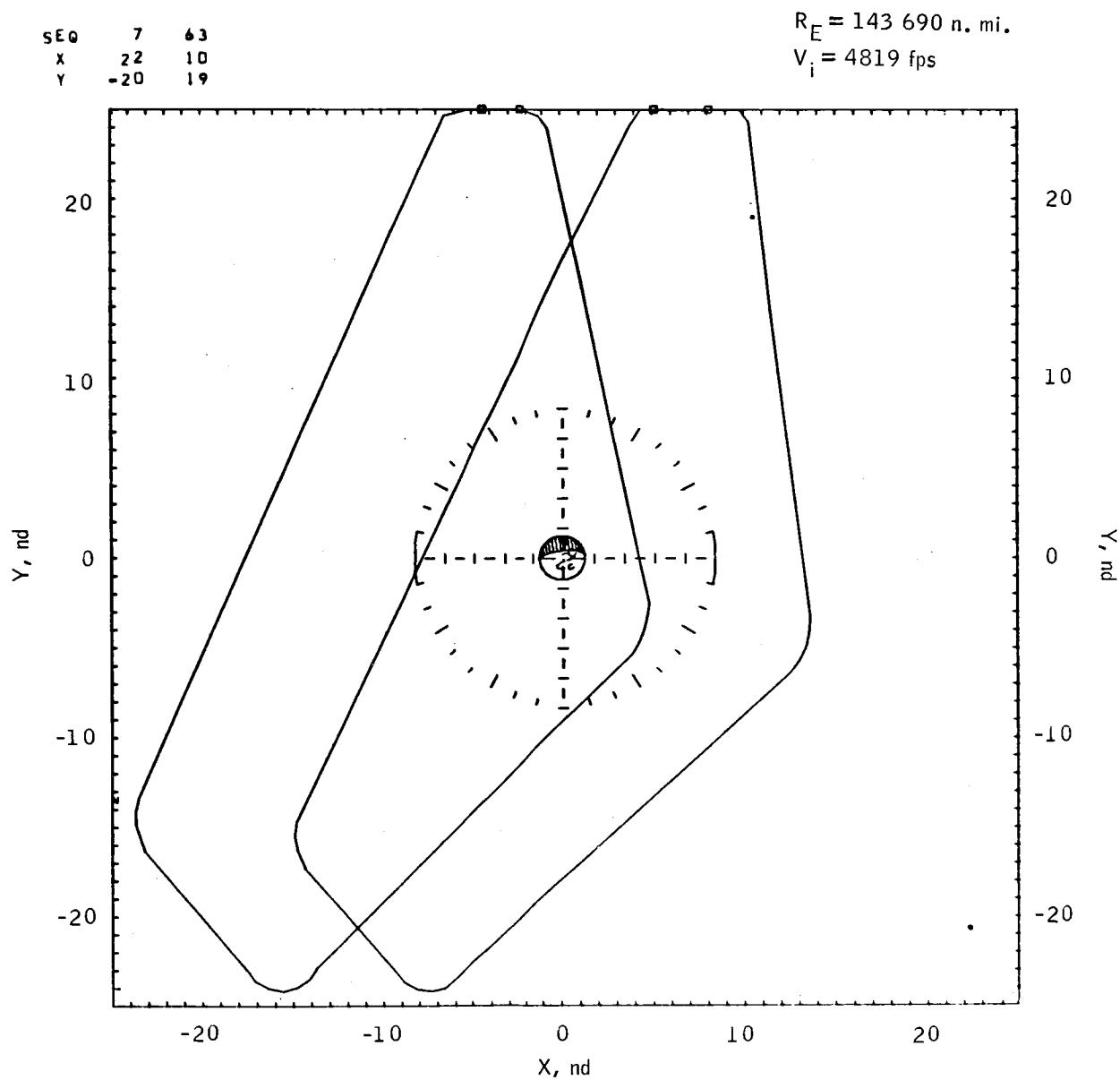
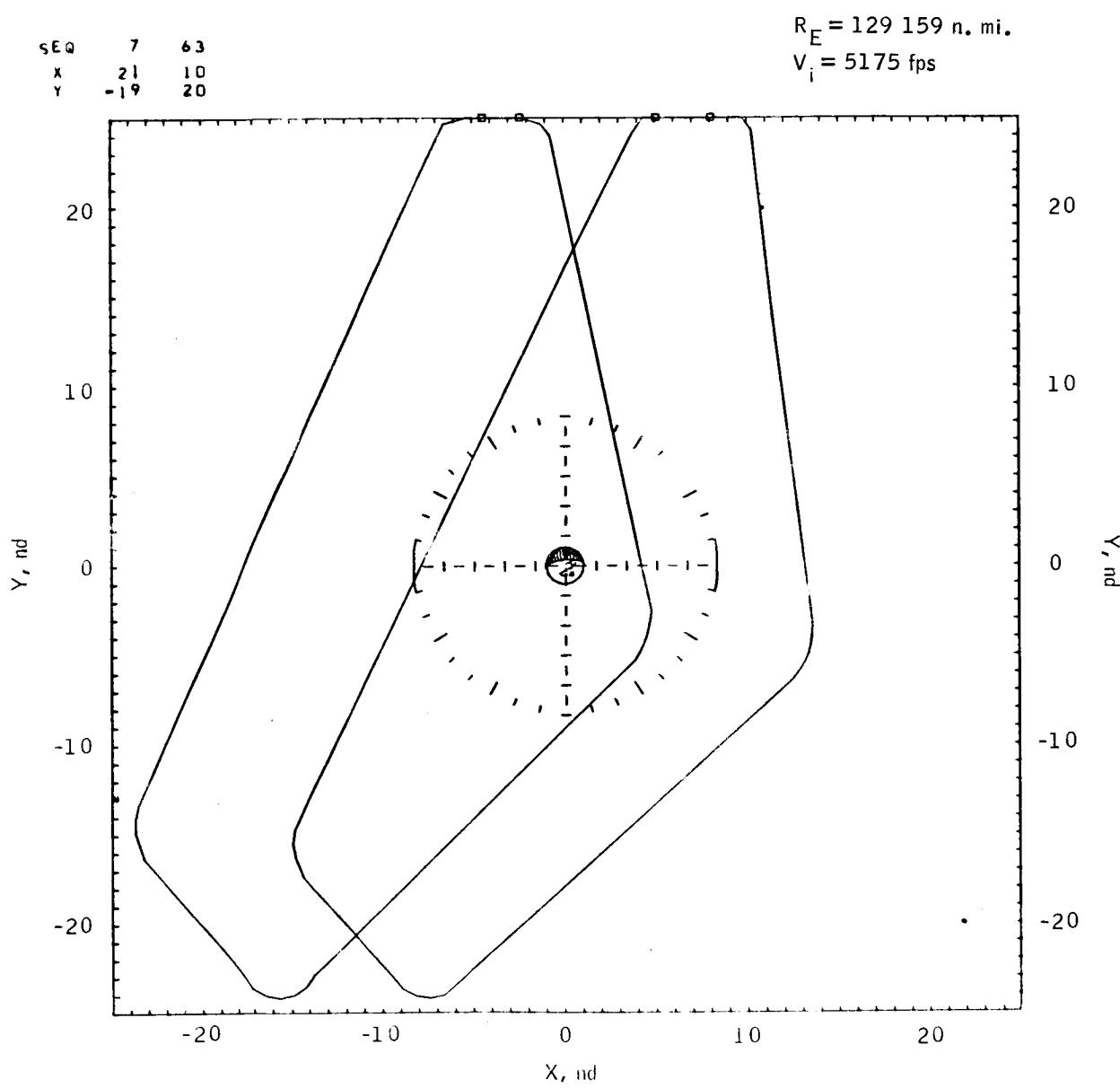
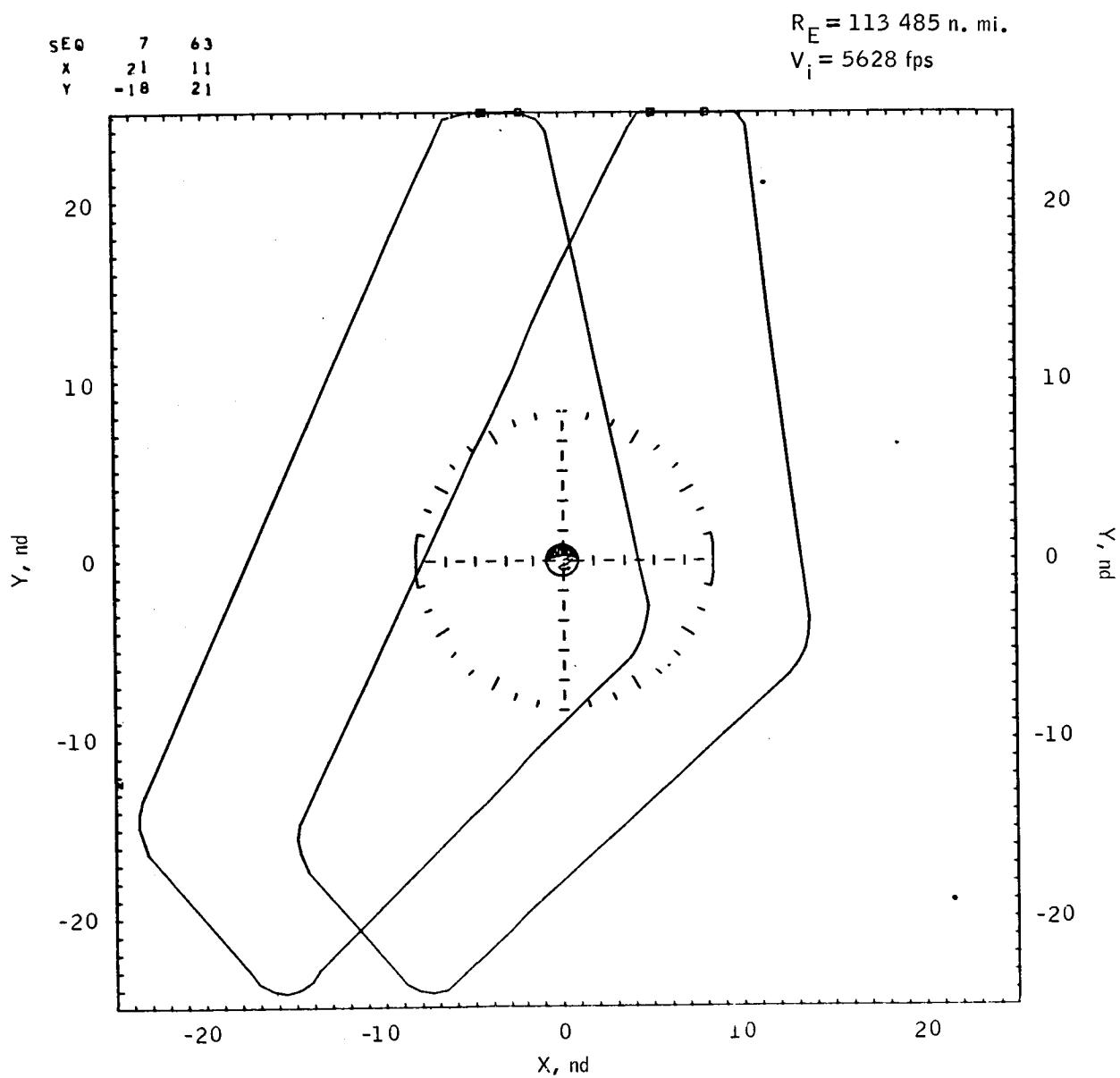


Figure 8.- Continued.



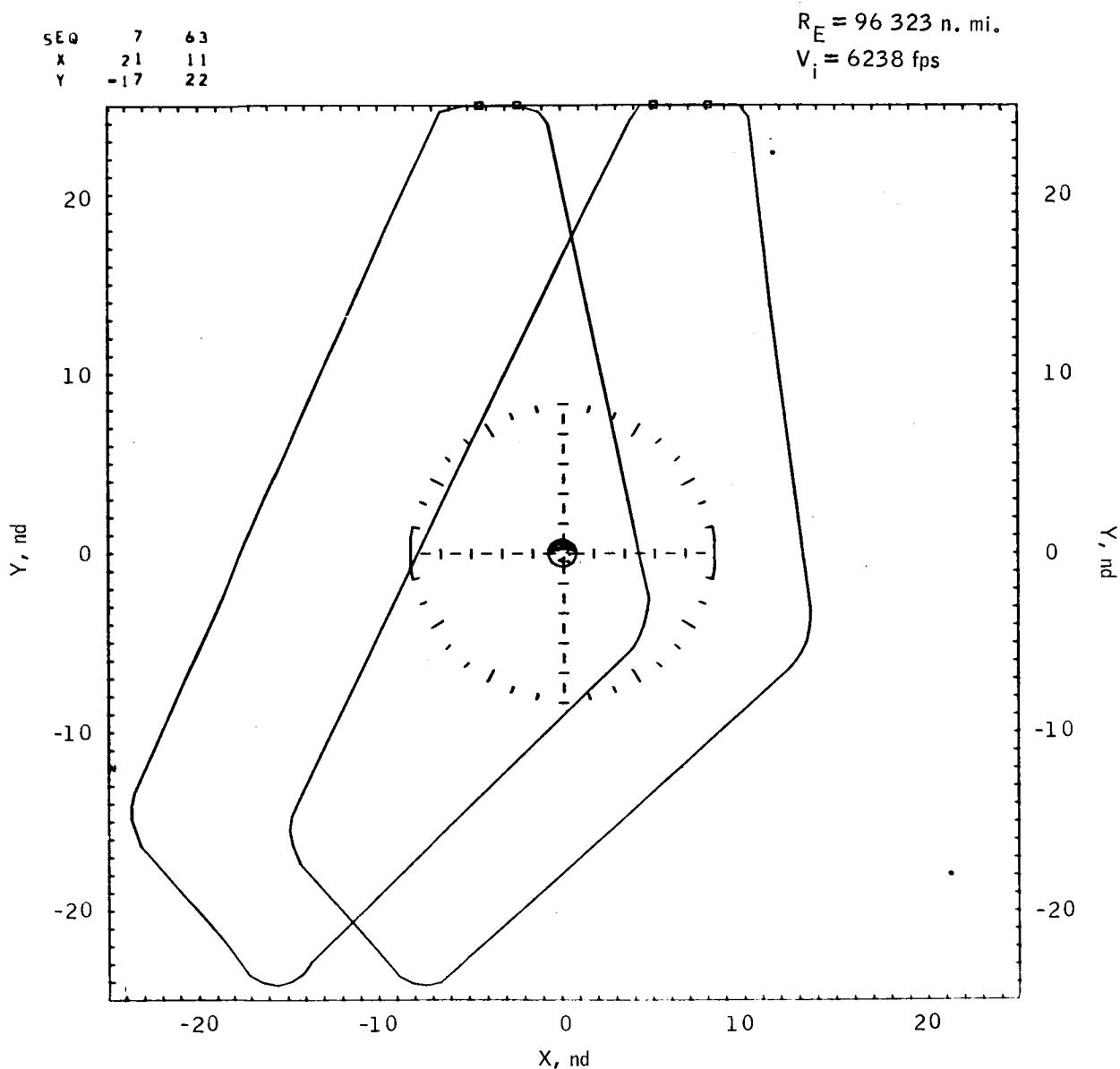
(i) Time from TEI cutoff = 30 hr.

Figure 8.- Continued.



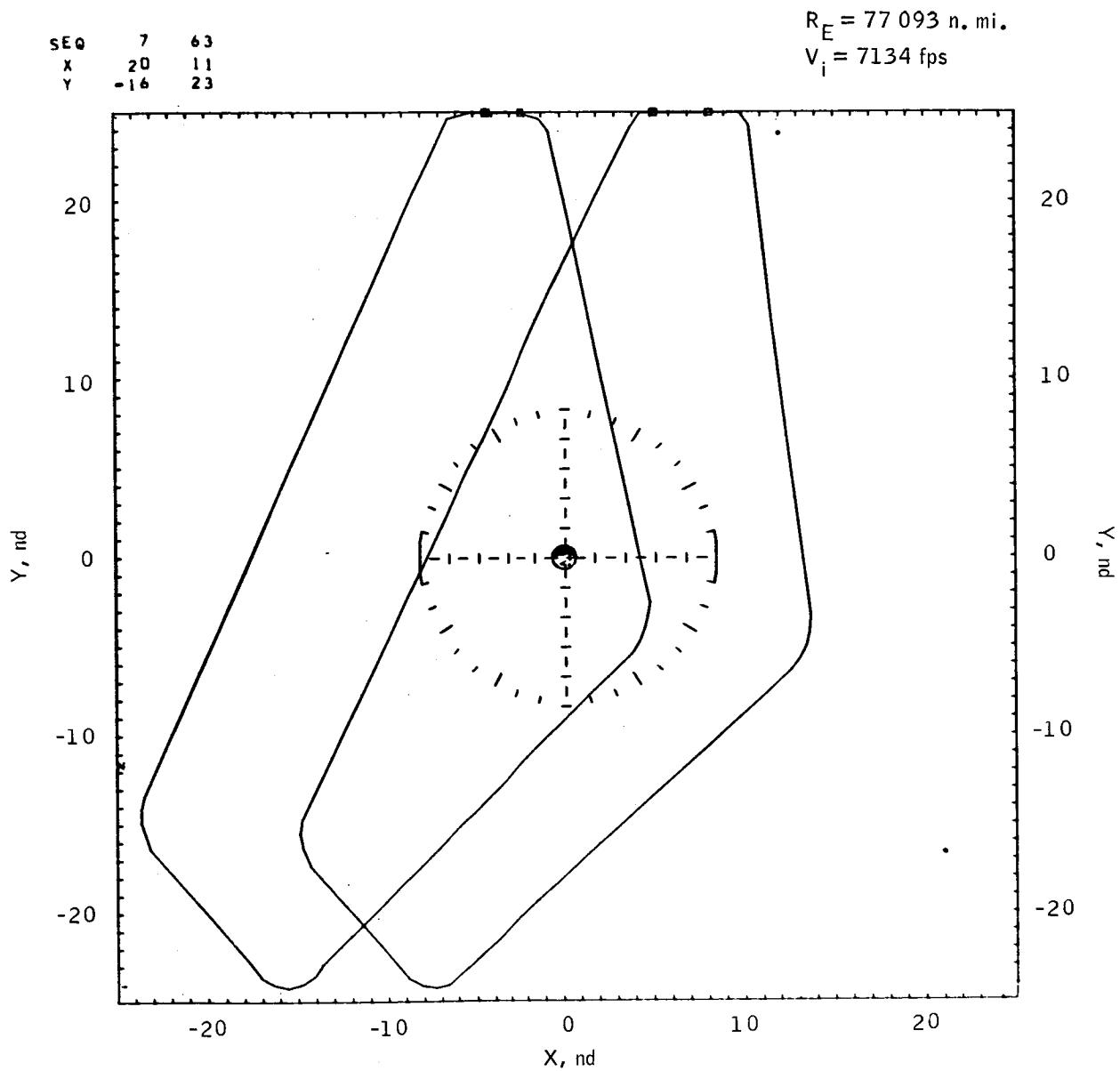
(j) Time from TEI cutoff = 35 hr.

Figure 8.- Continued.



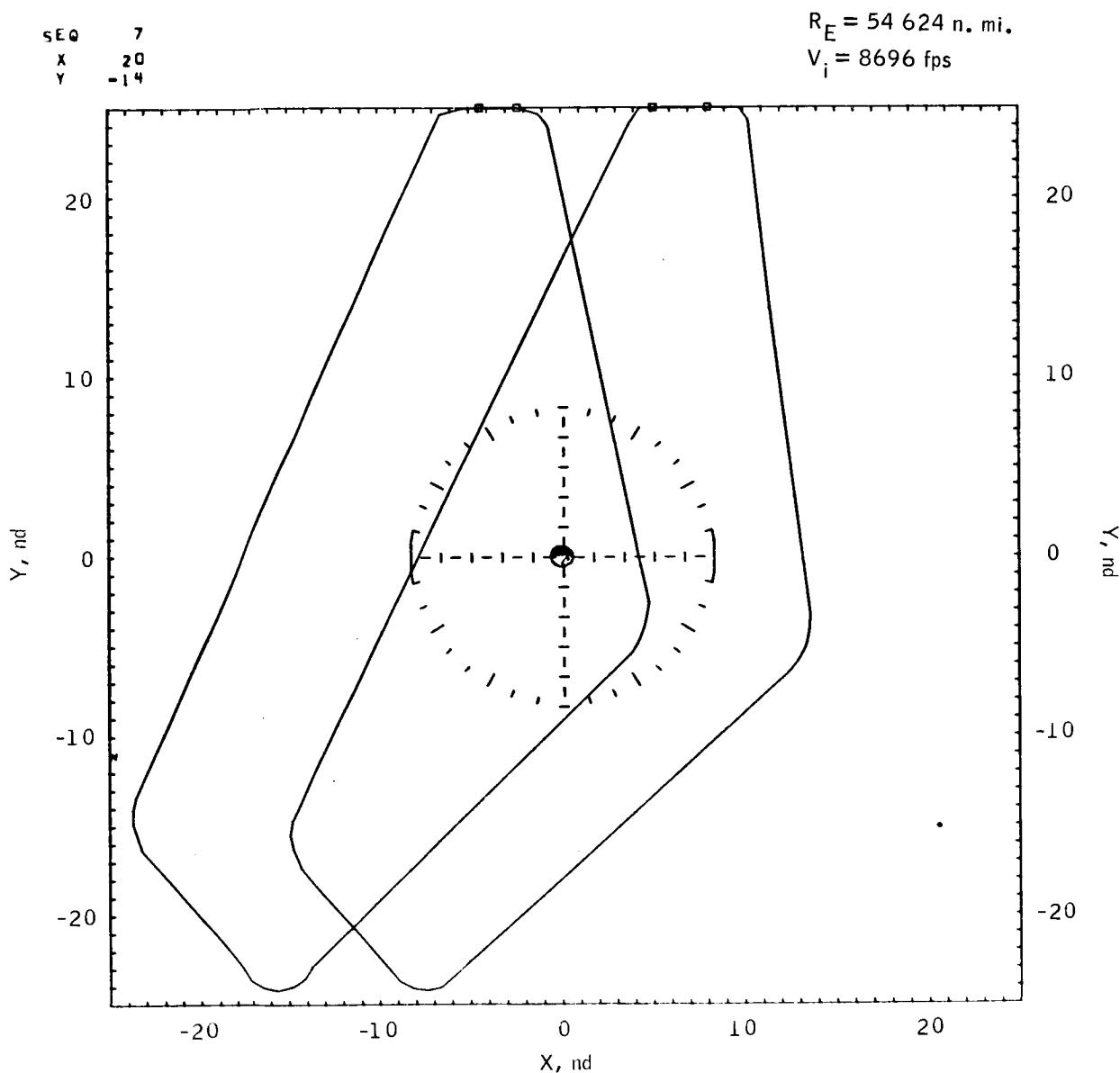
(k) Time from TEI cutoff = 40 hr.

Figure 8.- Continued.



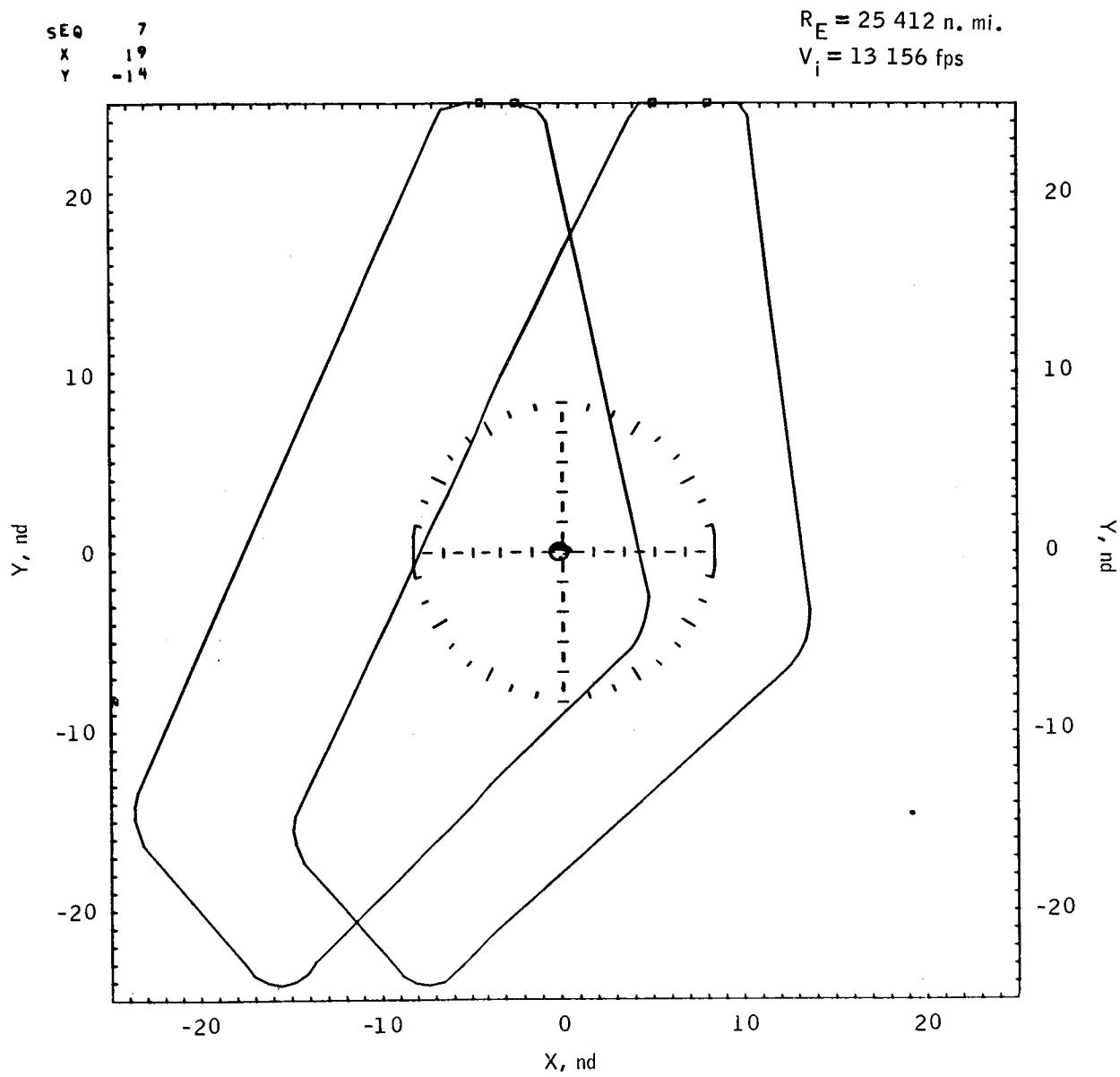
(I) Time from TEI cutoff = 45 hr.

Figure 8.- Continued.



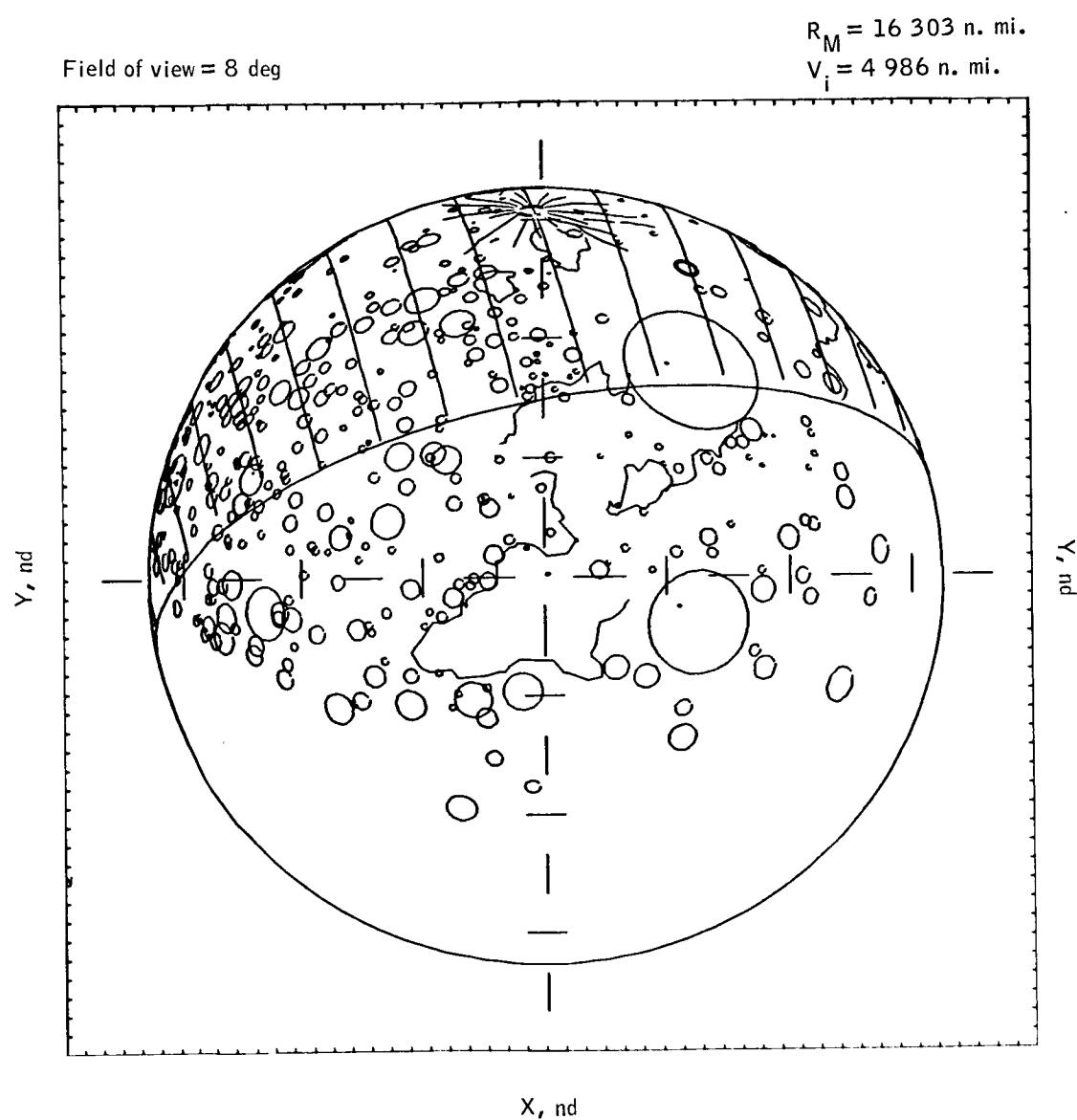
(m) Time from TEI cutoff = 50 hr.

Figure 8.- Continued.



(n) Time from TEI cutoff = 55 hr.

Figure 8.- Concluded.



(a) Time from TEI cutoff = 5 hr.

Figure 9.- Transearth coast (moon referenced) variable field of view.

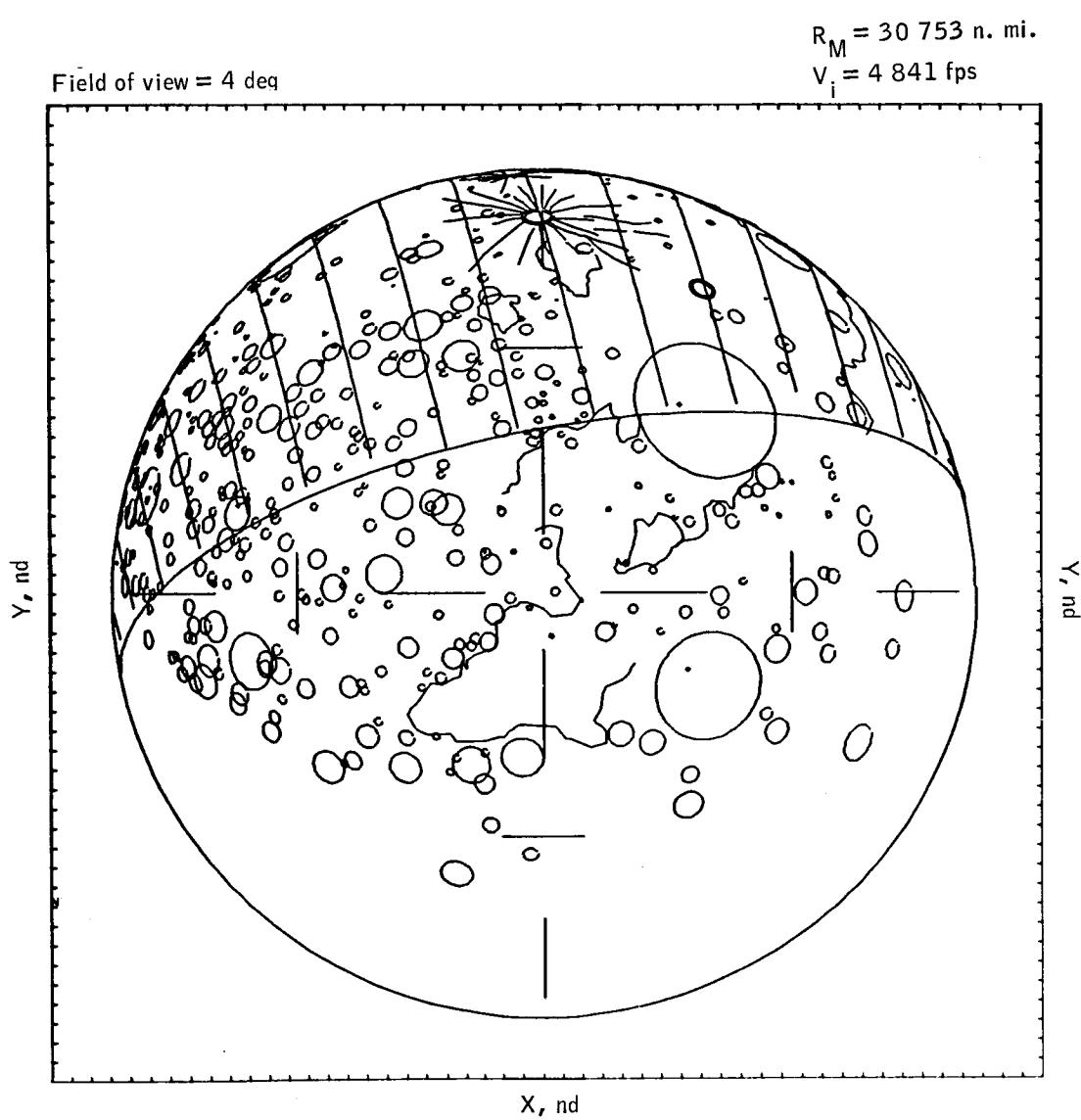
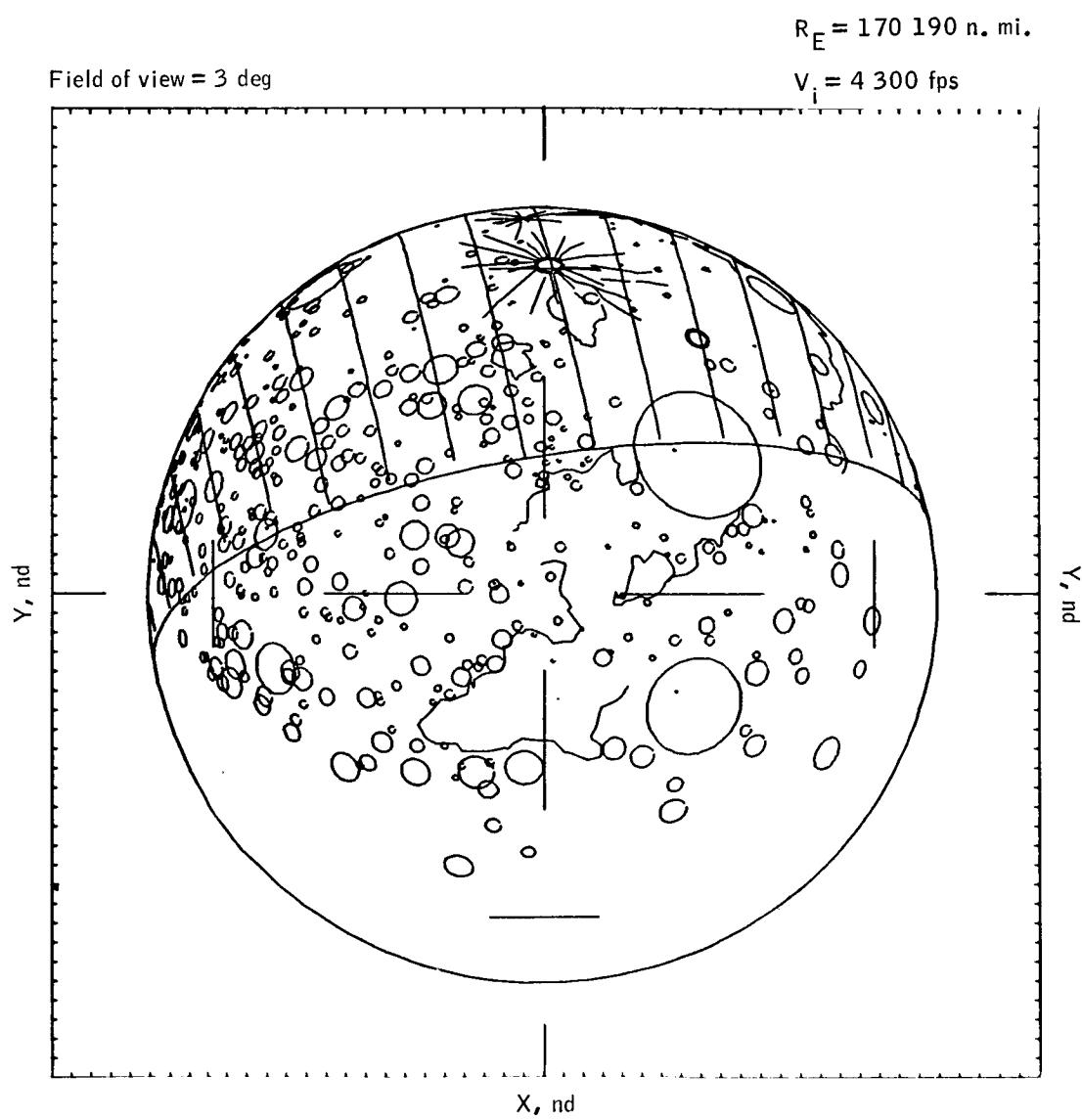


Figure 9.- Continued.



(c) Time from TEI cutoff = 15 hr.

Figure 9.- Continued.

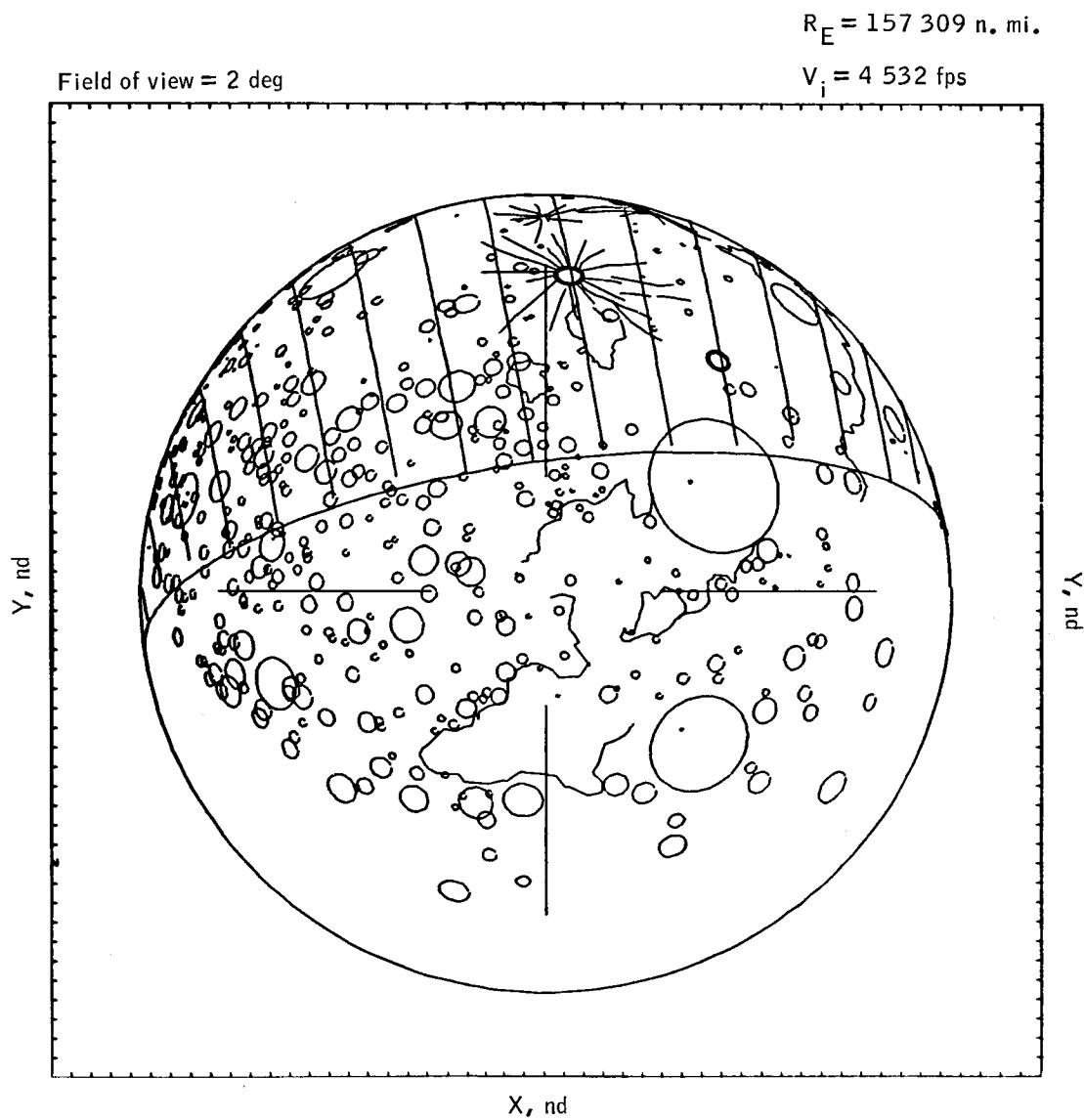


Figure 9.- Continued.

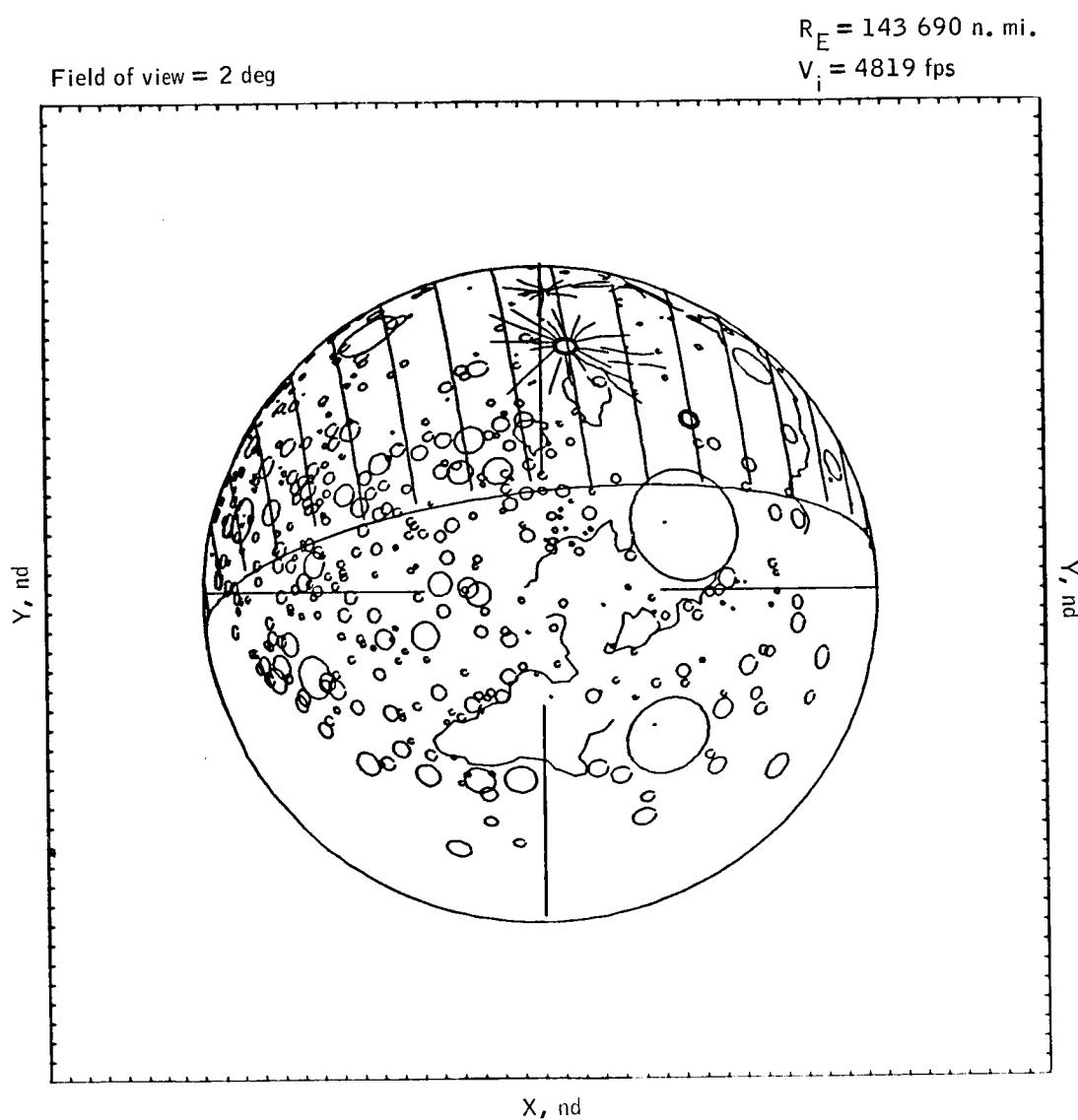
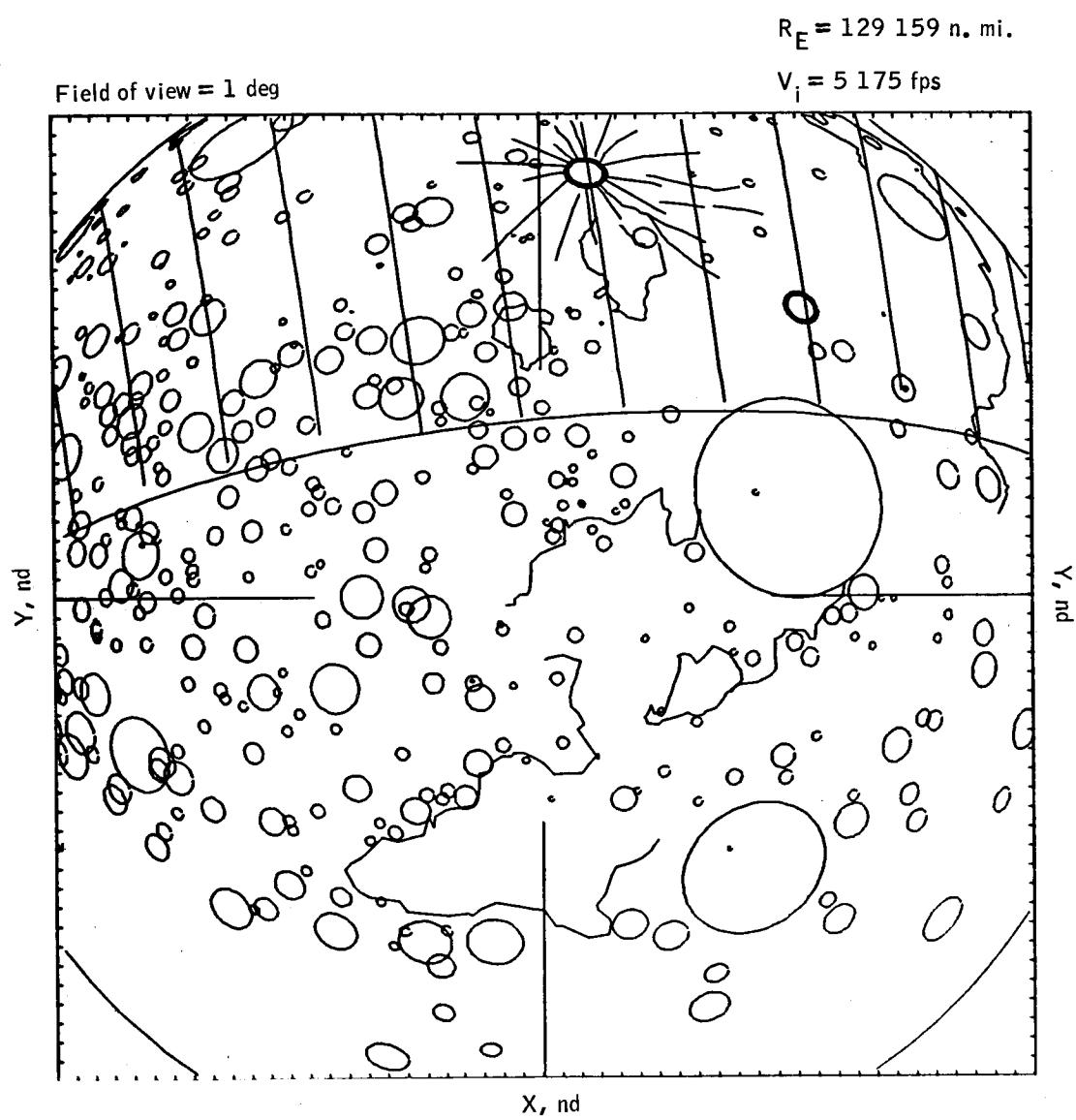


Figure 9.- Continued.



(f) Time from TEI cutoff = 30 hr.

Figure 9.- Continued.

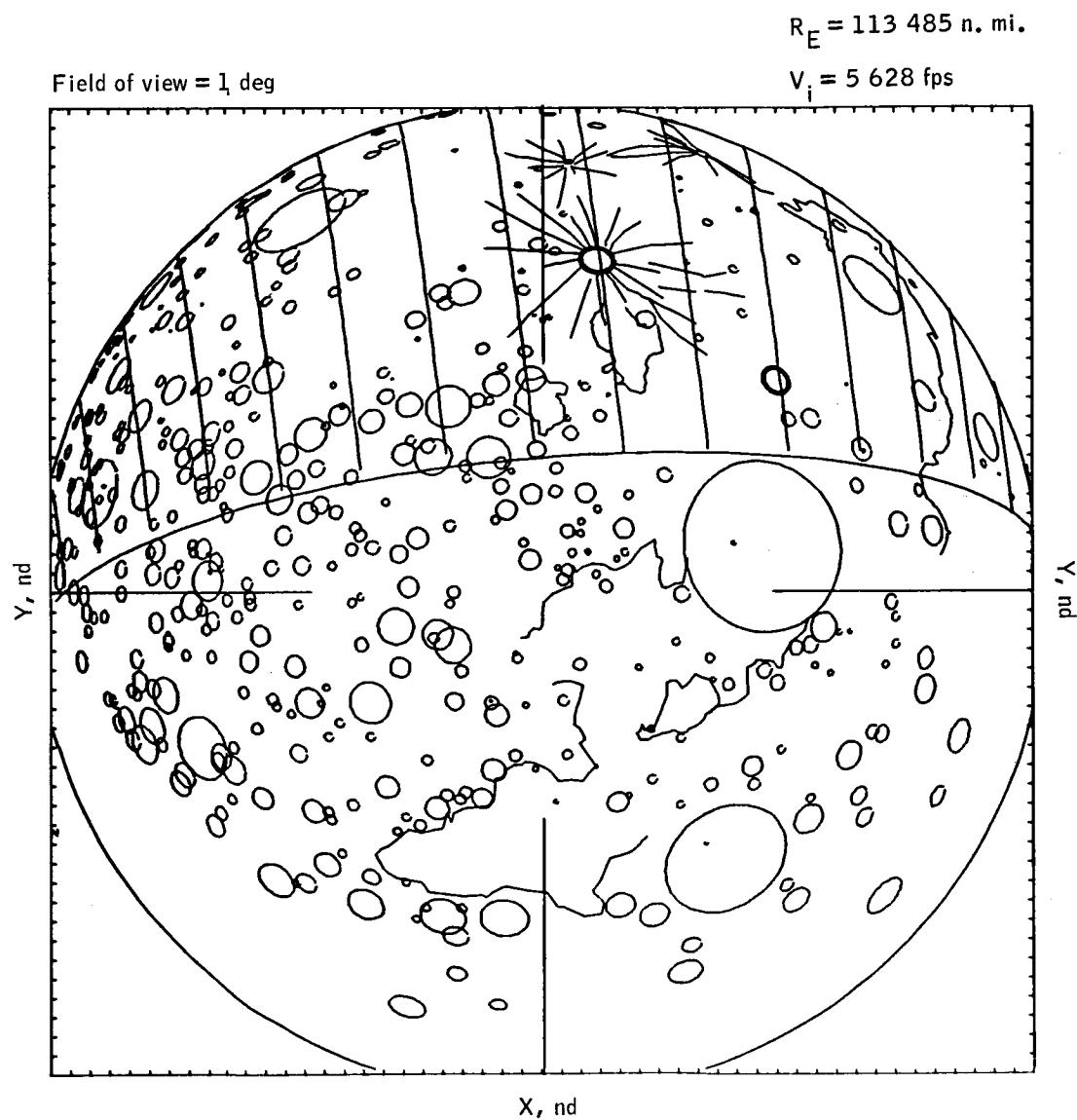


Figure 9.- Continued.

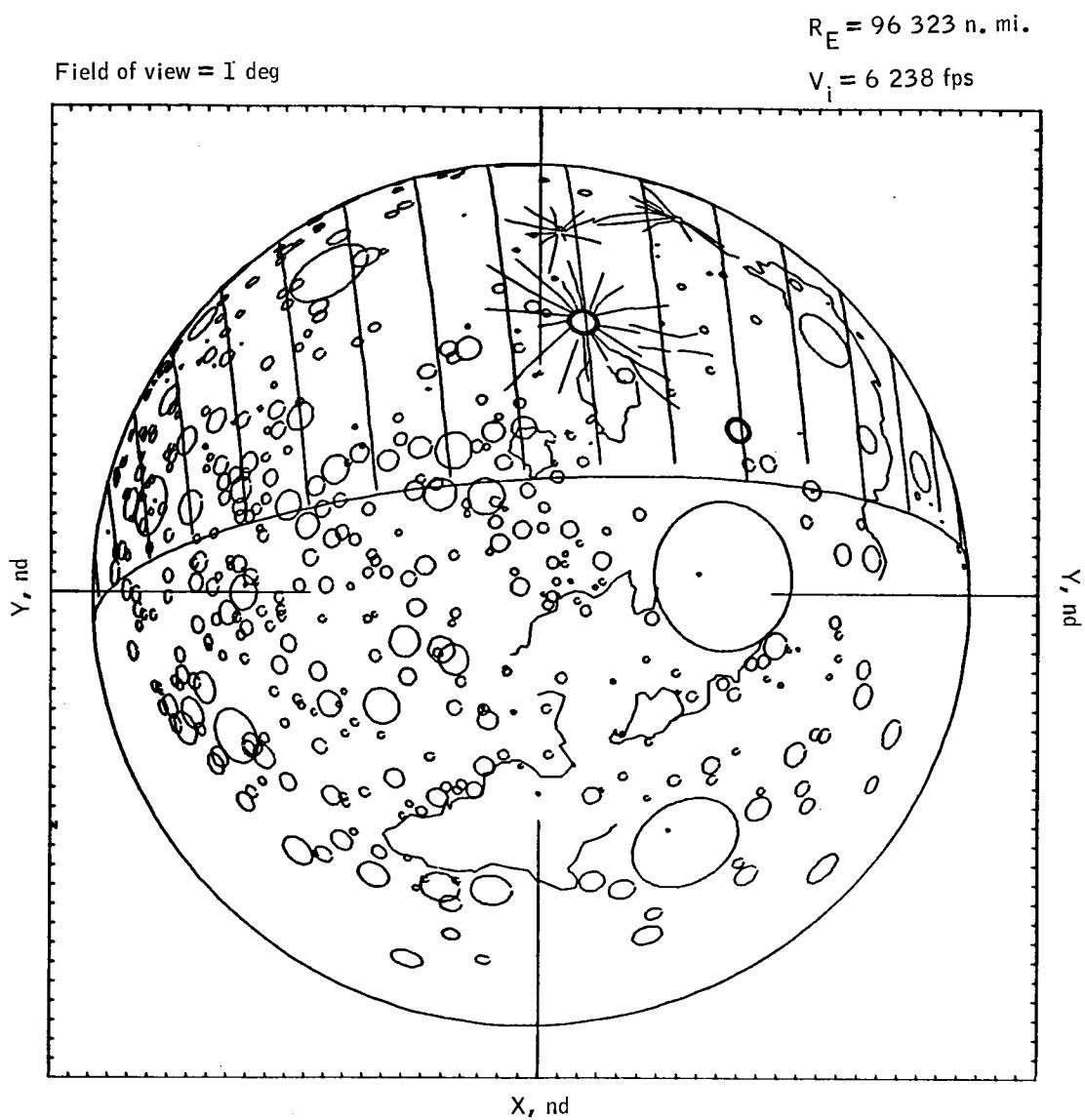


Figure 9.- Continued.

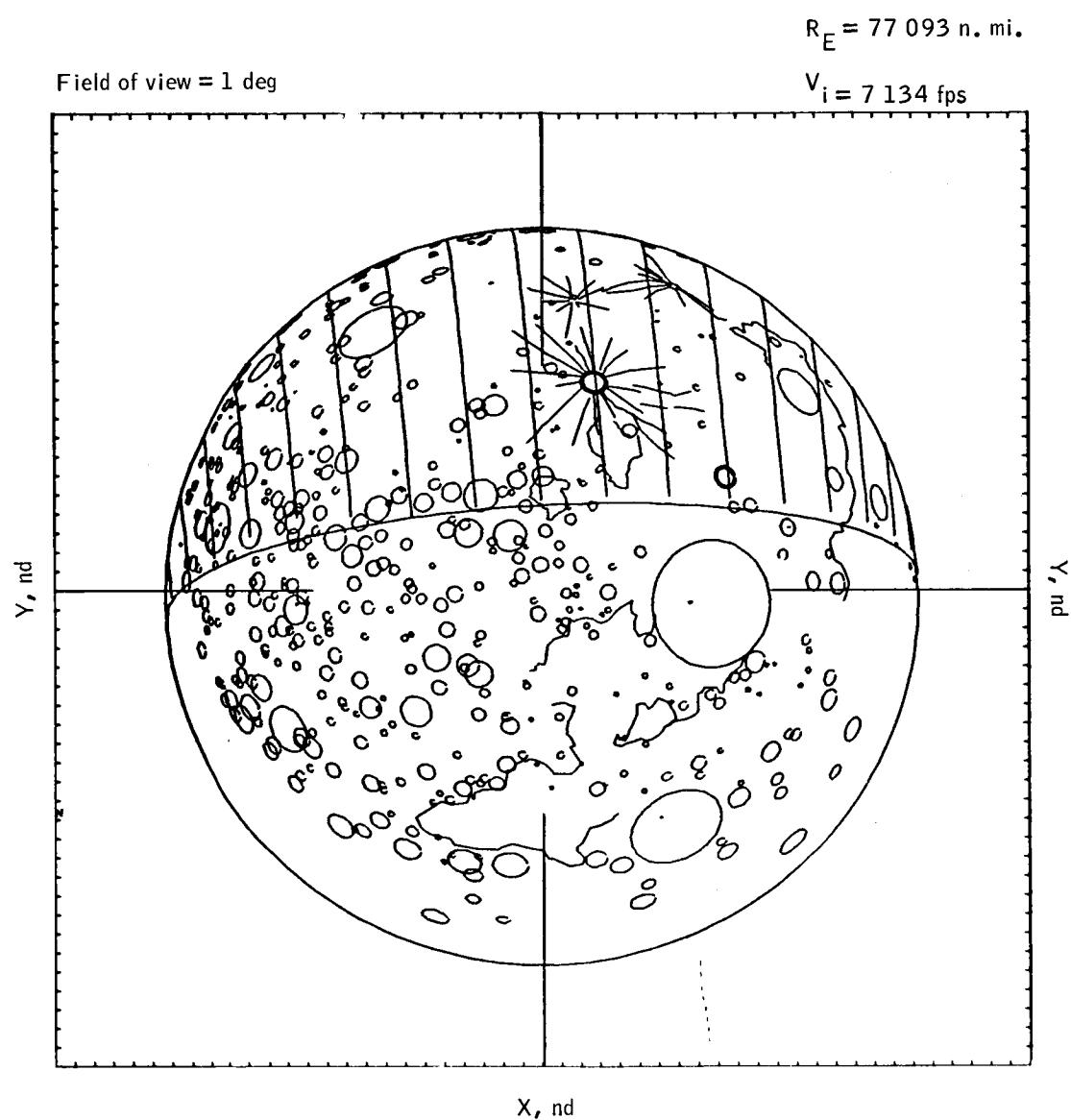


Figure 9.- Continued.

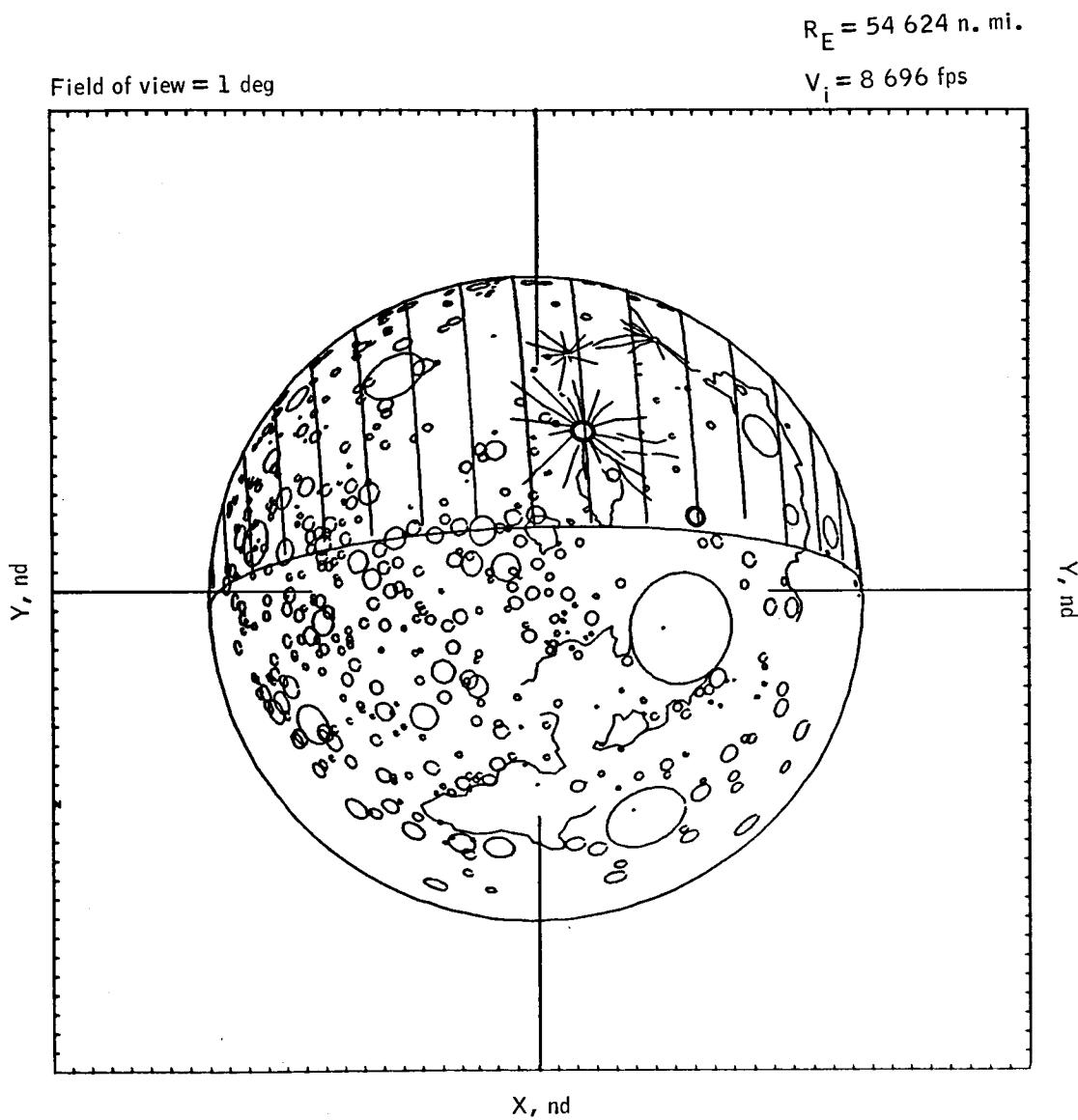


Figure 9.- Continued.

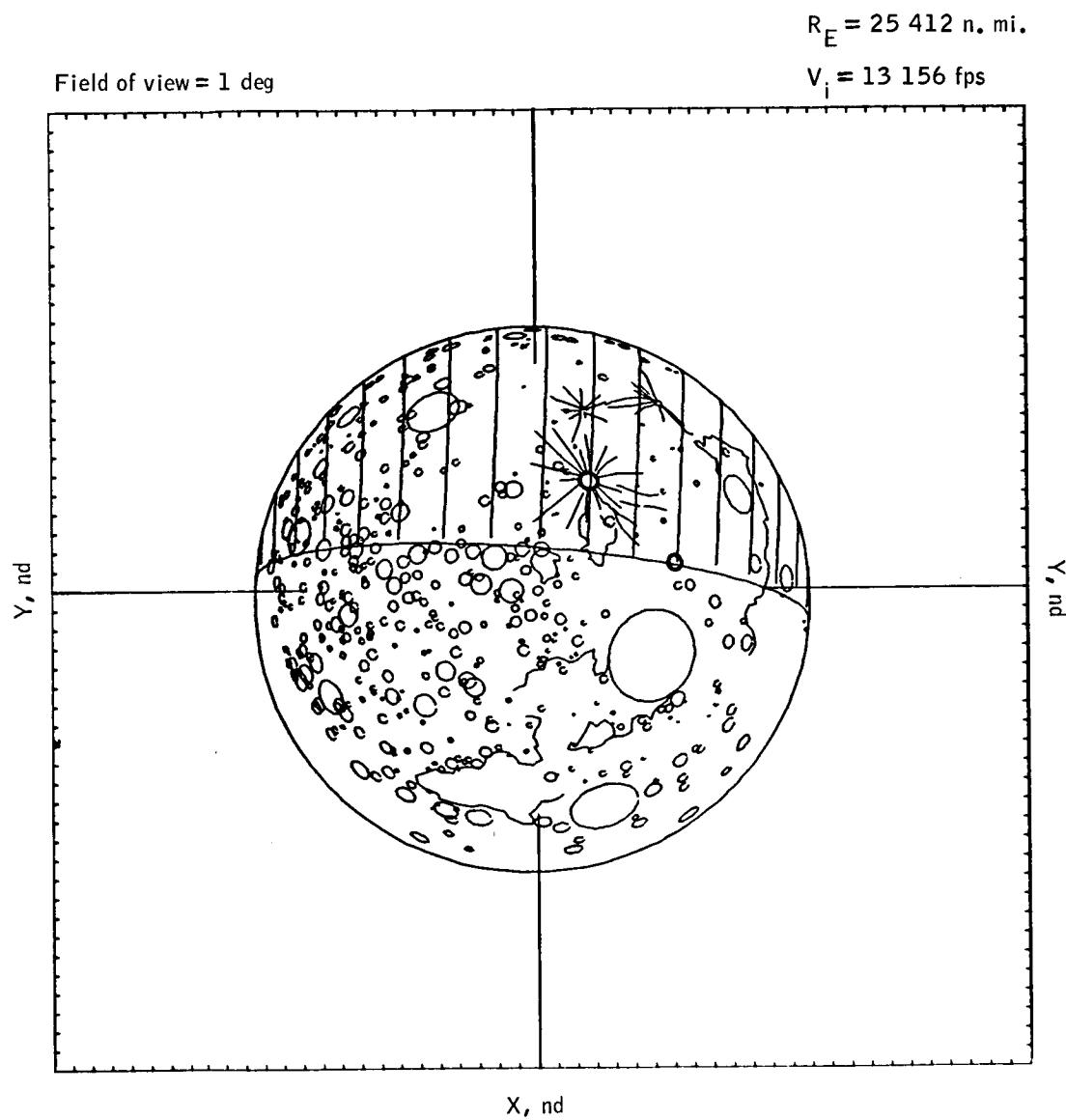
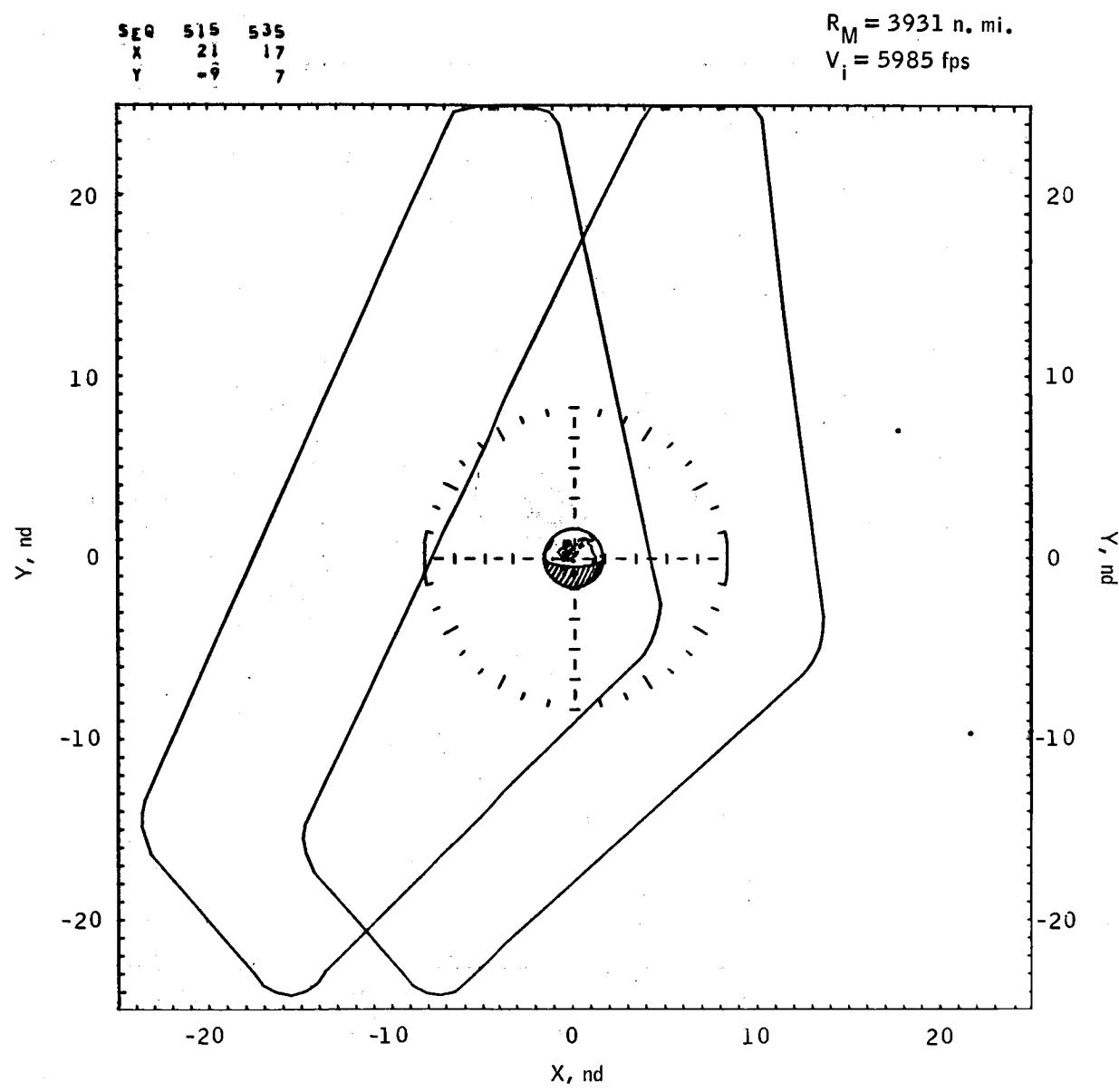


Figure 9.- Concluded.



(a) Time from TEI cutoff = 1 hr.

Figure 10.- Transearth coast (earth referenced) constant field of view.

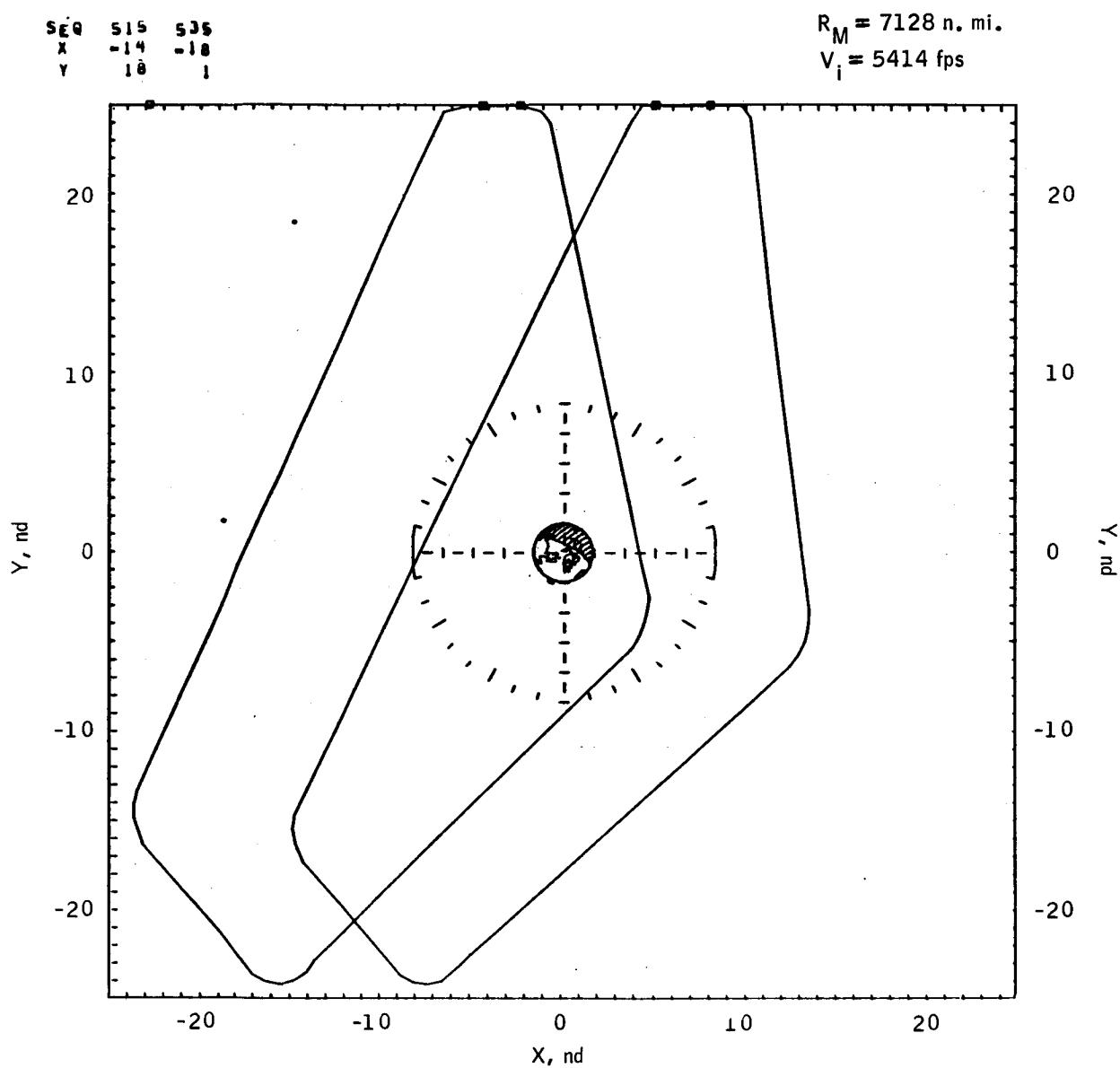


Figure 10.- Continued.

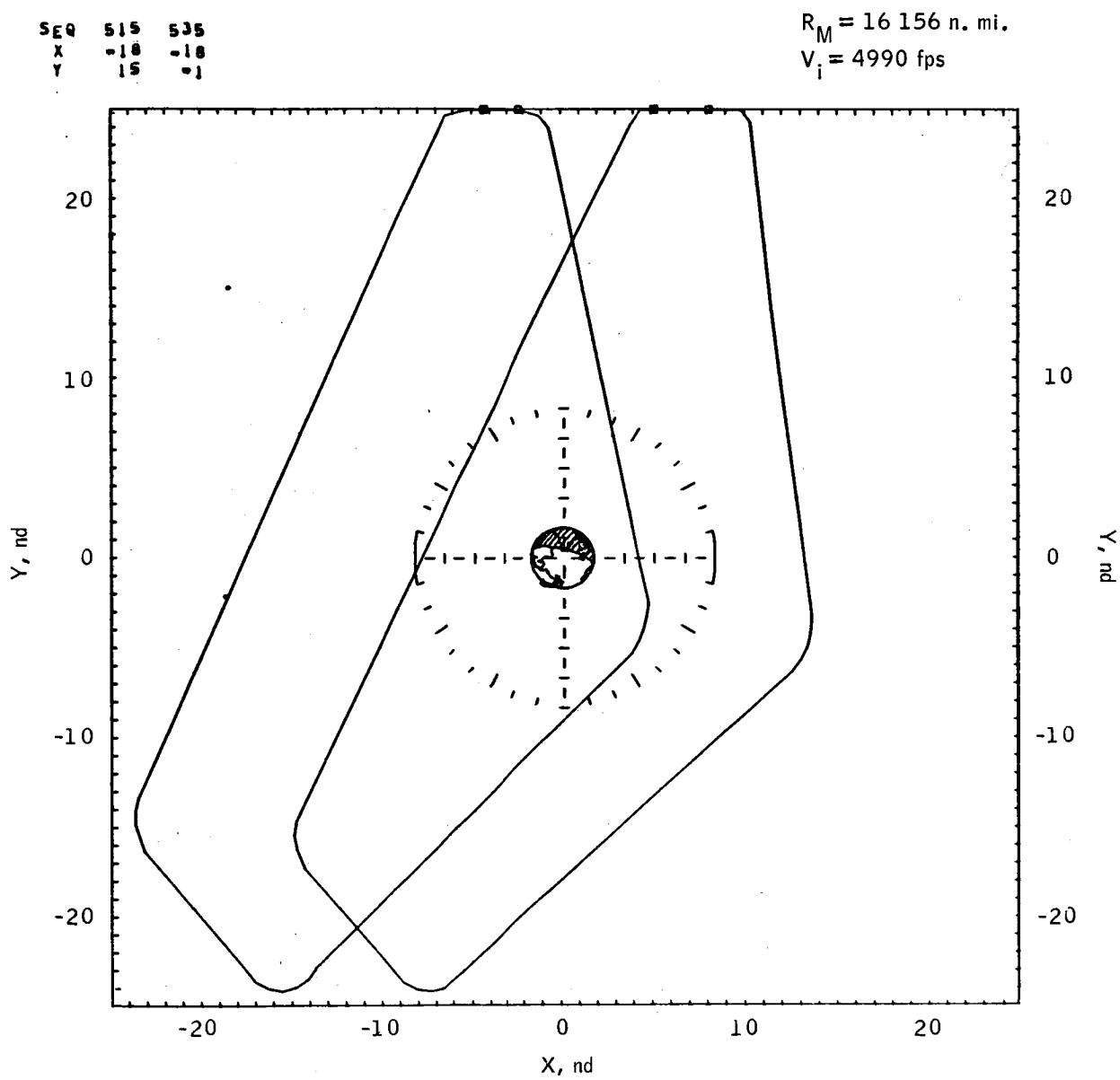


Figure 10.- Continued.

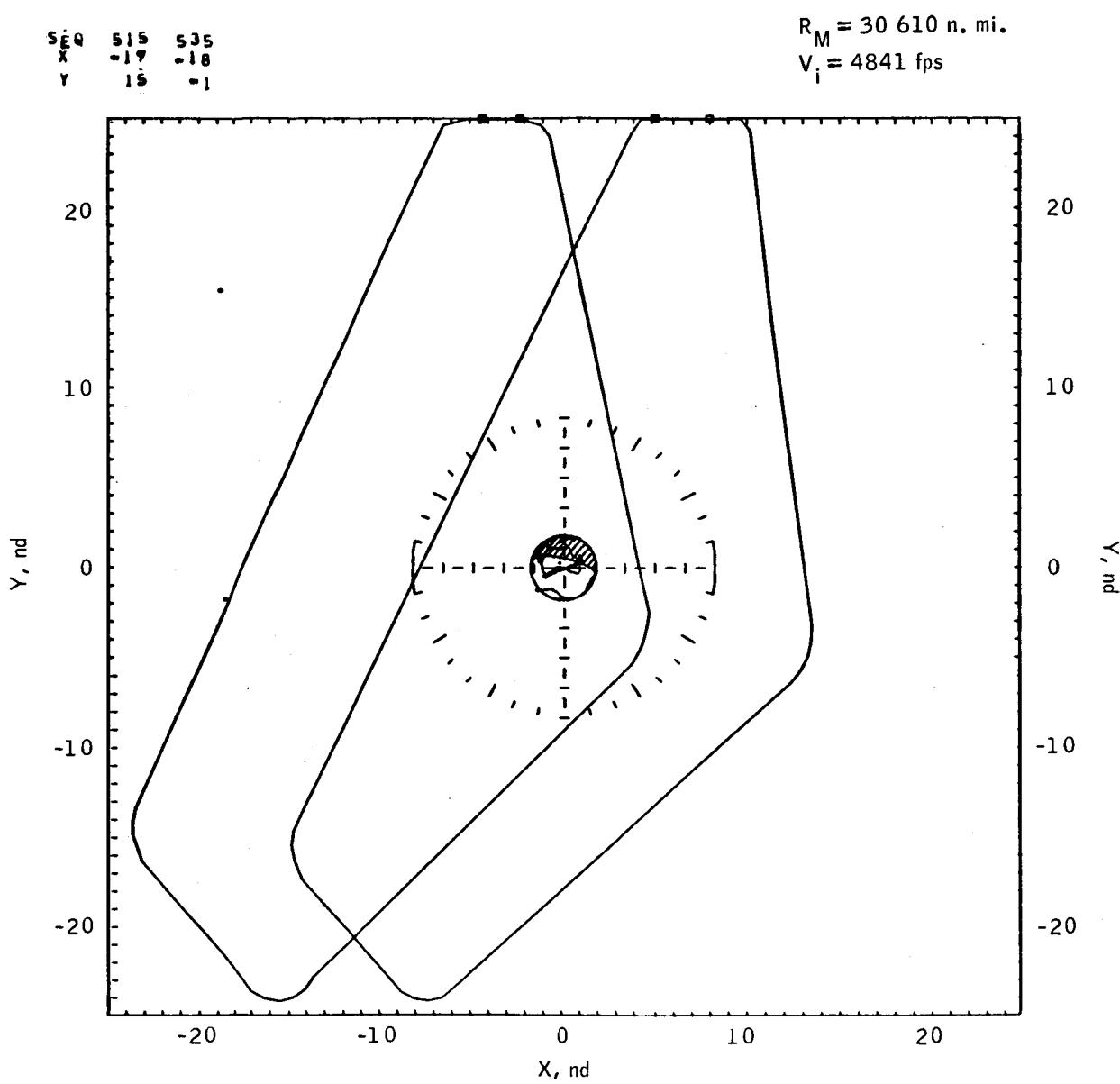


Figure 10.- Continued.

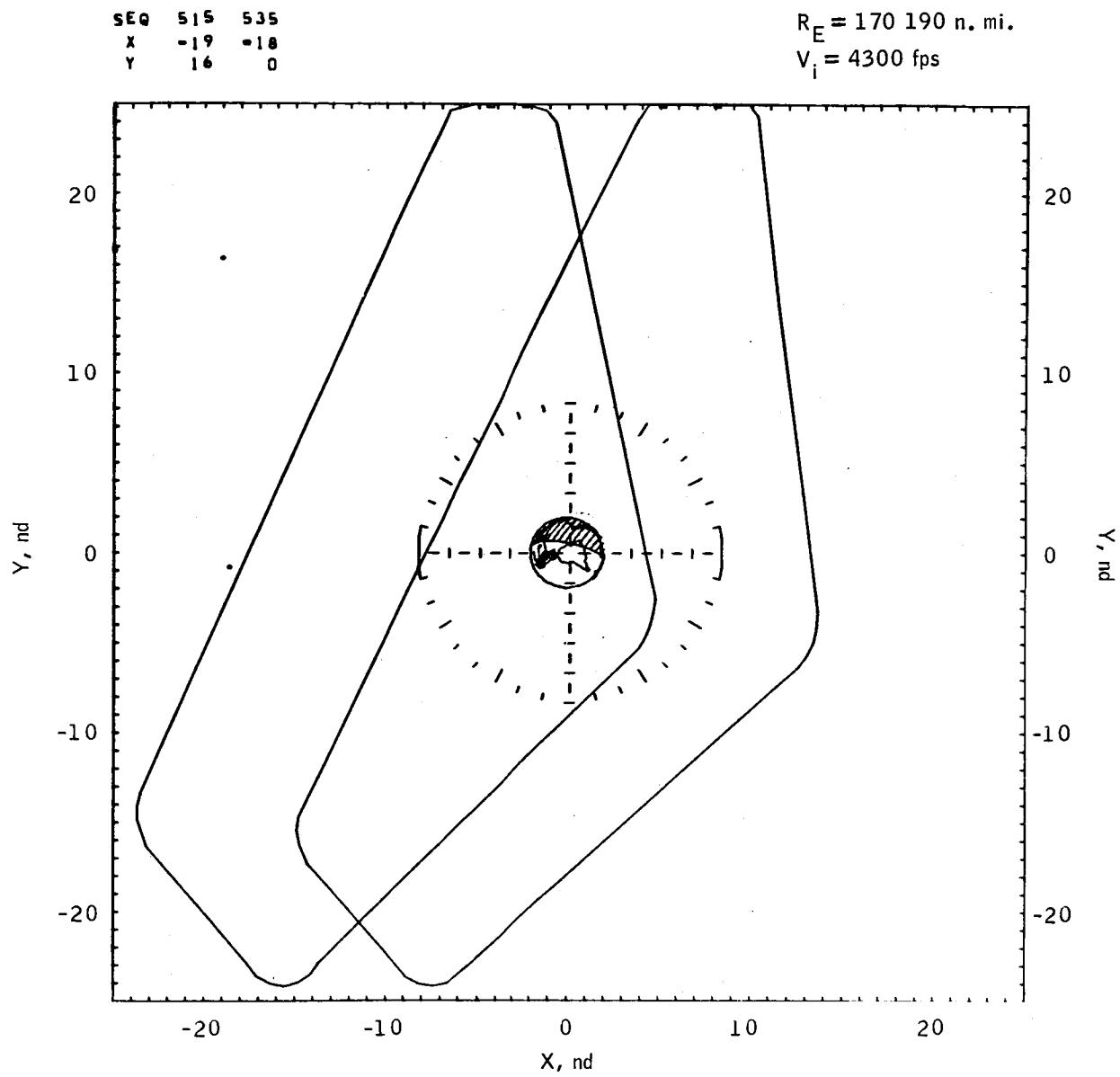
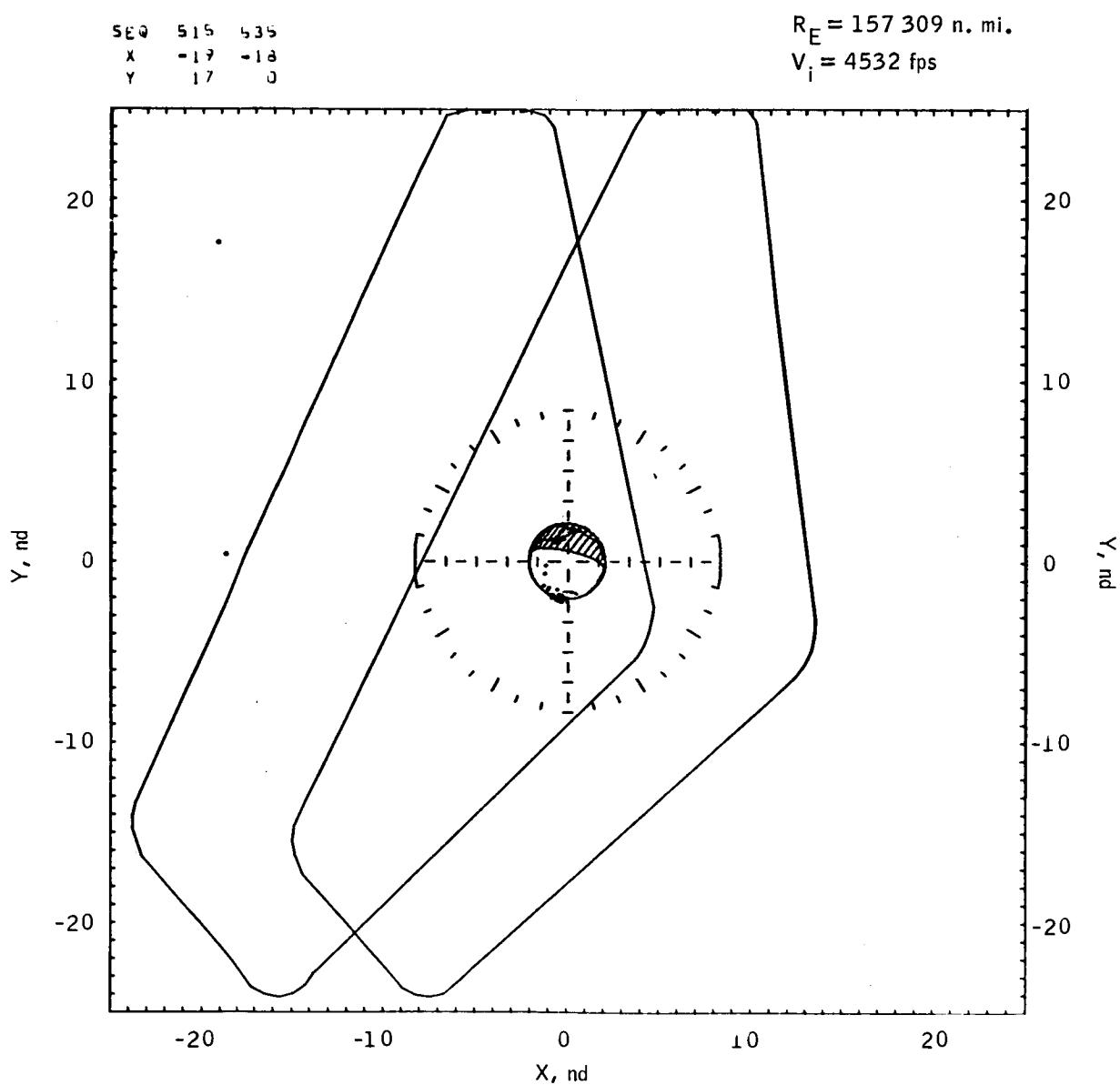


Figure 10.- Continued.



(f) Time from TEI cutoff = 20 hr.

Figure 10.- C0ntinued.

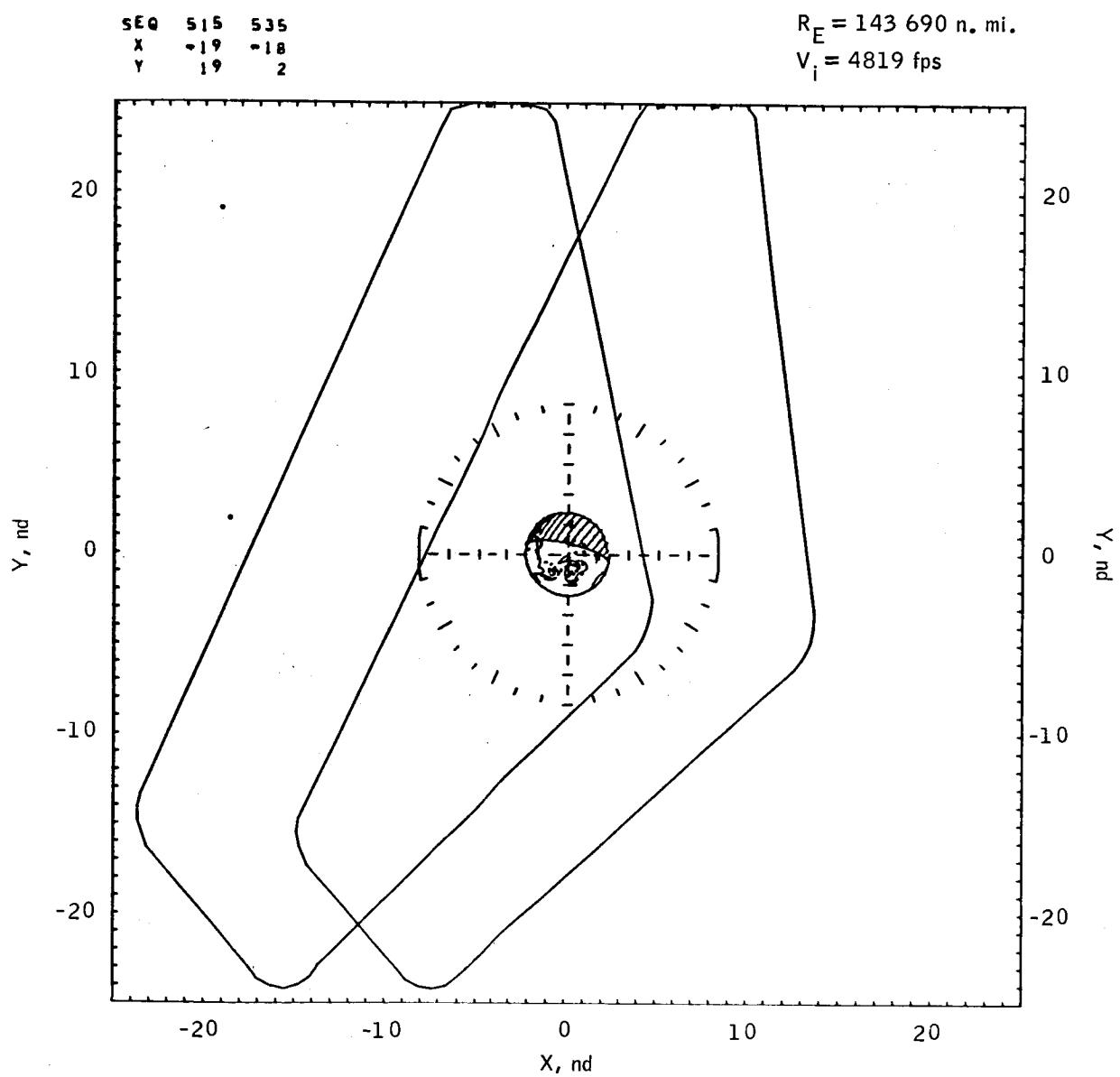
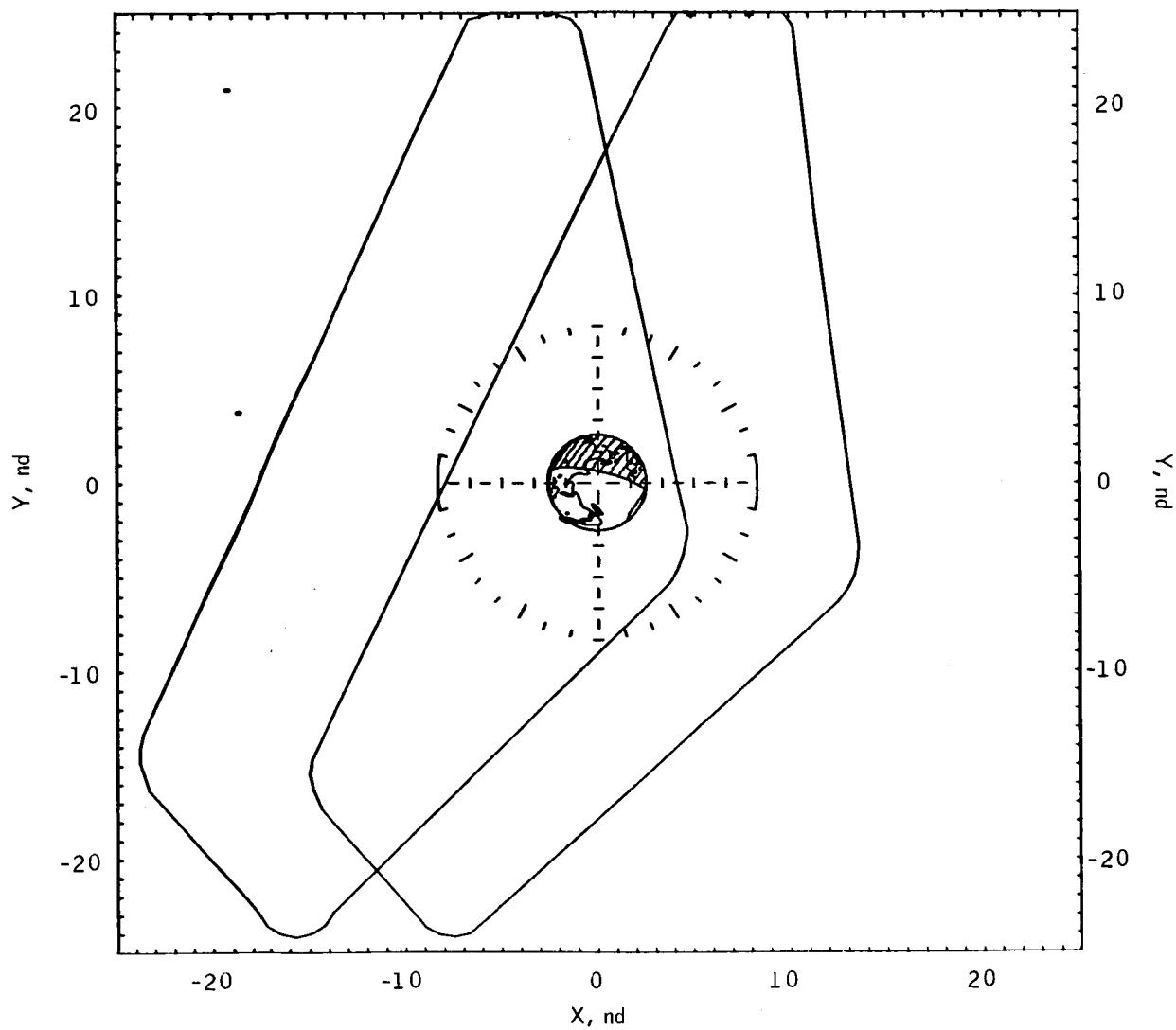


Figure 10.- Continued.

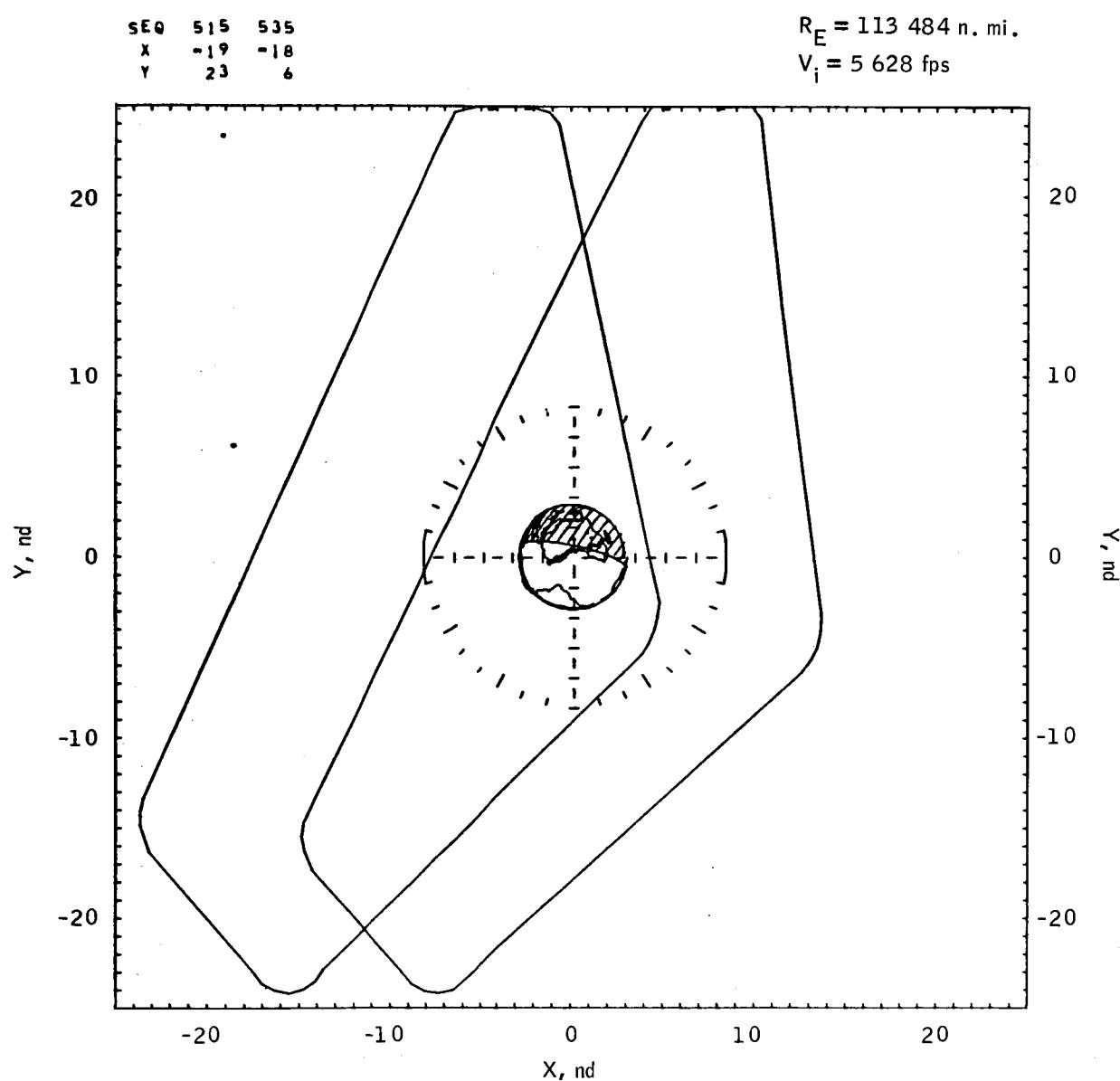
SEQ 515 535
 X -19 -18
 Y 21 3

$R_E = 129\,159$ n. mi.
 $V_i = 5\,175$ fpm



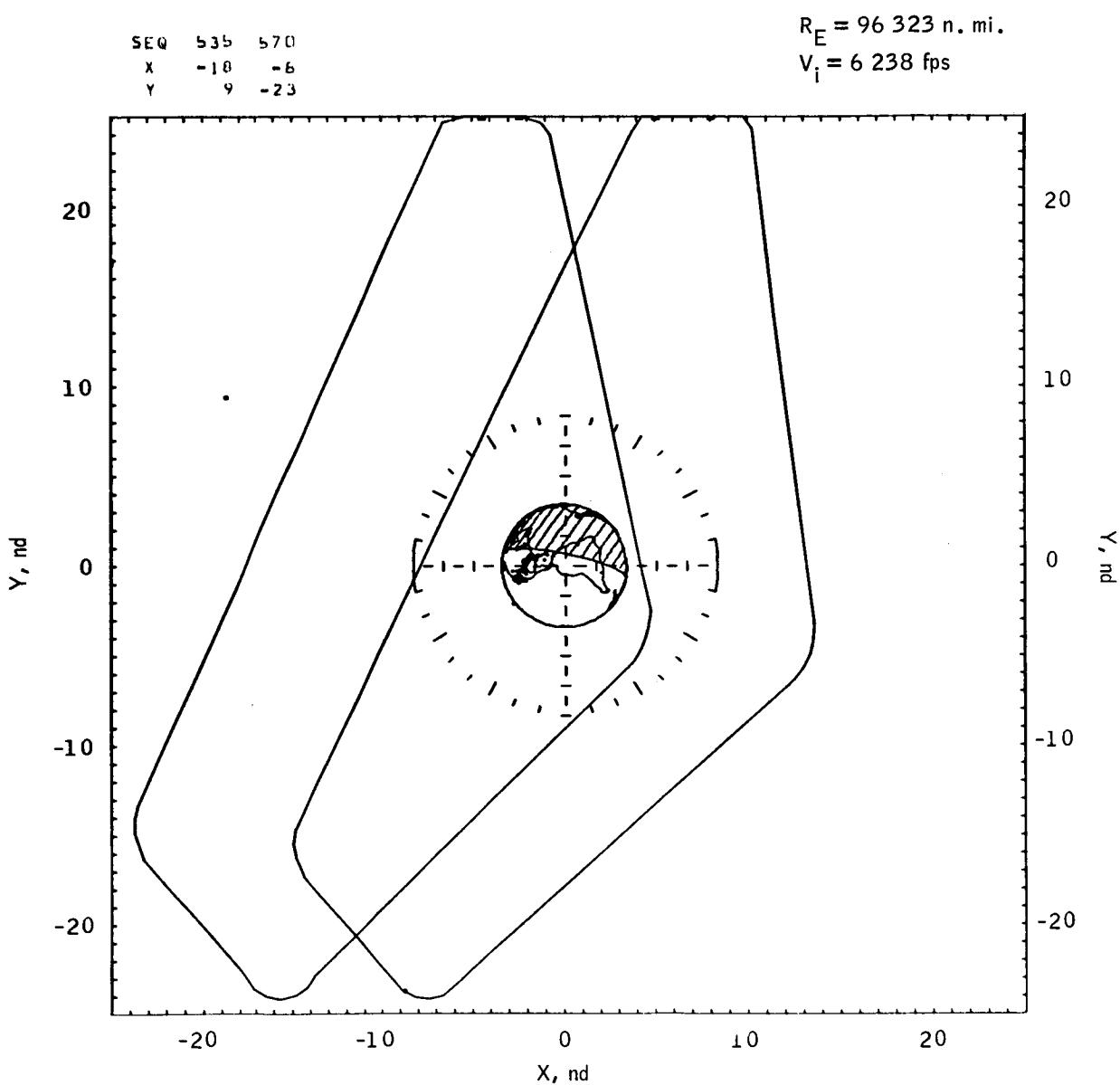
(h) Time from TEI cutoff = 30 hr.

Figure 10.- Continued.



(i) Time from TEI cutoff = 35 hr.

Figure 10.- Continued.



(j) Time from TEI cutoff = 40 hr.

Figure 10.- Continued.

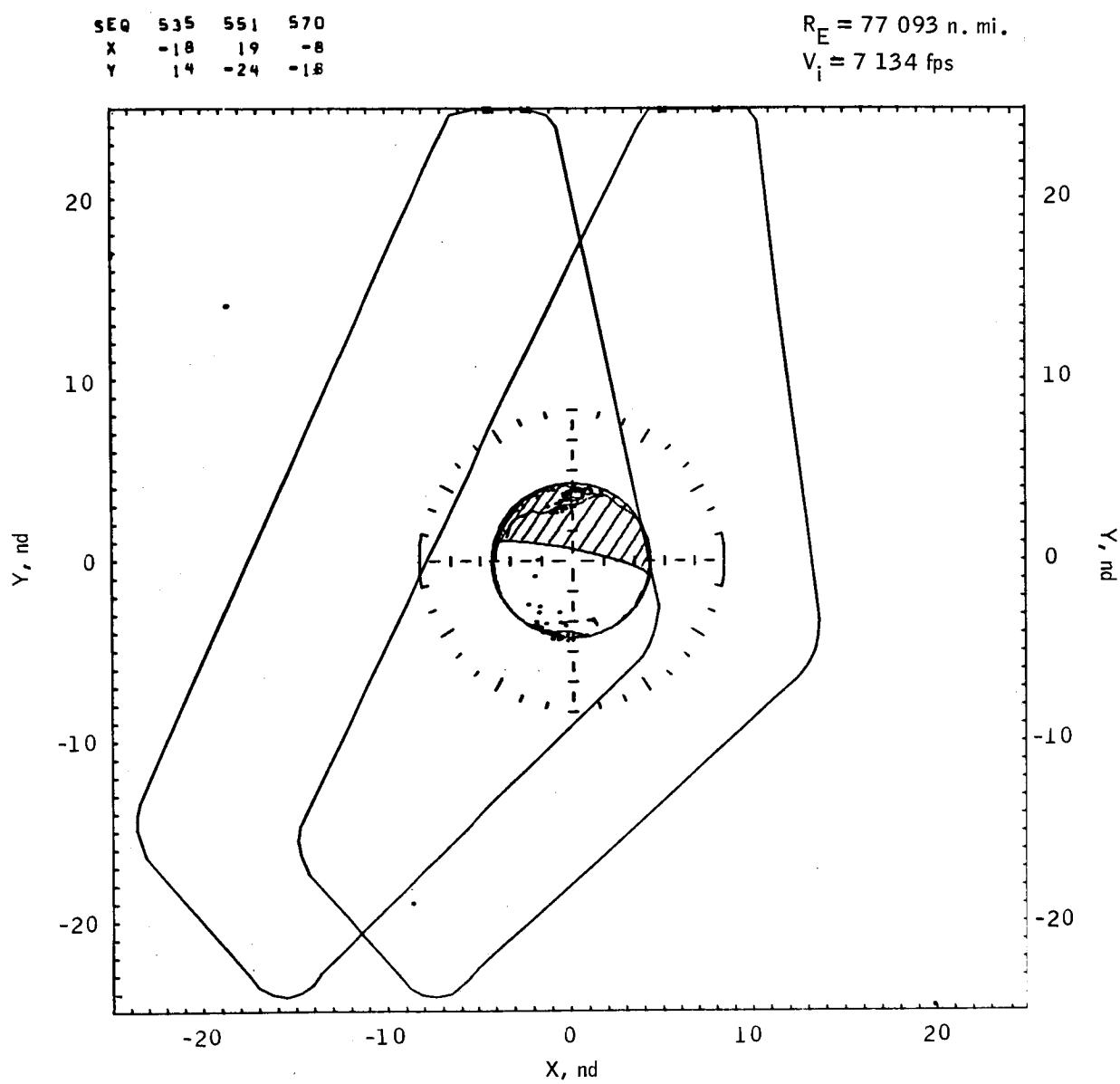


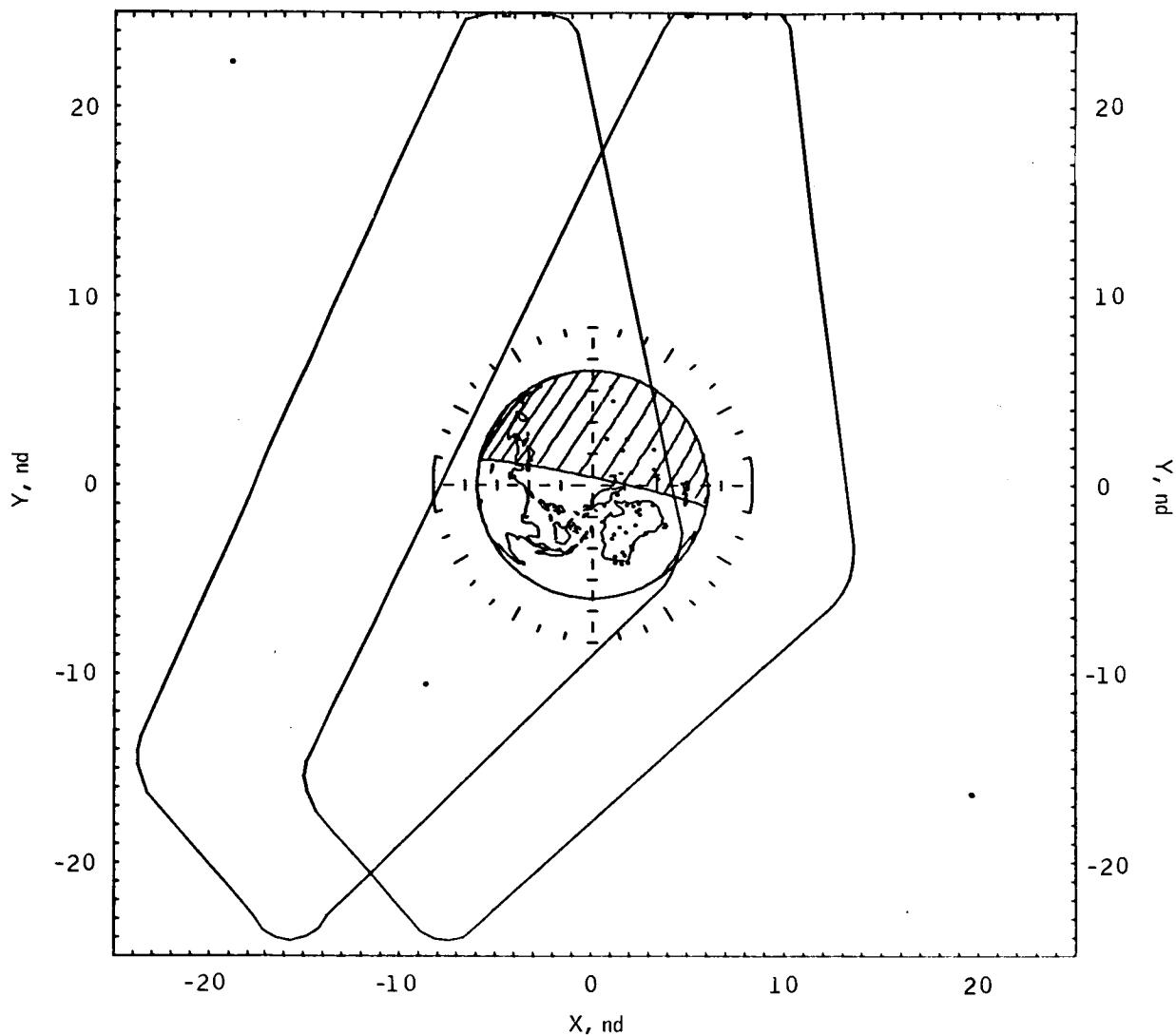
Figure 10.- Continued.

100

SEQ	535	551	570
X	-13	19	-8
Y	22	-16	-10

$$R_E = 54\,624 \text{ n. mi.}$$

$$V_i = 8\,694 \text{ fps}$$



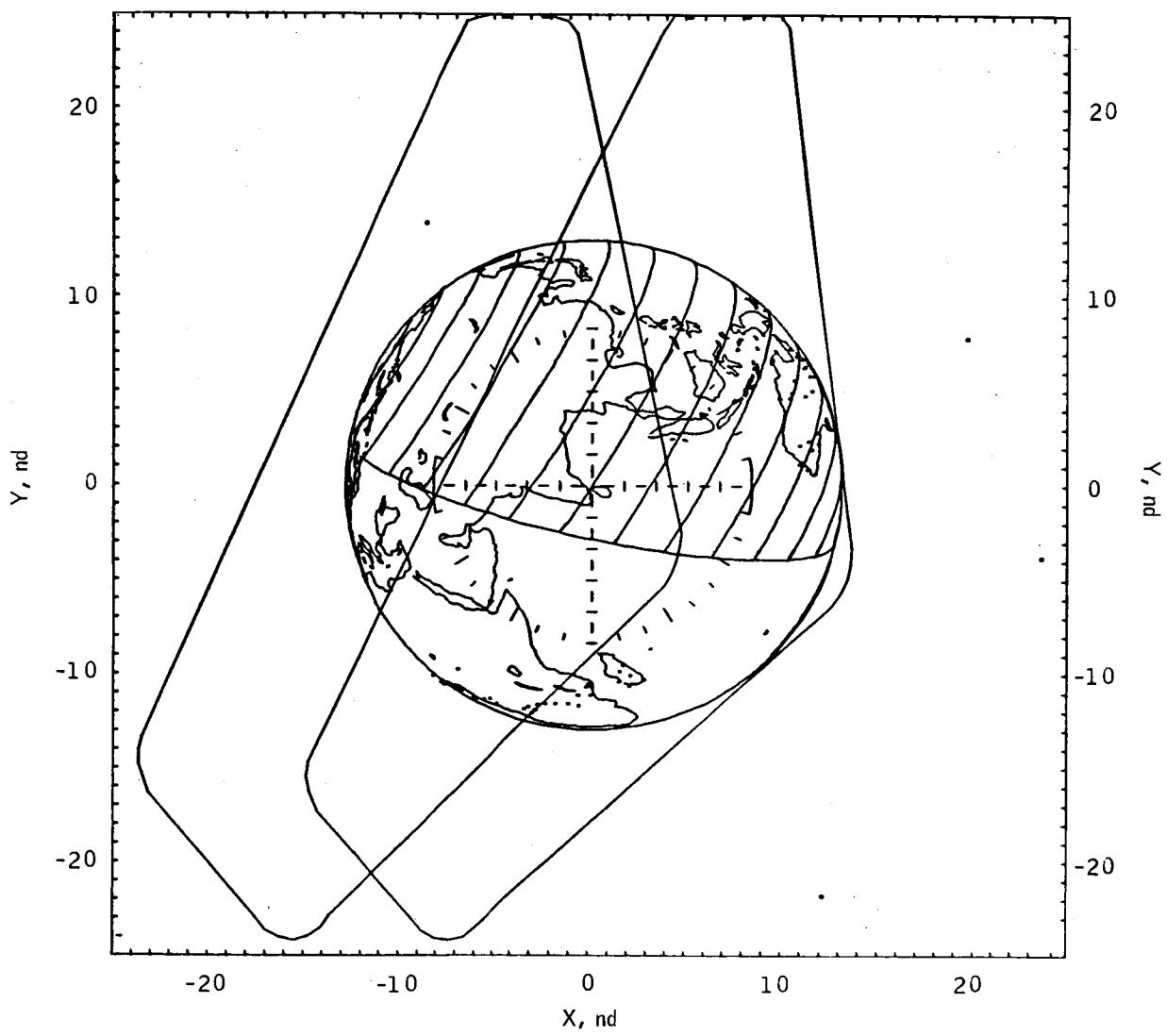
(I) Time from TEI cutoff = 50 hr.

Figure 10.- Continued.

SEQ	551	566	570	595
X	19	23	-8	11
Y	7	-3	14	-21

$$R_E = 25\ 412 \text{ n. mi.}$$

$$V_i = 13\ 156 \text{ fps}$$

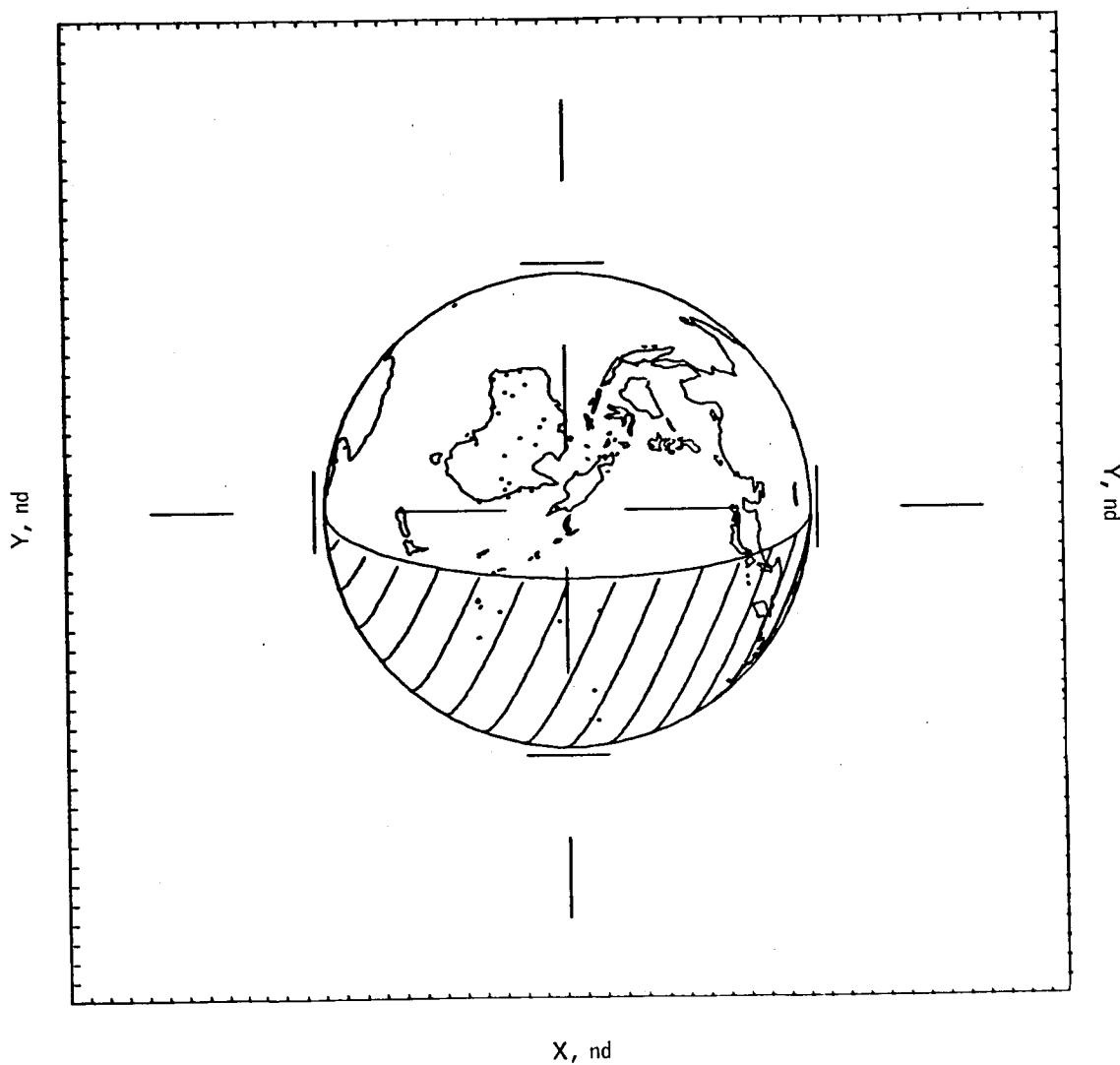


(m) Time from TEI cutoff = 55 hr.

Figure 10.- Concluded.

Field of view = 4 deg

$$R_M = 3931 \text{ n. mi.}$$
$$V_i = 5985 \text{ fps}$$

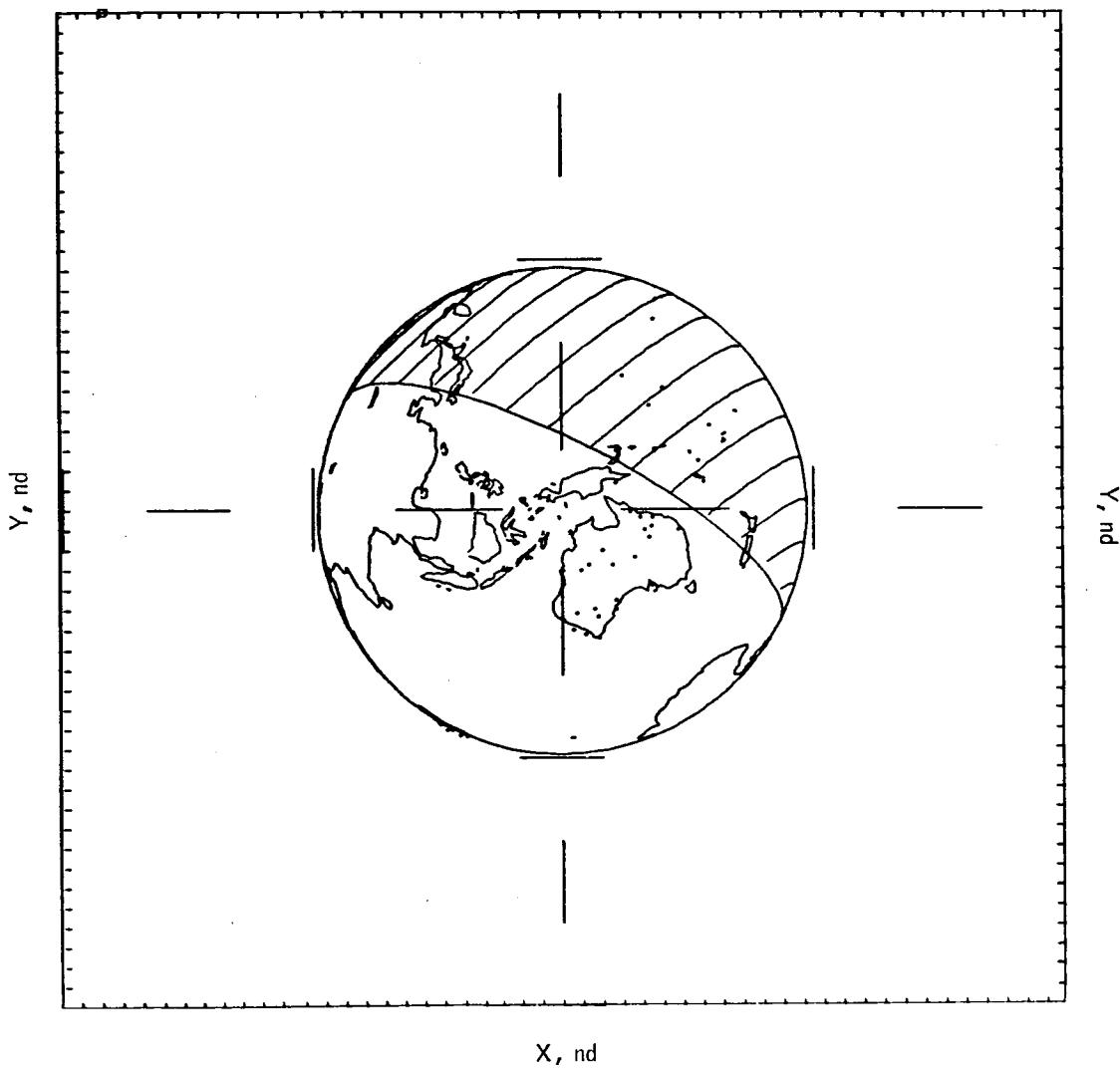


(a) Time from TEI cutoff = 1 hr.

Figure 11. - Transearth coast (earth referenced) variable field of view.

Field of view = 4 deg

$$R_M = 7128 \text{ n. mi.}$$
$$V_i = 5414 \text{ fps}$$

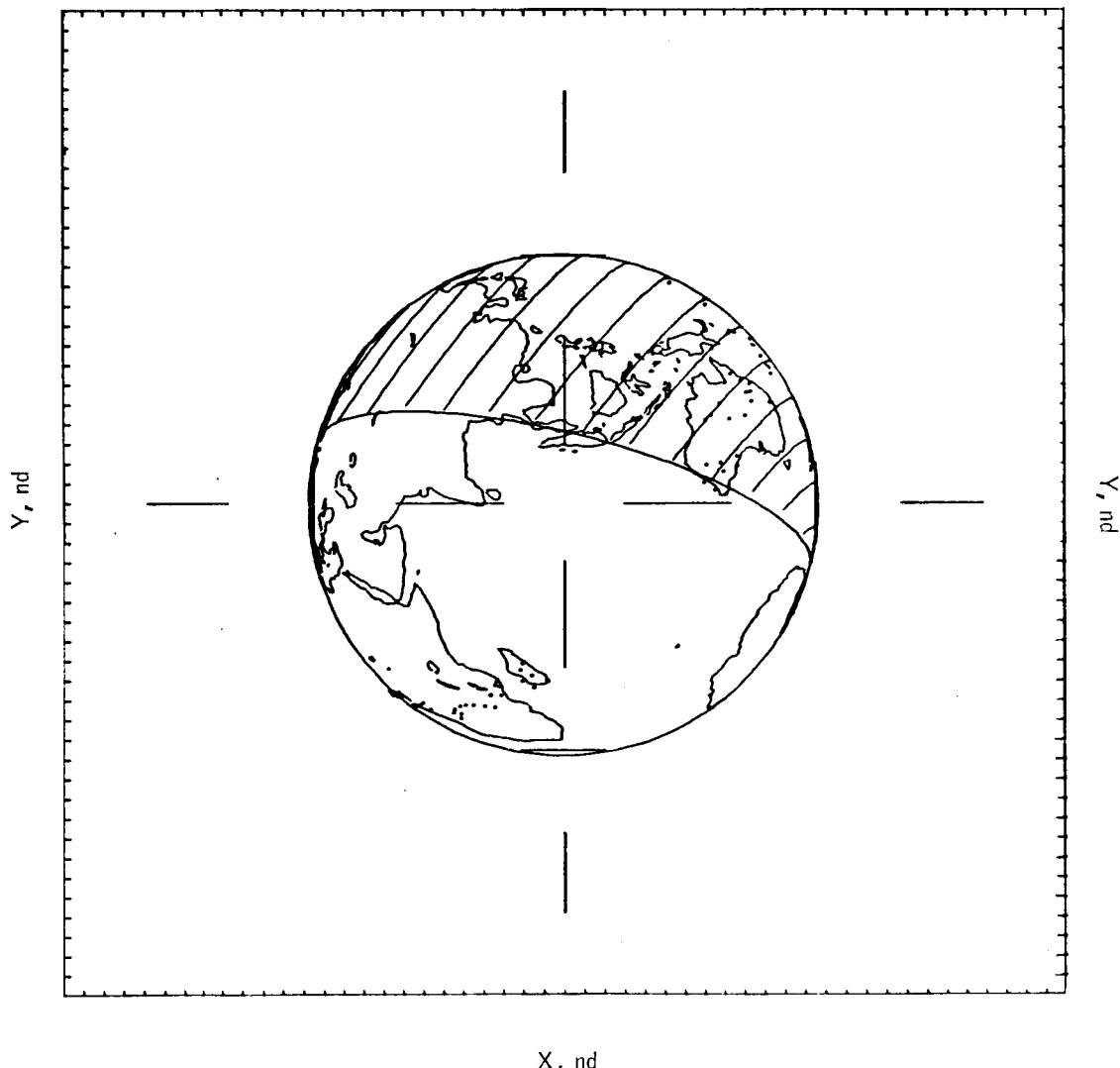


(b) Time from TEI cutoff = 2 hr.

Figure 11. - continued .

Field of view = 4 deg

$$R_M = 16\ 156 \text{ n. mi.}$$
$$V_i = 4\ 990 \text{ fps}$$



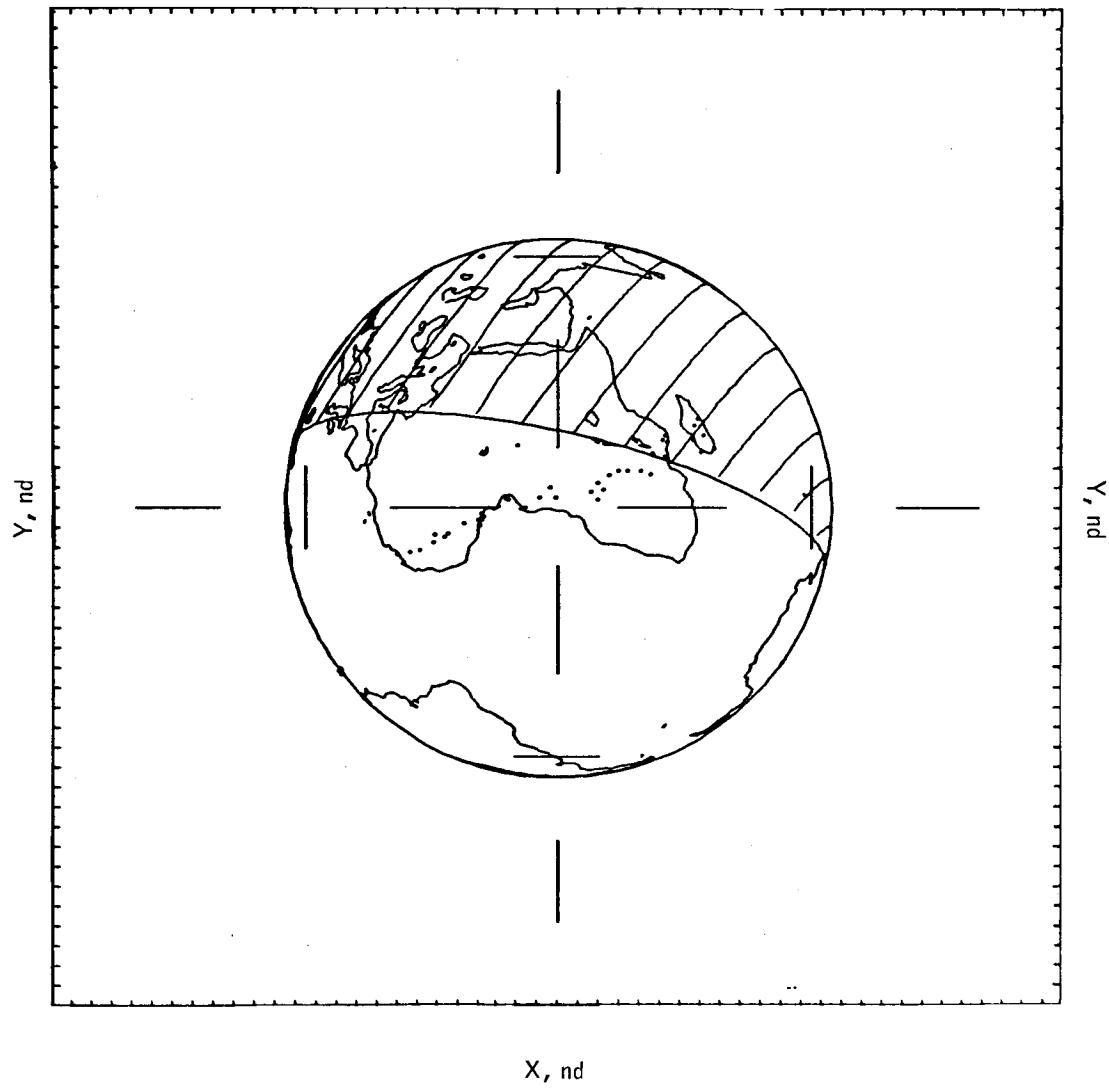
(c) Time from TEI cutoff = 5 hr.

Figure 11. - continued.

Field of view = 4 deg

$$R_M = 30\,610 \text{ n. mi.}$$

$$V_i = 4\,841 \text{ fps}$$



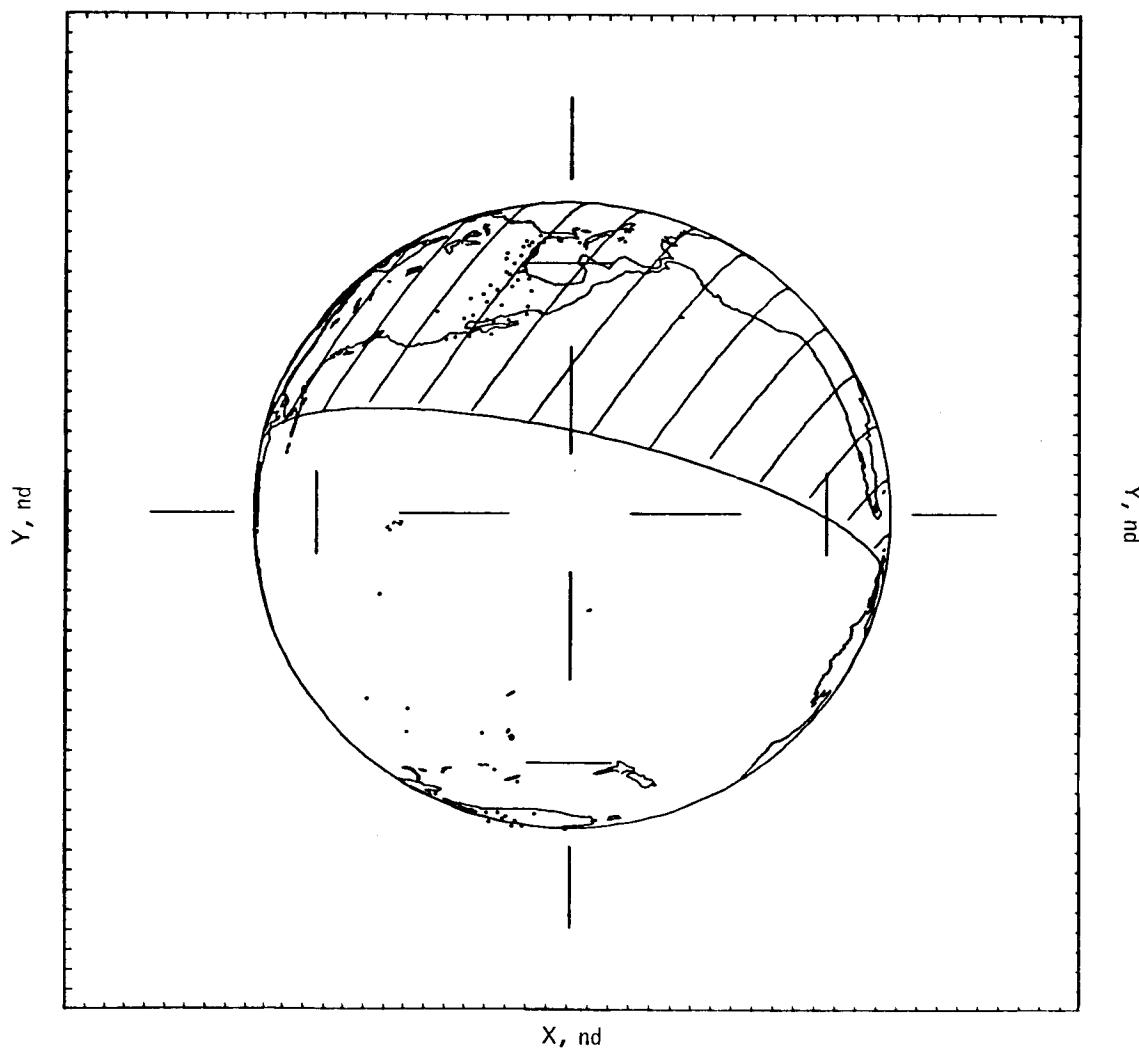
(d) Time from TEI cutoff = 10 hr.

Figure 11. - continued.

Field of view = 4 deg

$$R_E = 157\ 309 \text{ n. mi.}$$

$$V_i = 4\ 532 \text{ fps}$$

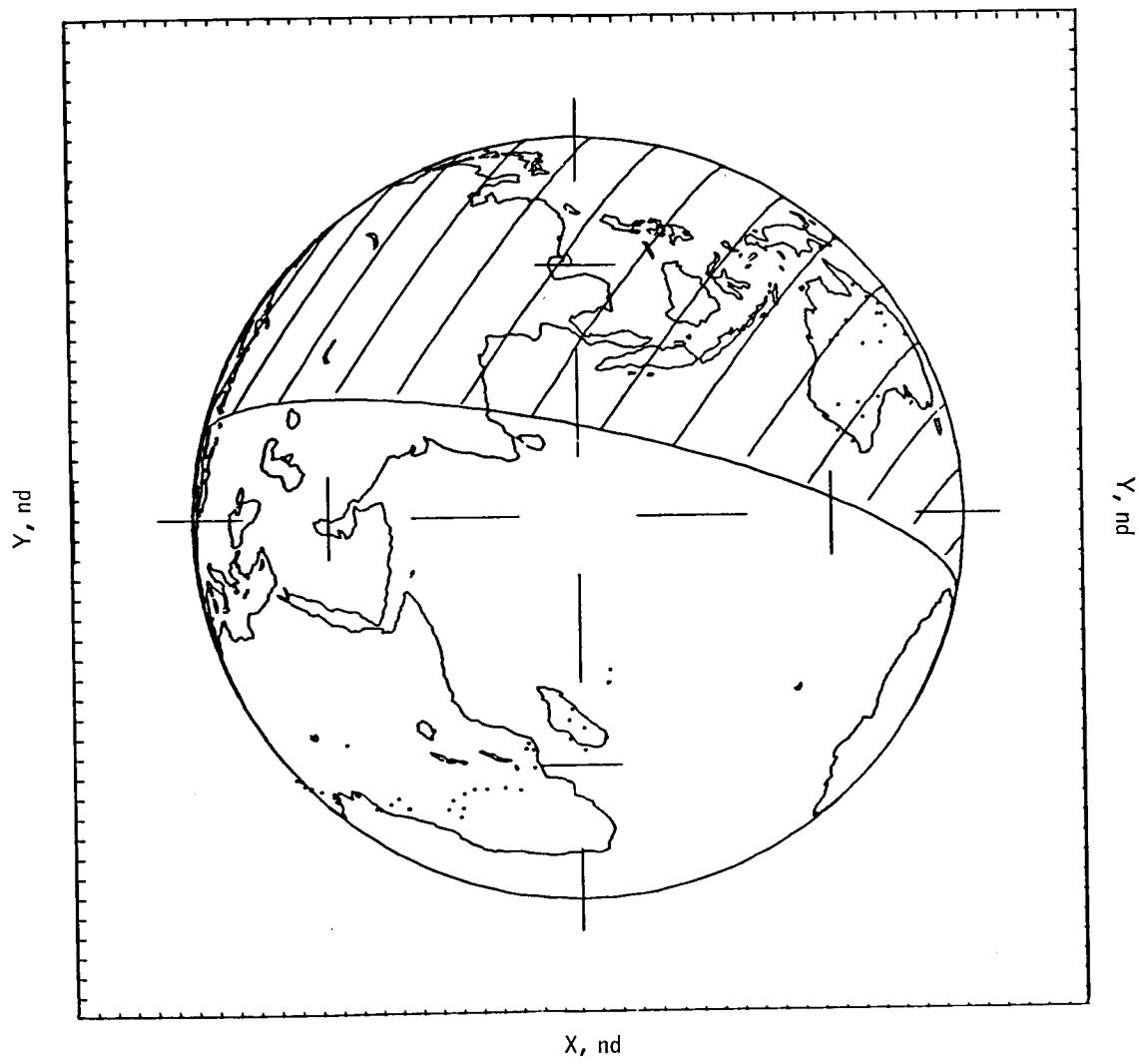


(e) Time from TEI cutoff = 20 hr.

Figure 11.- Continued.

$$R_E = 129\ 159 \text{ n. mi.}$$
$$V_i = 5\ 175 \text{ fps}$$

Field of view 4 deg

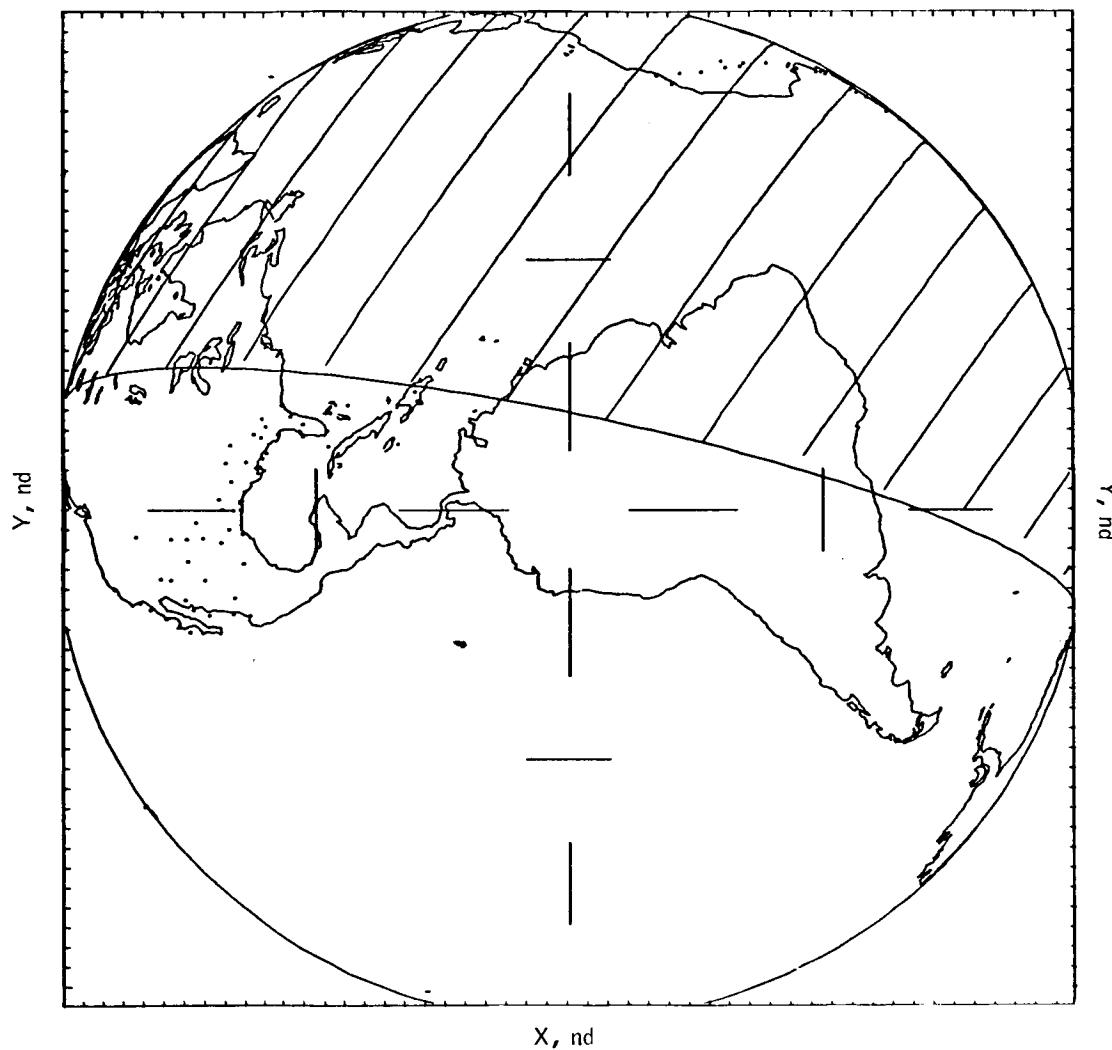


(f) Time from TEI cutoff = 30 hr.

Figure 11.- Continued.

$$R_E = 96\ 323 \text{ n. mi.}$$
$$V_i = 6\ 238 \text{ fps}$$

Field of view = 4 deg



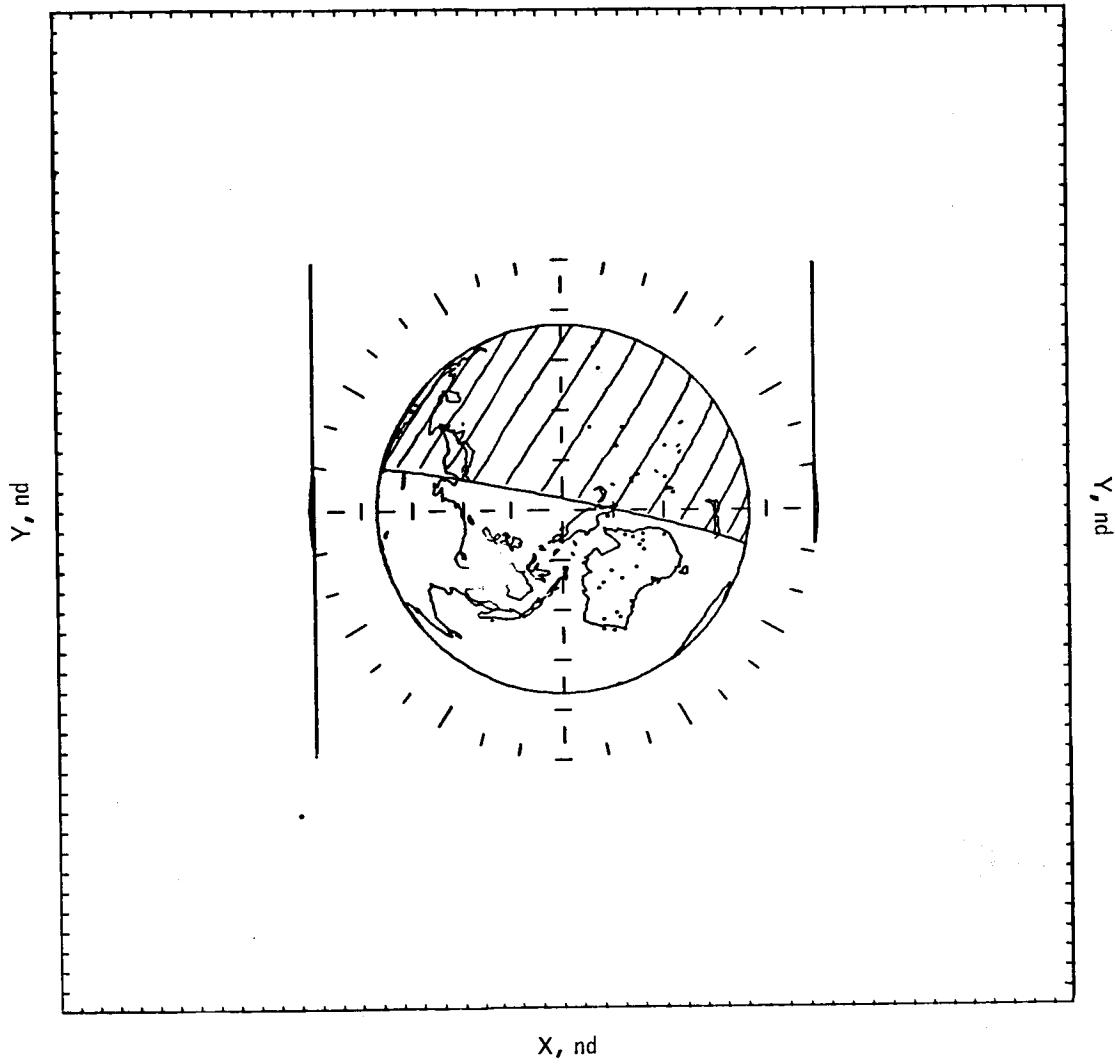
(g) Time from TEI cutoff = 40 hr.

Figure 11.- Continued.

$$R_E = 54\,624 \text{ n. mi.}$$

$$V_i = 8\,694 \text{ fps}$$

Field of view = 20 deg

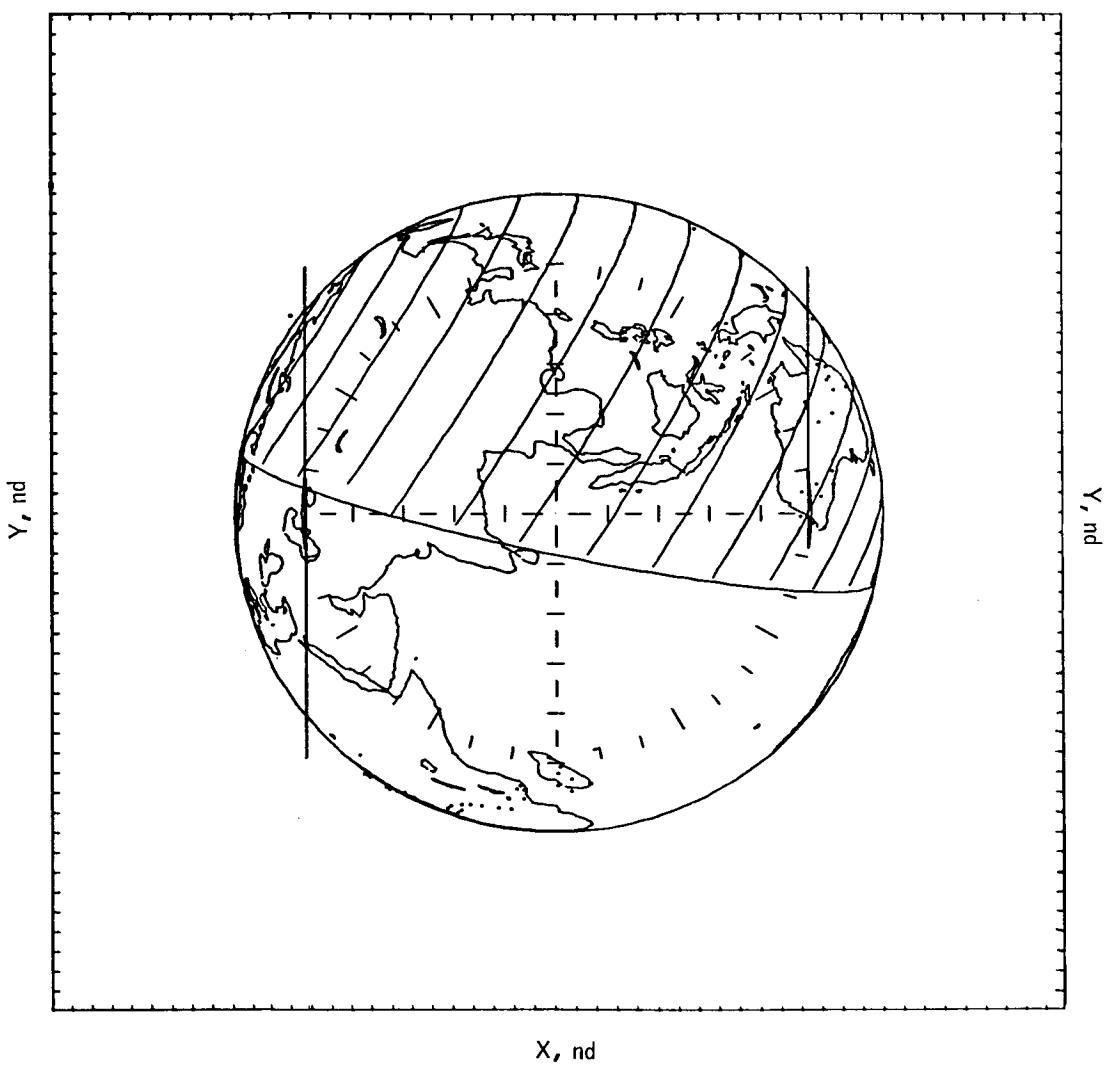


(h) Time from TEI cutoff = 50 hr.

Figure 11.- Continued.

Field of view = 20 deg

$$R_E = 32\ 229 \text{ n. mi.}$$
$$V_i = 11\ 598 \text{ fps}$$



(i) Time from TEI cutoff = 54 hr.

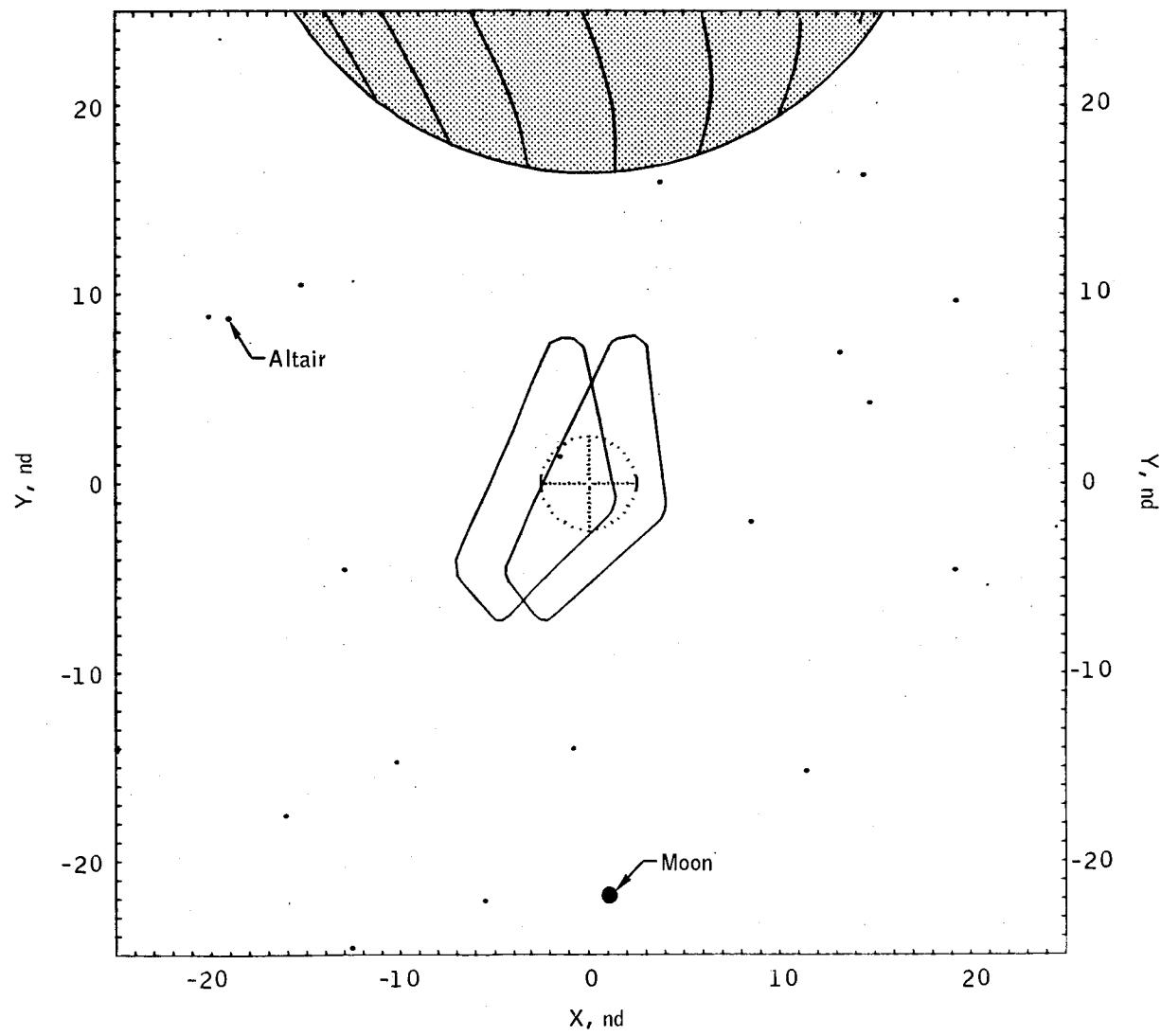
Figure 11.- Concluded.

SEQ	4	7	15	25	904	907	909	933	984	990	1001	1010	1028	1041	1044
X	-12	-5	19	11	-20	-19	-15	14	-13	-1	13	19	14	8	-16
Y	-24	-21	-4	-15	8	8	10	16	-4	1	7	9	4	-1	-17

1046
-10
-14

$$R_E = 5761 \text{ n. mi.}$$

$$V_i = 28197 \text{ fps}$$



(a) 17 minutes prior to entry.

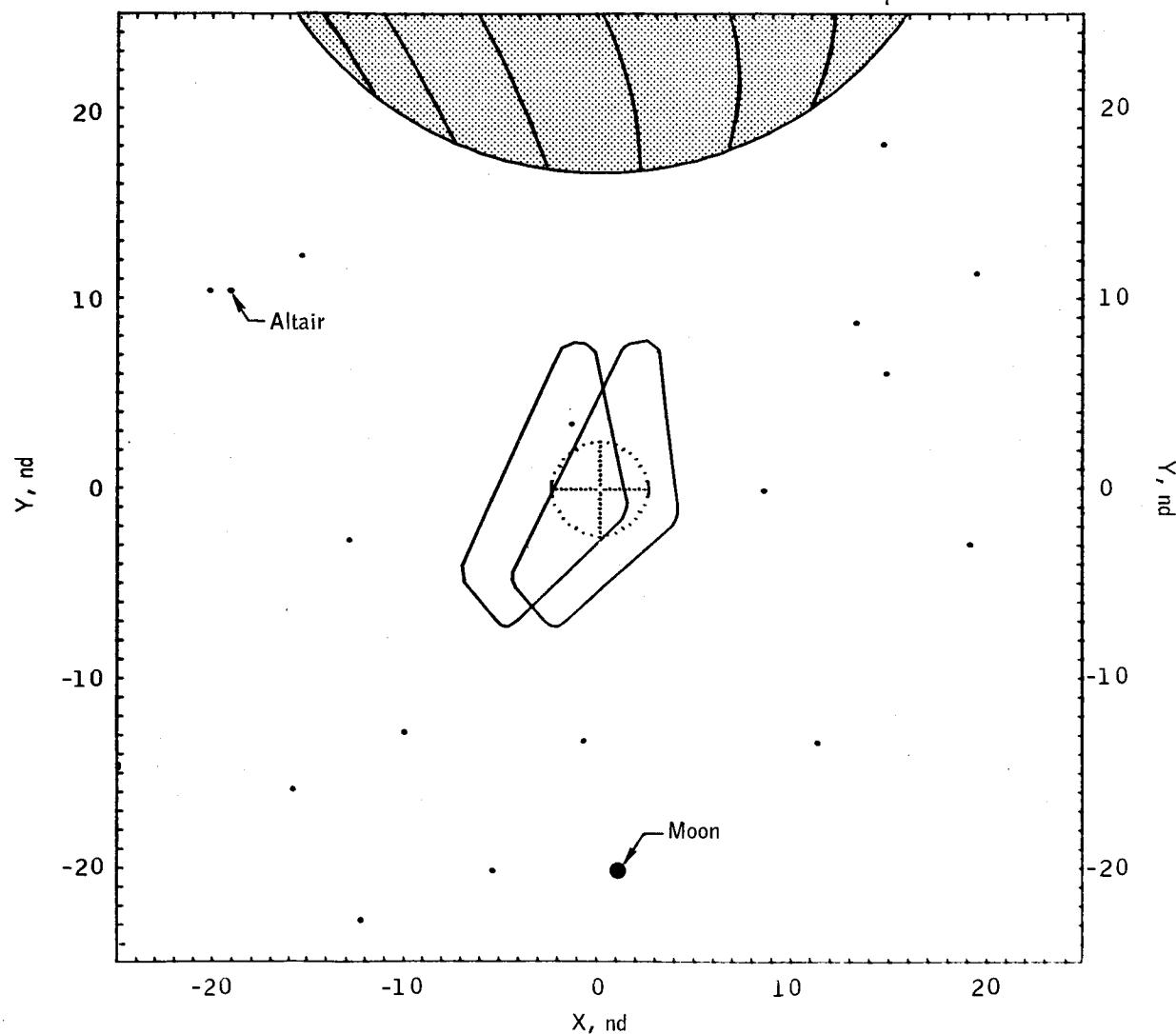
Figure 12.- Entry phase.

SEQ	4	7	15	25	904	907	909	933	984	990	1001	1010	1028	1041	1044
X	-12	-5	19	11	-20	-19	-15	14	-13	-1	13	19	14	8	-15
Y	-22	-19	-2	-13	10	10	12	18	-2	3	8	11	6	0	-15

1046
-10
-12

$$R_E = 5586 \text{ n. mi.}$$

$$V_i = 28639 \text{ fps}$$



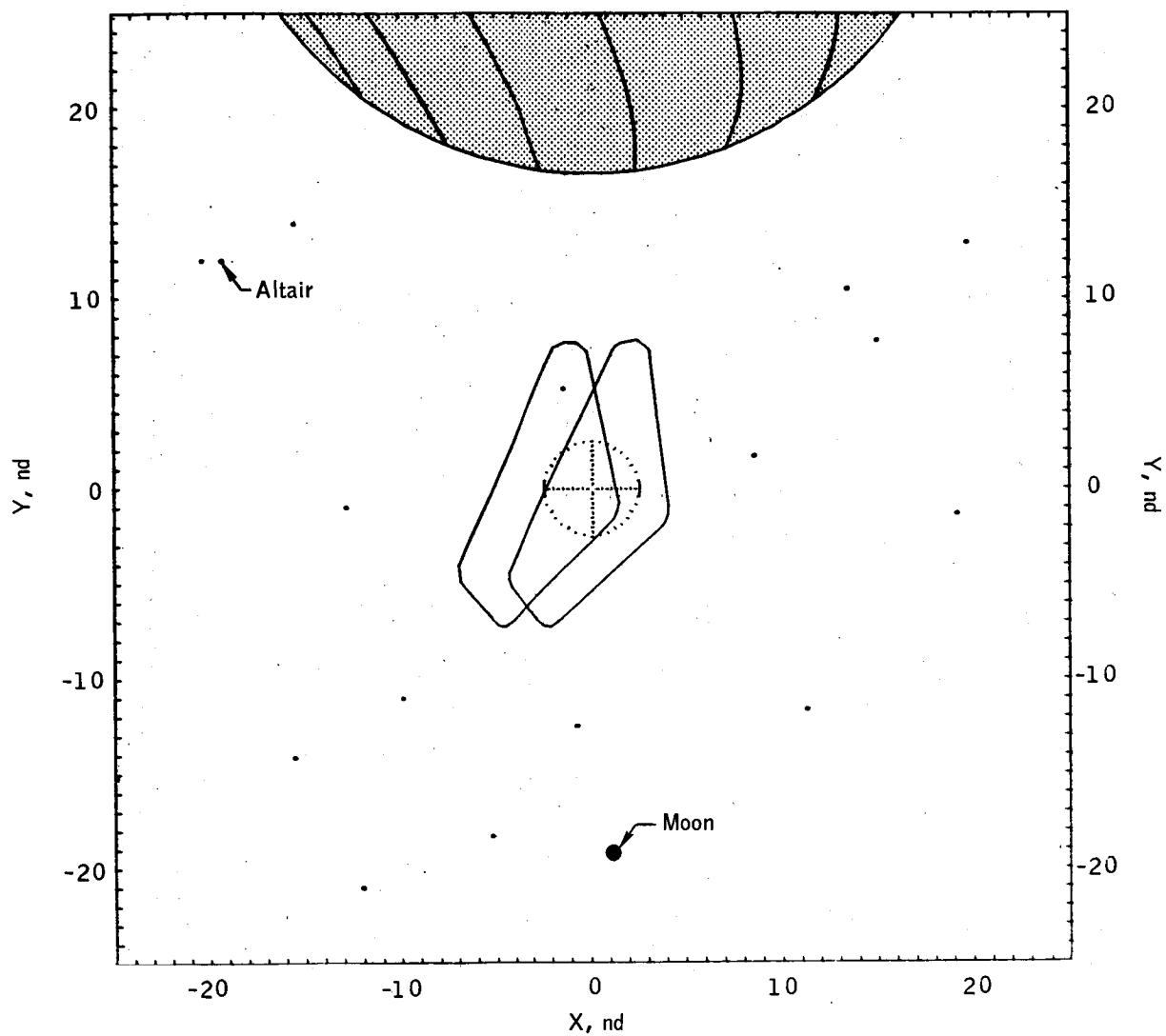
(b) 16 minutes prior to entry.

Figure 12.- Continued.

$$R_E = 5414 \text{ n. mi.}$$

$$V_i = 29098 \text{ fps}$$

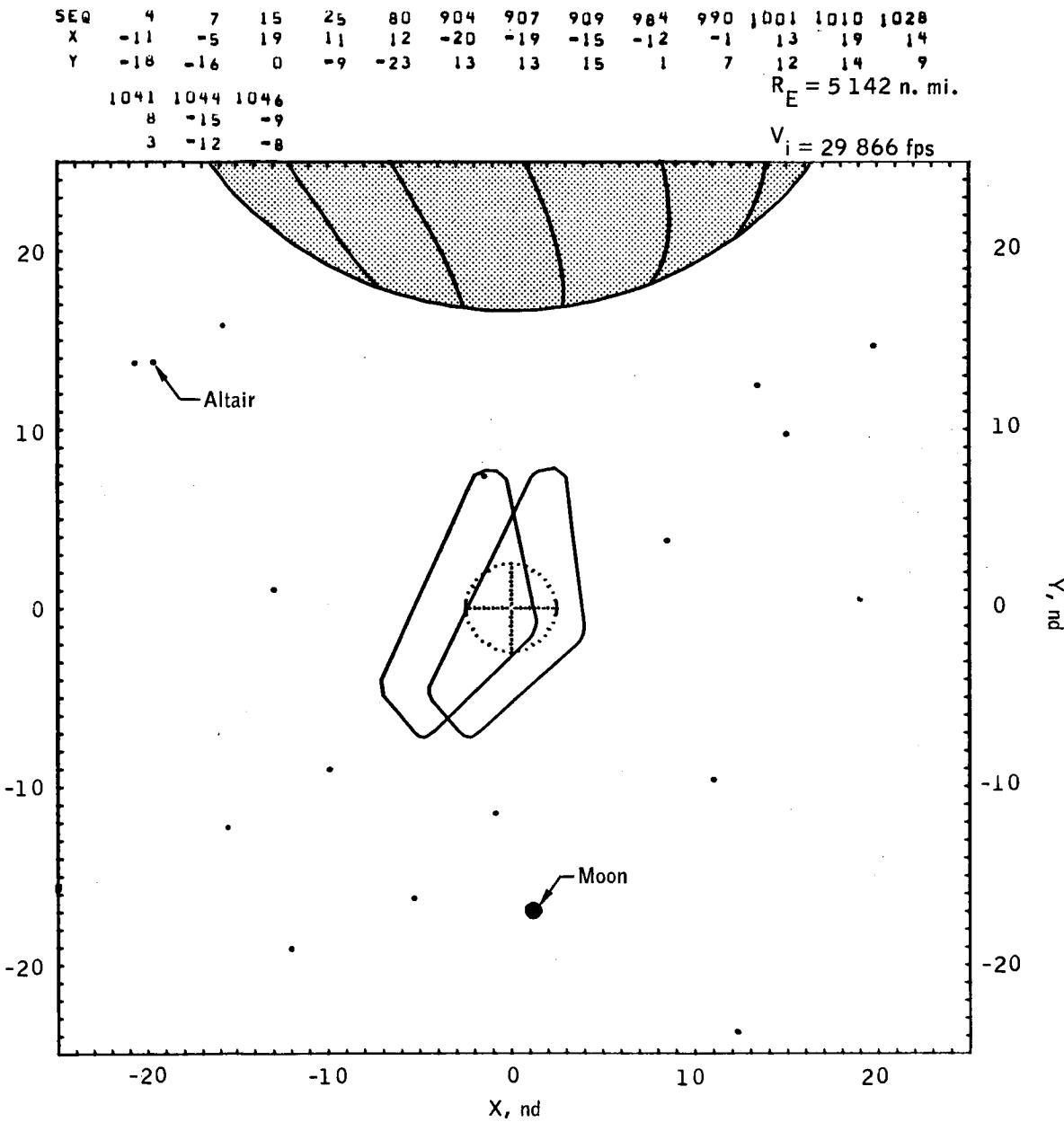
SE ₀	4	7	15	25	904	907	909	984	990	1001	1010	1028	1041	1044	1046
X	-12	-5	19	11	-20	-19	-15	-12	-1	13	19	14	8	-15	-10
Y	-20	-18	-1	-11	12	12	14	0	5	10	13	7	1	-13	-10



(c) 15 minutes prior to entry.

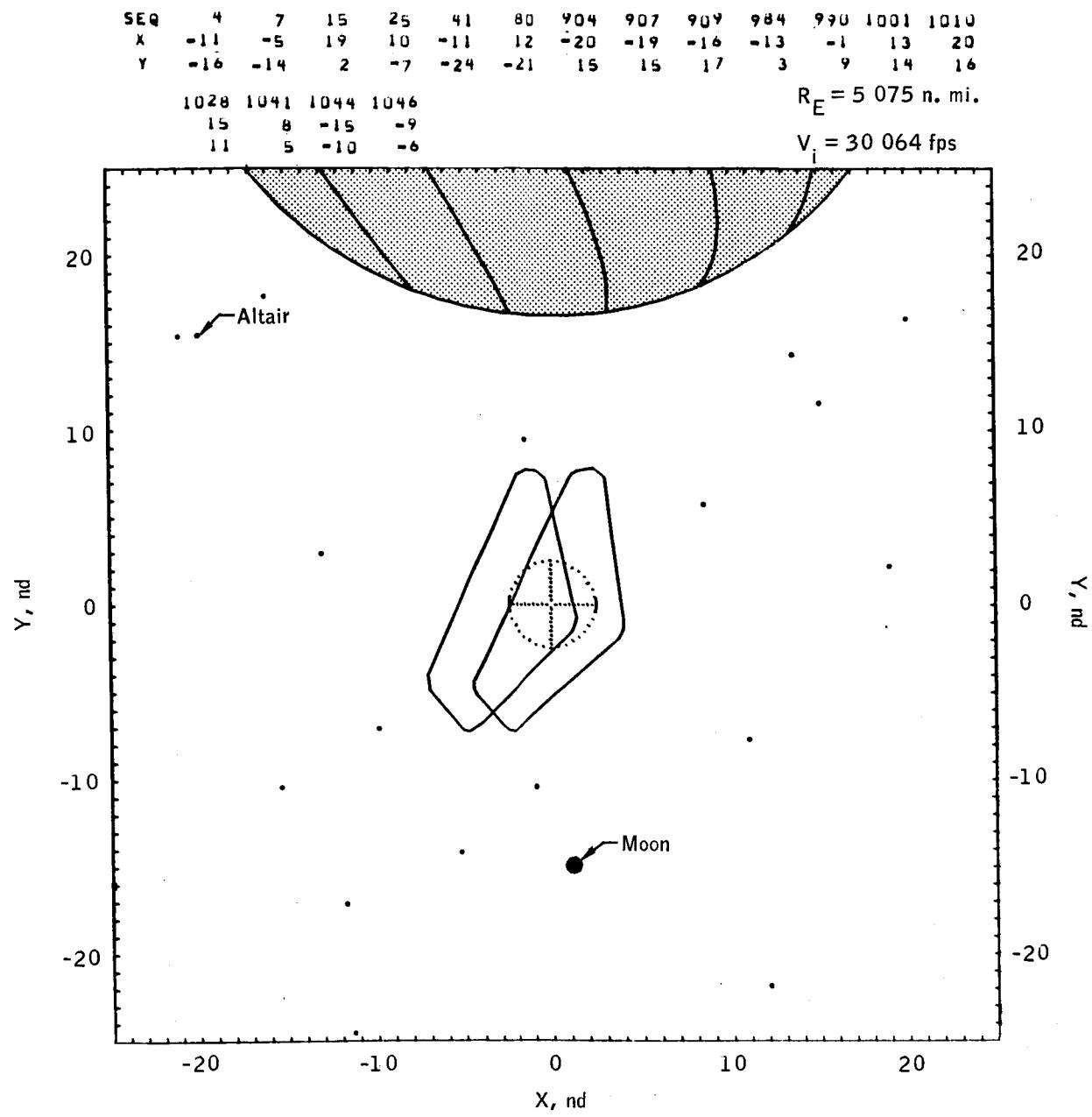
Figure 12.- Continued.

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(d) 14 minutes prior to entry.

Figure 12.- Continued.



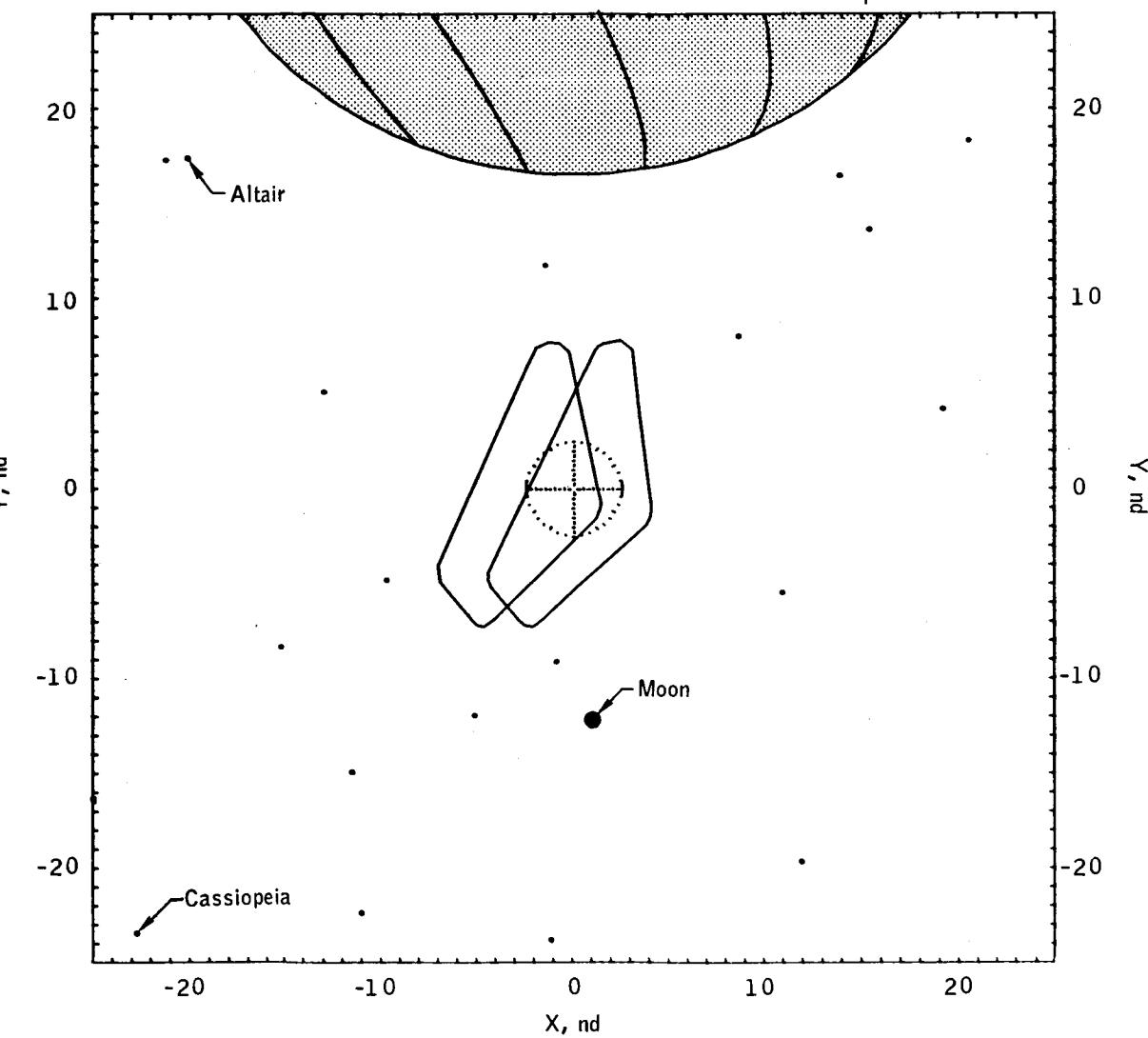
(e) 13 minutes prior to entry.

Figure 12. - Continued.

SEQ	4	7	15	22	25	41	63	80	904	907	984	990	1001	1010	1028
X	-11	-5	19	-22	10	-11	-1	11	-21	-20	-13	-1	13	20	15
Y	-14	-11	4	-23	-5	-22	-23	-19	17	17	5	11	16	18	13

$$R_E = 4910 \text{ n. mi.}$$

$$V_i = 30570 \text{ fps}$$



(f) 12 minutes prior to entry.

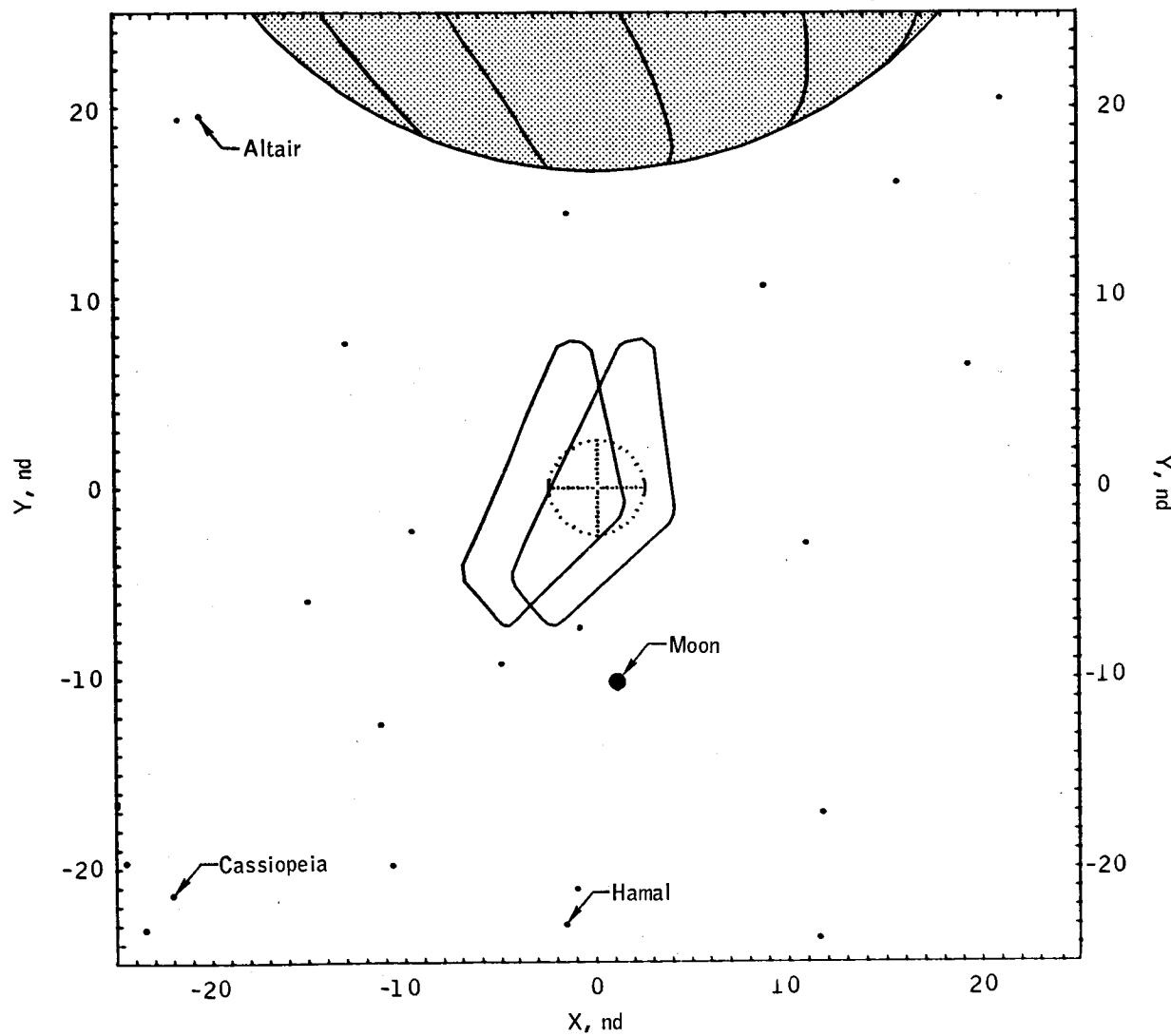
Figure 12.- Continued.

SEG	4	5	7	15	22	25	31	41	63	75	80	108	904	907	984
X	-11	-24	-5	19	-22	10	-23	-10	-1	-1	11	11	-21	-20	-13
Y	-12	-19	-9	6	-21	-2	-23	-19	-20	-22	-16	-23	19	19	7

990	1010	1028	1041	1044	1046
-1	20	15	8	-15	-9
14	20	16	10	-5	-2

$$R_E = 4748 \text{ n. mi.}$$

$$V_i = 31090 \text{ fps}$$



(g) 11 minutes prior to entry.

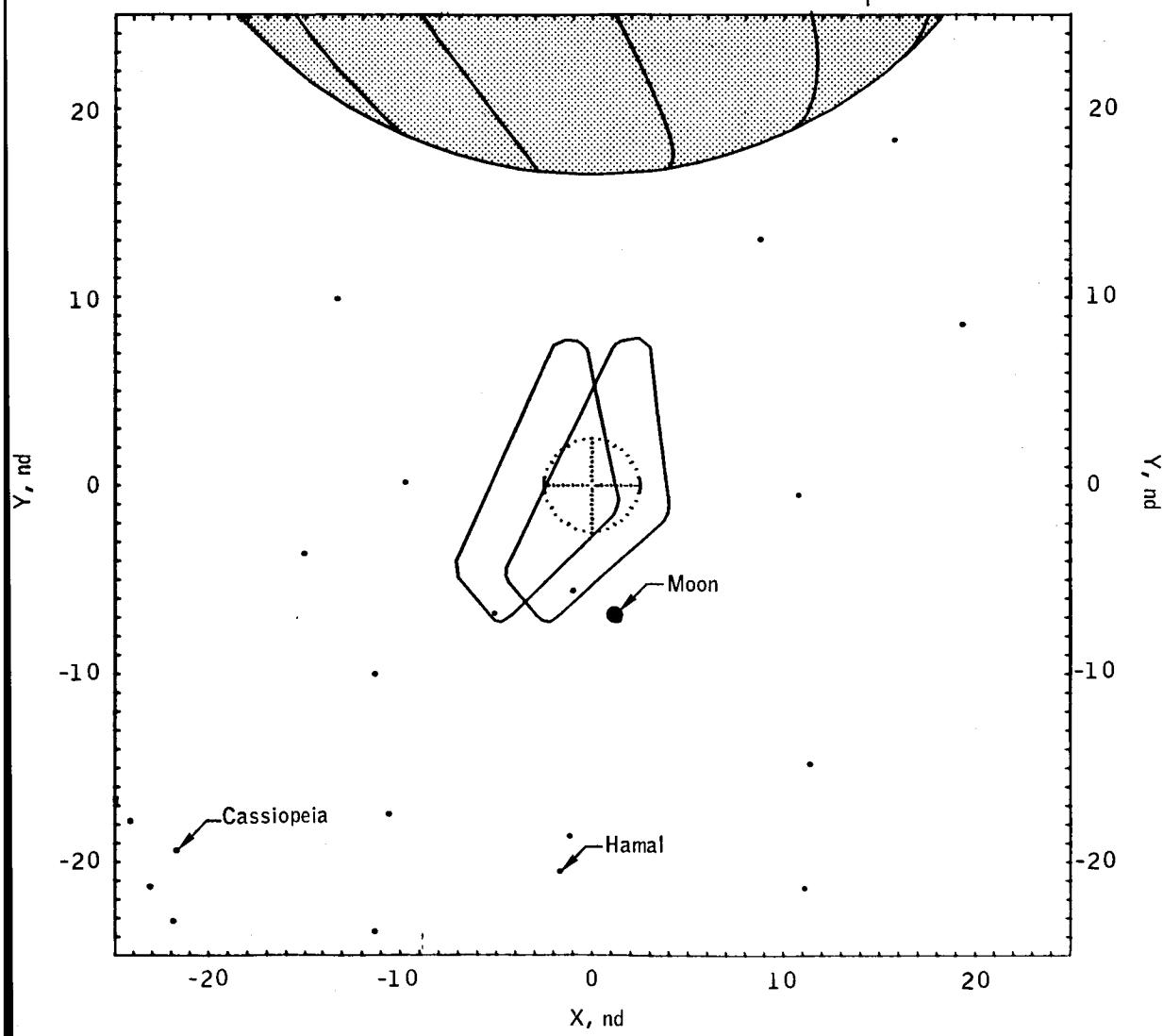
Figure 12.- Continued.

SEG	4	5	7	15	22	25	31	41	47	63	73	75	86	108	984
X	-11	-24	-5	19	-21	10	-23	-10	-21	-1	-11	-1	11	11	-13
Y	-9	-17	-6	8	-19	0	-21	-17	-23	-18	-20	-20	-14	-21	10

1028	1041	1044	1046
15	8	-15	-9
18	13	-3	0

$$R_E = 4591 \text{ n. mi.}$$

$$V_i = 31622 \text{ fps}$$



(h) 10 minutes prior to entry.

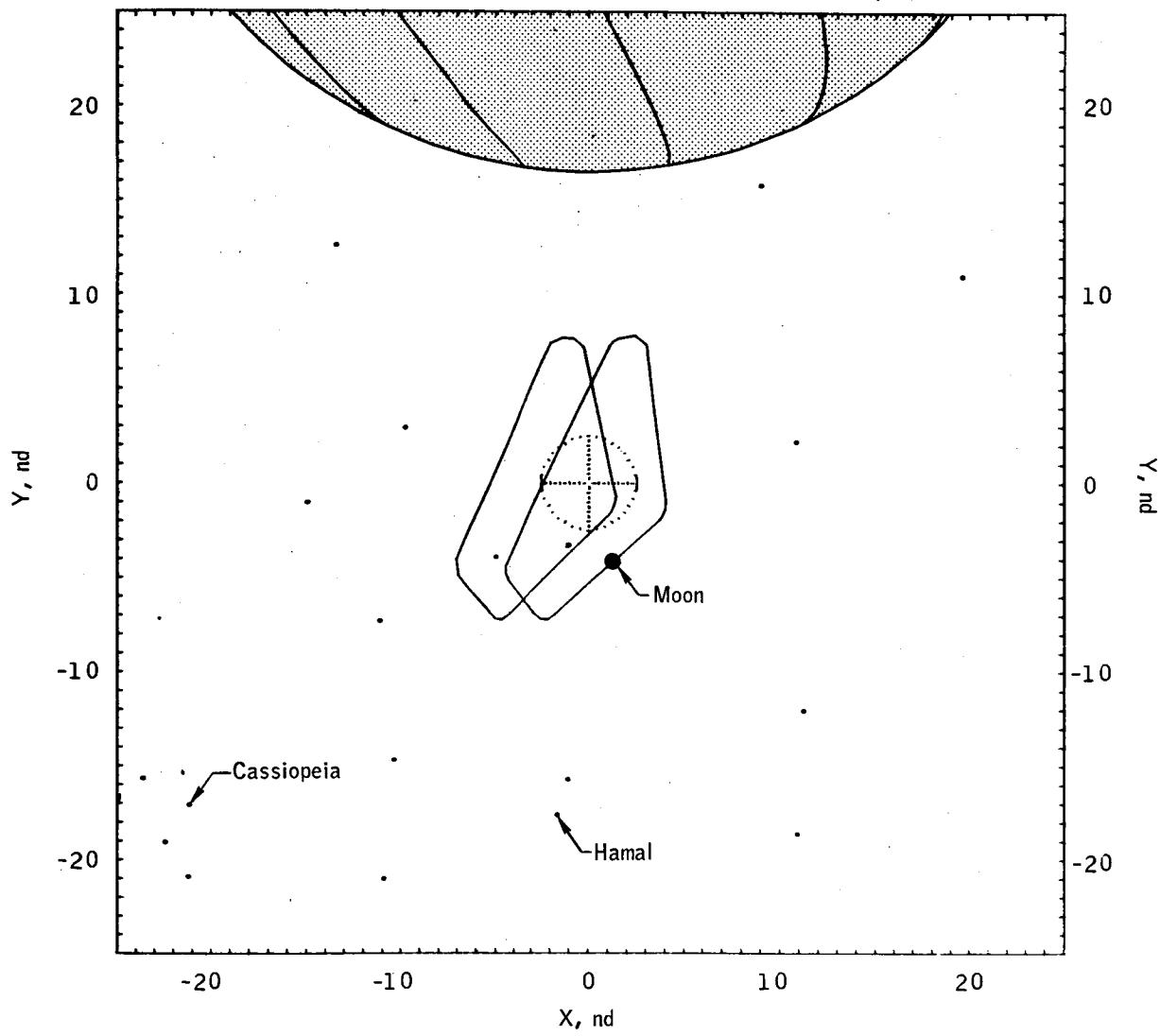
Figure 12.- Continued.

SEQ	4	5	7	15	22	25	31	41	47	63	73	75	80	108	984
X	-11	-23	-5	19	-21	10	-22	-10	-21	-1	-10	-1	11	10	-13
Y	-7	-15	-3	11	-16	2	-18	-14	-20	-15	-20	-17	-11	-18	12

1041 1044 1046
8 -14 -9
15 0 3

$$R_E = 4440 \text{ n. mi.}$$

$$V_i = 32161 \text{ fps}$$

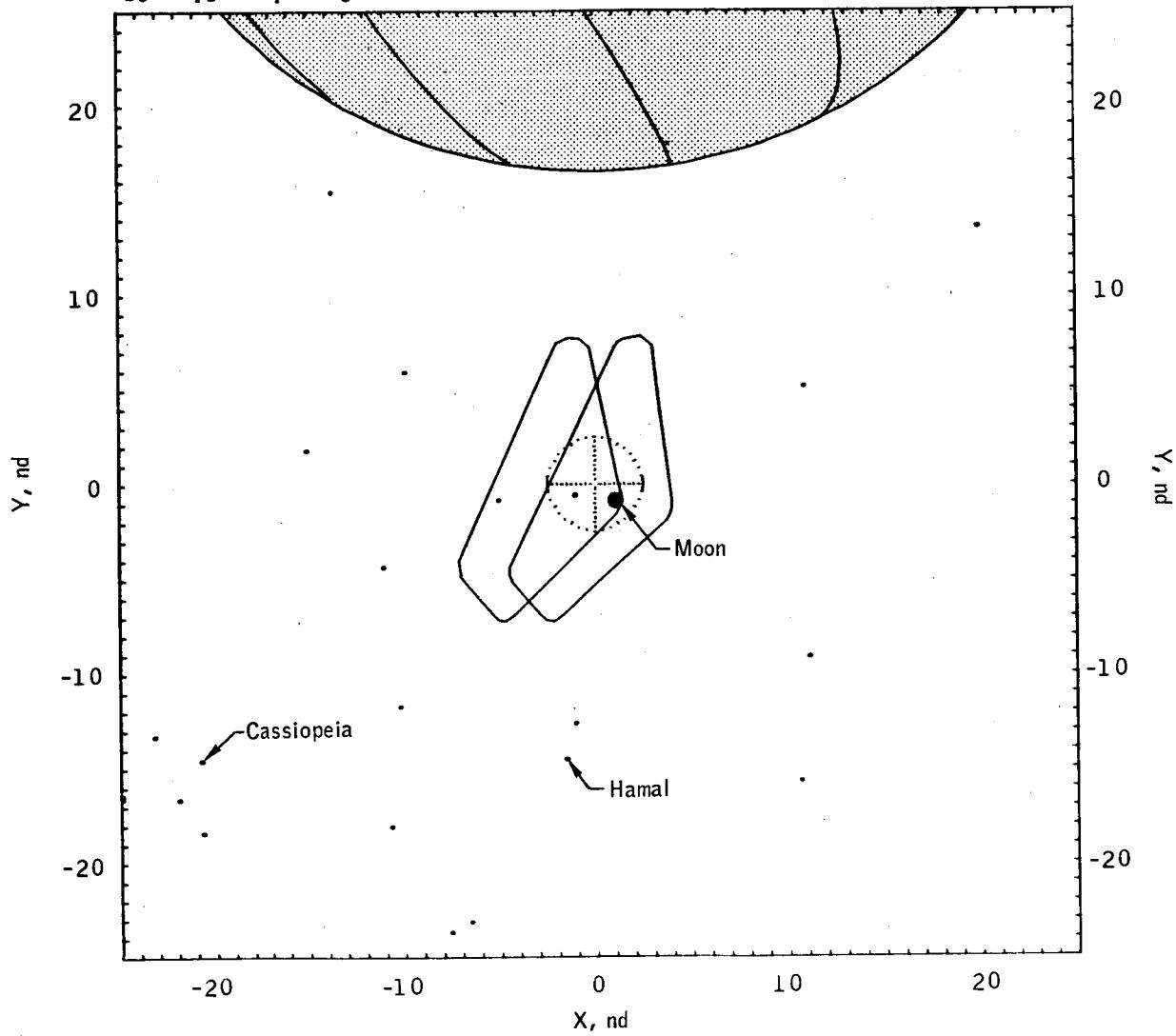


(i) 9 minutes prior to entry.

Figure 12.- Continued.

SEQ	4	5	7	15	22	25	31	41	47	63	73	75	80	108	111
X	-11	+23	-5	19	-20	10	-21	-10	-20	-1	-10	+1	11	10	-6
Y	-4	-13	0	13	-14	5	-16	-11	-18	-12	-17	-14	-8	+15	-22

$R_E = 4295$ n. mi.
 $V_i = 32705$ fps

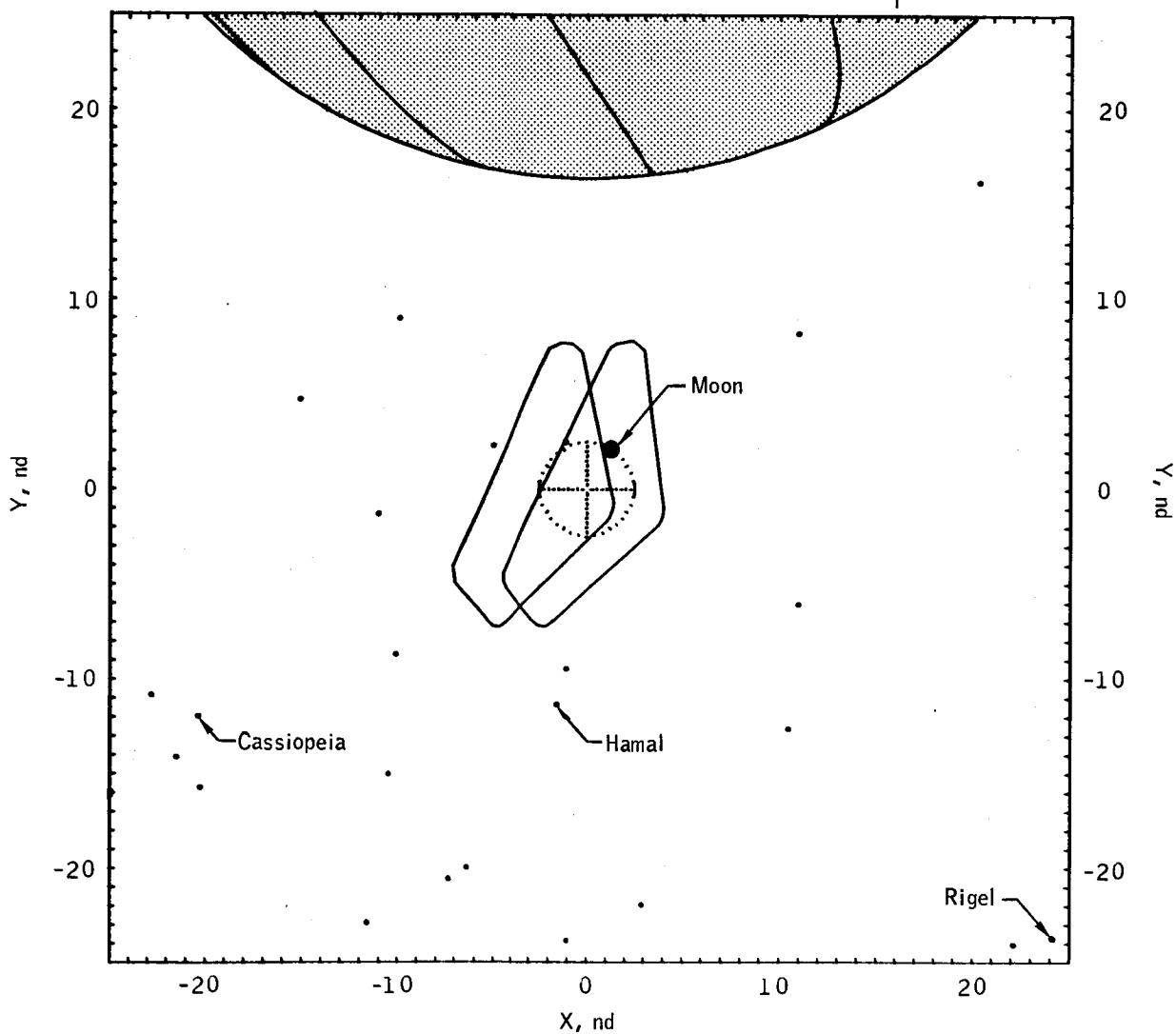


(j) 8 minutes prior to entry.

Figure 12.- Continued.

SEQ	4	5	7	15	22	25	31	41	47	63	73	75	80
X	-11	-22	-5	20	-20	10	-21	-10	-20	-1	-10	-1	10
Y	-1	-10	2	16	-11	8	-13	-8	-15	-9	-14	-11	-5
	108	111	112	120	144	150	215	221	1044	1046	$R_E = 4157$ n. mi.		
	10	-6	-7	-11	2	-1	22	24	-15	-9			
	-12	-19	-20	-22	-21	-23	-23	-23	4	9			

$V_i = 33248$ fps



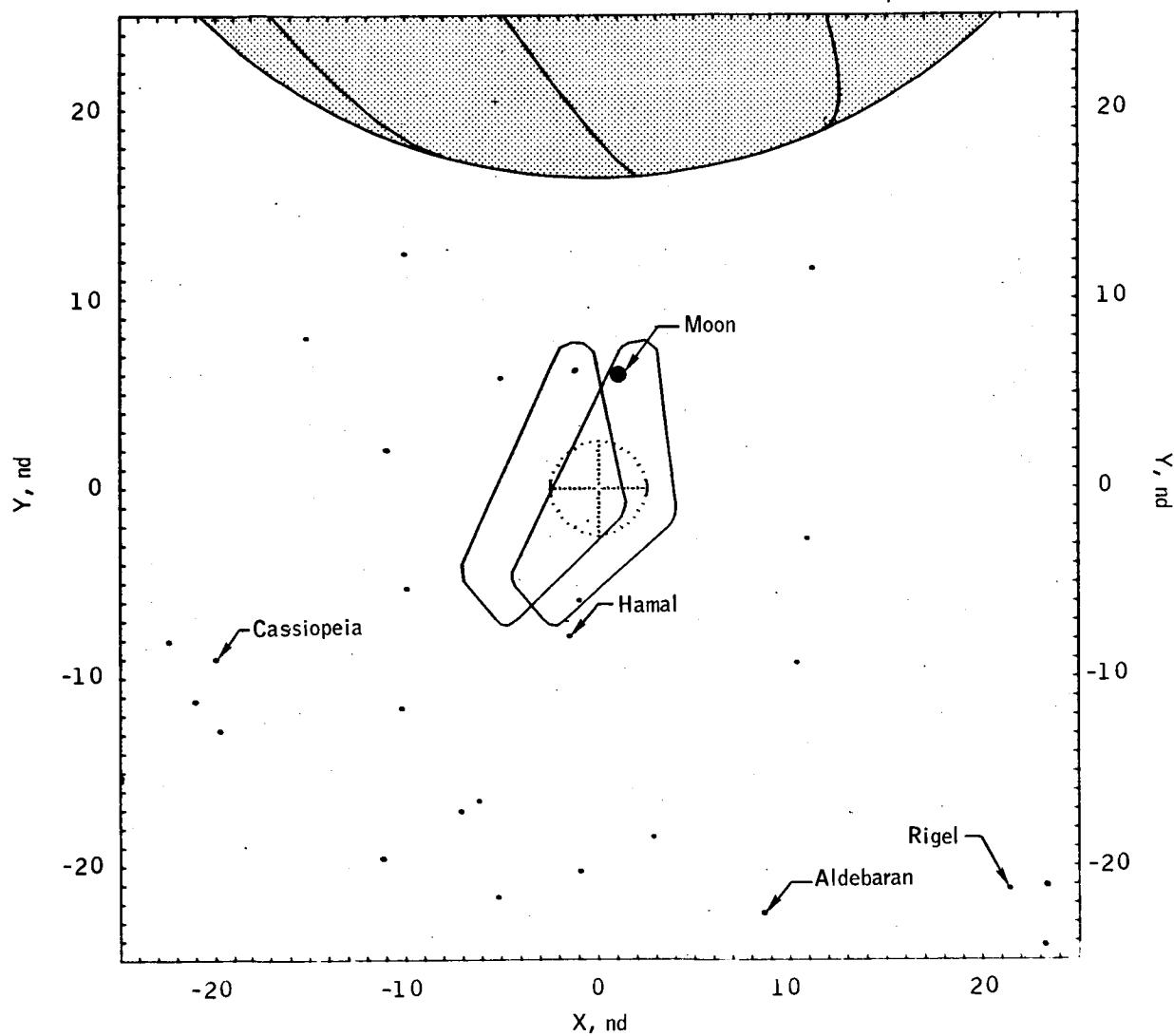
(k) 7 minutes prior to entry.

Figure 12.- Continued.

SEQ	4	5	7	22	25	31	41	47	63	73	75	80	108	111	112
X	-11	-22	-5	-20	11	-21	-9	-19	-1	-10	-1	10	10	-6	-7
Y	2	-7	5	-8	11	-11	-5	-12	-5	-11	-1	-2	-9	-16	-16

SEQ	120	144	150	151	186	215	221	245	1044	1046
X	-11	2	0	-5	8	21	23	23	-15	-10
Y	-19	-18	-20	-21	-22	-21	-20	-24	8	12

$R_E = 4028$ n. mi.
 $V_i = 33782$ fps



(II) 6 minutes prior to entry.

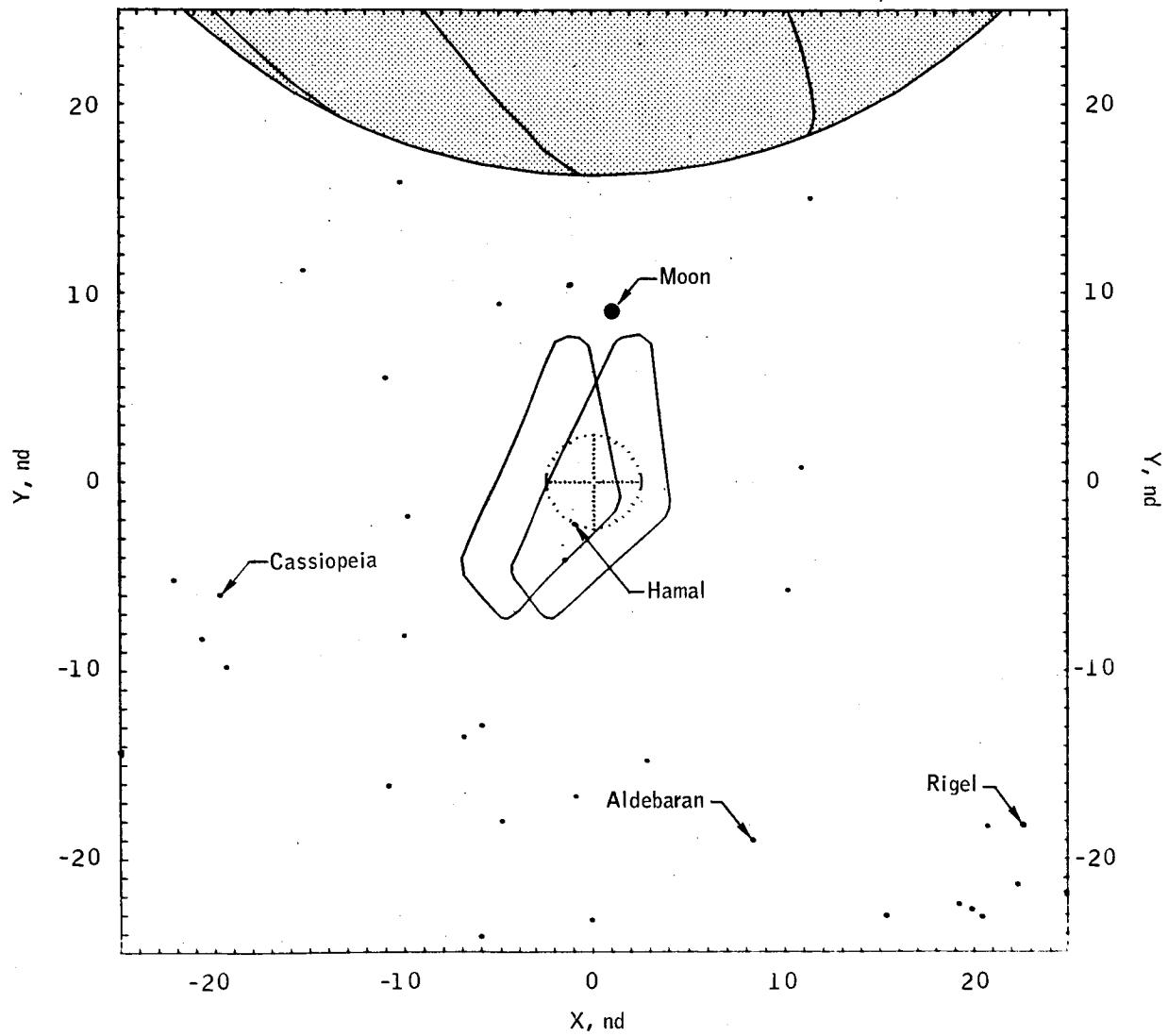
Figure 12.- Continued.

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SEW	4	5	7	22	25	31	41	47	63	73	75	80	108	111	112
X	-11	-22	-5	-19	11	-20	-9	-19	-1	-10	-1	10	10	-6	-6
Y	5	-5	9	-5	15	-8	-1	-9	-2	-7	-3	0	-5	-12	-13
SEW	120	144	150	151	186	205	207	215	221	230	237	245	246	252	256
X	-10	2	0	-5	8	0	-6	20	22	15	19	22	19	20	24
Y	-15	-14	-16	-17	-18	-23	-24	-18	-18	-22	-22	-21	-22	-22	-21
1044	1046														
	-15	-10													
	11	16													

$$R_E = 3909 \text{ n. mi.}$$

$$V_i = 34298 \text{ fpm}$$



(m) 5 minutes prior to entry.

Figure 12.- Continued.

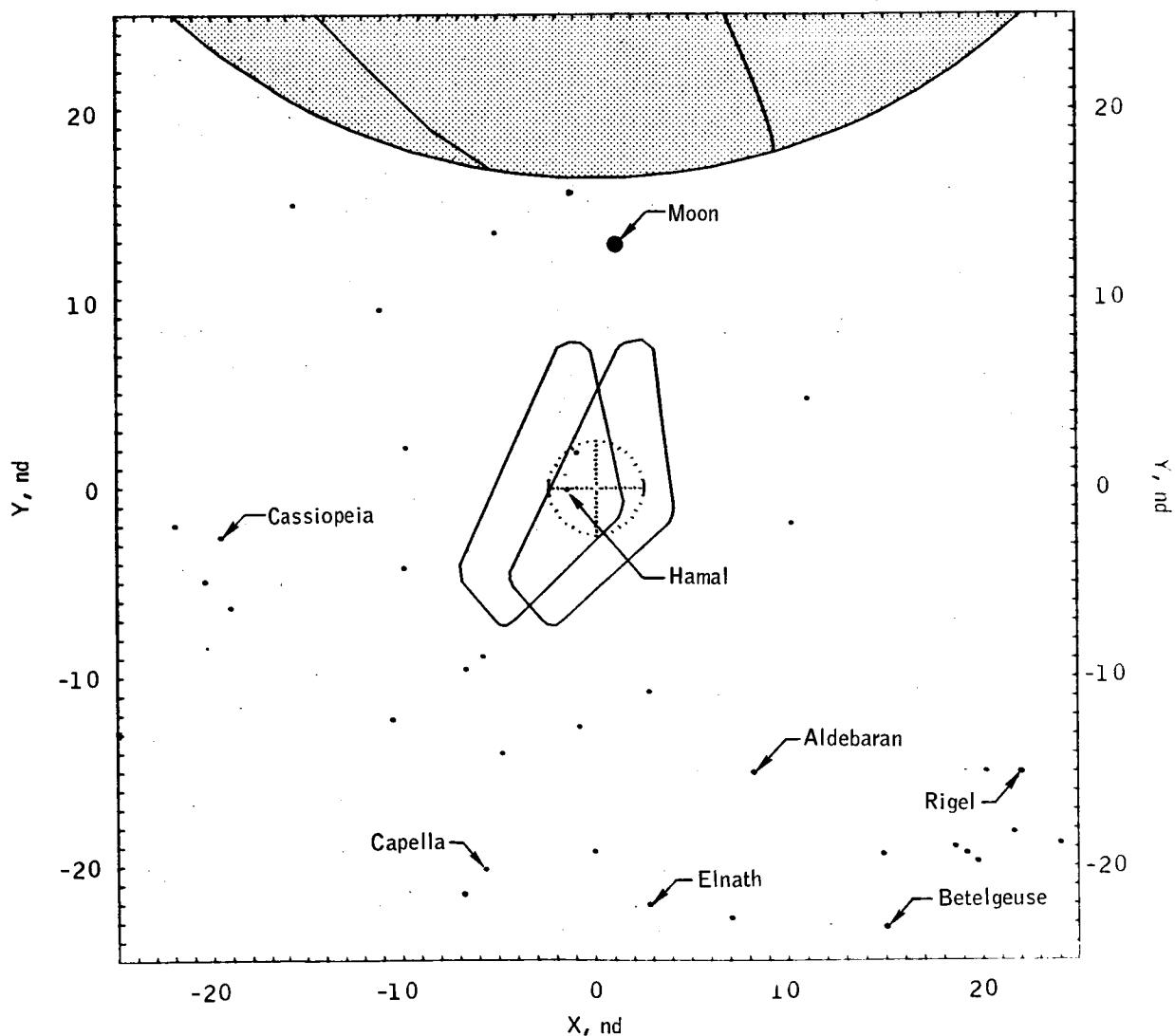
SEQ	4	5	7	22	31	41	47	63	73	75	80	108	111	112	120
X	-11	-22	-5	-19	-20	-9	-19	-1	-10	-1	10	10	-5	-6	-10
Y	9	-1	13	-2	-4	2	-6	2	-4	0	4	-1	-8	-9	-12

SEQ	151	186	205	207	215	221	222	230	231	237	245	246	248	252	256
X	-4	8	0	-5	20	21	-6	14	2	18	21	19	6	19	23
Y	-13	-14	-19	-19	-14	-14	-21	-19	-21	-18	-18	-19	-22	-19	-18

265	1044	144	150
14	-15	2	0
-23	15	-10	-12

$$R_E = 3800 \text{ n. mi.}$$

$$V_i = 34787 \text{ fps}$$



(n) 4 minutes prior to entry.

Figure 12.- Continued.

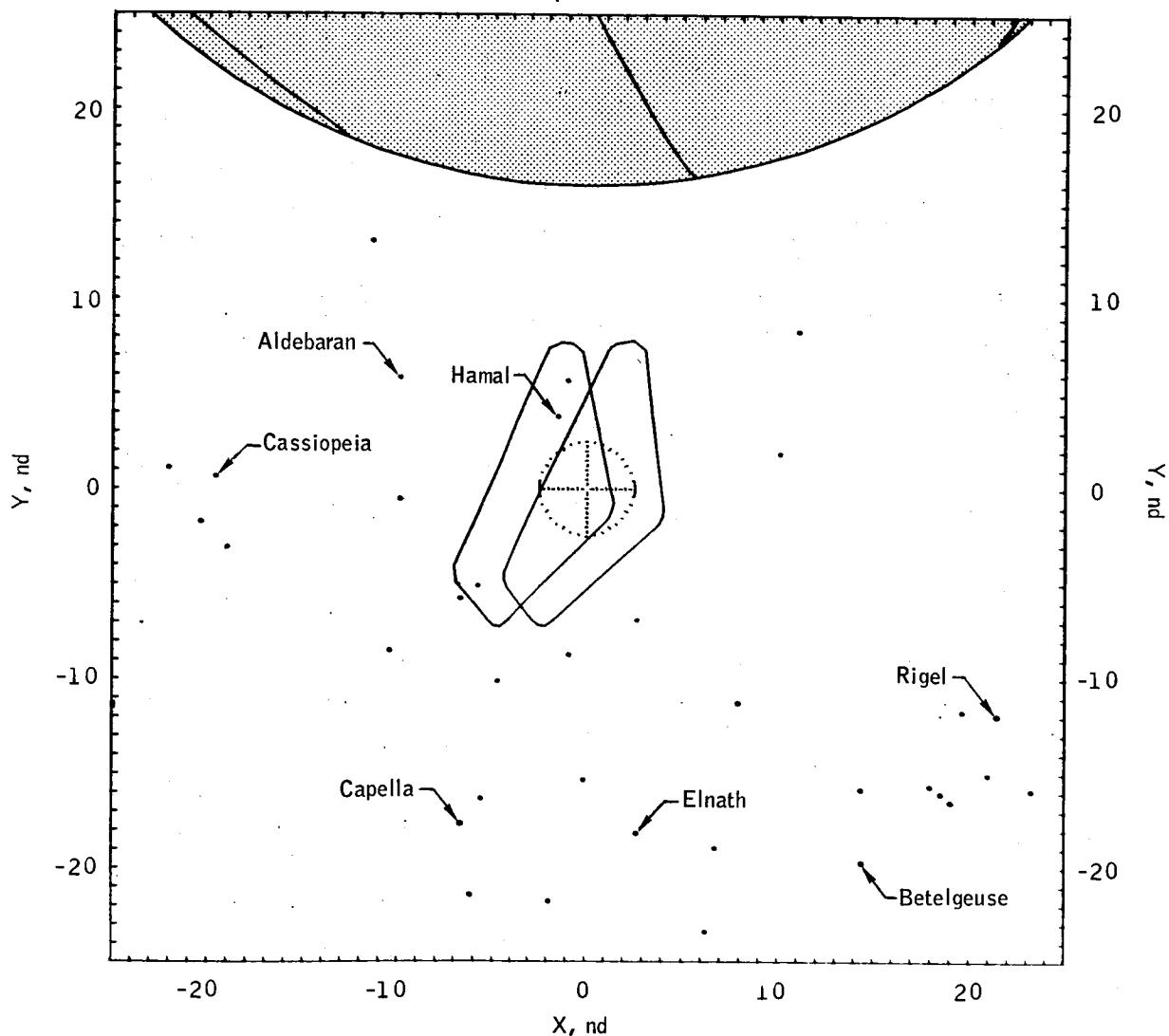
SEQ	4	5	22	31	41	47	63	73	75	80	108	111	112	120	144
X	-11	-22	-19	-20	-9	-19	-1	-9	-1	11	10	-5	-6	-10	2
Y	13	1	0	+1	6	-2	5	0	3	8	2	-4	-5	-8	-6

SEQ	150	161	186	205	207	215	221	222	230	231	237	245	246	248	262
X	0	-4	7	0	-5	-14	21	-6	-14	2	17	20	18	6	19
Y	-8	-9	-11	-15	-16	-11	-11	-17	-15	-18	-15	-14	-15	-18	-16

256	266	270	271	281
23	14	-6	-1	6
-15	-19	-21	-21	-23

$$R_E = 3705 \text{ n. mi.}$$

$$V_i = 35236 \text{ fpm}$$



(o) 3 minutes prior to entry.

Figure 12.- Continued.

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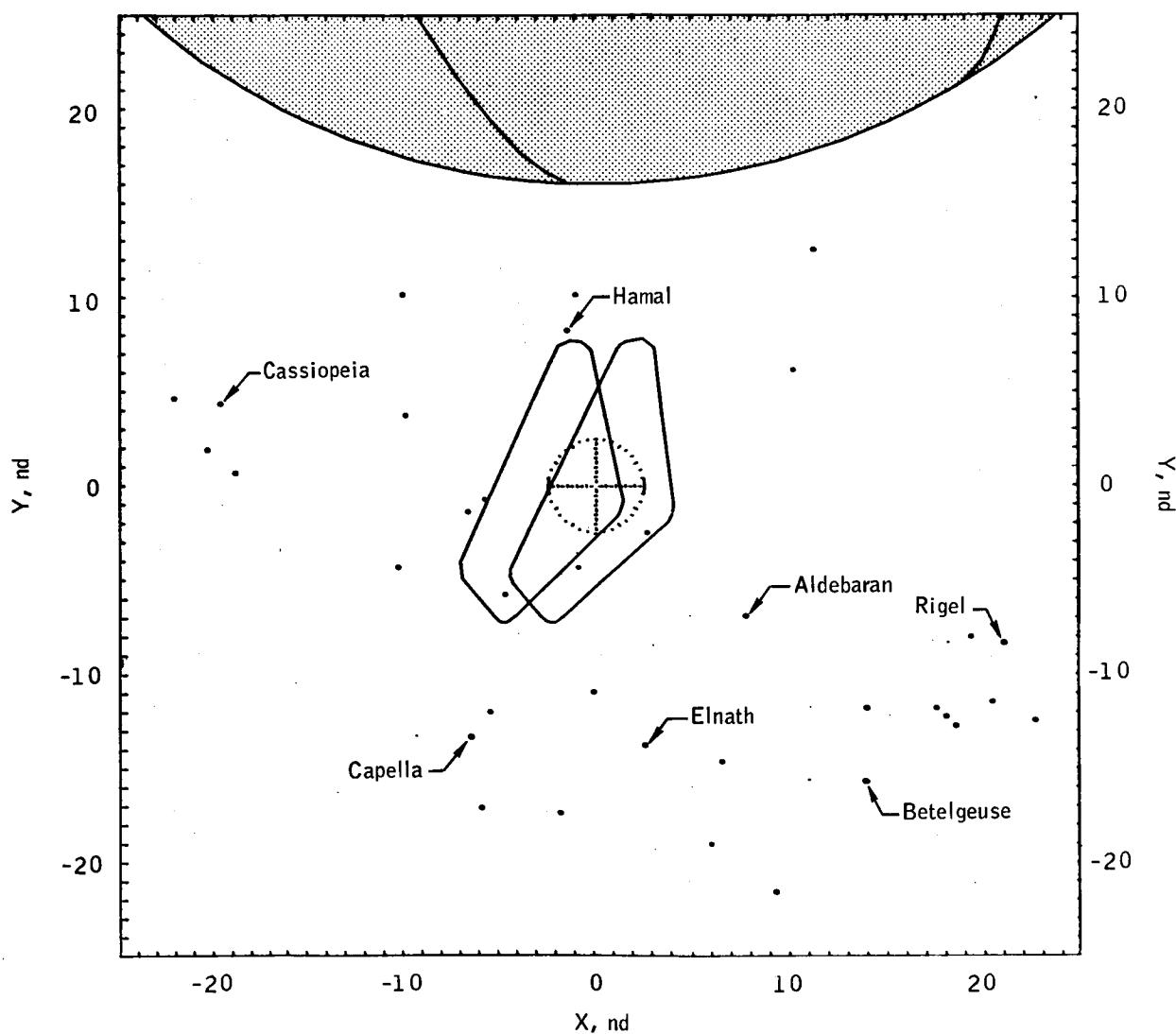
SEQ	5	22	31	41	47	63	73	75	80	108	111	112	120	144	150
X	-22	-19	-20	-10	-18	-1	-9	-1	11	10	-5	-6	-10	2	0
Y	4	4	2	10	0	10	3	8	12	6	0	-1	-4	-2	-4

SEQ	151	186	205	207	215	221	222	230	231	237	245	246	248	252	256
X	-4	7	0	-5	19	21	-6	13	2	17	20	18	6	18	22
Y	-5	-6	-10	-11	-7	-8	-13	-11	-13	-11	-11	-12	-14	-12	-12

265	270	271	281	301
13	-5	-1	5	9
-15	-16	-17	-18	-21

$$R_E = 3624 \text{ n. mi.}$$

$$V_i = 35634 \text{ fps}$$



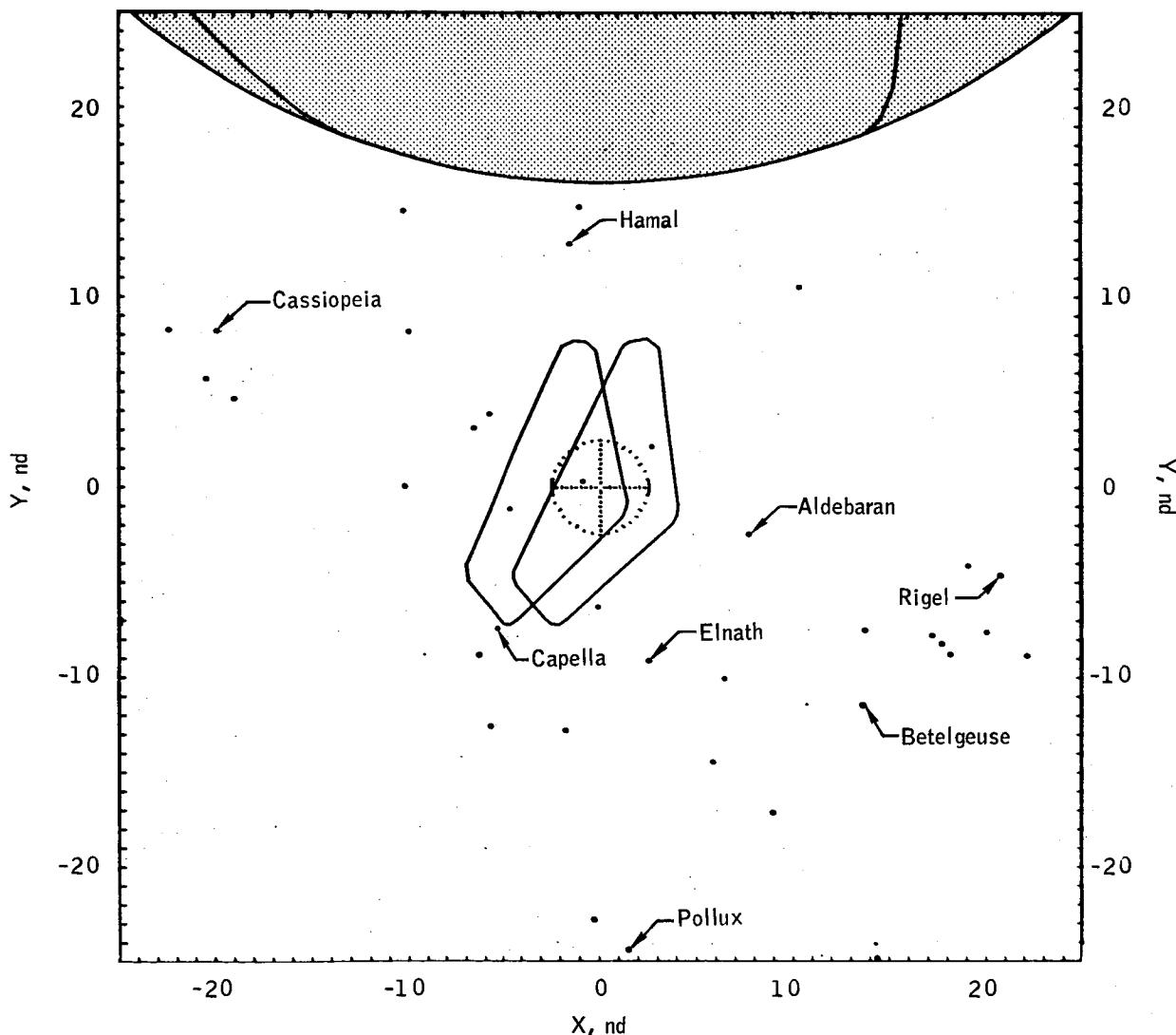
(p) 2 minutes prior to entry.

Figure 12.- Continued.

SEQ	5	22	31	41	47	63	73	75	108	111	112	120	144	150	161
X	-22	-19	-20	-10	-19	-1	-10	-1	10	-5	-6	-10	2	0	-4
Y	8	8	5	14	4	14	8	12	10	4	3	0	2	0	0
SEQ	186	205	207	215	221	222	230	231	237	245	246	248	252	256	265
X	7	0	-5	19	20	-6	13	2	17	20	17	6	18	22	13
Y	-2	-6	-7	-3	-4	-8	-7	-8	-7	-7	-8	-9	-8	-8	-11
SEQ	270	271	281	301	349	356	362								
X	-5	-1	5	8	0	14	1								
Y	-12	-12	-14	-16	-22	-24	-24								

$$R_E = 3558 \text{ n. mi.}$$

$$V_i = 35966 \text{ fps}$$



(q) 1 minute prior to entry.

Figure 12.- Continued.

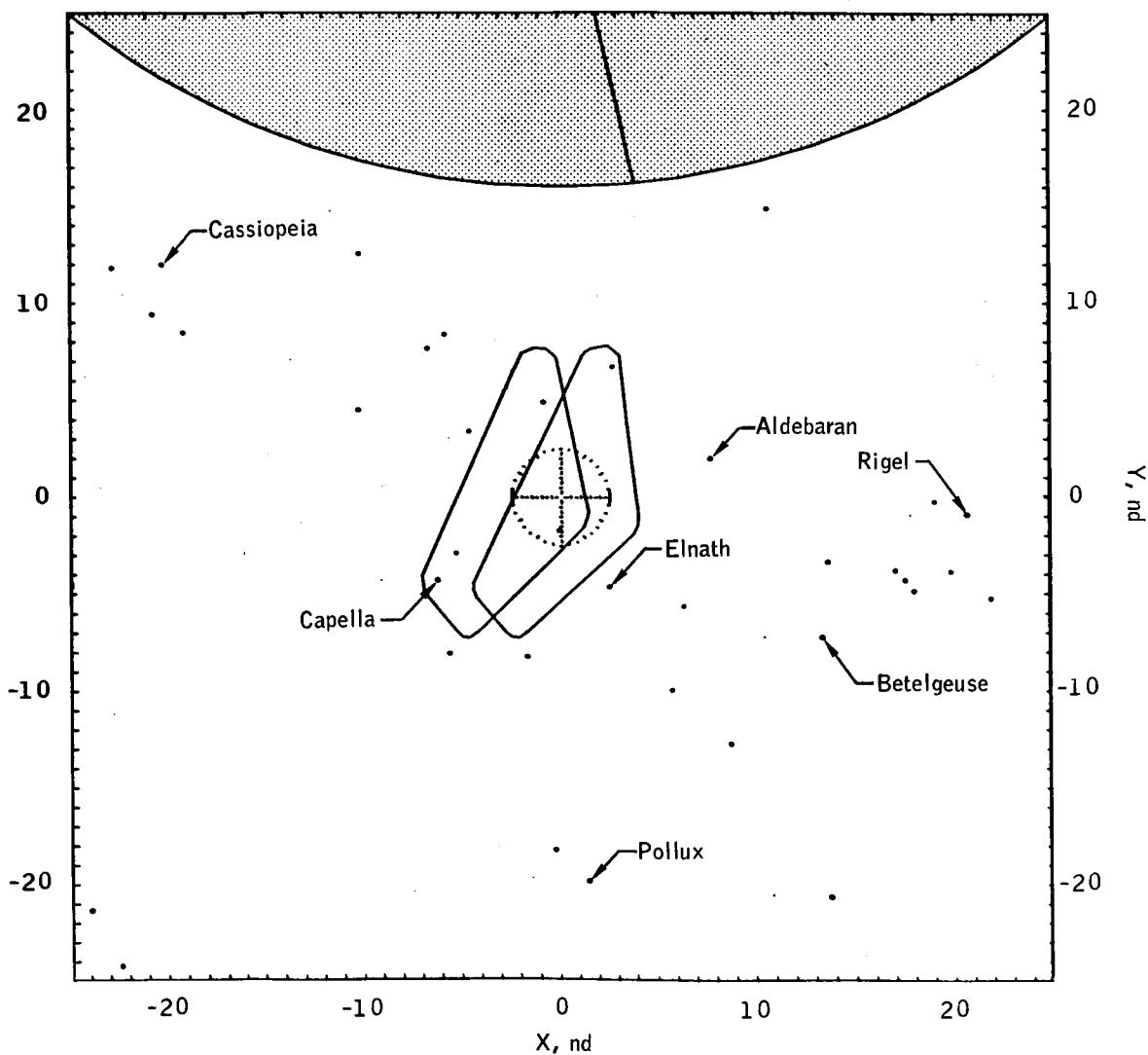
SEG	5	22	31	47	73	108	111	112	120	144	150	151	186	205	207	215
X	-22	-20	-20	-19	-10	10	-5	-6	-10	2	0	-4	7	0	-5	18
Y	12	12	9	8	12	15	8	7	4	6	5	3	2	-1	-2	0

SEG	22 ¹	222	230	231	237	245	246	248	252	256	265	270	271	281	301	345
X	20	-6	13	2	16	19	17	6	17	21	13	-5	-1	5	8	?
Y	0	-4	-3	-4	-3	-3	-4	-5	-4	-6	-7	-7	-8	-9	-12	-11

356 362 507 509
 13 1 -22 -24
 -20 -19 -24 -21

$$R_E = 3508 \text{ n. mi.}$$

$$V_i = 36220 \text{ fps}$$



(r) Entry.

Figure 12. - Concluded.

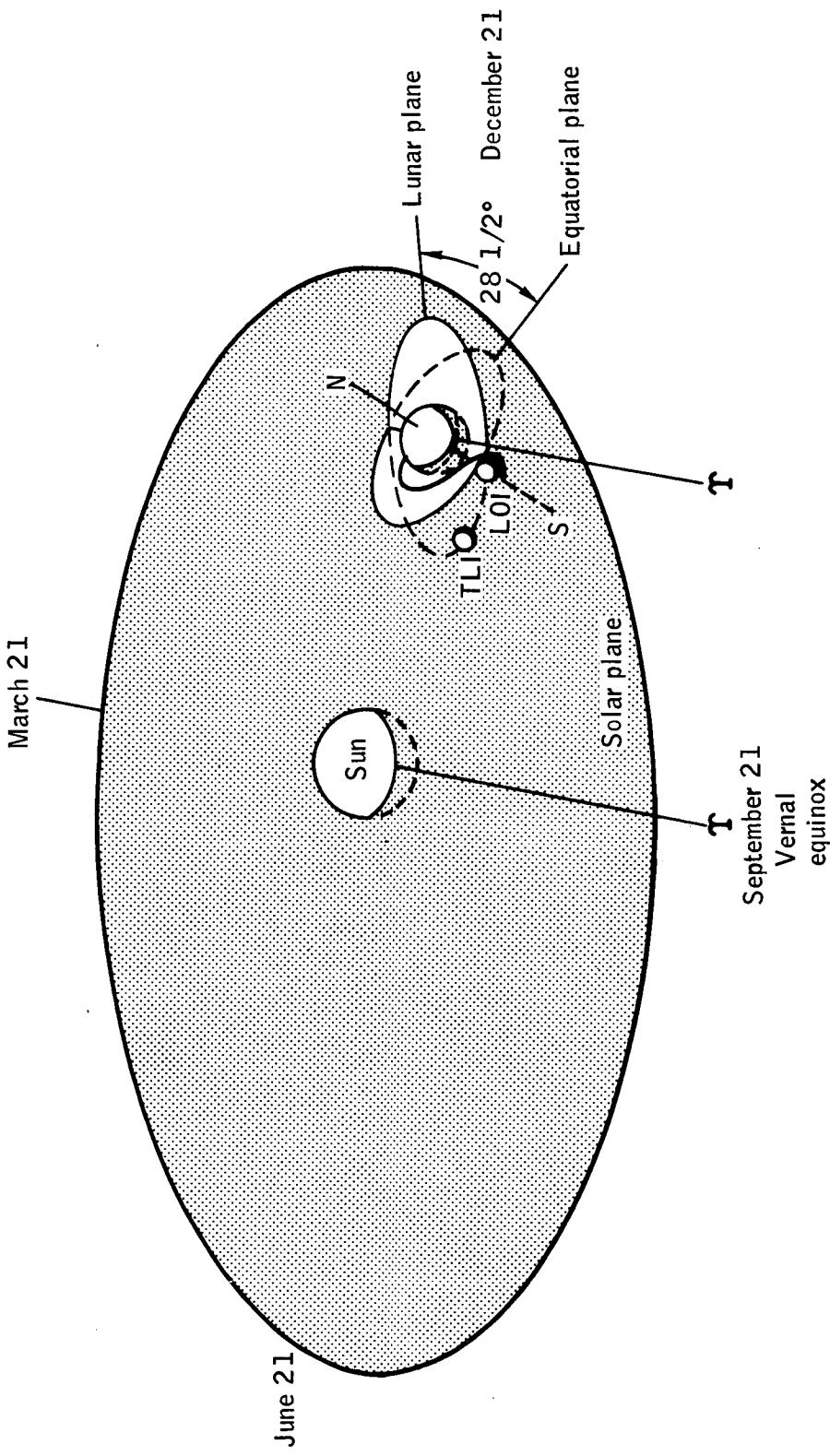


Figure 13.— Sun-centered inertial geometry of the Apollo 8 mission.

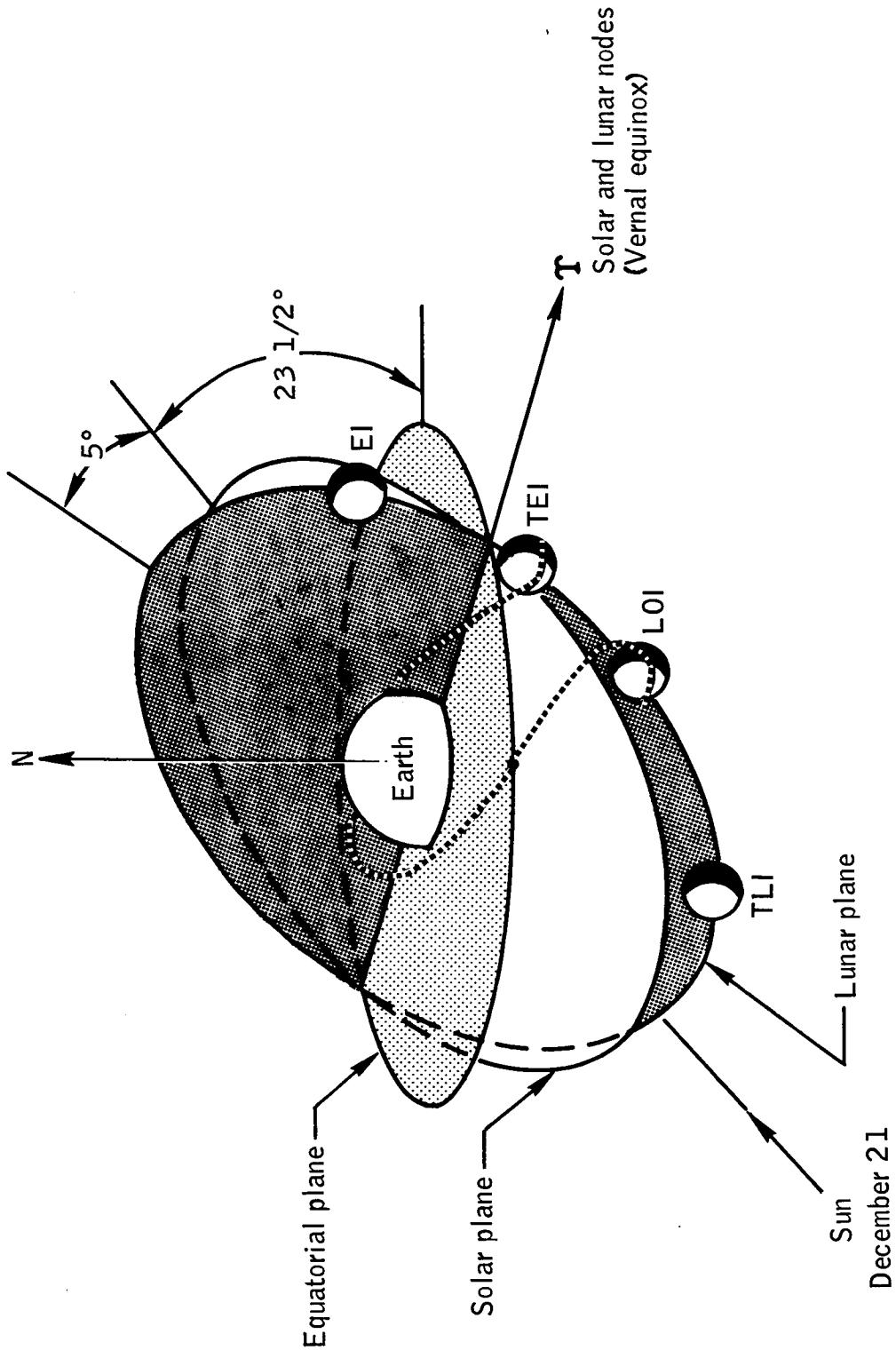


Figure 14.- Earth centered-inertial geometry of the Apollo 8 mission.

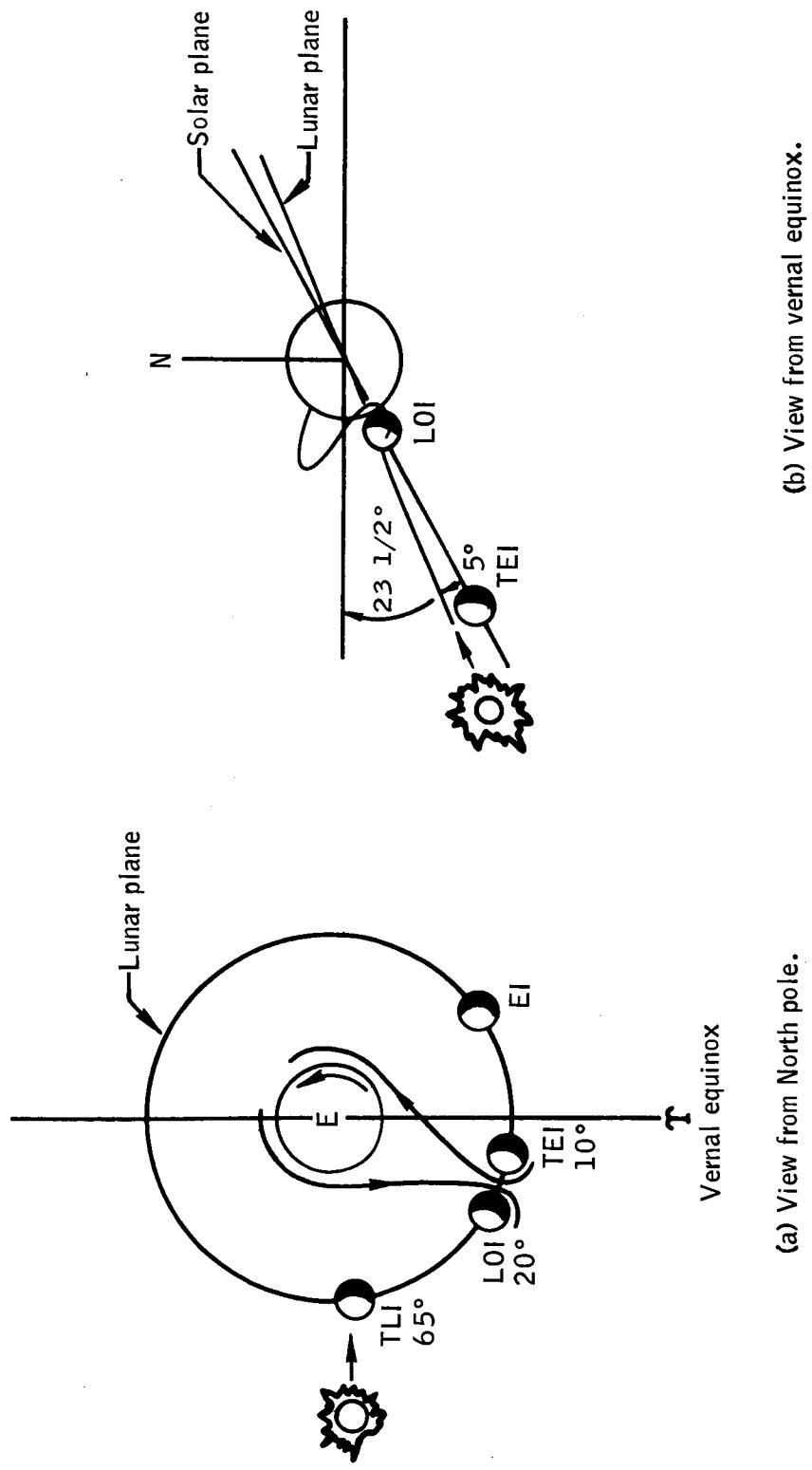


Figure 15. - Planar views of the Apollo 8 mission.

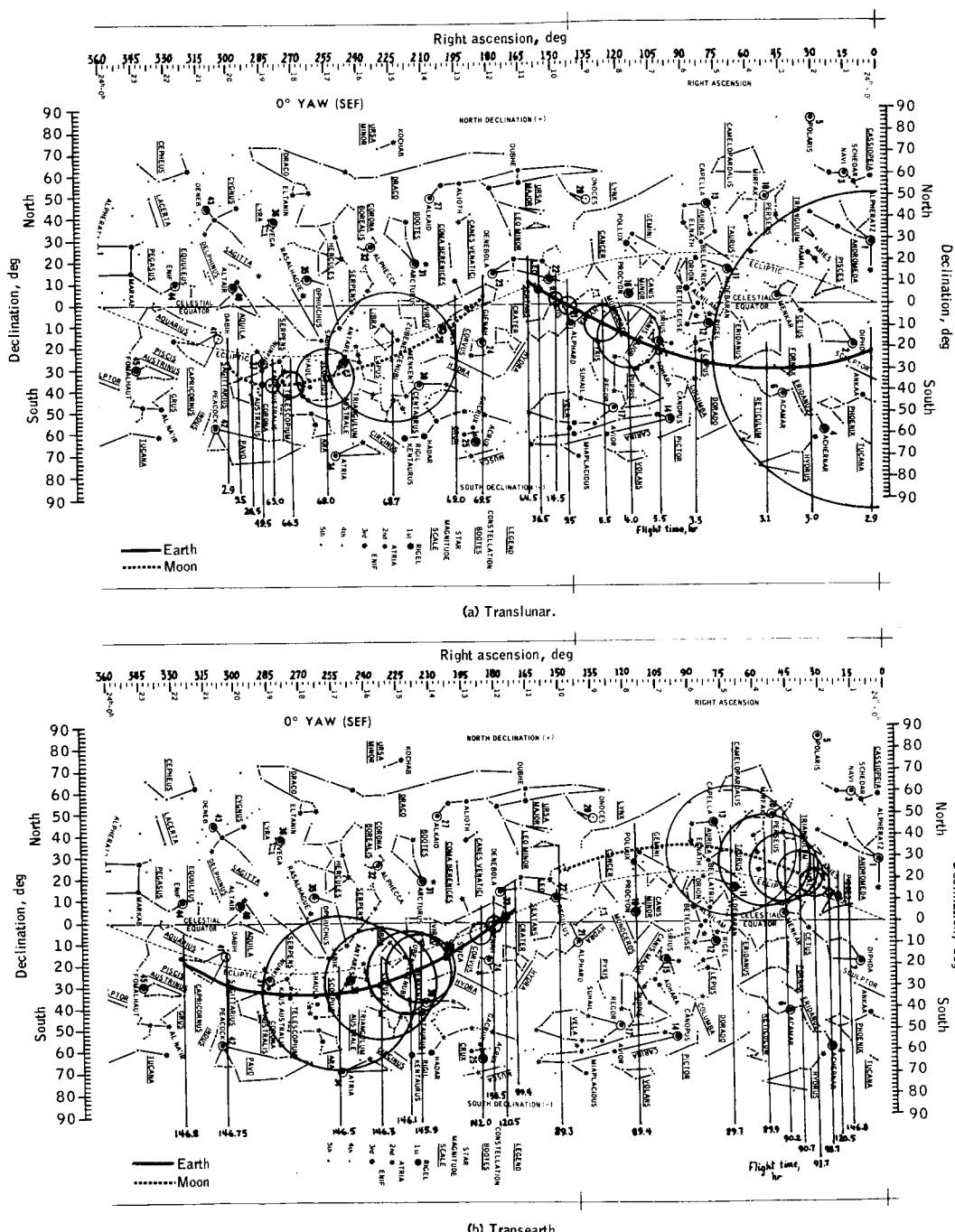


Figure 16.- Translunar and transearth star field.

TABLE I. - STAR IDENTIFICATION CATALOGUE - Continued

			SCD
610 24168	17.79125000	17.0	3.55
611 24364 GRUMIUM	17.88202700	47.0	2.0
612 24415	17.91752200	17.0	3.0
613 24432 ELTANIN	17.9179296000	52.0	3.90
614 24448	17.94405500	55.3	3.26
615 24468	17.95163800	17.0	3.90
616 24478	17.95269400	52.0	3.99
617 24502	17.97494400	55.0	15.0
618 24503	17.97719400	55.0	32.0
619 24509	17.98149900	55.0	2.42
620 24534	17.99910000	56.0	92.6
621 24605	18.00638800	57.0	9.0
622 24632 NASH	18.0065930500	57.0	6.0
623 24641	18.00641600	58.0	3.90
624 24645	18.00508300	58.0	33.6
625 24665	18.00680500	58.0	3.82
626 24693	18.0093833000	59.0	6.0
627 24695	18.009480500	59.0	3.50
628 24711	18.0097749900	59.0	3.0
629 24740	18.10291600	59.0	22.0
630 24767	18.12100000	60.0	4.40
631 24768	18.14833300	60.0	2.27
632 24956	18.19471000	61.0	57.2
633 24944	18.25427700	61.0	6.0
634 24961	18.26436000	61.0	5.92
635 25032	18.31055500	61.0	6.0
636 25024	18.31255500	61.0	6.0
637 25094	18.32494400	61.0	6.0
638 25045	18.33331600	61.0	6.0
639 25114	18.35432000	61.0	6.0
640 25122	18.36147200	61.0	6.0
641 25100 Kaus-Austr.	18.36413000	61.0	6.0
642 25116	18.37080300	61.0	6.0
643 25154	18.40630500	61.0	6.0
644 25180 Kaus-Bor.	18.43016400	61.0	6.0
645 25183	18.43558800	61.0	6.0
646 25220	18.45331000	61.0	6.0
647 25313	18.51672000	61.0	6.0
648 25305	18.55502000	61.0	6.0
649 25466 VEGA	18.59586100	61.0	6.0
650 25522	18.64925600	61.0	6.0
651 25580	18.67261000	61.0	6.0
652 25648	18.72033100	61.0	6.0
653 25661	18.72450000	61.0	6.0
654 25674	18.72608100	61.0	6.0
655 25686	18.73588800	61.0	6.0
656 25710	18.75527700	61.0	6.0
657 25734	18.75788800	61.0	6.0
658 25755	18.76022200	61.0	6.0
659 25847 SHELIAK	18.81311000	61.0	6.0
660 25823	18.81630500	61.0	6.0
661 25941 NUNKI	18.884471600	61.0	6.0
662 25954	18.88788500	61.0	6.0
663 25959	18.88777100	61.0	6.0
664 25930	18.88916000	61.0	6.0
665 25996	18.90447100	61.0	6.0
666 25991 ALYA	18.9077200	61.0	6.0
667 25991	18.9077200	61.0	6.0

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